Direct numerical simulations of a cylinder cutting a vortex

Steven Soriano

NASA Glenn Research Center, Cleveland, Ohio 44135, USA and Cullen College of Engineering, University of Houston, Houston, Texas 77204, USA

Rodolfo Ostilla-Mónico 10*

Departamento Ingeniería Mecánica y Diseño Industrial, Escuela Superior de Ingeniería, Universidad de Cádiz, Av. de la Universidad de Cádiz 10, 11519 Puerto Real, Spain and Cullen College of Engineering, University of Houston, Houston, Texas 77204, USA

(Received 10 January 2024; accepted 19 April 2024; published 8 May 2024)

The interaction between a vortex and an impacting body which is oriented normally to it is complex due to the interaction of inviscid and viscous three-dimensional mechanisms. To model this process, direct numerical simulations of a thin cylinder intersecting a columnar vortex oriented normally to it are conducted. By varying the impact parameter and the Reynolds number, the two regimes of interaction mentioned in the literature are distinguished: the weak and strong vortex regimes. Low impact parameters, representing strong vortices, led to ejection and interaction of secondary vorticity from the cylinder's boundary layer, while high impact parameters, representing weak vortices, led to approximately inviscid interaction of the cylinder with the primary vortex through deformations. No significant effect of the Reynolds number in the overall phenomenology is found, even if larger Reynolds numbers lead to the formation of increasingly smaller and more intense vortex structures in the parameter range studied. Finally, the hydrodynamic force curves on the cylinder are analyzed, showing that intense forces could be locally generated for some parameter regimes, but that the average force on the cylinder does not substantially deviate from baseline cases where no vortex was present. Our results shed light on the underlying mechanisms of vortex-body interactions and their dependence on various parameters.

DOI: 10.1103/PhysRevFluids.9.054701

I. INTRODUCTION

Vortices are one of the most characteristic structures present in fluid flows, and they have been referred to as "the sinews and muscles of fluid dynamics" [1]. At high Reynolds numbers, vortices are generally modeled as singular lines or sheets. A wide variety of flows such as aircraft or propeller-tip wakes or bubble rings can then be studied through the composition of singular solutions [1]. Much attention has been given to how vortex tubes, rings, or sheets interact with each other or with solid boundaries, including changes in the topology of singular solutions such as those seen in reconnecting vortex rings [2]. However, the purposeful breaking or "cutting" of topological vortex lines with solid objects has seen less work due to experimental and numerical challenges [3–5]. This flow phenomenology finds significant relevance in a specific context: blade-vortex interaction (BVI), which involves the intricate physical interaction when a rotating blade encounters a nearby vortex tube [6] but is also present in other physical systems such as Type II superconductors [7,8].

2469-990X/2024/9(5)/054701(21)

^{*}rodolfo.ostilla@uca.es

BVI holds widespread significance in various engineering applications, spanning wind turbines, helicopter rotors, marine propellers, and industrial fans [6], and due to its practical applications has been the workhorse of research in vortex line cutting in fluid dynamics. The majority of research on BVI concentrates on aerodynamic blades intersecting a straight vortex tube. This specific BVI scenario, referred to as perpendicular or normal BVI, occurs when the vortex and blade axes are perpendicular, and the blade cleaves through the vortex tube. Normal BVI is commonly encountered in applications such as helicopter rotors and wind turbines, where rotating blades confront vortices originating from prior blades or generated by their own motion [9-13]. Other orientations of the blade in relation to the vortex are possible, resulting in other types of BVI: grazing BVI, where the blade is perpendicular to the tube, but it moves in the direction of the core, and parallel BVI [14], where the blade is parallel to the tube and cuts the tube throughout its extent rather than at a single point. These are outside the scope of our study, which focuses on swift cutting of the core.

Experiments and simulations have shown that the interaction between a vortex tube and a body such as a blade or a cylinder depends on various parameters, such as body velocity, vortex core radius, vortex swirl velocity, body geometry and size, and turbulence within the body's boundary layer [15–21]. In particular, for a body impacting a vortex with no axial flow, three dimensionless parameters are relevant. These parameters are (1) the impact parameter ($I_P = 2\pi \sigma_0 V/\Gamma$), which is the ratio of the body's free-stream velocity to the maximum vortex swirl velocity, where σ_0 is the radius of the vortex, V the velocity of the body, and Γ the vortex's circulation; (2) the vortex Reynolds number ($\text{Re}_{\Gamma} = \Gamma/\nu$), where ν is the kinematic viscosity; and (3) the body thickness parameter (T), which is the ratio of a characteristic length such as the blade's curvature or the cylinder diameter D to the vortex core radius σ_0 [17,22].

Depending on both the response of the primary vortex to the body and about the interaction of the vortex with the body's boundary layer, response regimes can be constructed depending on the value the dimensionless parameters take. The response regimes are commonly referred to as the weak vortex regime, the strong vortex regime, and the bending regime. In this paper we will simulate cases in the weak and strong vortex regime, while the bending regime will not be investigated. However, we discuss here all three regimes for completeness.

The weak vortex regime occurs for thin bodies $(T = D/\sigma_0 \lesssim 1)$ when the vortex is sufficiently weak $(I_P > 0.2)$. This regime is characterized with very minimal boundary layer separation and ejection until after the body has penetrated the primary vortex [22]. Once the boundary layer has separated, the separated vorticity becomes entrained within the primary vortex and does not wrap around it, instead spreading away from the body and into the vortex core.

The strong vortex regime has been seen to occur when the vortex is sufficiently strong ($I_P < 0.08$) and is independent of the thickness parameter [17]. This regime is characterized with a large amount of interaction between the body's boundary layer and the vortex prior to the body's leading edge, making contact with the vortex. The ejected vorticity from the body's boundary layer will roll up into a series of vortex loops that will wrap around the primary vortex. The wrapping of the secondary vorticities creates a series of wave motions that will eventually disrupt the primary vortex. This procedure of primary vortex disruption may also occur when the body is several core radii away from the vortex, but only when the value of the impact parameter is relatively low [3].

The bending regime occurs when the vortex is sufficiently weak ($I_P > 0.2$) and the body is sufficiently thick (T > 5) [22,23]. Separation of the boundary layer does not occur before the body has traveled through the original position of the vortex. This regime also sees large-scale deformation of the primary vortex due to the inviscid interaction with the body. The core radius of the vortex will thin near the region where the body will impact the vortex due to stretching of the flow about the body. Once the vortex has deformed around the body, the body's boundary layer will separate and generate wave motions within the vortex eventually leading to a disruption of the vortex [3,17]. While experimental studies of this process exist, experiments are limited in their access to the full velocity and pressure fields. Furthermore, the effect of turbulence in the boundary layer on the interaction process is not well understood, partly due to the challenges in experimentally controlling inlet turbulence. Numerical simulations could in principle allow a detailed study of the development and evolution of the vortex formation and boundary layer detachment, determining the spatial and temporal forces present on the blade, as well as having full control over the turbulence in the system. They also have the advantage of being able to controlling input parameters separately, i.e., varying Re_{Γ} separately from the impact parameter, which is not easy in an experiment.

