Knudsen-to-Hydrodynamic Crossover in Liquid ³He in a High-Porosity Aerogel

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We present a combined experimental and theoretical study of the drag force acting on a high porosity aerogel immersed in liquid ³He and its effect on sound propagation. The drag force is characterized by the Knudsen number, which is defined as the ratio of the quasiparticle mean free path to the radius of an aerogel strand. Evidence of the Knudsen-hydrodynamic crossover is clearly demonstrated by a drastic change in the temperature dependence of ultrasound attenuation in 98% porosity aerogel. Our theoretical analysis shows that the frictional sound damping caused by the drag force is governed by distinct laws in the two regimes, providing excellent agreement with the experimental observation.

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The understanding of complex flow phenomena, such as in porous media, is a challenging goal of fluid physics. The pioneering work by Darcy [[1\]](#page-3-0) established empirically a linear relation between the fluid flux of laminar viscous flow in porous media and the total pressure drop in balance with viscous drag force. The Darcy law provides a constitutive relation in hydrodynamics and has been applied to various kinds of flow phenomena such as water flow in aquifers, blood flow in vessels, and petroleum flow through sandstone and gravel [\[2](#page-3-1)–[4\]](#page-3-2). In general, however, the hydrodynamic description breaks down when the mean free path (MFP) of fluid particles exceeds substantially the characteristic dimension of the porous material. In this case, the fluid mass is carried by the Knudsen diffusion [[5\]](#page-3-3), i.e., successive ballistic flights of individual particles followed by scattering at walls. The Knudsen diffusion, as well as the viscous flow, is of particular interest for gas transport in nanofabricated systems [\[6](#page-3-4)[,7](#page-3-5)], which have recently attracted significant attention in the context of nanofluidics [\[8\]](#page-3-6). The investigation of the MFP dependent flow regimes is also of fundamental importance towards a unified understanding of the transport properties in complex porous materials existing in nature.

Due to the unique porous structure of aerogel and the fascinating low temperature properties of liquid 3 He, the ³He-aerogel system presents an opportunity that allows investigations of a wide range of physical phenomena. The superfluid phase of 3 He in highly porous aerogel is an analog of unconventional superconductors with pairbreaking impurities [\[9](#page-3-7)–[14](#page-4-0)] and has been the subject of intense study since the first observations [\[9](#page-3-7)[,10](#page-3-8)] of superfluid transitions in this system. In the normal liquid, one can study the Fermi liquid behavior ranging from transport properties [\[11,](#page-3-9)[15–](#page-4-1)[19](#page-4-2)] to diverse flow phenomena [\[20\]](#page-4-3) in the presence of impurity-like disorder with a topologically antithetical geometry to the other conventional porous media. In this Letter, we present the first experimental and theoretical study on the Knudsen-hydrodynamic crossover in the normal liquid 3 He confined in 98% porosity silica aerogel.

Liquid 3 He is a paradigm of Fermi liquids in which the transport property is governed by low energy thermal excitations called quasiparticles. Since the collision processes between the quasiparticles are restricted by the Pauli exclusion principle, the quasiparticle MFP, l, can be much longer than the interparticle distance. The MFP increases with decreasing temperature T , following a $1/T^2$ dependence [[21](#page-4-4)], and approaches $\sim 10 \mu$ m near the superfluid transition temperature $T \approx 2 \text{ mK}$ (see Fig. 1) superfluid transition temperature, $T_c \approx 2 \text{ mK}$ (see Fig. [1\)](#page-1-0). On the other hand, aerogel, an interconnected network of thin strands of aggregated silica particles, introduces additional length scales: the strand diameter $2a$, the interstrand distance ζ , and the geometrical mean free path l_g .
Specifically in 98% porosity aeroed $2a \approx 3$ nm $\zeta \approx$ Specifically, in 98% porosity aerogel, $2a \approx 3$ nm, $\zeta \approx$
25–40 nm, and finally $l \approx 100-150$ nm. There is no 25–40 nm, and finally $l_g \approx 100$ –150 nm. There is no well-defined pore structure in aerogel in contrast to the nanochannels used in the previous works [\[6,](#page-3-4)[7](#page-3-5)] on the Knudsen-hydrodynamic crossover.

