

High-order lattice Boltzmann models for wall-bounded flows at finite Knudsen numbers

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We analyze a large number of high-order discrete velocity models for solving the Boltzmann–Bhatnagar–Gross–Krook equation for finite Knudsen number flows. Using the Chapman–Enskog formalism, we prove for isothermal flows a relation identifying the resolved flow regimes for low Mach numbers. Although high-order lattice Boltzmann models recover flow regimes beyond the Navier–Stokes level, we observe for several models significant deviations from reference results. We found this to be caused by their inability to recover the Maxwell boundary condition exactly. By using supplementary conditions for the gas–surface interaction it is shown how to systematically generate discrete velocity models of any order with the inherent ability to fulfill the diffuse Maxwell boundary condition accurately. Both high-order quadratures and an exact representation of the boundary condition turn out to be crucial for achieving reliable results. For Poiseuille flow, we can reproduce the mass flow and slip velocity up to the Knudsen number of 1. Moreover, for small Knudsen numbers, the Knudsen layer behavior is recovered.

DOI: [10.1103/PhysRevE.94.013304](https://doi.org/10.1103/PhysRevE.94.013304)**I. INTRODUCTION**

Fluid flow at very small scales has gained an increasing amount of attention recently due to its relevance for engineering applications in micro- and nanotechnologies [1,2], e.g., microelectromechanical systems (MEMS) and porous media. The characteristic length scale l_0 of the corresponding geometries is in the range of the mean free path length λ of the gas molecules. Such flows are often isothermal and characterized by extremely small Mach numbers. Nevertheless, these flows can become compressible because of considerable pressure gradients caused by viscous effects [3]. Based on the Chapman–Enskog (CE) expansion [4], the Knudsen number, defined as $\text{Kn} = \lambda/l_0$, can be considered a measure for the deviation of the flow behavior from thermodynamic equilibrium. For sufficiently large Kn nonequilibrium effects become important and the validity of the Navier–Stokes equation breaks down. In particular, the gas–surface interaction is very complex and cannot be described by the usual no-slip boundary condition. Within the Knudsen layer [5] the notion of the fluid as a continuum is no longer valid.

A fundamental description of hydrodynamics beyond the Navier–Stokes equation is provided by the Boltzmann equation valid for all values of Kn and all flow regimes [6]. Accurate simulations of finite Kn flows are achieved by low-level methods solving the Boltzmann equation numerically, e.g., Direct Simulation Monte Carlo (DSMC), which is traditionally used for high-Mach-number flows [7]. However, the application of the DSMC method to low-Mach-number microflows requires a large number of samples to reduce statistical errors and is computationally very time consuming.

Therefore, reduced-order models of the Boltzmann equation, such as the lattice Boltzmann (LB) approach, have become attractive alternatives [8–11]. The LB method is based on a reduction of the molecular velocities to a discrete velocity set in configuration space. Standard LB models, e.g., the $D3Q19$ model with 19 discrete velocities, were developed to describe

the Navier–Stokes fluid dynamics. Nowadays, they provide a well-established methodology for the computational modeling of various flow phenomena [12]. Furthermore, the LB method achieves promising results for microflow simulations [13–18]. In particular, the diffuse Maxwell boundary condition for the gas–surface interaction can be implemented at a kinetic level [19–24].

A systematic extension of the LB method to high-order hydrodynamics beyond the Navier–Stokes equation has been derived by Shan *et al.* [25]. These models are based on an expansion of the velocity space using Hermite polynomials in combination with appropriate Gauss–Hermite quadratures. First analytical solutions of the high-order LB model $D2Q16$ were presented by Ansumali *et al.* [26] for Couette flow and by Kim and Pitsch [27] for Poiseuille flow. The collection of LB models in the literature has grown successively (see, e.g., Refs. [28,29]), but makes no claim to be complete in any sense. It was shown [26,27] that a higher Gauss–Hermite quadrature order significantly improves the simulation accuracy for finite Kn flows compared to standard LB models. However, several studies [30–33] of high-order LB methods indicate that the accuracy for finite Kn flows does not monotonically increase with a higher Gauss–Hermite order and sensitively depends on the chosen discrete velocity set. Generally, it was observed that discrete velocity sets with an even number of velocities perform better than sets with the same Gauss–Hermite order but an odd number of velocities. Moreover, there are some LB models which show considerable deviations from DSMC results for finite Kn despite a high Gauss–Hermite quadrature order. It has been suggested that this is caused by gas–surface interactions [33,34]. The implementation of the diffuse Maxwell boundary condition using Gauss–Laguerre off-lattice quadrature models in Ref. [35] shows good results for Couette flow up to $\text{Kn} = 0.5$. By using an alternative framework, a high-order LB model with only 27 discrete velocities has been developed by Yudistiawan *et al.* [36] and it was shown that the corresponding moment system is quite similar to Grad’s 26-moment system. This off-lattice $D3Q27$ model is able to represent both Knudsen layer effects and the Knudsen minimum for Poiseuille flow.

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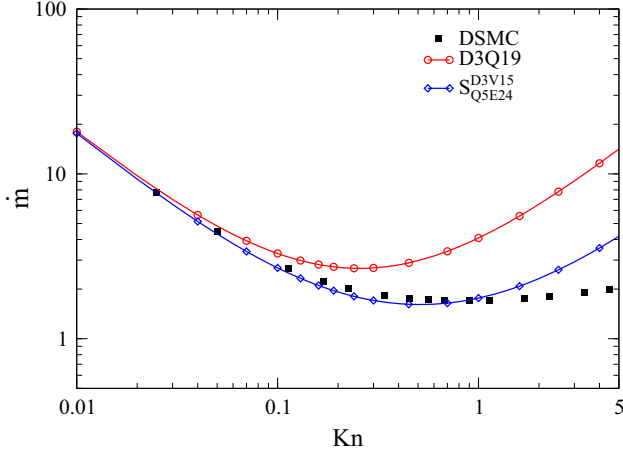


FIG. 1. Normalized mass flow rate for Poiseuille flow. Both LB models $D3Q19$ and S_{Q5E24}^{D3V15} are of quadrature order 5 but differ in the accuracy of the wall moment W_{zxx} and, consequently, in their capability to recover the diffuse Maxwell boundary condition.

As an introductory example, we consider the most commonly used LB model $D3Q19$ accompanied by the diffuse Maxwell boundary condition for unknown distribution functions at solid walls. It is well-known that the $D3Q19$ model shows deviations from reference results (e.g., DSMC) for increasing Kn . As shown in Fig. 1, the normalized mass flow rate for Poiseuille flow becomes inaccurate for $Kn \gtrsim 0.05$. One reason for this deficiency is the inability of the $D3Q19$ velocity set to represent the diffuse Maxwell boundary condition accurately. This drawback can be measured by half-space integrals at the wall [34]; we show in this paper that the integral W_{zxx} [cf. Eq. (62)] is most relevant. The standard $D3Q19$ model evaluates the latter yielding an error of 28%, while an alternative LB model S_{Q5E24}^{D3V15} (see Appendix A) with 15 velocities and the same quadrature order of 5 yields an error of only 2%. Figure 1 shows that the result for the mass flow rate of the latter model is much more accurate compared to that of $D3Q19$. This indicates that the LB model's representation of the diffuse Maxwell boundary condition requires high precision. Furthermore, due to the strong restriction of the configuration space, the standard LB velocity models capture only the first order of the CE expansion and therefore the applicability for describing finite Kn flows is limited [37]. Both models shown in Fig. 1 are not able to reproduce the Knudsen layer at solid walls where strong nonequilibrium effects are relevant. The description of finite Knudsen number flows must incorporate high-order flow regimes. It is therefore desirable to obtain high-order LB models which additionally are capable of representing the diffuse Maxwell boundary condition accurately.

In this work we systematically develop and investigate a large number of new high-order Gauss-Hermite LB models (on-lattice), which fulfill a constraint guaranteeing an accurate implementation of the diffuse Maxwell boundary condition. Consequently, these models ensure vanishing errors of the relevant half-space integrals. We prove a theorem using the CE expansion which specifies for low-Mach-number flows the recovered flow regimes beyond the Navier-Stokes regime depending on the Gauss-Hermite quadrature order. First sim-

ulation results for Poiseuille flow at finite Kn show that those high-order LB models which recover the diffuse Maxwell boundary condition exactly achieve excellent agreement with DSMC results. In particular, these models are able to describe Knudsen layer effects at solid walls. We emphasize that we do not resort to slip-boundary models in order to achieve these results. Finally, we recommend a preferred LB model with 96 discrete velocities and a seventh-order Gauss-Hermite quadrature order.

The paper is organized as follows. In Sec. II, a brief review of LB models and the generation of high-order discrete velocity models is presented. In Sec. III, we derive the low-Mach-number theorem mentioned above. In Sec. IV, we present the systematic generation of high-order LB models with the inherent ability to capture the diffuse Maxwell boundary condition accurately. In Sec. V, we discuss the numerical implementation based on second-order accurate discretization in space and time. In Sec. VI, we compare first simulation results for Poiseuille flow at finite Kn with DSMC data and discuss the ability of the new LB models to describe the Knudsen layer behavior at solid walls. In Sec. VII, a summary of the major conclusions is given.

II. DEFINITION OF LATTICE BOLTZMANN MODELS

We consider the one-particle velocity distribution function $f(\xi)$ governed by the Boltzmann equation with the Bhatnagar-Gross-Krook (BGK) collision operator,

$$(\partial_t + \xi \cdot \nabla) f(\xi) = -\frac{1}{\tau} [f(\xi) - f^{(0)}(\xi, \rho, \mathbf{u})]. \quad (1)$$

The distribution function $f(\xi)$ relaxes to the equilibrium function

$$f^{(0)}(\xi, \rho, \mathbf{u}) = \frac{\rho}{(2\pi\theta)^{D/2}} \exp\left[-\frac{(\mathbf{u} - \xi)^2}{2\theta}\right] \quad (2)$$

on the time scale τ . At each point in space and time the macroscopic quantities are defined as velocity moments in D -dimensional space,

$$M_{i_1 \dots i_n} = \int d^D \xi f(\xi) \xi_{i_1} \dots \xi_{i_n}, \quad (3)$$

such as the density ρ , the velocity \mathbf{u} , and the temperature θ ,

$$\rho = M, \quad (4a)$$

$$\rho u_i = M_i, \quad (4b)$$

$$\rho(D\theta + u_i u_i) = M_{ii}. \quad (4c)$$

Repeated indices are summed over. All quantities are dimensionless and expressed in units of characteristic scales, i.e., the length scale l_0 , the reference density ρ_0 , and the isothermal speed of sound $c_0 = \sqrt{R\theta_0}$, with the specific gas constant R and the reference temperature θ_0 . Since we focus on flows with a low Mach number $Ma = |\mathbf{u}|$, we assume constant temperature and set $\theta = 1$ henceforth. The time t as well as the relaxation time τ are expressed in units of $t_0 = l_0/c_0$.