Inviscid filament models are the oldest method used to simulate BVI, due to their straightforwardness and effectiveness, but present well-known limitations: they require extensions to account for variations in the vortex core area and to ensure no-penetration conditions on the surfaces; and they are inviscid, which means that they cannot be used to analyze vortex-induced boundary layer separation [3,23–26], a crucial phenomenon in the weak vortex regime. Nonetheless, they are useful for predicting how the body will penetrate the vortex or how the vortex will react to the motion of the body, and can also be used to estimate the pressure on the body surface during the interaction with a vortex up to a certain distance from the body [23,27]. A number of extensions to the filament model which incorporate two-dimensional boundary layer dynamics have been conducted, which hint at the type of complex flow phenomena which arise [28,29]. These were later extended to include three-dimensional boundary layers [30,31]. However, they remain limited by the fact that the underlying filament models present singularities and cannot fully resolve the dynamics [31].

The highest fidelity simulations of normal BVI available in the literature employ threedimensional direct numerical simulations of a vortex impacting a blade [5,32]. These studies have detailed the way a vortex cuts through a blade, and the way this cutting generates circulation and a lift force. Reference [5] was limited in the range of Reynolds numbers they could simulate, as well as the choice of large impact parameters due to the lack of adequate computational power. Reference [32] simulates larger Reynolds numbers by fixing the blade Reynolds number to be $\text{Re}_b = 1000$ and varying the strength of the vortex. This study focuses on large impact parameters, $I_P > 0.5$, and does not access the strong vortex regime.

As can be gathered from above, a detailed numerical study of the separation process in the weak vortex regime and the transition from weak to strong vortex regimes is missing. The use of DNS in vortex-body interaction studies has the potential to clarify previously unclear areas such as the viscous interaction of the body with the vortex [32]. Simulations also allow careful control of the Reynolds number and can assess questions of Reynolds number independence unanswered in experiments [17]. In this paper, we set out to address these knowledge gaps by conducting three-dimensional direct numerical simulations (DNS) of tube cutting using stateof-the-art numerical techniques. We will simulate a simplified problem: a thin wire (cylinder) impacting normally a high-Reynolds-number vortex. We will study the interaction process by varying two parameters: the impact parameter I_P and the circulation Reynolds number Re_{Γ}. Values of I_P will be chosen such that simulations can access the weak and strong vortex regimes, while varying Re_{Γ} will modify the secondary vorticity of the cylinder, as well potentially triggering more hydrodynamical instabilities in the primary tubes which set in at values of $\text{Re}_{\Gamma} \sim O(10^3)$ [33]. We will focus on key flow phenomena observed during the interaction, calculate the induced force on the cylinder as it cuts through the vortex, and relate how major flow structures influence the force curve.

The paper is organized as follows: Sec. II details the numerical methods used in the manuscript. Section III presents the main results from the flow field, including the transition between strong and weak vortex regimes (Sec. III A) and in-depth analysis of the strong (Sec. III B) and weak vortex regimes (Sec. III C). Section IV examines the force in the wire. Finally, Sec. V presents a summary of the findings as well as an outlook for future work.



FIG. 1. Schematic view of the computational domain. The z direction is shortened for clarity.

II. NUMERICAL METHODS

To study BVI, we directly simulate the the incompressible Navier-Stokes equations:

n--

$$\frac{\partial \mathbf{u}}{\partial t} + \mathbf{u} \cdot \nabla \mathbf{u} = -\rho^{-1} \nabla p + \nu \nabla^2 \mathbf{u} + \mathbf{f}, \tag{1}$$

$$\nabla \cdot \mathbf{u} = 0, \tag{2}$$

where **u** is the velocity, *t* is time, ρ is the fluid density, ν is the fluid kinematic viscosity, and **f** is a body force originating from the the immersed boundary method (IBM) forcing used to model the cylinder. A schematic of the system is presented in Fig. 1, which shows a cylinder of diameter *D*, with an axis parallel to the *z* axis, that travels in the *x* direction at a velocity *V* towards a vortex of core size σ_0 which has an axis parallel to the *y* direction. This vortex is initialized at the initial time with a Lamb-Ossen (Gaussian) velocity profile with circulation Γ . The vortex circulation Γ and the vortex radius σ_0 are used to nondimensionalize the system.

The computational domain is taken as triply periodic, with periodic lengths $\mathcal{L}_x \times \mathcal{L}_y \times \mathcal{L}_z = 30\sigma_0 \times 30\sigma_0 \times 90\sigma_0$. The larger length in \mathcal{L}_z allows for the force on the object to return to baseline values; this effect is fully discussed in Sec. IV. The simulation is initialized with a zero velocity field except for the Lamb-Ossen vortex with the core extending in the *y* direction and centered at (x = 0, z = 0). The cylinder starts from rest and travels for $10\sigma_0$ at *V* velocity before reaching the vortex's core line. Because the distance is the same, this means that for varying cylinder velocity (through changing I_P), the cylinder reaches the vortex at different times. After cutting the axis of the vortex, the cylinder then travels to the opposing end of the box. This allows us to capture all stages of the interaction with the vortex.

The diameter of the cylinder is fixed as $0.95\sigma_0$, keeping the thickness parameter $T = D/\sigma_0 \approx 1$ in the thin regime. The impact parameter is varied between $I_P \in (0.05, 0.25)$. With this value of T, the degree of (inviscid) vortex bending that occurs before the ejection of boundary layer vorticity is determined by the impact parameter. The range of values of I_P chosen, which allows us to explore the weak and strong interaction regimes for low and high values of I_P , respectively. Finally, the circulation-based Reynolds numbers are varied in the range $\text{Re}_{\Gamma} \in (1000, 3000)$. This will impact the flow in two ways: the boundary layer of the cylinder will be thinner with higher vorticity, and the increased Reynolds number may also trigger hydrodynamical instabilities in the primary tube, as many instabilities in tubes set in at values of Re_{Γ} which are between 1000 and 3000 [33].

The associated cylinder Reynolds number $\text{Re}_c = VD/\nu$ is calculated as $\text{Re}_c = \Gamma I_P T/(2\pi\nu) = \text{Re}_{\Gamma} I_P T/(2\pi)$ and varies in the range $\text{Re}_c \in (7.96, 119)$. Table I summarizes the control parameters used for all the runs in this paper. We note that while the values of Re_c used here appear substantially lower than those used in Ref. [32], the reference uses a chord-based Reynolds number, while this

 T	I_P	Re_{Γ}	Re _c
0.95	0.05	1000	7.56
0.95	0.15	1000	22.7
0.95	0.25	1000	37.8
0.95	0.05	2000	15.1
0.95	0.25	2000	75.6
0.95	0.05	3000	22.7
0.95	0.25	3000	113

TABLE I. Control parameters for the simulations of tube-wire interaction.

paper uses something analogous to a "thickness"-based Reynolds number leading to an order of magnitude discrepancy when comparing Re_c to the blade Reynolds number in Ref. [32].