To understand the effect of the high porosity aerogel on the mass transport in liquid 3 He, we performed extensive ultrasound transmission measurements over a wide temperature range up to 200 mK. Based on the experimental results, we shall show that the transport properties in aerogel are characterized by two length scales of aerogel, the strand radius a and the geometrical MFP l_{ϱ} , separating three transport regimes as summarized in Fig. [1](#page-1-0).

FIG. 1 (color online). Characteristic length scales of 98% porosity aerogel and the temperature dependence of the quasiparticle mean free path, for three pressures, $l \propto 1/T^2$ in pure liquid 3He. The MFP was extracted from the experimental viscosity data [\[29\]](#page-4-5). The whole temperature range is divided into three transport regimes classified by the hydrodynamic, quasi-Knudsen, and full Knudsen regime. For $T \le 15$ mK (full Knudsen regime) the quasiparticle MFP is limited by the geometrical MFP l_g . Above \sim 15 mK, the mutual quasiparticle
collision becomes dominant, but the drag force is still in the collision becomes dominant, but the drag force is still in the Knudsen regime as far as $l > a$ (quasi-Kundusen regime). Around 150 mK (hydrodynamic regime), where $l \sim a$, the Knudsen-hydrodynamic crossover of the drag force occurs Knudsen-hydrodynamic crossover of the drag force occurs. The figures on the right show a schematic illustration of two extreme flow regimes.

The acoustic cavity used in this experiment is formed by two longitudinal $LiNbO₃$ transducers used as a transmitter and a receiver. The transducers are separated by a Macor spacer maintaining a 3.05 ± 0.02 mm parallel gap. The pair of transducers were carefully selected to achieve the best match in the fundamental frequency (9.5 MHz) and the frequency response. The 98% porosity aerogel sample was grown between the transducers in situ to prevent any irregular reflections at the transducer-aerogel boundaries. The temperature was determined by a melting pressure thermometer for $T > 1$ mK and a Pt NMR thermometer for $T < 1$ mK. The pressure of the cell was monitored throughout the measurements by a capacitive pressure sensor incorporated as a part of the experimental cell. A commercial spectrometer, LIBRA/NMRKIT II (Tecmag Inc., Houston, TX), was used to excite the transmitter and observe the propagated sound signal from the receiver. A 1 μ s pulse was sent to the transmitter and the signal from the receiver was amplified with a preamplifier (Miteq AU1114) before being transferred to the spectrometer. A detailed description of the experimental setup can be found elsewhere [\[15\]](#page-4-1).

In Fig. [2](#page-1-1), the receiver signal outputs acquired at 29 bar are plotted for selected temperatures. The signals for all temperatures rise around 8 μ s. Below 0.5 mK, in the low attenuation regime, one can see the three echoes following the initial peak. From the time-of-flight measurements, the

FIG. 2 (color online). Receiver signal outputs at 29 bar for selected temperatures. The superfluid transition in aerogel is demarcated by an arrow.

sound velocity c was obtained to be $c = 330$ m/s at 29 bar. The temperature dependence of the sound velocity is less than our measurement error for the whole temperature range presented in this work.

The relative sound attenuation α was determined using two different methods. One was to use the peak value of the initial signal and the other was to use the integrated signal of the initial peak from the rising edge to 23 μ s. Both methods produced consistent results within 1% in the low attenuation regime and within 5% in the high attenuation regime. The relative attenuation was then converted into the absolute attenuation using a reference value at 0.4 mK where 3 echoes were clearly observed owing to the low attenuation in the superfluid phase. The peaks of the echoes exhibit an exponential decay, enabling us to determine accurately the absolute attenuation. In Fig. [3](#page-1-2), the

FIG. 3 (color online). Ultrasound attenuation ($\omega/2\pi$ = ⁹:5 MHz) in the normal fluid above the aerogel superfluid transition. The solid symbols represent the experimental data at three different pressures. The solid lines are the results of fit to the data using Eqs. [\(7\)](#page-3-10)–([9\)](#page-3-11) with the Cunningham constant $A =$ 6/5. The fitting parameters are $(l_{tr}, l_a) = (115.4, 157.2)$ nm for 10 bar, (108.7,147.1) nm for 21 bar, and (100.0,135.4) nm for 29 bar. The dashed lines are the sound attenuation calculated in the Knudsen (α_{∞}) and hydrodynamic (α_0) limit (see text).

attenuation data obtained by the second method are plotted as a function of temperature T. The data at three different pressures are shown above the superfluid transition in aerogel. The attenuation α is almost independent of T near T_c but decreases gradually with increasing T. This behavior is in good agreement with those observed in previous ultrasound attenuation measurements [\[11\]](#page-3-9). However, our higher temperature measurements revealed a new behavior which exhibits a minimum around 50 mK followed by a rather rapid increase in attenuation with temperature.