Following the work of Grad [38,39], Shan and He [11], and Shan *et al.* [25], we discretize the velocity space by expanding the distribution function in the Hilbert space of

tensorial Hermite polynomials up to an arbitrary order N ,

$$f(\boldsymbol{\xi}) \approx f^N(\boldsymbol{\xi}) = \omega(\boldsymbol{\xi}) \sum_{n=0}^N \frac{1}{n!} a_{i_1 \dots i_n} \mathcal{H}_{i_1 \dots i_n}^{(n)}(\boldsymbol{\xi}), \quad (5)$$

where the Hermite polynomials $\mathcal{H}_{i_1 \dots i_n}^{(n)}(\boldsymbol{\xi})$ are introduced by the recurrence relation

$$\mathcal{H}_{i_1 \dots i_{n+1}}^{(n+1)}(\boldsymbol{\xi}) = (\xi_{i_{n+1}} - \partial_{i_{n+1}}^{(\boldsymbol{\xi})}) \mathcal{H}_{i_1 \dots i_n}^{(n)}(\boldsymbol{\xi}), \quad (6a)$$

$$\mathcal{H}^{(0)}(\boldsymbol{\xi}) = 1. \quad (6b)$$

The lowest-order term of the expansion (5) is given by the weight function ω ,

$$\omega(\boldsymbol{\xi}) = \frac{1}{(2\pi)^{D/2}} e^{-\boldsymbol{\xi}^2/2}, \quad (7)$$

and the expansion coefficients

$$a_{i_1 \dots i_n} = \int d^D \xi f^N(\boldsymbol{\xi}) \mathcal{H}_{i_1 \dots i_n}^{(n)}(\boldsymbol{\xi}) \quad (8)$$

correspond to the moments of the distribution function.

In the configuration space of a discrete velocity model the molecular velocities $\boldsymbol{\xi}_\alpha$ are here restricted to a Cartesian lattice X with uniform spacing c , called lattice speed, such that the components $\xi_{\alpha i}/c$ are integer valued. A subset of X containing a number V of these velocities, the stencil $S \subset X$,

$$S = \{\boldsymbol{\xi}_\alpha | \alpha = 1, \dots, V\}, \quad (9)$$

is used with the corresponding weights w_α to compute the integral in Eq. (3) by quadrature,

$$\begin{aligned} M_{i_1 \dots i_n} &= \int d^D \xi \omega(\boldsymbol{\xi}) P_{i_1 \dots i_n}(\boldsymbol{\xi}) \\ &= \sum_{\alpha} w_{\alpha} P_{i_1 \dots i_n}(\boldsymbol{\xi}_{\alpha}), \end{aligned} \quad (10)$$

with the function $P_{i_1 \dots i_n}(\boldsymbol{\xi}) = f(\boldsymbol{\xi}) \xi_{i_1} \dots \xi_{i_n} / \omega(\boldsymbol{\xi})$. For a polynomial $P_{i_1 \dots i_n}(\boldsymbol{\xi})$ of degree Q the quadrature (10) is exact as long as it satisfies the orthogonality relation

$$\begin{aligned} &\int d^D \xi \omega(\boldsymbol{\xi}) \mathcal{H}_{i_1 \dots i_n}^{(n)}(\boldsymbol{\xi}) \mathcal{H}_{j_1 \dots j_m}^{(m)}(\boldsymbol{\xi}) \\ &= \sum_{\alpha=1}^V w_{\alpha} \mathcal{H}_{i_1 \dots i_n}^{(n)}(\boldsymbol{\xi}_{\alpha}) \mathcal{H}_{j_1 \dots j_m}^{(m)}(\boldsymbol{\xi}_{\alpha}) \\ &= \begin{cases} 1 & \text{if } (i_1, \dots, i_n) = \text{perm}((j_1, \dots, j_m)), \\ 0 & \text{else,} \end{cases} \\ &n + m \leq Q, \end{aligned} \quad (11)$$

up to the Q th order, where $\text{perm}(\mathbf{j})$ denotes a (odd or even) permutation of the vector $\mathbf{j} = (j_1, \dots, j_m)$. This can be readily seen by expanding $P_{i_1 \dots i_n}(\boldsymbol{\xi})$ using Hermite polynomials

$$M_{i_1 \dots i_n} = \sum_{m=0}^Q p_{i_1 \dots i_n, j_1 \dots j_m} \int d^D \xi \omega(\boldsymbol{\xi}) \mathcal{H}_{j_1 \dots j_m}^{(m)}(\boldsymbol{\xi}), \quad (12)$$

with some coefficients $p_{i_1 \dots i_n, j_1 \dots j_m}$. If Eq. (11) holds, the Q th-order polynomial in Eq. (12) is evaluated exactly by the quadrature and hence Eq. (10) holds as well. We then refer to the stencil S along with its weights w_α as a *discrete velocity*

model (DVM) with *quadrature order* Q . Note that we write the moments (10) as sums on discrete velocities, implicitly assuming a sufficiently high quadrature order, unless indicated.

A LB model solves for the variables

$$f_{\alpha} = w_{\alpha} \frac{f(\boldsymbol{\xi}_{\alpha})}{\omega(\boldsymbol{\xi}_{\alpha})} \quad (13)$$

in the LB-BGK equation

$$(\partial_t + \xi_{\alpha i} \partial_i) f_{\alpha} = -\frac{1}{\tau} (f_{\alpha} - f_{\alpha}^{(0)}), \quad (14)$$

using a DVM for the evaluation of the macroscopic variables

$$\rho = \sum_{\alpha} f_{\alpha}, \quad (15a)$$

$$\rho u_i = \sum_{\alpha} f_{\alpha} \xi_{\alpha i}, \quad (15b)$$

$$\rho(D\theta + u_i u_i) = \sum_{\alpha} f_{\alpha} \xi_{\alpha i} \xi_{\alpha i}. \quad (15c)$$

The latter serve the LB model to determine the equilibrium function

$$\begin{aligned} f_{\alpha}^{(0)}(\rho, \mathbf{u}) &= w_{\alpha} \frac{f^{(0)}(\boldsymbol{\xi}_{\alpha}, \rho, \mathbf{u})}{\omega(\boldsymbol{\xi}_{\alpha})} \\ &= w_{\alpha} \sum_{n=0}^N \frac{1}{n!} a_{i_1 \dots i_n}^{(0)} \mathcal{H}_{i_1 \dots i_n}^{(n)}(\boldsymbol{\xi}_{\alpha}), \end{aligned} \quad (16)$$

expanded to a *Hermite order* N , where the Hermite coefficients for isothermal flows are given by

$$a_{i_1 \dots i_n}^{(0)} = \rho u_{i_1} \dots u_{i_n}. \quad (17)$$

For any stencil S , the weights w_α are obtained by solving the set (11) of linear equations. We decompose the stencil $S = \cup_{g=1}^G S_g$ into a number G of velocity sets (*groups*), each group S_g containing V_g velocities (such that $V = \sum_g V_g$) generated by the symmetries of the lattice. These groups S_g can be obtained by reflecting a single $\boldsymbol{\xi}^{(g)} \in S_g$ on those hyperplanes of the lattice which reproduce the lattice itself upon reflection. Hence, the velocity weights must be identical in each group,¹

$$w_{\alpha} = \bar{w}_g \quad \forall \{\alpha | \boldsymbol{\xi}_{\alpha} \in S_g\}. \quad (18)$$

It is efficient to rewrite Eq. (11) into the form

$$K_{\alpha\beta}(S) w_{\beta} = 0 \quad \forall \alpha, \quad \sum_{\alpha=1}^V w_{\alpha} = 1, \quad (19)$$

where K is a symmetric matrix with elements

$$K_{\alpha\beta}(S) = \sum_{m=1}^Q H_{i_1 \dots i_m}^{(m)}(\boldsymbol{\xi}_{\alpha}) H_{i_1 \dots i_m}^{(m)}(\boldsymbol{\xi}_{\beta}) \quad (20)$$

and where the Hermite tensor indices are contracted. In view of Eq. (12), it is sufficient to solve Eq. (11) for $n = 0$ and

¹Note that the metric norm of the velocities, $|\boldsymbol{\xi}|^2 = \xi_i \xi_i$, is not useful to identify the weights. For example, the velocities (3, 0, 0) and (2, 2, 1) have the same norm; however, they are not part of the same group.

$m \leq Q$. Obviously, Eq. (19) then follows from Eq. (11) by multiplication of $\mathcal{H}_{i_1 \dots i_m}^{(m)}(\xi_\beta)$ and summing on m . The reverse is true because Eq. (19) can be written, after multiplication by w_α , as

$$0 = w_\alpha K_{\alpha\beta}(S) w_\beta = \sum_{m=1}^Q \left[\sum_{\alpha} \mathcal{H}_{i_1 \dots i_m}^{(m)}(\xi_\alpha) w_\alpha \right]^2,$$

which requires that each part of the sum be zero. To obtain the matrix $K(S)$, the scalar product of two Hermite tensor polynomials is obtained by the recurrence formula (no sum on m)

$$\begin{aligned} & \mathcal{H}_{i_1 \dots i_{m+1}}^{(m+1)}(\xi) \mathcal{H}_{i_1 \dots i_{m+1}}^{(m+1)}(\eta) \\ &= (\xi_{i_{m+1}} \eta_{i_{m+1}} - \xi_{i_{m+1}} \partial_{i_{m+1}}^{(\eta)} - \eta_{i_{m+1}} \partial_{i_{m+1}}^{(\xi)} + \partial_{i_{m+1}}^{(\xi)} \partial_{i_{m+1}}^{(\eta)}) \\ & \quad \times \mathcal{H}_{i_1 \dots i_m}^{(m)}(\xi) \mathcal{H}_{i_1 \dots i_m}^{(m)}(\eta), \end{aligned} \quad (21)$$

which follows from Eq. (6).