Simulations are run using the open-source code AFiD [34]. This code uses energy-conserving second-order centered finite differences to spatially discretize the domain. Time marching is done through a third-order Runge-Kutta for the nonlinear terms and a second-order Crank-Nicolson for the viscous terms. The spatial resolution is taken as $624 \times 624 \times 1872$, meaning the grid is equally spaced in all directions, with a grid length $\sim 0.05\sigma_0 \approx 0.05D$. Resolution adequacy is checked by monitoring numerical dissipation to be less than 2%. To increase the confidence in the mesh adequacy, the Appendix shows a comparison between a coarser mesh, the mesh used in this paper, and a finer mesh for the $I_P = 0.25$, Re_{Γ} = 3000 case.

The cylinder is incorporated through the immersed boundary method (IBM) using a movingleast-squares formulation [35]. This algorithm has been used previously for turbulent simulations with rigid, moving, and deforming boundaries, and has been well validated [36-38]. The surface is discretized using 388 800 triangles. The triangle skewness is <0.3, and the average edge length is about 70% of the grid spacing. This ensures adequate resolution without consuming excessive computational resources [35].

III. FLOW DYNAMICS FOR TUBE-WIRE INTERACTION

A. The distinction between weak and strong vortex regimes

This subsection examines cases at $\text{Re}_{\Gamma} = 1000$, varying I_P to illustrate the difference between the strong and weak vortex regimes. The Q criterium [39,40] will be used to highlight the primary vortex. This is defined as

$$Q = -\frac{\partial u_j}{\partial x_i} \frac{\partial u_i}{\partial x_j} = \frac{1}{2}\omega^2 - \sigma^2,$$
(3)

where $\omega^2 = (\nabla \times \mathbf{u})^2$ is the vorticity magnitude squared, and $\sigma_{ij} = \frac{1}{2}(\partial_i u_j + \partial_j u_i)$ the strain tensor, with $\sigma^2 = \sigma_{ij}\sigma_{ij}$. *Q* represents the source term in the pressure-Poisson equation [41] and is useful for distinguishing rotation-dominated regions of the flow, which will tend to have positive values of *Q*, from strain-dominated regions, which have negative values of *Q*. Visualizing regions of positive *Q* through time will highlight the dynamics of the primary vortex, which is rotation-dominated, while leaving out the boundary layers. Later, $|\omega|$ visualizations will be presented to emphasize the dynamics of the wire's boundary layer and secondary vortices and its interaction with the primary vortex.

Figure 2 depicts a volume rendering of Q for a simulation within the strong vortex regime (Re_{Γ} = 1000, I_P = 0.05). As discussed in Ref. [17] and elsewhere, the vortex "pre-senses" the wire's presence prior to its approach to the core. Even before the wire penetrates the core, the central area of the vortex is disrupted. As the wire gradually traverses the core, numerous secondary structures coil around the primary tube; a detailed analysis of these occurs later during vorticity



FIG. 2. Volume rendering visualization of the time evolution of Q [Eq. (3)] for a simulation in the strong vortex regime with $\text{Re}_{\Gamma} = 1000$, $I_P = 0.05$. Regions in blue indicate regions of strongly positive Q. The wire is shown as a solid black object.

visualization. By the end of the interaction, little remains of the tube the wire passed through. The BVI process has significantly diminished the vortex's strength by its conclusion.

Figure 3 depicts a similar volume rendering of Q in the weak vortex regime (Re_{Γ} = 1000, I_P = 0.25). In this regime the primary vortex does not "pre-sense" the wire, instead interacting with it solely when it is in close proximity to the vortex core. The vortex swiftly traverses the tube, causing significant deformation to the core. This deformation generates upwards- and downwards-propagating waves. As time progresses, the wire completes its passage through the primary vortex tube, leaving behind a minor wake and a distorted tube. Surprisingly, despite the distortion, the "weak" vortex tube displays better resilience compared to the scenario involving a strong vortex. By the end of the BVI process, the waves have transversed the tube, and a distorted vortex tube remains behind. Counterintuitively this means that at Re_{Γ} = 1000, a vortex tube endures the "cut" of its core more effectively when encountering a faster moving cutting object. Primarily, this outcome emerges because the interaction with a slower wire is prolonged, allowing the vortex more time to engage with secondary vorticity originating from the wire. This extended interaction disrupts the primary vortex to a greater extent than the short interaction of the weak vortex regime.

Figure 4 highlights this result by showing Q visualizations for the four simulated cases at various I_P values after the wire's passage. In the case of the smallest I_P , the vortex experiences complete disruption. However, for $I_P = 0.15$ and 0.25, the vortex undergoes deformation, exhibiting propagating waves along its axis, yet retains some coherence. The $I_P = 0.1$ scenario lies between these extremes, displaying weakening in certain regions and deformation in others. Notably, the extent and nature of disruption in the primary core serve as markers distinguishing between weak



FIG. 3. Volume rendering visualization of the time evolution of Q [Eq. (3)] for a simulation in the weak vortex regime with $\text{Re}_{\Gamma} = 1000$, $I_P = 0.25$. Regions in blue indicate regions of strongly positive Q. The wire is shown as a solid black object.



FIG. 4. Volume rendering visualization of Q [Eq. (3)] after the wire has passed through the cylinder axis for Re_{Γ} = 1000 and I_P = 0.05, 0.1, 0.15, and 0.25 (left to right). Regions in blue indicate regions of strongly positive Q. The wire is shown as a solid black object.

and strong vortex regimes. The behavior observed in Fig. 4 suggests a smooth transition between these regimes taking place around $I_P \approx 0.1$. This was observed in experiments, and a value of $I_P \approx 0.08$ was given for the transition between regimes in Ref. [17].

B. The strong vortex regime $(I_P = 0.05)$

As the weak and strong vortex regimes show very different dynamics, they will be discussed separately, taking the lowest and highest values of I_P as paradigmatic for each regime. This subsection discusses the strong vortex regime, with $I_P = 0.05$.

To investigate the role of secondary vorticity in the vortex-wire interaction, Fig. 5 presents volumetric representations of the vorticity magnitude $|\omega|$ at the early stages of interaction for both $Re_{\Gamma} = 1000$ and $Re_{\Gamma} = 3000$. This stage of the interaction is rich in details, and the visualizations



FIG. 5. Volume rendering visualization of the time evolution of $|\omega|$ during the early stages for a simulation in the strong vortex regime $I_P = 0.05$ with $\text{Re}_{\Gamma} = 1000$ (top row) and $\text{Re}_{\Gamma} = 3000$ (bottom row). Regions shown in orange-brownish are regions of high vorticity.