As the first step of analysis of the experimental results, let us estimate l_e in 98% aerogel. The simplest estimate can be made by modeling the aerogel as a dilute random distribution of spheres with radius *a* (sphere model). Noting that the sphere density n_a is related to the aerogel porosity ϕ by $(4/3)\pi a^3n_a = 1 - \phi$, we obtain $l_g =$ $1/\pi a^2 n_a = (4/3)a/(1 - \phi)$. Substitution of $a = 1.5$ nm and $\phi = 0.98$ yields $l_g = 100$ nm. A more realistic model of randomly oriented cylinders [\[22\]](#page-4-6) also leads to a similar value $l_g = 2a/(1 - \phi) = 150$ nm. These estimations are consistent with those from an empirical formula extracted from computer simulations based on a diffusion limited cluster aggregation model [[23](#page-4-7)]. In Fig. [1,](#page-1-0) l_g is compared with $l \propto 1/T^2$ in pure liquid ³He. We observe $l_g \approx l$ at around 15 mK.

At low temperatures where $l > l_g$, the transport properties are governed by the geometry of the aerogel structure. This point is clearly demonstrated by regarding the quasiparticle-quasiparticle and quasiparticle-aerogel collisions as two independent scattering sources. Then, for example, the effective viscosity coefficient in the ³He-aerogel system has the form

$$
\eta_{\rm eff} = \frac{1}{5} n p_F (1/l_a + 1/l)^{-1} = \frac{1}{5} \frac{n p_F l_a}{1 + T^2/T_a^2},\qquad(1)
$$

with *n* being the ³He number density, p_F the Fermi momentum, l_a a temperature independent MFP of the order of l_g , and T_a the temperature at which $l = l_a$. At high temperatures, $T \gg T_a$, Eq. [\(1](#page-2-0)) is reduced to the conventional gas-kinetic formula for a pure Fermi liquid. As temperature decreases, the quasiparticle scattering processes become gradually dominated by the aerogel-quasiparticle collisions, so that η_{eff} saturates to the geometry-limited value of $(1/5)$ n p_Fl_a . Similar saturation behavior is observed for the spin-diffusion coefficient of 3 He in 98% aerogel [[18\]](#page-4-8).

Now we discuss the sound propagation in aerogel. The sound frequency $\omega = 2\pi \times 9.5$ MHz used in our measurements should be compared to the effective quasiparticle scattering rate, $1/\tau_{\text{eff}} = v_F(1/l_a + 1/l)$. The Fermi
velocity v_F in liquid ³He is \approx 50 m/s [24]. Since $1/\tau_{\text{eff}} >$ velocity v_F in liquid ³He is \simeq 50 m/s [\[24](#page-4-9)]. Since $1/\tau_{\text{eff}} >$
 $v_F/l \ge 10^8$ s⁻¹ our ultrasound measurements are in the $v_F/l_a \approx 10^8 \text{ s}^{-1}$, our ultrasound measurements are in the collision-dominated regime, $\omega < 1/\tau_{\text{eff}}$. Therefore, the sound propagation can then be described by the Navier-Stokes equation

$$
\rho \partial_t \mathbf{v} = -\nabla P + \eta_{\text{eff}} (\Delta \mathbf{v} + \frac{1}{3} \nabla \text{div} \mathbf{v}) - F, \qquad (2)
$$

where $\rho = mn$ is the ³He mass density, **v** the velocity field of liquid 3 He, P pressure, and F the drag force (per unit volume) exerted on aerogel. This drag force gives rise to the coupled oscillations of liquid 3 He and aerogel [\[16,](#page-4-10)[17\]](#page-4-11). Since the sound wavelength of interest is much longer than the distance between aerogel strands, the aerogel oscillations can be described within a continuum approximation,

$$
\rho_a \partial_t^2 \mathbf{u}_a = \rho_a c_a^2 \Delta \mathbf{u}_a + \mathbf{F}, \tag{3}
$$

where \mathbf{u}_a , $\rho_a \approx 0.044$ g/cm³, and $c_a \approx 50$ m/s are the displacement vector, the mass density, and the speed of longitudinal sound in aerogel itself, respectively.