Finding the weights w_α for a stencil reduces to the task of finding the null space of the symmetric matrix $K(S)$ and normalizing according to Eq. (19). As a special feature, the matrix $K(S)$ has a parametric dependence on c (through S) such that for some discrete values $c^* \in \mathbb{R}$ its nullity is increased. This means that by populating the stencil with sufficiently many velocity groups, we find for each quadrature order Q a minimal number G of velocity groups such that for $c = c^*$

$$\dim \ker K(S) = 1, \quad (22)$$

while for $c \neq c^*$ the matrix $K(S)$ is regular and no solution of Eq. (19) can be found. The solutions of Eq. (19) which obey Eq. (22) with $c = c^*$ are referred to as *minimal DVMs*. We only consider DVMs with positive weights. Fixing the quadrature order Q and the number V of velocities, there is a countable infinity of minimal DVMs, since S can be chosen from the (virtually) unbounded lattice X . However, by introducing the integer-valued *energy* E of a stencil and limiting it from above,

$$E = \frac{1}{2c^2} \sum_{\alpha=1}^V \xi_\alpha \cdot \xi_\alpha \leq E_{max}, \quad (23)$$

the number of available Q th-order minimal DVMs becomes finite. We are thus able to determine the complete set of minimal DVMs inside the energy sphere defined by $E \leq E_{max}$. The notation

$$S_{Q(Q)E(E)}^{D(D)V(V)}, \quad \text{e.g., } D3Q19 \rightarrow S_{Q5E15}^{D3V19}, \quad (24)$$

is introduced for the DVMs. As a reminder, we refer to the spatial dimension as D , the quadrature order as Q , the stencil's energy as E , and the number of velocities as V . In the sequel, we consider three spatial dimensions, $D = 3$. Although small values of V are desirable for a high computational performance, it is unclear which choice of the stencil is the most accurate for finite Kn number flows. It is shown in the next sections what are the essential features of a DVM for the resolution of high-order hydrodynamic regimes.

III. LATTICE-BOLTZMANN HYDRODYNAMICS

In this section we study the capability of LB models with Gauss-Hermite quadrature order Q and Hermite order N to capture isothermal ($\theta = 1$) microflows beyond the Navier-Stokes flow regime by using the CE expansion. We restrict our analysis to low-Ma flows. The flow phenomena thus described bear fixed values of the Reynolds number Re , Kn , and Ma , where $Ma \ll 1$ is a small number allowing an expansion.

By taking moments of the discrete LB-BGK equation (14) with respect to the discrete particle velocities ξ_α macroscopic moment equations can be derived. For the density ρ and the momentum $\rho \mathbf{u}$, we obtain the evolution equations

$$\partial_t \rho + \partial_i (\rho u_i) = 0, \quad (25a)$$

$$\partial_t (\rho u_i) + \partial_j \Pi_{ij} = 0, \quad (25b)$$

with the momentum flux tensor

$$\Pi_{ij} = \sum_{\alpha} f_{\alpha} \xi_{\alpha i} \xi_{\alpha j}. \quad (26)$$

Equations (25a) and (25b) represent mass and momentum conservation, respectively, guaranteed by invariants of the BGK collision operator. Corresponding equations for higher moments, e.g., Π_{ij} , are derived by taking moments of Eq. (14) with more than one particle velocity ξ_α .

A. Chapman-Enskog analysis

The CE analysis [4,40] is a perturbative method to solve the Boltzmann equation and is based on the assumption that the distribution function f_α deviates only slightly from the equilibrium $f_\alpha^{(0)}$. The CE expansion introduces a small parameter ϵ into the collision time $\tau \rightarrow \epsilon \tau$, which controls the perturbative analysis and is absorbed into τ after finishing the analysis. Physically, the parameter ϵ may be identified with the Knudsen number, which measures the deviation from equilibrium. The first, second, and third orders correspond to the Navier-Stokes, the Burnett, and super-Burnett flow regimes, respectively. Consequently, the LB-BGK equation (14) becomes

$$(\partial_t + \xi_{\alpha i} \partial_i) f_\alpha = -\frac{1}{\epsilon \tau} (f_\alpha - f_\alpha^{(0)}) \quad (27)$$

and the distribution function is expanded in powers of ϵ ,

$$f_\alpha = f_\alpha^{(0)} + \epsilon f_\alpha^{(1)} + \epsilon^2 f_\alpha^{(2)} + \dots \quad (28)$$

The CE expansion is a multiple-scale expansion of both f and t ,

$$\partial_t = \partial_t^{(0)} + \epsilon \partial_t^{(1)} + \epsilon^2 \partial_t^{(2)} + \dots, \quad (29)$$

with the solvability conditions

$$\sum_{\alpha} f_\alpha^{(n)} = 0, \quad \sum_{\alpha} f_\alpha^{(n)} \xi_{\alpha i} = 0, \quad n \geq 1. \quad (30)$$

Therefore, high-order contributions $f_\alpha^{(n)} (n \geq 1)$ of the CE expansion do not contribute to the macroscopic density and flow velocity. By inserting the CE ansatz (28) and (29) into the Boltzmann equation (27) we find the general solution

for $n \geq 1$

$$f_\alpha^{(n)} = -\tau \left[\partial_i \xi_{\alpha i} f_\alpha^{(n-1)} + \sum_{m=0}^{n-1} \partial_t^{(m)} f_\alpha^{(n-m-1)} \right]. \quad (31)$$

Furthermore, the solvability conditions (30) are equivalent to the relations

$$\partial_t^{(0)} \begin{bmatrix} \rho \\ u_i \end{bmatrix} = \begin{bmatrix} -\partial_i(\rho u_i) \\ -u_j \partial_j u_i - \frac{1}{\rho} \partial_i \rho \end{bmatrix}, \quad (32a)$$

$$\partial_t^{(n)} \begin{bmatrix} \rho \\ u_i \end{bmatrix} = \begin{bmatrix} 0 \\ -\frac{1}{\rho} \partial_j \Pi_{ij}^{(n)} \end{bmatrix}, \quad n \geq 1, \quad (32b)$$

where

$$\Pi_{ij}^{(n)} = \sum_{\alpha} f_\alpha^{(n)} \xi_{\alpha i} \xi_{\alpha j}. \quad (33)$$

The CE analysis is expected to be valid for flow regimes where the system is not too far from equilibrium [41].

B. Navier-Stokes and Burnett flow regimes

The Navier-Stokes momentum flux tensor $\Pi_{ij}^{(1)}$ can be calculated by taking the second moment of Eq. (31) for $n = 1$

$$\Pi_{ij}^{(1)} = -\tau (\partial_t^{(0)} \Pi_{ij}^{(0)} + \partial_k Q_{ijk}^{(0)}), \quad (34)$$

with

$$Q_{ijk}^{(n)} = \sum_{\alpha} f_\alpha^{(n)} \xi_{\alpha i} \xi_{\alpha j} \xi_{\alpha k}. \quad (35)$$

Using Hermite polynomials and the orthogonality relation (11), we obtain

$$\Pi_{ij}^{(0)} = \rho \delta_{ij} + \rho u_i u_j \quad (36)$$

and

$$Q_{ijk}^{(0)} = \rho [u_i \delta_{jk} + u_j \delta_{ik} + u_k \delta_{ij}] + \rho u_i u_j u_k. \quad (37)$$

Inserting these results into Eq. (34) and evaluating the time derivative with respect to Eq. (32a) yields the Navier-Stokes momentum flux tensor $\Pi_{ij}^{(1)}$ for isothermal flows,

$$\Pi_{ij}^{(1)} = -\tau \rho (\partial_i u_j + \partial_j u_i). \quad (38)$$

The standard LB models with accuracy order $Q = 5$ (e.g., D3Q19) do not capture the last term in Eq. (37) and thus cause an error $|\Delta \Pi_{ij}^{(1)}| = \tau |\partial_k (\rho u_i u_j u_k)| = O(\text{Ma}^3)$ [42]. On the other hand, if a DVM with quadrature order $Q \geq 6$ and Hermite order $N = 3$ is used, we exactly recover the Navier-Stokes momentum flux tensor $\Pi_{ij}^{(1)}$. Evidently, the momentum flux tensor is $O(\text{Ma})$ and higher-order Ma terms do not contribute when considering low Ma values.

The Burnett momentum flux tensor $\Pi_{ij}^{(2)}$ can be calculated by taking the second moment of Eq. (31) for $n = 2$

$$\Pi_{ij}^{(2)} = -\tau (\partial_t^{(0)} \Pi_{ij}^{(1)} + \partial_t^{(1)} \Pi_{ij}^{(0)} + \partial_k Q_{ijk}^{(1)}). \quad (39)$$

The third moment of Eq. (31) for $n = 1$ yields

$$Q_{ijk}^{(1)} = -\tau (\partial_t^{(0)} Q_{ijk}^{(0)} + \partial_n R_{ijkn}^{(0)}), \quad (40)$$

with

$$R_{ijkn}^{(0)} = \sum_{\alpha} f_\alpha^{(0)} \xi_{\alpha i} \xi_{\alpha j} \xi_{\alpha k} \xi_{\alpha n}. \quad (41)$$

By applying the Hermite polynomial the tensor $R_{ijkn}^{(0)}$ can be expressed as

$$\begin{aligned} R_{ijkn}^{(0)} = & \rho (\delta_{ij} \delta_{kn} + \delta_{ik} \delta_{jn} + \delta_{in} \delta_{jk}) \\ & + \rho (u_i u_j \delta_{kn} + u_i u_k \delta_{jn} + u_i u_n \delta_{jk} + u_j u_k \delta_{in} \\ & + u_j u_n \delta_{ik} + u_k u_n \delta_{ij}) + \rho u_i u_j u_k u_n. \end{aligned} \quad (42)$$

Inserting this result together with relation (37) into Eq. (40), we can compute the tensor $Q_{ijk}^{(1)}$, which is required for the evaluation of $\Pi_{ij}^{(2)}$. After an extensive calculation of all terms in Eq. (39) by considering the multiple-scale time derivatives $\partial_t^{(0)}$ and $\partial_t^{(1)}$, we obtain the Burnett momentum flux tensor for isothermal flows,

$$\Pi_{ij}^{(2)} = 2\rho\tau^2 \left[(\partial_k u_i)(\partial_k u_j) + \frac{1}{\rho^2} (\partial_i \rho)(\partial_j \rho) - \frac{1}{\rho} \partial_i \partial_j \rho \right]. \quad (43)$$

The Burnett tensor for an incompressible flow field is $\Pi_{ij}^{(2)} = O(\text{Ma}^2)$ and thus does not contribute to the momentum dynamics in the low-Ma regime. However, for finite Kn the flow behavior can become compressible even for low Ma [3] and in that case we obtain contributions from the last term of Eq. (43). We show in the following section that these low-Ma terms are recovered even for LB models with accuracy order $Q = 5$.