FIG. 6. Vorticity magnitude $|\omega|$ at the z = 0 plane evolution just before impact in the strong vortex regime with $I_P = 0.05$ at $\text{Re}_{\Gamma} = 1000$ (top row) and $\text{Re}_{\Gamma} = 3000$ (bottom row). Snapshots are at $t\Gamma/D^2 = 940$ (left), 1140 (middle), and 1340 (right). Abbreviations: "DSV" detached secondary vorticity, "PVN" primary vortex necking.

align with the experimental images and descriptions in Ref. [17]: as the vortex approaches the tube, vorticity from its boundary layer detaches and begins to curl around the primary tube. This is marked with a blue arrow at $t\Gamma/D^2 = 740$. Even before the wire approaches the vortex core, this detached vorticity loops once around the main vortex, distorting the core, as seen at $t\Gamma/D^2 = 940$. This is because the wrapped vorticity induces velocities which commence a compression or "necking" of the core. This phenomenon is accentuated as the cylinder approaches the primary vortex. In the impact zone, the vortex and cylinder have oppositely signed vorticity, which, when pressed together, further thins the neck region. The primary vorticity increases to conserve circulation, and a region with increased vorticity magnitude becomes apparent at $t\Gamma/D^2 = 1140$ (marked with a black arrow) and $t\Gamma/D^2 = 1340$. At the same time, the detached vorticity propagates further along the vortex axis, nearly reaching the computational domain's periodic boundaries as seen in the last snapshot at $t\Gamma/D^2 = 1540$.

The observations detailed here mirror those documented experimentally in Ref. [17], while expanding upon the experimental groundwork in several crucial ways. First, Fig. 5 portrays vorticity three-dimensionally rather than photographs of dye visualization. This representation enables us to illustrate the propagation of secondary vorticity through the axis, a phenomenon which is not clear in the experiments, as well as showing the new vortex filaments produced. Second, we demonstrate that secondary vorticity arises without the need to introduce a source of turbulence, through either background flow or vibrations of the structure. Even starting from simple initial conditions, such as a Lamb-Oseen vortex, the secondary vorticity originating solely from the wire's boundary layer is enough to faithfully reproduce the significant disruptions in the primary vortex seen experimentally. Last, we note that our results are roughly symmetrical around the cutting plane in our results due to the absence of axial flow. This shows that the presence of a weak axial flow does not change the phenomenology of strong vortex BVI very much. We also note that for both Reynolds numbers, the flow phenomenology at the early stages is very similar, including the extent of wrapping and speed of propagation of secondary vorticity around the primary vortex, even if the vorticity itself appears much sharper due to the reduced viscosity. This facilitates the comparison to the experiment.

To focus on the early stages of the collision, Fig. 6 provides more qualitative data by representing the vorticity magnitude plotted at the central x-y plane at three time instants as the wire is



FIG. 7. Vorticity magnitude $|\omega|$ at the y = 0 midplane evolution for a simulation in the strong vortex regime at $I_P = 0.05$ with $\text{Re}_{\Gamma} = 1000$ (top row) and with $\text{Re}_{\Gamma} = 3000$ (bottom row). Snapshots are at $t\Gamma/D^2 = 740$ (left), 940 (middle), 1340 (right). Abbreviations: "DSV," detached secondary vorticity; "PVN," primary vortex necking.

approaching the tube. The early detachment of secondary vorticity is seen in the left panels of Fig. 6. Significant detachment and wrapping are present even if the wire is several diameters away from the core. The detached vorticity appears much sharper at $Re_{\Gamma} = 3000$, but the degree of wrapping is similar, as will be confirmed below. The middle panels show how the primary vortex has already been distorted before the wire begins to penetrate into the vortex. The wrapped vorticity induces velocities which begin to thin out the center of the vortex. This process is accentuated as the physical interaction begins: the core is further compressed producing a "necking" seen in the right panels. We note that at $Re_{\Gamma} = 1000$, even if the core is compressed, the vorticity intensification is smaller (or nonpresent). The reasons for this will be analyzed below.

Another view of the process is shown in Fig. 7, which presents the vorticity magnitude plotted at the central *x*-*z* plane. The snapshots in the left column confirm that the dynamics of the secondary vorticity detachment process are approximately Reynolds independent: the degree of wrapping is the same for both Reynolds numbers. The middle panel also shows how the vortex is distorted by the wrapped vorticity before the close interaction begins. Finally, the right panel shows the degree of "necking" present in the primary vortex as the body comes close to the vortex. The middle and right panels also reveal the main effect of Re_{Γ} : while at $\text{Re}_{\Gamma} = 1000$ the primary vortex remains more or less smooth, at $\text{Re}_{\Gamma} = 3000$ the vorticity profile is highly deformed as vorticity concentrates in thin sheets outside the center. We also note that the core has been substantially displaced from its original position.

However, these differences do not result in distinct outcomes of the BVI. Figure 8 shows the vorticity modulus for the late stages of the interaction, once the wire has cut the vortex core. As discussed in Sec. III A, in the strong vortex case the vortex remnants are much weaker than the original primary vortex. Instead, the flow appears to consist of a turbulent cloud with vortices oriented in many directions.

Another view of the remnants is shown in Fig. 9, which presents the vorticity at the y = 0 plane at a late stage. For $\text{Re}_{\Gamma} = 1000$, the remnants of the primary vortex can still be distinguished. It has been displaced, six to seven vortex radii from its initial position, and has a circulation which has been reduced to a quarter of its original strength. For $\text{Re}_{\Gamma} = 3000$, the remnants are extremely



FIG. 8. Volume visualization of the time evolution of $|\omega|$ during the late stages for a simulation in the strong vortex regime $I_P = 0.05$ with $\text{Re}_{\Gamma} = 1000$ (top row) and $\text{Re}_{\Gamma} = 3000$ (bottom row). Regions shown in orange-brownish are regions of high vorticity.

weak, and the circulation cannot be adequately measured as a single remnant cannot be identified. For both Re_{Γ} , the remnants are dissipated by viscosity a few time-units later.



FIG. 9. Vorticity magnitude $|\omega|$ at the y = 0 midplane evolution and $t\Gamma/D^2 = 2140$ for a simulation in the strong vortex regime at $I_P = 0.05$ with $\text{Re}_{\Gamma} = 1000$ (left) and with $\text{Re}_{\Gamma} = 3000$ (right). Abbreviation: "PVR," primary vortex remnant.



FIG. 10. Volume rendering visualization of the time evolution of $|\omega|$ during the early stages for a simulation in the weak vortex regime $I_P = 0.25$ with $\text{Re}_{\Gamma} = 1000$ (top row) and $\text{Re}_{\Gamma} = 3000$ (bottom row). Regions shown in orange-brownish are regions of high vorticity.

C. The weak vortex regime ($I_P = 0.25$)

Turning to the weak vortex regime, Fig. 10 shows volume plots of the vorticity magnitude $|\omega|$ for $I_P = 0.25$ and the two Reynolds numbers simulated at the early stages of the interaction. As in the strong vortex regime, two sources of vorticity are evident: the primary vortex and the wire's boundary layer. However, the overall flow evolution looks very different when compared to Fig. 5. The presence of a weaker vortex results in lower induced velocities on the wire. The wire's boundary layer is only beginning to lift off in the leftmost panels at $t\Gamma/D^2 = 200$ and 220, and has no time to curl around and deform the primary vortex before impact. This can be appreciated by contrasting with the visualization at $t\Gamma/D^2 = 740$ in Fig. 5 where the primary core is seen to have deformed on the right side.