To proceed further, we need the constitutive relation for **F**. Let Kn be the Knudsen number defined by Kn = $l/a \propto$ $1/T²$. Since the smallest length scale of aerogel is the strand radius a, the hydrodynamic description of the drag force is justified for $Kn < 1$. In the hydrodynamic limit, $Kn \rightarrow 0$, the drag force density obeys the Darcy law

$$
F_0 = -\frac{\eta}{\kappa} \mathbf{v}_{\text{rel}}.\tag{4}
$$

Ĵ Here $\mathbf{v}_{rel} = \mathbf{v} - \partial_t \mathbf{u}_a$ is the relative velocity field of liquid v^3 He and aerogel $n = (1/5)n_{rel}$ the viscosity of pure ³He and aerogel, $\eta = (1/5)np_Fl$ the viscosity of pure liquid ³He, and κ the permeability. In the sphere model for aerogel, the Darcy law is equivalent to the Stokes law $F_0/n_a = 6\pi a \eta v_{rel}$, and then the permeability takes the form, $\kappa = al_{\varrho}/6$.

In the opposite limit, $Kn \rightarrow \infty$, where the Darcy law breaks down, the drag force is determined by the collisions between aerogel and individual quasiparticles (see Fig. [1\)](#page-1-0). Ichikawa et al. [[16](#page-4-10)] discussed the drag force on aerogel by modeling it as randomly distributed impurities. By calculating the momentum transfer from liquid 3 He to aerogel using the impurity-quasiparticle collision integral, they obtained the following expression for the drag force density:

$$
F_{\infty} = \frac{np_F}{l_{\text{tr}}} \mathbf{v}_{\text{rel}},
$$
 (5)

where l_{tr} is the transport MFP. Equation [\(5\)](#page-2-1) has an anticipated form of the drag force density in the Knudsen limit, $Kn \rightarrow \infty$. Specifically, consider a situation in which an object with a cross section σ is moving with a constant
velocity $-\mathbf{v}$, in a "gas" of quasiparticles with momentum velocity $-v_{rel}$ in a "gas" of quasiparticles with momentum p_F . The body would experience collisions with the quasiparticles of number $n\sigma |\mathbf{v}_{rel}|$ in unit time and experience a
force $\sim n_{\rm F}n\sigma \mathbf{v}$. Dividing it by the total volume V and force $\sim p_F n \sigma v_{rel}$. Dividing it by the total volume V and replacing σ/V by $1/l$, we reach Eq. (5) It should be replacing σ/V by $1/l_{tr}$, we reach Eq. ([5\)](#page-2-1). It should be
noted however that since *l*, depends on the details of the noted, however, that since l_{tr} depends on the details of the object, e.g., its shape and the irregularity of the surface, it is difficult to estimate l_{tr} accurately for complex porous materials like aerogel. We should therefore regard l_{tr} as a phenomenological parameter comparable to the geometrical MFP l_g .

In order to obtain the expression for the drag force for arbitrary values of Kn, we put $\mathbf{F} = \mathbf{F}_0/(1 + A \mathbf{K}_0)$ and determine the dimensionless constant A so that the Knudsen-limit formula Eq. [\(5\)](#page-2-1) is recovered. A similar procedure was used by Cunningham [\[25\]](#page-4-12) in his theory for the drag force on a sphere immersed in classical fluids, and the resulting Stokes-Cunningham formula was applied successfully to Millikan's oil-drop experiment in air [\[26\]](#page-4-13). In the present problem, the Cunningham constant A takes the form of $A = al_{tr}/5\kappa$. For the sphere model, we have $A = 6l_{tr}/5l_{g}$, which illustrates that A is a constant of order unity. For the analysis of the experimental results, it is convenient to write \bf{F} in the following form:

$$
F = \frac{1}{1 + 1/A \text{Kn}} F_{\infty} = \frac{1}{1 + T^2/A T_1^2} \frac{n p_F}{l_{\text{tr}}} \mathbf{v}_{\text{rel}}.
$$
 (6)