C. Low-Mach truncation error

Depending on the quadrature accuracy order Q a LB model is able to recover different flow regimes. In this section we analyze the quadrature error with respect to the power of Ma and discuss the ability to capture high-order flow regimes especially for low-Ma flows. For this purpose we consider the k th moment of the n th CE level, which is defined by

$$M_{i_1 \dots i_k}^{(n)} = \sum_{\alpha} f_\alpha^{(n)} \xi_{\alpha i_1} \dots \xi_{\alpha i_k}, \quad (44)$$

where $k \geq 2$ because of the solvability conditions (30). An equation for this moment can be derived by taking the k th moment of Eq. (31) with respect to the discrete particle velocities ξ_α ,

$$\begin{aligned} M_{i_1 \dots i_k}^{(n)} = & -\tau [\partial_t^{(0)} M_{i_1 \dots i_k}^{(n-1)} + \partial_t^{(1)} M_{i_1 \dots i_k}^{(n-2)} + \dots \\ & + \partial_t^{(n-1)} M_{i_1 \dots i_k}^{(0)} + \partial_j M_{i_1 \dots i_k j}^{(n-1)}]. \end{aligned} \quad (45)$$

Based on this relation it can be easily shown by induction that the moment $M_{i_1 \dots i_k}^{(n)}$ is completely determined by moments of the equilibrium distribution. Because of a finite accuracy order of the Gauss-Hermite quadrature, higher moments of the CE expansion are only approximately captured and therefore high-order flow regimes are not accurately recovered. In order to discuss such a truncation error for a moment $M_{i_1 \dots i_k}^{(n)}$ we define the quadrature error

$$\Delta M_{i_1 \dots i_k}^{(n)} = M_{i_1 \dots i_k}^{(n)} - M_{i_1 \dots i_k}^{(n)DVM}, \quad (46)$$

where $M_{i_1 \dots i_k}^{(n)DVM}$ is a possibly inaccurate moment, calculated by a DVM.

Due to the fact that a general CE moment $M_{i_1 \dots i_k}^{(n)}$ can be expressed by equilibrium moments, it is important to consider, in a first step, the quadrature error for equilibrium moments

$$M_{i_1 \dots i_k}^{(0)} = \sum_{\alpha} f_{\alpha}^{(0)} \xi_{\alpha i_1} \dots \xi_{\alpha i_k}, \quad (47)$$

with the equilibrium function given by Eq. (16). Note that the Hermite coefficients [see Eq. (17)] yield

$$a_{i_1 \dots i_n}^{(0)} = O(\text{Ma}^n). \quad (48)$$

The product $\xi_{\alpha i_1} \dots \xi_{\alpha i_k}$ in Eq. (47) can be expressed by the Hermite polynomial $\mathcal{H}_{i_1 \dots i_k}^{(k)}(\xi_{\alpha})$, $\mathcal{H}_{i_1 \dots i_{k-2}}^{(k-2)}(\xi_{\alpha})$, and terms with lower-order Hermite polynomials,

$$\begin{aligned} \xi_{\alpha i_1} \dots \xi_{\alpha i_k} &= \mathcal{H}_{i_1 \dots i_k}^{(k)}(\xi_{\alpha}) \\ &+ \sum_{r < s} \mathcal{H}_{i_1 \dots i_{r-1} i_{r+1} \dots i_{s-1} i_{s+1} \dots i_k}^{(k-2)}(\xi_{\alpha}) \delta_{i_r i_s} + \dots \end{aligned} \quad (49)$$

Therefore, we obtain

$$\begin{aligned} M_{i_1 \dots i_k}^{(0)} &= \sum_{n=0}^N \frac{1}{n!} a_{j_1 \dots j_n}^{(0)} \sum_{\alpha} w_{\alpha} \left[\mathcal{H}_{j_1 \dots j_n}^{(n)}(\xi_{\alpha}) \mathcal{H}_{i_1 \dots i_k}^{(k)}(\xi_{\alpha}) \right. \\ &+ \sum_{r < s} \delta_{i_r i_s} \mathcal{H}_{j_1 \dots j_n}^{(n)}(\xi_{\alpha}) \\ &\left. \times \mathcal{H}_{i_1 \dots i_{r-1} i_{r+1} \dots i_{s-1} i_{s+1} \dots i_k}^{(k-2)}(\xi_{\alpha}) + \dots \right]. \end{aligned} \quad (50)$$

An exact evaluation of the moment $M_{i_1 \dots i_k}^{(0)}$ requires a sufficiently high Hermite order $N \geq k$ of the equilibrium (16) to ensure that the contributions of all Hermite polynomials in the brackets of Eq. (50) are included [28]. Due to the orthogonality relation (11) higher Hermite polynomials in the equilibrium, $\mathcal{H}_{i_1 \dots i_n}^{(n)}$ with $n > k$, do not contribute to the moment $M_{i_1 \dots i_k}^{(0)}$. In addition, we need an adequate Gauss-Hermite quadrature with an accuracy order $Q \geq k + N$ to guarantee the correct evaluation of all Hermite contractions in Eq. (50). Otherwise, we obtain quadrature errors, which are analyzed in detail in Appendix C. The result of this error estimate for equilibrium moments $M_{i_1 \dots i_k}^{(0)}$ is given by

$$\Delta M_{i_1 \dots i_k}^{(0)} = \begin{cases} 0 & \text{for } k + N \leq Q, k \leq N, \\ O(\text{Ma}^{N+1}) & \text{for } k + N \leq Q, k > N, \\ O(\text{Ma}^{Q-k+1}) & \text{for } k - 1 \leq Q < k + N, \\ O(\text{Ma}^0) & \text{for } Q < k - 1. \end{cases} \quad (51)$$

At this point, it is worth discussing the optimal choice of the Hermite order N of the equilibrium function (16) in order to ensure an exact evaluation of all $M_{i_1 \dots i_k}^{(0)}$ with maximal k . Using Eq. (51) this is achieved for $N = \frac{1}{2}Q$. Usually Q is an odd number and N is an integer, which allows either $N = \frac{1}{2}(Q + 1)$ or $N = \frac{1}{2}(Q - 1)$.

For an odd Gauss-Hermite quadrature order Q we fix in the following the Hermite order of the equilibrium (16) to $N = \frac{1}{2}(Q - 1)$. For this particular choice we obtain for the

quadrature error (51) the relation

$$\begin{aligned} \Delta M_{i_1 \dots i_k}^{(0)} &= \begin{cases} 0 & \text{for } 2k + 1 \leq Q, \\ O(\text{Ma}^{Q-k+1}) & \text{for } k - 1 \leq Q < 2k + 1, \\ O(\text{Ma}^0) & \text{for } Q < k - 1, \end{cases} \\ &= \Theta(2k - Q) O(\text{Ma}^{\max(Q-k+1, 0)}), \end{aligned} \quad (52)$$

where Θ is the Heaviside step function. If we choose the other possibility, $N = \frac{1}{2}(Q + 1)$, we get for the estimate of the quadrature error the same result (52). However, we want to point out that the quadrature error for $N = \frac{1}{2}(Q - 1)$ is different from the choice $N = \frac{1}{2}(Q + 1)$; however, the leading power of Ma in the error term is the same.

The dominant quadrature error (with respect to Ma) of a quantity consisting of several equilibrium moments is determined by the highest equilibrium moment $M_{i_1 \dots i_k}^{(0)}$. Based on the recurrence relation (45) for a CE moment $M_{i_1 \dots i_k}^{(n)}$ and the error estimate (52) for equilibrium moments, we are able to prove the following theorem.

Theorem. For any LB model with an odd Gauss-Hermite quadrature order Q and a Hermite order $N = (Q - 1)/2$ of the equilibrium (16), the truncation error of the k th velocity moment $M_{i_1 \dots i_k}^{(n)}$ in the n th CE level is given, for low- Ma values, by

$$\Delta M_{i_1 \dots i_k}^{(n)} = (-\tau)^n \partial_{j_1} \dots \partial_{j_n} \Delta M_{i_1 \dots i_k j_1 \dots j_n}^{(0)} + \text{subleading terms} \quad (53)$$

and can be estimated by

$$\Delta M_{i_1 \dots i_k}^{(n)} = \Theta[2(k + n) - Q] O(\text{Ma}^{\max(Q-k-n+1, 0)}), \quad (54)$$

where $k \geq 2$ and Θ is the Heaviside step function.

The theorem is proved by induction in Appendix D. It is in agreement with the accuracy determinations of LB models given by Shan *et al.* [25]. Moreover, the theorem presented here makes it possible to identify the recovered flow regimes for low- Ma values by analyzing the truncation error.