Consequently, the wire's boundary layer does not play a significant role in this scenario. This lack of interaction before the wire approaches the core serves as a key characteristic of the weak vortex regime and has historically facilitated its modeling using inviscid methods. Instead, the physics in this regime are dominated by the displacement and deformation of the core, which is propagated through axial waves, as illustrated in the visualizations at $t\Gamma/D^2 = 240$ and onwards. The core is deformed due to the physical presence of the wire, which forces oppositely signed vorticity to approach the core in a process reminiscent of vortical reconnection [32] (marked by a blue arrow in Fig. 10). This displaces the vortex tube and stretches it into a thin "neck" until the core is eventually "cut." This process is more marked in the Re_{\substack} = 3000 case, where the vorticity is intensified substantially.

Simultaneously, the core deformations close to the wire propagate through waves along the primary tube's axis, as was previously observed in Fig. 3. These waves do not substantially modify the radius of the primary vortex, a phenomenon observed in the simulations and experiments of weak-vortex BVI with axial flow in Ref. [22]. Here there is a crucial difference from the experiments with axial flow which introduces a notable asymmetry in wave propagation. Our simulations instead show a symmetrical propagation originating from the cylinder impact zone.



FIG. 11. Vorticity magnitude $|\omega|$ at the z = 0 plane evolution just before impact in the weak vortex regime with $I_P = 0.25$ at $\text{Re}_{\Gamma} = 1000$ (top row) and $\text{Re}_{\Gamma} = 3000$ (bottom row). Snapshots are at $t\Gamma/D^2 = 220$ (left), 240 (middle), and 260 (right). Abbreviations: "PVN," primary vortex necking; "DSV," detached secondary vorticity.

As in the previous section, a closer look is provided by looking at two-dimensional cuts. Figure 11 represents the vorticity magnitude plotted at the central x-y plane at three time instants as the wire is approaching the tube for the two Reynolds numbers, and Fig. 12 shows the vorticity through the x-z plane for the same time instants and the two Reynolds numbers.

The Reynolds number dependence is first reflected as thinner and more intense boundary layers. For the first panel at $t\Gamma/D^2 = 220$, the secondary vorticity has not had time to completely wrap around the core and is not visible in the z = 0 cut. The primary vortex has been deformed by the



FIG. 12. Vorticity magnitude $|\omega|$ at the y = 0 midplane evolution for a simulation in the weak vortex regime at $I_P = 0.25$ with Re_{Γ} = 1000 (top row) and Re_{Γ} = 3000 (bottom row). Snapshots are at $t\Gamma/D^2 = 220$ (left), 240 (middle), and 260 (right). Abbreviations: "DSV," detached secondary vorticity; "OV," opposite vorticity; "PV," primary vortex.



FIG. 13. Volume visualization of the time evolution of $|\omega|$ during the late stages for a simulation in the weak vortex regime $I_P = 0.25$ with $\text{Re}_{\Gamma} = 1000$ (top row) and $\text{Re}_{\Gamma} = 3000$ (bottom row). Regions shown in orange-brownish are regions of high vorticity.

presence of the wire, which is pushing oppositely signed vorticity towards it. Notably, although the vorticity has less time to detach, it exhibits a higher value than in Fig. 7. This can be attributed to the nondimensionalization using Γ/σ as a velocity scale, where higher values of I_P lead to greater wire velocities, resulting in larger vorticity values for a boundary layer when normalized using $\Gamma-\sigma$ units.

By $t\Gamma/D^2 = 240$, the secondary vorticity has had time to begin to wrap around the primary vortex. As suggested by Ref. [22], the primary vortex can entrain a small quantity of vorticity from the boundary layer during the interaction. However, it is almost indiscernible for the panel at Re_{\Gamma} = 1000, and for Re_{\Gamma} = 3000, while visible, it is eclipsed by the vorticity intensification due to the core stretching and deformation. The intensification of vorticity is particularly marked for the higher Reynolds number, a phenomenon which is again reminiscent of reconnection, where increases in Reynolds number produce increasingly thinner vortex sheets of large vorticity magnitudes [42]. However, once viscosity takes over, annihilating or reconnecting the primary and opposite vorticity, this does not seem to have a bearing on the rest of the flow. This can explain why flow models which neglect secondary vorticity can adequately capture the interaction within this regime: the phenomenology is primarily driven by the deformations induced by the moving object, rather than the annihilation of vorticity.

By the latest time, $t\Gamma/D^2 = 260$, the cylinder is cutting the vortex, and almost no vorticity is visible in the plots close to the wire. By returning to the three-dimensional visualization of Fig. 10, it is clear that the cut has taken place. In these simulations, some of the phenomena usually attributed to the bending regime can be observed, i.e., deformations of the primary vortex, and the stretching (or "necking") of the core near the impact location. This hints at the fact that the transition between strong vortex and bending regimes, which is outside the scope of this paper, will also be smooth.

The flow phenomenology after the wire has completely transversed the vortex is shown in full three-dimensional visualizations in Fig. 13 and in a selected cut in the y = 0 plane in Fig. 14. As the wire passes through the vortex, it leaves behind a wake. The waves reach the limit of the

FIG. 14. Vorticity magnitude $|\omega|$ at the y = 0 midplane evolution and $t\Gamma/D^2 = 400$ for a simulation in the weak vortex regime at $I_P = 0.25$ with Re_{Γ} = 1000 (left) and with Re_{Γ} = 3000 (right).

computational domain around $t\Gamma/D^2 = 400$. The primary vortex reconstitutes itself, even if vortex loops are seen to detach from it, especially at high Reynolds number. The resulting distribution of vorticity is different from the original Gaussian, as it is now concentrated in a thin sheet and displaced to the right by a few vortex cores. Despite this, a comparison of the initial and final circulation of the vortex for $\text{Re}_{\Gamma} = 1000$ reveals only a 5% loss throughout the entire process, while at $\text{Re}_{\Gamma} = 3000$ the core has lost 15% of its original circulation, very likely to the small-scale structures which arise and dissipate. This showcases the vortex's ability to withstand the interaction significantly better than the case at $I_P = 0.05$, which suffered a 75% loss of its original circulation at $\text{Re}_{\Gamma} = 1000$ and was not significantly detectable at $\text{Re}_{\Gamma} = 3000$.

The loss of circulation after the interaction has taken place (Γ_f) can be quantified for all cases at Re_{\(\Gamma\)} = 1000. Figure 15 shows Γ_f/Γ_0 , Γ_0 being the original circulation of the vortex. The amount of circulation in the vortex by the end of the interaction monotonically increases with I_P in the range studied. We note that this trend cannot persist, as the circulation must saturate at $\Gamma_f \approx \Gamma_0$ at higher values of I_P , as it is not expected that the wire imparts additional circulation to the vortex.

From these findings, it becomes evident that the primary control parameter in BVI is the impact parameter I_P . Varying the Reynolds number results in vorticity intensification and amplified distortion along the primary vortex, driven by either secondary vorticity or wave propagation. Also, at the later stages finer flow structures and turbulent clouds appear at higher Reynolds number, due to the increased instability of the vortices. However, these alterations do not significantly impact the fundamental collision features, at least up to Re_{Γ} = 3000. Furthermore, our simulations

FIG. 15. Loss of circulation during the interaction quantified as the ratio between final and initial circulations as a function of I_P for the four cases simulated at $\text{Re}_{\Gamma} = 1000$.