Here, T_1 is the temperature at which Kn = 1. The crossover temperature T_1 is well above T_a , as can be seen in Fig. [1.](#page-1-0)

The Cunningham-type formula implicitly assumes a smooth transition of the drag force between the hydrodynamic and Knudsen regimes. Virtanen and Thuneberg [\[27](#page-4-14)[,28\]](#page-4-15) recently reported a detailed numerical study of the drag force on a vibrating wire immersed in ³He-⁴He mixtures. Although the system is different from that of present interest, a common feature can be found in their results, showing a smooth transition in the dissipative part of the mechanical impedance Z at low frequencies.

Having obtained the system of equations, we can now analyze the experiment. Considering the longitudinal sound of the plane wave form $\propto e^{i(q \cdot r - \omega t)}$ and taking account of an experimental fact that $|c_a q/\omega|^2 \approx 0.02 \ll 1$ one obtains 1, one obtains

$$
\alpha = \text{Im}q = \alpha_v + \alpha_f, \tag{7}
$$

$$
\alpha_v = \frac{2\omega^2 p_F l_a}{15mcc_1^2} \frac{1}{1 + T^2/T_a^2},\tag{8}
$$

$$
\alpha_f = \frac{\omega^2 m l_{\rm tr}}{2 c p_F} \frac{(\rho_a/\rho)^2}{1 + \rho_a/\rho} \left(1 + \frac{T^2}{A T_1^2}\right),\tag{9}
$$

where $c = \omega/Req = c_1/(1 + \rho_a/\rho)^{1/2}$ and c_1 are the sound velocities in the presence and absence of aerogel sound velocities in the presence and absence of aerogel, respectively. The sound velocity c is reduced from c_1 by a factor $(1 + \rho/\rho_a)^{-1/2}$ due to the extra inertia from
the dragged aerogel Using $c_1 = 398$ m/s and $\rho =$ the dragged aerogel. Using $c_1 = 398$ m/s and $\rho =$ 0.114 g/cm³ at 29 bar [\[24\]](#page-4-9), we obtain $c = 338$ m/s, in good agreement with the experiment. c_1 and therefore c are almost independent of temperature, which also agrees with our observation. In Eq. [\(7\)](#page-3-10) for sound attenuation, α_{ν} and α_f are associated with the viscous damping and the frictional damping caused by the drag force, respectively. The theoretical calculation represented by the solid lines in Fig. [3](#page-1-2) shows excellent agreement with the experiment using the fitting parameters l_{tr} and l_a as listed in the figure caption and $A = 6/5$. The role of the Cunningham constant A is to shift the crossover temperature in Eq. ([9](#page-3-11)). The best fit to the experimental data is achieved by taking A to be around $6/5$, corresponding to the sphere model with $l_{\text{tr}} = l_g.$

To gain insights into the observed behavior, we plot in Fig. [3](#page-1-2) the sound attenuation in the two limiting cases for 29 bar: the Knudsen (α_{∞}) and the hydrodynamic (α_0) regimes by taking the limits of $T_1 \rightarrow \infty$ and $T_1 \rightarrow 0$, respectively in Eq. ([9\)](#page-3-11). The difference between the two limits comes mainly from the different T dependence of the frictional damping α_f . In the full Knudsen regime where $T \ll T_a$ and effectively $T_1 \rightarrow \infty$, both α_v and α_f are temperature independent. This simply reflects the fact that the effective quasiparticle MFP is limited by the geometrical MFP l_{ϱ} . The decrease of the Knudsen attenuation above T_a is due to the decrease of α_v . The low temperature experimental data are well described by the Knudsen behavior. However, as T increases, a significant deviation occurs especially around $T = T_1$. With increasing T further above T_1 , the experimental data approach asymptotically the hydrodynamic behavior, $\alpha \approx \alpha_0 \propto T^2$. The crossover temperature T_1 increases with decreasing pressure (see Fig. [1\)](#page-1-0) due to the pressure dependent quasiparticle mean free path. Consequently, this effect shifts the attenuation minimum to higher temperature for lower pressure as observed in the experiment.

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