In the following we discuss the macroscopic momentum dynamics

$$\partial_t(\rho u_i) + \partial_j \Pi_{ij} = 0, \quad (55)$$

with the momentum flux tensor,

$$\Pi_{ij} = \sum_n \Pi_{ij}^{(n)}, \quad (56)$$

and analyze the recovered flow regime of a LB model with Gauss-Hermite accuracy order Q by using the theorem. Based on the error estimate (54), the relevant error of the momentum flux tensor of the n th flow regime $\Pi_{ij}^{(n)}$ with respect to the Ma power is given by

$$\Delta \Pi_{ij}^{(n)} = \Theta(2n + 4 - Q) O(\text{Ma}^{\max(Q-n-1, 0)}). \quad (57)$$

LB models with quadrature order $Q = 5$. For LB models with quadrature order $Q = 5$ the Navier-Stokes momentum flux tensor $\Pi_{ij}^{(1)}$ is not evaluated exactly for all Ma numbers, because the highest equilibrium moment $Q_{ijk}^{(0)}$ is only recovered up to an error $O(\text{Ma}^3)$ in accordance with relation (54). The leading momentum flux error is given by $\Delta \Pi_{ij}^{(1)} = -\tau \partial_k \Delta Q_{ijk}^{(0)} = O(\text{Ma}^3)$. It is interesting to notice

that even LB models with accuracy $Q = 5$ recover the Burnett momentum flux tensor $\Pi_{ij}^{(2)}$ for low Ma values. Because of relation (57) we obtain $\Delta\Pi_{ij}^{(2)} = O(\text{Ma}^2)$, which indicates that low-Ma contributions are recovered. Considering the isothermal Burnett tensor we observe that only the last term on the right-hand side of Eq. (43) includes low-Ma contributions for compressible finite Kn flows.

LB models with quadrature order $Q = 7$. LB models with $Q = 7$ recover the Navier-Stokes momentum flux tensor (38) exactly, whereas the Burnett tensor includes an error of $\Delta\Pi_{ij}^{(2)} = O(\text{Ma}^4)$ according to relation (57). This is caused by an inaccurate evaluation of the equilibrium moment $R_{ijkn}^{(0)}$. Furthermore, the momentum flux tensor $\Pi_{ij}^{(3)}$ is captured up to an $O(\text{Ma}^3)$ error and $\Pi_{ij}^{(4)}$ up to an $O(\text{Ma}^2)$ error. Therefore, LB models with quadrature accuracy order $Q = 7$ recover the momentum dynamics up to the fourth flow regime ($\Pi_{ij}^{(4)}$) for low-Ma flows.

LB models with arbitrary quadrature order Q . In general, the error estimate (57) states that for low-Ma flows LB models with Gauss-Hermite quadrature order Q capture the correct momentum dynamics up to the $(Q - 3)$ th flow regime, because of $\Delta\Pi_{ij}^{(Q-3)} = O(\text{Ma}^2)$. On the other hand, higher flow regimes are not described correctly. For example, the error of $\Delta\Pi_{ij}^{(Q-2)} = O(\text{Ma})$ corrupts the leading-order terms of the momentum dynamics. Fortunately, such effects are suppressed by a term $O(\text{Kn}^{Q-2})$, which is sufficiently small for $\text{Kn} < 1$.

IV. GAS-SURFACE INTERACTION

For wall-bounded flows at finite Kn the gas-surface interaction is of crucial importance because of its influence on the Knudsen layer. It is known that the diffuse Maxwell reflection model [6] is sufficiently accurate to describe flows for a wide range of Kn. Its numerical implementation in the LB framework has been reported in Refs. [21–24,43]. Although high-order LB models recover flow regimes beyond the Navier-Stokes level, there are some models, e.g., $D3Q121$, which show significant deviations to reference results for wall-bounded flows at finite Kn (see Sec. VI). The reason for this failure is shown to be caused by the inability of these models to recover the diffuse Maxwell boundary condition accurately. We define wall moments and introduce a wall accuracy order and thus assess the capability of a LB model to capture the diffuse Maxwell boundary condition.

In this section we discuss a systematic way to generate DVMs which inherently exhibit the diffuse Maxwell boundary condition.

A. Diffuse Maxwell boundary condition

The diffuse Maxwell reflection model suggests that particles emitted from the solid surface do not depend on anything prior to their surface impact and their velocities are normalized by the equilibrium distribution. This notion infers that the scattering kernel only depends on the emitted velocities. We can then write [21] the distribution function of particles emitted by the wall as

$$f(\boldsymbol{\xi}) = \Psi f^{(0)}(\boldsymbol{\xi}, \rho_w, 0) \quad \text{for } \mathbf{n} \cdot \boldsymbol{\xi} > 0, \quad (58)$$

where $\mathbf{n} = \mathbf{e}_z$ is the inward wall normal vector and the scalar Ψ ensures mass conservation across the (impermeable and stationary) surface,

$$\Psi = \frac{\int_{\mathbf{n} \cdot \boldsymbol{\xi}' < 0} d^3 \boldsymbol{\xi}' f(\boldsymbol{\xi}') |\mathbf{n} \cdot \boldsymbol{\xi}'|}{\int_{\mathbf{n} \cdot \boldsymbol{\xi} > 0} d^3 \boldsymbol{\xi} f^{(0)}(\boldsymbol{\xi}, \rho_w, 0) |\mathbf{n} \cdot \boldsymbol{\xi}|}. \quad (59)$$

The equilibrium function $f^{(0)}(\boldsymbol{\xi}, \rho_w, 0)$ is evaluated for the density ρ_w and the macroscopic velocity $\mathbf{u} = 0$ at the wall. It was shown in Ref. [30] that for steady unidirectional flows $\Psi = 1$. The discussion of this section allows Ψ to be arbitrary.

B. Equilibrium wall moments

At the wall, the velocity moments (3) are decomposed into a part connected to the fluid domain and a part which is influenced by the wall,

$$M_{i_1 \dots i_n} = \int_{\mathbf{n} \cdot \boldsymbol{\xi} < 0} d^3 \boldsymbol{\xi} f(\boldsymbol{\xi}) \xi_{i_1} \dots \xi_{i_n} + \int_{\mathbf{n} \cdot \boldsymbol{\xi} > 0} d^3 \boldsymbol{\xi} f(\boldsymbol{\xi}) \xi_{i_1} \dots \xi_{i_n}. \quad (60)$$

The second integral is completely determined by distributions coming from the wall representing the gas-surface interaction. Using Eqs. (58) and (59), we can write

$$M_{i_1 \dots i_n} = \int_{\mathbf{n} \cdot \boldsymbol{\xi} < 0} d^3 \boldsymbol{\xi} f(\boldsymbol{\xi}) \xi_{i_1} \dots \xi_{i_n} + \Psi \rho_w W_{i_1 \dots i_n}, \quad (61)$$

where the equilibrium wall moments $W_{i_1 \dots i_n}$ were introduced as

$$W_{i_1 \dots i_n} = \int d^3 \boldsymbol{\xi} \Theta(\mathbf{n} \cdot \boldsymbol{\xi}) \omega(\boldsymbol{\xi}) \xi_{i_1} \dots \xi_{i_n}. \quad (62)$$

The capability of a DVM to capture the Maxwell boundary condition can now be investigated by analyzing the equilibrium wall moments. It is straightforward to obtain exact solutions for $W_{i_1 \dots i_n}$, expressing them by the scalar integrals

$$\begin{aligned} & \int_{-\infty}^{\infty} d\xi_x \int_{-\infty}^{\infty} d\xi_y \int_0^{\infty} d\xi_z \omega(\boldsymbol{\xi}) \xi_z^{m_z} \xi_y^{m_y} \xi_x^{m_x} \\ &= \frac{\sqrt{2}^{m_x + m_y + m_z}}{8\sqrt{\pi}^3} [1 + (-1)^{m_x}] [1 + (-1)^{m_y}] \\ & \times \Gamma\left(\frac{m_x + 1}{2}\right) \Gamma\left(\frac{m_y + 1}{2}\right) \Gamma\left(\frac{m_z + 1}{2}\right), \quad (63) \end{aligned}$$

with the Euler Gamma function Γ . We now show how to evaluate the equilibrium wall moments (62) numerically with a discrete velocity model.

C. Discrete velocity models with wall constraints

In the discrete velocity space the wall moments (62) are computed by quadrature. Note, however, that the evaluation of n th wall moments $W_{i_1 \dots i_n}$ cannot be performed in an exact manner by Gauss-Hermite quadratures with order $Q = n$. This is due to the nonanalyticity of the Heaviside function at the wall where an expansion of the integrand does not exist. Although alternative quadratures can be introduced using functions orthogonal in the half-space [35,44,45], we would like to cope with this difficulty within the conventional framework

of Gauss-Hermite quadratures. We employ the quadrature prescription

$$W_{i_1 \dots i_n}^{DVM} = \sum_{\mathbf{n} \cdot \xi_{\alpha} > 0} w_{\alpha} \xi_{\alpha i_1} \dots \xi_{\alpha i_n}, \quad (64)$$

and demand that—while defining the stencil—the supplementary constraints

$$W_{i_1 \dots i_n}^{DVM} = W_{i_1 \dots i_n} \quad (65)$$

be fulfilled, aside from the orthogonality condition (11). We thus ensure that the wall moments (62) are computed exactly. To this end the kernel of the matrix K , see Eq. (20), is augmented by adding groups of velocities, yielding $\{w_{\alpha}^{(1)}, w_{\alpha}^{(2)}, \dots, w_{\alpha}^{(J)}\}$ as an orthogonal basis of the J -dimensional null space. Any weight vector

$$w_{\alpha} = x_j w_{\alpha}^{(j)} \quad (66)$$

in this null space then defines a quadrature where the freedom in the coefficients x_j is used to implement the wall constraints (65). We achieve this by solving the set of linear equations

$$A_{kj} x_j = b_k, \quad 1 \leq k \leq L, \quad (67)$$

where the vector \mathbf{b} comprises the nontrivial equilibrium wall moments of Eq. (65),

$$\begin{aligned} \mathbf{b} = & (W, W_z, W_{xx}, W_{zzz}, W_{zxx}, W_{xxxx}, W_{xyy}, \\ & W_{zzzz}, W_{zzzx}, W_{zxzx}, W_{zxyy}, W_{xxxxx}, \\ & W_{xxxxy}, W_{zzzzz}, W_{zzzzx}, W_{zzzxx}, \\ & W_{zzzxy}, W_{zxxxx}, W_{zxxxxy}, \dots), \end{aligned} \quad (68)$$

and the matrix A is given by

$$A_{kj} = \sum_{\mathbf{n} \cdot \xi_{\alpha} > 0} T_k^{\alpha} w_{\alpha}^{(j)}, \quad (69)$$

with the symbols T_k^{α} defined as

$$\begin{aligned} \mathbf{T}^{\alpha} = & (1, \xi_{\alpha z}, \xi_{\alpha x}^2, \xi_{\alpha z}^3, \xi_{\alpha z} \xi_{\alpha x}^2, \xi_{\alpha x}^4, \xi_{\alpha x}^2 \xi_{\alpha y}^2, \\ & \xi_{\alpha z}^5, \xi_{\alpha z}^3 \xi_{\alpha x}^2, \xi_{\alpha z} \xi_{\alpha x}^4, \xi_{\alpha z} \xi_{\alpha x}^2 \xi_{\alpha y}^2, \xi_{\alpha x}^6, \\ & \xi_{\alpha x}^4 \xi_{\alpha y}^2, \xi_{\alpha z}^7, \xi_{\alpha z}^5 \xi_{\alpha x}^2, \xi_{\alpha z}^3 \xi_{\alpha x}^4, \\ & \xi_{\alpha z}^3 \xi_{\alpha x}^2 \xi_{\alpha y}^2, \xi_{\alpha z} \xi_{\alpha x}^6, \xi_{\alpha z} \xi_{\alpha x}^4 \xi_{\alpha y}^2, \dots). \end{aligned} \quad (70)$$