FIG. 16. Left: Pressure coefficient along the cylinder for Re_{Γ} = 1000, I_P = 0.25 at time $t\Gamma/D^2$ = 160. Middle: Force coefficient along the cylinder axis for the same case. Left: Evolution of drag coefficient with time for the full cylinder (orange) and half the cylinder (blue).

successfully reproduce and provide reasoning for most experimental outcomes, demonstrating that even in the absence of turbulence, the phenomenology of weak and strong BVI can be faithfully replicated.

IV. FORCES ON THE WIRE

A. Initial remarks

This section discusses an area which is much less analyzed in BVI: measurement of the force on the cylinder. However, an adequate definition of the force is not simple. Two things have to be taken into account: first, the force coming from BVI is a deviation from a constant drag which resists the movement of the cylinder at a constant velocity in a quiescent flow. Second, this deviation will be localized to the point of interaction, and if the forces for a quasi-infinite cylinder or blade undergoing BVI are added up, they would be approximately the same for those where there is no BVI, as the constant drag would overwhelm the forces from BVI.

To elaborate on this, the left panel of Fig. 16 shows the force coefficient distribution $C_{f,x} = f_x/(\frac{1}{2}\rho[\Gamma/\sigma]^2)$ on the cylinder for the weak vortex case ($I_P = 0.25$, $\text{Re}_{\Gamma} = 1000$) before the cylinder has reached the vortex core. We note that this representation is highly distorted as the cylinder is very long. However, it is useful to note a few things. First, the distribution is symmetric around $\theta = 0$, which corresponds to the leading edge of the cylinder. This is a natural product of the symmetry of the system. Second, while the region of strongest interaction is closely localized to the impact point, as could be expected, the return to asymptotic values is rather slow. Finally, a rather small region of the cylinder sees an overall force in the direction of motion of motion of the cylinder ($C_{f,x} > 0$). This is due to the fact that locally the flow direction can be reversed when the velocity induced due to the vortex is larger than the cylinder motion.

These results are emphasized in the middle panel of Fig. 16, which shows the force on the cylinder along its axis $C_f(z)$. The baseline case where no vortex is present is shown as a solid line. The forces can be seen to be somewhat symmetrical around z = 0, i.e., the collision point. The overall force again shows positive values, coincident with regions where the velocity the cylinder sees is reversed. Finally, a long tail decay is observed: the forces do not return to the baseline case even at the edges of the cylinder. We note that the simulations yield qualitatively similar curves to those presented in the simplified models of Ref. [23], with the presence of a suction peaks and regions of increased pressure. However, comparison is difficult due to the simplifying assumptions made of the flow and the different parameter regimes investigated.

As a final step, the average drag coefficient can be calculated. If the entire extent of the cylinder is used, a value of $C_F = 0.449$ is obtained. If the cylinder half closest to the vortex is used, and the two quarters close to the edges are discarded, a value of $C_F = 0.428$ is obtained. While both values are almost indistinguishable from the baseline value of $C_F = 0.436$, one deviates upwards

FIG. 17. Force coefficient along the cylinder axis at several time instances. Left column is cases at $I_P = 0.05$, while right column is cases at $I_P = 0.25$. Top row is at $\text{Re}_{\Gamma} = 1000$, while bottom row is at $\text{Re}_{\Gamma} = 3000$. Symbols: left column ($I_P = 0.05$), light to dark: $t\Gamma/D^2 = 700$ (light red), 1100 (dark red), 1500 (black), 1900 (dark blue), and 2300 (blue). Right column ($I_P = 0.25$): $t\Gamma/D^2 = 160$ (light red), 200 (dark red), 240 (black), 280 (dark blue), and 320 (light blue). The horizontal thin line represents the baseline value.

and the other downwards. These problems of adequately calculating C_F are showcased in the right panel of Fig. 16, which shows the time evolution of C_F calculated using the full cylinder and half the cylinder. Both values show similar trends of increasing and decreasing during stages of the collision. However, they can be either above or below the baseline value depending on the cylinder segment considered for calculation. In any case, the deviations are mild and reach only a maximum of 20% drag increase for the half-cylinder calculation. We also note that some small oscillations are present in the force calculation, which have a frequency close to the natural frequency of the cylinder D/V.

Due to the problems of adequately defining C_F , we choose not to represent it for other cases and will show only the evolution of $C_f(z)$ for a few select cases below.

B. Results

In this section $C_f(z)$ is calculated for a few selected time instants for four characteristic cases. These are two cases for each of the strong ($I_P = 0.05$) and weak ($I_P = 0.25$) vortex regimes, at the two Reynolds numbers simulated. These are shown in Fig. 17.

The discussion starts with the case with $I_P = 0.05$ and $\text{Re}_{\Gamma} = 1000$. The curves are relatively similar to those shown in the previous subsection. The curves deviate from the baseline value, and they are roughly antisymmetrical around the collision point z = 0. The deviations increase as the cylinder approaches the vortex, but they reach a maximum at distinct times: the peak force inversion ($C_f > 0$) happens earlier than the peak suction ($C_f < 0$). In the same manner, the values in the force inversion side return to the baseline values earlier than in the suction side. This is probably due to the fact that the cylinder boundary layer is being lifted up earlier from the force inversion side, so the interaction happens first on this half of the cylinder. Finally, the deviations from the baseline value are much more significant in this case: after all, the vortex is strong. This means that it will create forces on the cylinder which are much larger than those originating solely from the ordinary drag force.

Turning to the case with $I_P = 0.25$ and $\text{Re}_{\Gamma} = 1000$, the behavior of the force has some similarities but also some differences when compared to the earlier case. The main difference is that the forces in the regions of large positive C_f fluctuate much less: it attains a maximum value and remains close to that maximum value for most of the interaction. This suggests a probable absence of significant boundary layer liftoff. On the cylinder half experiencing force reversal, temporal variations are more pronounced yet localized due to the weaker vortex. In essence, little more can be extrapolated from these observations.

Finally, the bottom panels of Fig. 17 show the Reynolds number dependence of the force, which can be contrasted to the top row. There is a relatively weak dependence of behavior on Reynolds number. The primary difference lies in the smaller baseline values of C_F , yet both curves exhibit analogous behavior. Close to the impact point, spatial oscillations in $C_f(z)$ become more discernible, likely owing to the influence of smaller and more intense vortical structures on the cylinder.

In essence, our investigation into the forces on the cylinder reveals that while locally BVI can generate substantial forces, especially within the strong vortex regime, these forces tend to average out across the cylinder halves facing either positive or negative vortex velocities. Consequently, the average force remains relatively consistent.