The index k in Eq. (67) runs from 1 to L , where for $Q = 7$ we find $L = 19$ nontrivial equilibrium wall moments. For $Q > 7$, additional components need to be appended to the vectors \mathbf{b} and \mathbf{T}^{α} .²

²The wall moments are determined for $n \leq Q$, i.e., the quadrature's accuracy at the wall is not required to exceed the accuracy in the bulk. Note that the integrals in Eq. (63) vanish by symmetry if m_x or m_y are odd. Since this symmetry is preserved by the stencils, the quadrature then automatically yields correct results. For even m_x and m_y , the equilibrium wall moments (62) are automatically exact for even $m_z = 2, 4, \dots$ [see the discussion following Eq. (78)]. The components of the vectors \mathbf{b} and \mathbf{T}^{α} are nested in the order of increasing n , decreasing m_z , and decreasing m_x , where $m_x \in \{0, 2, 4, \dots\}$, $m_y \in \{0, 2, 4, \dots \mid m_y \leq m_x\}$, $m_z \in \{0, 1, 3, 5, \dots\}$, and $m_x + m_y + m_z \leq Q$.

By solving the supplementary Eq. (67) we obtain stencils suitable as Q -th-order quadratures for the gas-surface interaction as well as the bulk flow. As shown in Sec. VI, LB models of a given quadrature order Q show strongly different behavior depending on the quantities

$$\sigma_{i_1 \dots i_n} = (W_{i_1 \dots i_n} - W_{i_1 \dots i_n}^{DVM}) / W_{i_1 \dots i_n}, \quad (71)$$

which measure the numerical error of the gas-surface interaction. It is convenient to use the expression

$$\sigma_{\Sigma} = \frac{\sum_{n=0}^Q \sum_{i_1 \dots i_n} |\sigma_{i_1 \dots i_n}| e^{-n}}{\sum_{n=0}^Q \sum_{i_1 \dots i_n} e^{-n}} \quad (72)$$

assessing the net effect of nontrivial wall errors.

We put forward the so-called *wall accuracy index* Λ_{index} for indicating a DVM's ability to evaluate the wall integrals (62). The wall accuracy index is defined as the binary number

$$\Lambda_{index} = (\lambda_L \dots \lambda_2 \lambda_1)_2, \quad (73)$$

where the λ_k are numerical Booleans which determine whether the k th wall constraint in Eq. (67) is satisfied,

$$\lambda_k = \begin{cases} 0 & \text{for } A_{kj} x_j \neq b_k, \\ 1 & \text{for } A_{kj} x_j = b_k. \end{cases} \quad (74)$$

For instance, a DVM with one single wall constraint, $\sigma_{zxx} = 0$, can be characterized by the wall accuracy index

$$\Lambda_{index} = 0000000000000010000_2 = 16. \quad (75)$$

In Eq. (75), it is shown that the wall accuracy index can be given by a decimal number as well. We thus use Λ_{index} as an abbreviation bearing lots of information on the validity of the wall constraints (65).

Furthermore, we define the *wall accuracy order* Λ by the highest rank of equilibrium wall moments (62) up which the quadrature is exact,

$$\sigma_{i_1 \dots i_n} = 0 \quad \forall \{n, i_1 \dots i_n \mid n = 0, 1, \dots, \Lambda\}. \quad (76)$$

If Eq. (76) is false for any $\Lambda \geq 0$, we set $\Lambda = -1$. It can be shown (see Appendix E) that a DVM with wall accuracy order Λ is able to exactly evaluate the Maxwell boundary factor Ψ [see Eq. (59)] up to the $(\Lambda - N - 1)$ th CE level.

D. Complete sets of discrete velocity models

Several sets of DVMs, denoted by \mathcal{S}_n , $n = 1, 2, \dots$, are introduced in this paper to alleviate the discussion on efficiency and accuracy of the DVMs. An overview of these sets is given in Table I. Each set is defined by its dimension D , its quadrature order Q , the upper limit E_{max} of the stencil's energy, and the wall accuracy index Λ_{index} . A set \mathcal{S}_n contains a number A of minimal DVMs with V_{min} being the minimal velocity count. While in one spatial dimension, it is well-known how to find V_{min} for Gauss-Hermite quadratures, in higher dimensions, this is not obvious. For most common quadratures in three dimensions, $V > V_{min}$.

A set of DVMs can only be complete if an upper bound on the stencil energy is provided. Defining the set \mathcal{S}_1 , we arbitrarily chose $E_{max} = 500$ but, of course, this number may be increased. \mathcal{S}_1 includes 4677 three-dimensional, seventh-order DVMs. The lowest velocity count within \mathcal{S}_1 yields $V_{min} = 38$.

TABLE I. Overview of DVM sets \mathcal{S}_n which comprise a number A of minimal DVMs with dimension D , quadrature order Q , and upper limit E_{max} for the stencil energy. The sixth column shows the minimal velocity count V_{min} . For each set \mathcal{S}_n , it is indicated in the seventh column which wall moment errors vanish. Finally, the wall accuracy index Λ_{index} is given. The symbol “—” is used where a quantity is not considered.

	D	Q	E_{max}	A	V_{min}	$\sigma_{\dots} = 0$	Λ_{index}
\mathcal{S}_1	3	7	500	4677	38	—	0
\mathcal{S}_2	3	9	625	618	79	—	0
\mathcal{S}_3	3	7	—	500	—	σ_{zxx}	16
\mathcal{S}_4	3	7	2500	45863	80	$\sigma, \sigma_{ij}, \sigma_{ijnm}, \sigma_{ijnmpq}$	6245
\mathcal{S}_5	2	7	250	1188	16	—	0
\mathcal{S}_6	2	9	300	592	33	—	0
\mathcal{S}_7	2	7	1000	21952	20	$\sigma, \sigma_{ij}, \sigma_{ijnm}, \sigma_{ijnmpq}$	549

There are two models of this kind, displayed in Tables VI and VII. All tables of DVMs are deferred to the Appendix. The set \mathcal{S}_2 comprises 618 three-dimensional, ninth-order DVMs with $E_{max} = 625$. The lowest velocity count yields $V_{min} = 79$. This DVM, which is noted in Table X, complements previously known DVMs with minimal velocities [28]. A widely used ninth-order DVM is the one shown in Table XI, also known as $D3Q121$ [24].

The two DVM sets \mathcal{S}_3 and \mathcal{S}_4 shown in Table I are designed for wall-bounded flows, taking into account some of the equilibrium wall moments (65); hence, $\Lambda_{index} > 0$. In particular, the DVM set \mathcal{S}_3 comprises 500 stencils for $D = 3$ and $Q = 7$ with the supplementary constraint

$$\sigma_{zxx} = 0. \quad (77)$$

As shown in Sec. VI, a minimal equilibrium wall moment error $|\sigma_{zxx}|$ is crucial to the resolution of correct mass flow and slip velocity in finite Kn flows. Further optimization (see below) yields the stencils S_{Q7E672}^{D3V77} and $S_{Q7E1023}^{D3V107}$ shown in Tables XII and XIII. Note that these are *augmented DVMs*: They are based on DVMs for minimal DVMs and the last two velocity groups are added to account for the wall constraint (77). The wall accuracy order for \mathcal{S}_3 yields $\Lambda = -1$.

For the DVM set \mathcal{S}_4 , where $D = 3$ and $Q = 7$, we split the domain of numerical integration across the wall, i.e., the stencils do not contain velocities parallel to the wall, $\mathbf{n} \cdot \boldsymbol{\xi}_\alpha \neq 0 \forall \alpha$. A noteworthy property of these models is that they enforce wall interaction of the stencil nearest to the wall. The occurrence of ballistic particles as described in Ref. [15] is thus avoided. We refer to the stencils without wall-parallel velocities as *scattering stencils*.

Considering Q th-order polynomials $P(\boldsymbol{\xi})$ invariant for

$$\mathbf{n} \cdot \boldsymbol{\xi} \longrightarrow -\mathbf{n} \cdot \boldsymbol{\xi} \quad (78)$$

(even functions included), the relation

$$\begin{aligned} \int d^3\xi \Theta(\mathbf{n} \cdot \boldsymbol{\xi}) \omega(\boldsymbol{\xi}) P(\boldsymbol{\xi}) &= \frac{1}{2} \int d^3\xi \omega(\boldsymbol{\xi}) P(\boldsymbol{\xi}) \\ &= \frac{1}{2} \sum_{\alpha} w_{\alpha} P(\boldsymbol{\xi}_{\alpha}) \end{aligned}$$

$$= \frac{1}{2} \sum_{\mathbf{n} \cdot \boldsymbol{\xi}_{\alpha} = 0} w_{\alpha} P(\boldsymbol{\xi}_{\alpha}) + \sum_{\mathbf{n} \cdot \boldsymbol{\xi}_{\alpha} > 0} w_{\alpha} P(\boldsymbol{\xi}_{\alpha}) \quad (79)$$

is exact for quadratures with order Q . For scattering stencils, the first sum in Eq. (79) is absent by definition ($\mathbf{n} \cdot \boldsymbol{\xi}_{\alpha} \neq 0$) and therefore the customary wall quadrature prescription (64) yields exact results for even wall moments,

$$\sigma_{i_1 \dots i_{2n}} = 0 \quad \forall \{n, i_1 \dots i_{2n} \mid 0 \leq 2n \leq Q\}. \quad (80)$$

On the other hand, wall-parallel and zero velocities infer a quadrature error using Eq. (64) because their contribution to Eq. (79) is neglected. This circumstance is an advantage of scattering stencils such as $D2Q16$ and it also explains why DVMs with an even number of velocities are superior to others in the context of wall-bounded flows. Such stencils necessarily lack the rest velocity $\boldsymbol{\xi} = 0$ which typically comes with a relatively large weight w_0 in the quadrature. The rest velocity causes a considerable error neglecting the term $\frac{1}{2} w_0 P(0)$ in Eq. (79) while using the wall quadrature prescription (64). This error is suppressed by stencils with an even number of velocities.

The wall accuracy order of \mathcal{S}_4 is found, according to Eq. (76), to be $\Lambda = 0$. On the other hand, the wall accuracy index yields

$$\Lambda_{index} = 0000001100001100101_2 = 6245. \quad (81)$$

The energies of the stencils in \mathcal{S}_4 are limited by $E_{max} = 2500$, which yields 45 863 DVMs with the property (80). Among those, we picked out the DVMs $S_{Q7E1932}^{D3A96}$ and $S_{Q7E1764}^{D3A112}$, for reasons explained in Sec. VI, and we present them in Tables XIV and XV.

Finally, Table I lists the sets \mathcal{S}_5 , \mathcal{S}_6 , and \mathcal{S}_7 , which concern the popular research area of LB models in 2 spatial dimensions. Details about these models can be found in Appendix B.

V. NUMERICAL METHOD

For the numerical solution of our test case shown below, we include an external force F_{α} in the LB-BGK equation (14) by writing

$$(\partial_t + \xi_{\alpha i} \partial_i) f_{\alpha} = -\frac{1}{\tau} (f_{\alpha} - f_{\alpha}^{eq}), \quad (82)$$

where we introduced the generalized equilibrium function

$$f_{\alpha}^{eq} = f_{\alpha}^{(0)} + \tau F_{\alpha}, \quad (83)$$

which must be expanded using Hermite polynomials [25]. Using the Hermite order $N = 3$, this yields for the equilibrium

$$\begin{aligned} f_{\alpha}^{(0)}(\rho, \mathbf{u}) &= \rho w_{\alpha} \left\{ 1 + u_i \xi_{\alpha i} + \frac{1}{2} [(u_i \xi_{\alpha i})^2 - u_i u_i] \right. \\ &\quad \left. + \frac{1}{6} u_i \xi_{\alpha i} [(u_j \xi_{\alpha j})^2 - 3u_j u_j] \right\} \end{aligned} \quad (84)$$

and for the body force term

$$\begin{aligned} F_{\alpha} &= \rho w_{\alpha} [g_i \xi_{\alpha i} + (g_i \xi_{\alpha i} u_j \xi_{\alpha j} - u_i g_i)] \\ &\quad + \frac{w_{\alpha}}{2} (\Pi_{ij} - \rho \delta_{ij}) [g_k \xi_{\alpha k} \mathcal{H}_{ij}^{(2)}(\boldsymbol{\xi}_{\alpha}) - 2g_i \xi_{\alpha j}]. \end{aligned} \quad (85)$$

Here \mathbf{g} denotes the acceleration of an external body force field. The numerical solution of Eq. (82) in space \mathbf{x} and time t can be achieved by integrating along a characteristic for a

time interval Δt using the trapezium rule

$$\begin{aligned} & f_\alpha(\mathbf{x} + \boldsymbol{\xi}_\alpha \Delta t, t + \Delta t) - f_\alpha(\mathbf{x}, t) \\ &= -\frac{1}{\tau} \int_0^{\Delta t} dt' [f_\alpha(\mathbf{x} + \boldsymbol{\xi}_\alpha t', t + t') - f_\alpha^{eq}(\mathbf{x} + \boldsymbol{\xi}_\alpha t', t + t')] \\ &= -\frac{1}{\tau} \frac{\Delta t}{2} \{ [f_\alpha(\mathbf{x}, t) - f_\alpha^{eq}(\mathbf{x}, t)] \\ &\quad + [f_\alpha(\mathbf{x} + \boldsymbol{\xi}_\alpha \Delta t, t + \Delta t) - f_\alpha^{eq}(\mathbf{x} + \boldsymbol{\xi}_\alpha \Delta t, t + \Delta t)] \} \\ &\quad + O(\Delta t^3), \end{aligned} \quad (86)$$

resulting in a second-order implicit differencing scheme. By introducing a modified distribution function [46],

$$\bar{f}_\alpha = f_\alpha + \frac{\Delta t}{2\tau} (f_\alpha - f_\alpha^{eq}), \quad (87)$$

the scheme (86) is transformed into a fully explicit scheme of second-order accuracy,

$$\begin{aligned} & \bar{f}_\alpha(\mathbf{x} + \boldsymbol{\xi}_\alpha \Delta t, t + \Delta t) \\ &= \bar{f}_\alpha(\mathbf{x}, t) - \frac{\Delta t}{\tau + \frac{\Delta t}{2}} [\bar{f}_\alpha(\mathbf{x}, t) - f_\alpha^{eq}(\mathbf{x}, t)]. \end{aligned} \quad (88)$$

The macroscopic density ρ and velocity \mathbf{u} are given by

$$\rho = \sum_\alpha f_\alpha = \sum_\alpha \bar{f}_\alpha, \quad (89a)$$

$$\rho u_i = \sum_\alpha f_\alpha \xi_{\alpha i} = \sum_\alpha \bar{f}_\alpha \xi_{\alpha i} + \frac{\Delta t}{2} \rho g_i, \quad (89b)$$

where we have used $\sum_\alpha F_\alpha \xi_{\alpha i} = \rho g_i$.

The relaxation time τ in the LB-BGK equation (14) is chosen to adjust the viscosity of the bulk flow. Based on the Navier-Stokes momentum flux tensor (38) τ is determined by

$$\tau = \nu = \frac{\tilde{\nu}}{c_0 l_0}, \quad (90)$$

where $\tilde{\nu}$ is the kinematic viscosity in physical units. We write the quantities ν , λ with tildes if they are expressed in physical units.

Because of the nature of intermolecular collisions there is no well-defined definition of the mean free path. The conventional solution to this problem is to consider a model gas with hard sphere molecules, where the mean free path can be expressed exactly [4,6,47]. For the present study we use Cercignani's definition of the mean free path based on the viscosity [6,47]

$$\tilde{\lambda} = \sqrt{\frac{\pi}{2}} \frac{\tilde{\nu}}{c_0}, \quad (91)$$

which is very close to the analytical result of a hard sphere gas.³ Thus, Kn is given by

$$\text{Kn} = \frac{\tilde{\lambda}}{l_0} = \sqrt{\frac{\pi}{2}} \frac{\tilde{\nu}}{c_0 l_0} = \sqrt{\frac{\pi}{2}} \tau, \quad (92)$$

which is—up to a factor—identical to the relaxation time τ .

³The CE result for the mean free path in a hard sphere gas is $\tilde{\lambda} = \frac{16}{5\sqrt{2\pi}} \frac{\tilde{\nu}}{c_0}$ [48].

For the interaction of gas molecules with a solid wall surface we implement the diffuse Maxwell boundary condition (58) for a nonmoving rigid wall using the expanded equilibrium function (84)

$$f_\alpha(\mathbf{x}_w, t) = \Psi f_\alpha^{(0)}(\rho_w, \mathbf{u} = 0), \quad (93)$$

with

$$\Psi = \frac{\sum_{\mathbf{n} \cdot \boldsymbol{\xi}_\alpha < 0} f_\alpha |\mathbf{n} \cdot \boldsymbol{\xi}_\alpha|}{\sum_{\mathbf{n} \cdot \boldsymbol{\xi}_\beta > 0} f_\beta^{(0)}(\rho_w, 0) |\mathbf{n} \cdot \boldsymbol{\xi}_\beta|}. \quad (94)$$

Unknown distribution functions near solid walls which cannot be calculated by the standard propagation step are set to the distribution function of the diffusive Maxwell boundary condition (93). For the test case investigated here, the density is uniform and the flow behavior is steady and unidirectional, which implies $\Psi = 1$ [30]. As for the half-way bounce-back scheme, the wall is located at a distance of half a lattice spacing from the first fluid collision center [22,43]. For a numerical calculation the boundary condition (93) needs to be transformed with Eqs. (83) and (87) into a corresponding form for the modified distribution function \bar{f}_α .

VI. POISEUILLE FLOW

In this section, we check DVMs for their capability of describing Poiseuille flow for various values of Kn. The DVMs considered here are organized in the sets \mathcal{S}_1 , \mathcal{S}_3 , and \mathcal{S}_4 shown in Table I, to be discussed in Secs. VIA, VIB, and VIC, respectively. This order is chosen with increasing value of the wall accuracy index Λ_{index} representing a more accurate computation of the wall moments. It should be possible to find a DVM with all wall moments evaluated exactly, i.e., wall accuracy order $\Lambda = Q$, following the procedure given in Sec. IVC and solving the wall equation (67) as a constraint. However, since the stencil would then become prohibitively large, we relax some of these constraints and minimize the remaining wall errors instead.