V. SUMMARY AND CONCLUSIONS

Results from direct numerical simulations of normal body-vortex interaction using a thin cylinder have been presented. The focus was on the transition and the differences in the weak and strong vortex regimes. Three findings were reported:

(1) The simulations confirm that the transition between weak and strong vortex regimes is a smooth transition controlled by I_P and primarily driven by the capacity of the cylinder's boundary layer to detach and wrap around the primary vortex before impact takes place.

(2) In the strong vortex regime, this detached, or secondary, vorticity propagates along the vortex's axis, interacting with it and heavily debilitating it, with the result that by the end of the interaction the vortex has almost dissipated.

(3) In the weak vortex regime, the cylinder interacts only with the vortex once the body reaches the tube. It then deforms it, propagating waves along the axis. The behavior of the waves was significantly different from those observed when axial flow is present [17].

The Reynolds number dependence of the flow was also investigated by simulating two cases at $\text{Re}_{\Gamma} = 3000$. The effect of Re_{Γ} on the system was reflected in two aspects: at higher Re_{Γ} the boundary layers are thinner and have more vorticity, and vortices are more unstable and prone to breaking down. Three main results were found:

(1) In the strong vortex regime, qualitative differences in the flow can be seen before the impact of the cylinder. The secondary vorticity, which is now concentrated in smaller and more intense structures, has a stronger interaction with the primary tube leading to more deformations in the neck area. These differences are absent from the weak vortex regime due to the reduced importance of secondary vorticity originating from the boundary layer in the early stages of the interaction.

(2) In the weak vortex regime, the stretching and necking of the primary vortex due to the cylinder forcing oppositely signed vorticity close to it is enhanced. However, this does not result in significantly different outcomes.

(3) After the interaction, the vortex remnants tend to break down rather than recompose again into a primary tube. The simulations indicate that even in the absence of background perturbations, the flow left behind the cylinder will be turbulent.

FIG. 18. Vorticity magnitude for the $I_P = 0.25$ and $\text{Re}_{\Gamma} = 3000$ case at $t\Gamma^2/D = 220$ and z = 0 (top), at $t\Gamma^2/D = 220$ and y = 0 (middle) and at $t\Gamma^2/D = 400$ and y = 0 (bottom). The left, middle, and right columns show the coarse, medium, and fine resolutions, respectively.

The results presented in this manuscript qualitatively reproduce existing experimental data [17,22] when the presence of axial flow is not important. The simulations also confirm that the adequate modeling of the boundary layer is necessary to capture the intricacies of the strong vortex regime.

The distribution of pressure forces on the cylinder was also reported. The results show local deviations from the baseline value (considered as the one where no vortex is present), which are concentrated around the impact point. The deviations are large, to the point that the force can locally reverse direction, with the cylinder feeling a force in the direction of the motion. However, in general the forces average out on both sides of the interaction, and the integrated force coefficients are very similar to the baseline value.

As the introduction stated, the starting point of the study was on "cutting" vortex lines. However, these simulations showed that the phenomenology is much more complicated, especially for the strong vortex case, and that the interaction deviates from what one could expect from ideal vortex models. While the scope of the paper has been restricted to normal BVI, with minor modifications the code could be extended to simulating parallel and grazing BVI, and regimes in between. Another possible extension of this work is the question of what happens when a vortex ring and not a vortex tube is cut. Very few studies of cutting a vortex ring with a wire have been made [4]. Structures such as "elephant ears" have been reported, which to our knowledge have not been related in the wider literature on BVI.

FIG. 19. Force coefficient at $t\Gamma^2/D = 220$ for the $I_P = 0.25$ and $\text{Re}_{\Gamma} = 3000$ case for the three grids. A horizontal line shows the baseline value when no vortex is present. Symbols: blue, coarse grid; orange, medium grid; green, fine grid.

ACKNOWLEDGMENTS

We thank the Research Computing Data Core (RCDC) at the University of Houston for providing computing resources.

APPENDIX: GRID DEPENDENCE STUDIES

To assess the adequateness of the grid, two additional simulations of the case with $I_P = 0.25$ and $\text{Re}_{\Gamma} = 3000$ were conducted with coarse and fine grids. These parameters were chosen as this simulation case was the most unfavorable as the cylinder Reynolds number was highest ($\text{Re}_c =$ 113), so it provides an upper limit for the problems with the grid. The coarse grid is obtain by halving the resolution in each spatial direction, i.e., $312 \times 312 \times 936$ points, while the fine grid is obtained by increasing the resolution by a factor $1.5 \times$ in every direction, i.e., $936 \times 936 \times 2808$.

Figure 18 shows the vorticity magnitude for the three grids alongside two plane cuts. The first two rows show the variable at $t\Gamma^2/D = 220$: this time instant was chosen because it is close to peak dissipation, the tube has been stretched and displaced, and vorticity has intensified, which makes it challenging to resolve. Cuts are shown for both z = 0, i.e., across the wire, and y = 0, i.e., across the vortex core. Clear signatures of underresolution can be seen for the coarse mesh: there are marks of numerical dispersion behind the cylinder in the top-left panel, as well as deformations in the vortex core not seen for the medium and fine grids. While some differences to appear between the medium and fine grids, they are not as significant. This can be quantified by measuring the total circulation in the vortex cores shown in the right column: the coarse grid has a circulation which is 25% lower than the medium grid, while the fine grid has a circulation which is 2% higher than the medium grid. This confirms our assessment that the medium grid can be considered is adequate to within 2%, but that further coarsening gives results which are inaccurate due to numerical dispersion.

Finally, the bottom row shows the vorticity magnitude at $t\Gamma^2/D = 400$ through the y = 0 plane. This is after the interaction has finished, so the differences between cases should be maximum. Once again, the medium and fine grids show similarities, while the coarse grid shows a picture that is somewhat similar to the other two grids, yet with vorticity concentrated in different points.

A further check was conducted by assessing the variation in the force on the wire. Figure 19 shows $C_f(z)$ at the same time instant for the three meshes. Again, differences can be seen between the three cases, but the difference between the coarse grid and the medium and fine grids is much larger than the difference between medium and fine grids. The integrated force coefficients are

 $C_F = 0.24$, $C_F = 0.31$, and $C_F = 0.33$ for the coarse, medium, and fine grids, respectively. This shows that the quantities on the wire are slightly more sensitive than the vorticity statistics. However, the curves all show similar phenomenology, with a strong suction peak, and a strong increase close to the vortex core. Hence, we do not expect the results from the medium mesh to miss any important physics, while the coarse mesh shows C_f dropping below the baseline value, questioning its use.

To summarize, the medium mesh shows small discrepancies when compared to a finer mesh, especially when measuring the force on the wire. However, the overall phenomenology is well captured, unlike what is seen for the coarser mesh, where the flow topology is changed. We also reiterate that this is the most unfavourable instance, and yet the medium grid is able to produce reasonable results with a much lower computational cost.

- [1] P. G. Saffman, *Vortex Dynamics* (Cambridge University Press, New York, 1995).
- [2] S. Kida and M. Takaoka, Vortex reconnection, Annu. Rev. Fluid Mech. 26, 169 (1994).
- [3] J. Marshall and R. Yalamanchili, Vortex cutting by a blade. II Computations of vortex response, AIAA J. 32, 1428 (1994).
- [4] T. Naitoh, B. Sun, and H. Yamada, A vortex ring travelling across a thin circular cylinder, Fluid Dyn. Res. 15, 43 (1995).
- [5] X. Liu and J. Marshall, Blade penetration into a vortex core with and without axial core flow, J. Fluid Mech. 519, 81 (1999).
- [6] D. Rockwell, Vortex-body interactions, Annu. Rev. Fluid Mech. 30, 199 (1998).
- [7] A. Palau, R. Dinner, J. H. Durrell, and M. G. Blamire, Vortex breaking and cutting in type II superconductors, Phys. Rev. Lett. 101, 097002 (2008).
- [8] A. Glatz, V. K. Vlasko-Vlasov, W. K. Kwok, and G. W. Crabtree, Vortex cutting in superconductors, Phys. Rev. B 94, 064505 (2016).
- [9] A. Jameson, The analysis of propeller-wing flow interaction, NASA Symposium Proceedings SP-228 (1970), pp. 721–749.
- [10] D. P. Witkowski, A. K. Lee, and J. P. Sullivan, Aerodynamic interaction between propellers and wings, J. Aircr. 26, 829 (1989).
- [11] J. G. Leishman and A. Bagai, Challenges in understanding the vortex dynamics of helicopter rotor wakes, AIAA J. 36, 1130 (1998).
- [12] H. Y. Yung, Rotor blade-vortex interaction noise, Prog. Aerospace Sci. 36, 97 (2000).
- [13] E. Amet, T. Maître, C. Pellone, and J.-L. Achard, 2D Numerical simulations of blade-vortex interaction in a Darrieus turbine, J. Fluids Eng. 131, 111103 (2009).
- [14] M. Wilder and D. Telionis, Parallel blade–vortex interaction, J. Fluids Struct. 12, 801 (1998).
- [15] A. R. Ahmadi, An experimental investigation of blade-vortex interaction at normal incidence, J. Aircr. 23, 47 (1986).
- [16] J. Kim and N. Komerath, Summary of the interaction of a rotor wake with a circular cylinder, AIAA J. 33, 470 (1995).
- [17] S. Krishnamoorthy and J. Marshall, Three-dimensional blade–vortex interaction in the strong vortex regime, Phys. Fluids 10, 2828 (1998).
- [18] A. Gossler and J. Marshall, Simulation of normal vortex-cylinder interaction in a viscous fluid, J. Fluid Mech. 431, 371 (2001).
- [19] M. Felli, C. Roberto, and G. Guj, Experimental analysis of the flow field around a propeller-rudder configuration, Exp. Fluids 46, 147 (2009).
- [20] D. Peng and J. W. Gregory, Vortex dynamics during blade-vortex interactions, Phys. Fluids 27, 053104 (2015).

- [21] M. Felli, Underlying mechanisms of propeller wake interaction with a wing, J. Fluid Mech. 908, A10 (2021).
- [22] J. S. Marshall and S. Krishnamoorthy, On the instantaneous cutting of a columnar vortex with non-zero axial flow, J. Fluid Mech. 351, 41 (1997).
- [23] H. Affes and A. Conlisk, Model for rotor tip vortex-airframe interaction. I. Theory, AIAA J. 31, 2263 (1993).
- [24] D. W. Moore and P. G. Saffman, The motion of a vortex filament with axial flow, Philos. Trans. R. Soc. London A 272, 403 (1972).
- [25] T. Lundgren and W. Ashurst, Area-varying waves on curved vortex tubes with application to vortex breakdown, J. Fluid Mech. 200, 283 (1989).
- [26] J. Marshall, A general theory of curved vortices with circular cross-section and variable core area, J. Fluid Mech. 229, 311 (1991).
- [27] H. Affes, A. Conlisk, J. Kim, and N. Komerath, Model for rotor tip vortex-airframe interaction. II. Comparison with experiment, AIAA J. 31, 2274 (1993).
- [28] T. Doligalski and J. Walker, The boundary layer induced by a convected two-dimensional vortex, J. Fluid Mech. 139, 1 (1984).
- [29] A. Luton, S. Ragab, and D. P. Telionis, Interaction of spanwise vortices with a boundary layer, Phys. Fluids 7, 2757 (1995).
- [30] L. L. Van Dommelen and S. J. Cowley, On the Lagrangian description of unsteady boundary-layer separation. Part 1. General theory, J. Fluid Mech. 210, 593 (1990).
- [31] H. Affes, Z. Xiao, and A. Conlisk, The boundary-layer flow due to a vortex approaching a cylinder, J. Fluid Mech. **275**, 33 (1994).
- [32] D. C. Saunders and J. S. Marshall, Vorticity reconnection during vortex cutting by a blade, J. Fluid Mech. 782, 37 (2015).
- [33] T. Leweke, S. Le Dizes, and C. H. Williamson, Dynamics and instabilities of vortex pairs, Annu. Rev. Fluid Mech. 48, 507 (2016).
- [34] E. P. van der Poel, R. Ostilla-Mónico, J. Donners, and R. Verzicco, A pencil distributed finite difference code for strongly turbulent wall-bounded flows, Comput. Fluids 116, 10 (2015).
- [35] V. Spandan, V. Meschini, R. Ostilla-Mónico, D. Lohse, G. Querzoli, M. D. de Tullio, and R. Verzicco, A parallel interaction potential approach coupled with the immersed boundary method for fully resolved simulations of deformable interfaces and membranes, J. Comput. Phys. 348, 567 (2017).
- [36] V. Spandan, R. Verzicco, and D. Lohse, Physical mechanisms governing drag reduction in turbulent Taylor–Couette flow with finite-size deformable bubbles, J. Fluid Mech. **849**, R3 (2018).
- [37] X. Zhu, R. A. Verschoof, D. Bakhuis, S. G. Huisman, R. Verzicco, C. Sun, and D. Lohse, Wall roughness induces asymptotic ultimate turbulence, Nat. Phys. 14, 417 (2018).
- [38] P. Berghout, X. Zhu, D. Chung, R. Verzicco, R. J. Stevens, and D. Lohse, Direct numerical simulations of Taylor–Couette turbulence: The effects of sand grain roughness, J. Fluid Mech. 873, 260 (2019).
- [39] J. C. Hunt, A. A. Wray, and P. Moin, in Eddies, Streams, and Convergence Zones in Turbulent Flows, Studying Turbulence Using Numerical Simulation Databases, *Proceedings of the 1988 Summer Program* (1988), https://ntrs.nasa.gov/citations/19890015184.
- [40] C. Meneveau, Lagrangian dynamics and models of the velocity gradient tensor in turbulent flows, Annu. Rev. Fluid Mech. 43, 219 (2011).
- [41] A. Pumir, A numerical study of pressure fluctuations in three-dimensional, incompressible, homogeneous, isotropic turbulence, Phys. Fluids **6**, 2071 (1994).
- [42] F. Hussain and K. Duraisamy, Mechanics of viscous vortex reconnection, Phys. Fluids 23, 021701 (2011).