Two parallel plates are located at $z = \pm l_0/2$ and the flow is driven by a constant pressure gradient, ρg , in the x direction. We use periodic boundary conditions in the x and y directions. The body force is small enough so that we can assume low-Ma flow. Due to symmetry, the only nontrivial velocity component is $u(z) = u_x(\mathbf{x})$. Below, we discuss the slip velocity at the wall,

$$u_s = \frac{u(l_0/2)}{u(0)}, \quad (95)$$

as well as the normalized mass flow rate,

$$\dot{m} = \frac{1}{4u_c \text{Kn}} \frac{1}{B} \int_0^B dy \frac{1}{l_0} \int_{-l_0/2}^{l_0/2} dz u(z), \quad (96)$$

where B is the extension of the flow domain in the y direction. The Navier-Stokes equation with no-slip boundary condition, $u_s = 0$, yields the centerline velocity $u_c = u(0) = g l_0^2 / (8\tilde{\nu})$

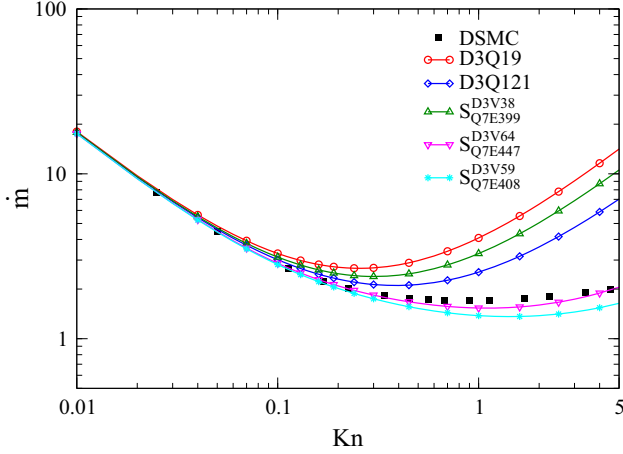


FIG. 2. Mass flow rate for common LB models and some exclusive \mathcal{S}_1 models ($\Delta_{index} = 0$).

and the mass flow rate

$$\dot{m} = \frac{1}{6Kn}. \quad (97)$$

A. DVMs of set \mathcal{S}_1 without wall constraints

Figures 2 and 3 show the mass flow rate and the slip velocity at the wall for two known LB models, $D3Q19$ and $D3Q121$, and some new models of set \mathcal{S}_1 compared with DSMC data. For small Kn the $D3Q19$ model (accuracy order $Q = 5$) and the $D3Q121$ model (accuracy order $Q = 9$) agree very well with the reference data. However, for higher Kn both models exhibit strong deviations from the DSMC results. Similarly, the new model S_{Q7E399}^{D3V38} with an accuracy order $Q = 7$ and a minimal number of 38 velocities fails for higher Kn . Although the high-order models $D3Q121$ and S_{Q7E399}^{D3V38} recover flow regimes beyond the Navier-Stokes level (for a detailed discussion see Sec. III), the results remain unsatisfactory for finite Kn .

The new LB models of set \mathcal{S}_1 are of Gaussian quadrature order $Q = 7$ and are thus able to recover the momentum

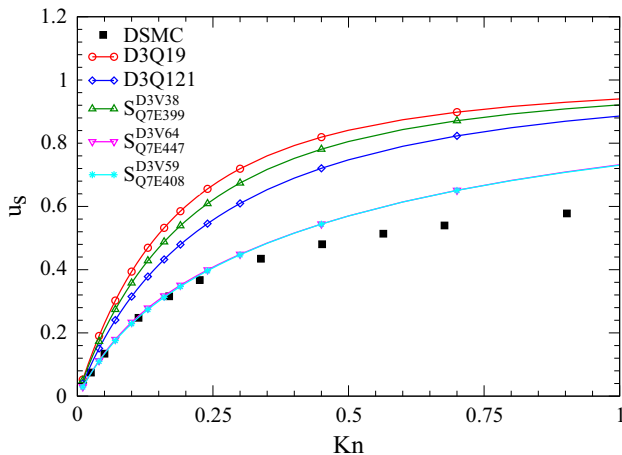


FIG. 3. Slip velocity for common LB models and some exclusive \mathcal{S}_1 models ($\Delta_{index} = 0$).

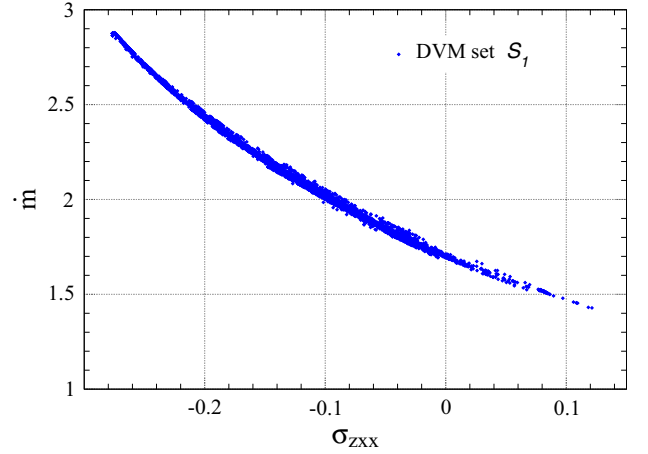


FIG. 4. Correlation between σ_{zxx} and \dot{m} for $Kn = 0.4514$. Every dot represents a DVM within \mathcal{S}_1 .

dynamics for small Ma up to the fourth flow regime [$\Delta\Pi_{ij}^{(4)} = O(Ma^2)$]. If we analyze these models, we observe for finite Kn , nevertheless, quite different results and considerable deviations from the DSMC results for both the mass flow rate and the slip velocity. The Gaussian quadrature order Q is very important to recover high-order flow regimes, but not sufficient to guarantee accurate results of a LB model for finite Kn . This is also found for high-order LB models in $D = 2$ dimensions [30].

The reason for the failure of many high-order LB models for finite Kn is their weak ability to recover the diffuse Maxwell boundary condition exactly. We assess this ability by considering the errors of the wall moments in Table II. The exact definition of the models listed here can be found in Appendix A. We here identified DVMs of the set \mathcal{S}_1 with wall moment errors as small as possible, which thus ensure a good realization of the diffuse Maxwell boundary condition. For the two different models S_{Q7E447}^{D3V64} and S_{Q7E408}^{D3V59} some wall moment errors are given in Table II and the flow results are shown in Figs. 2 and 3. Due to smaller wall moment errors, we observe for both models a significantly higher accuracy for finite Kn .

The errors in the slip velocity are nearly the same, whereas the predicted mass flow rate of the model S_{Q7E447}^{D3V64} is more accurate. By using the CE analysis (see Sec. III) one can

TABLE II. Wall moment errors of standard LB model $D3Q19$, the common DVM S_{Q9E594}^{D3V121} (also known as $D3Q121$), as well as some exclusive \mathcal{S}_1 models. Q is the quadrature order.

DVM	Q	σ	σ_z	σ_{xx}	σ_{zxx}	σ_Σ
$D3Q19$	5	-0.6667	-0.2764	-0.6667	-0.2764	0.5506
$D3Q121$	9	-0.4767	-0.1292	-0.4767	-0.1292	0.3685
S_{Q7E399}^{D3V38}	7	-0.4902	-0.1505	-0.5926	-0.2109	0.4042
S_{Q7E447}^{D3V64}	7	-0.0443	-0.0750	-0.1212	-0.0258	0.0621
S_{Q7E408}^{D3V59}	7	-0.0367	0.0932	-0.0794	0.0658	0.0576

TABLE III. Wall moment errors of some exclusive \mathcal{S}_3 models, where Q is the quadrature order.

LB model	Q	σ	σ_z	σ_{xx}	σ_{zxx}	σ_{zzxx}	σ_{xxzzzz}	σ_Σ
$S_{Q7E1023}^{D3V107}$	7	-0.0998	0.0247	-0.1482	0.0	0.0027	-0.0083	0.0849
S_{Q7E672}^{D3V77}	7	-0.0816	0.0343	-0.1339	0.0	0.0132	-0.0710	0.0765

show that the relevant low-Ma contributions to the momentum dynamics up to the super-Burnett flow regime are affected by the third and fifth equilibrium moments. Therefore, the higher accuracy of model S_{Q7E447}^{D3V64} in the mass flow prediction is caused by a more accurate representation of the third wall moment W_{zxx} (see Table II).

If we consider the influence of the wall moment error σ_{zxx} on the mass flow rate within all \mathcal{S}_1 models, we observe a strong correlation between σ_{zxx} and \dot{m} . This is shown in Fig. 4 for a Knudsen number of $\text{Kn} = 0.4514$. Pearson’s linear correlation coefficient yields the value -0.98 , exceeding in magnitude all other correlations of \dot{m} . It is interesting to notice that all models with a nearly vanishing wall moment error σ_{zxx} predict a mass flow rate which is very close to the DSMC results of $\dot{m}_{DSMC} = 1.76$ for $\text{Kn} = 0.4514$.

These results confirm that a reliable LB model for finite Kn flows requires both a high quadrature order Q and small wall moment errors, which guarantees a precise realization of the diffuse Maxwell boundary condition.

B. DVMs of set \mathcal{S}_3 with wall constraints

In this section we discuss the set \mathcal{S}_3 of DVMs. These models are of Gaussian order $Q = 7$ and include a wall moment constraint which ensures that the σ_{zxx} error vanishes. Consequently, the wall accuracy index has the value $\Lambda_{index} = 16$; see Eq. (75). From this set of models we select the $S_{Q7E1023}^{D3V107}$ and S_{Q7E672}^{D3V77} models, which additionally have a small overall wall moment error σ_Σ (see Table III). The results for the mass flow rate, shown in Fig. 5, are in excellent agreement

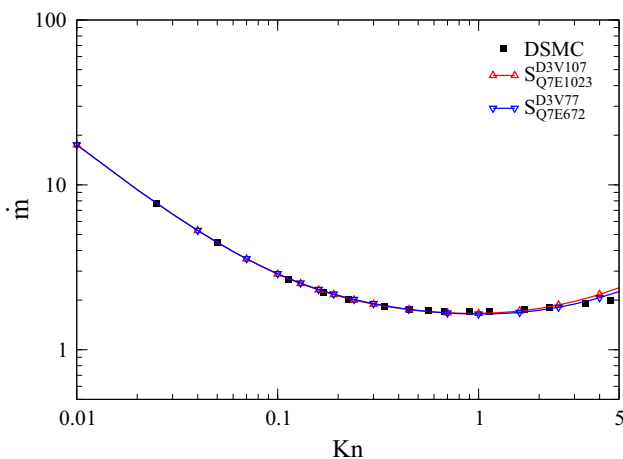


FIG. 5. Mass flow rate of selected \mathcal{S}_3 models ($\Lambda_{index} = 16$).

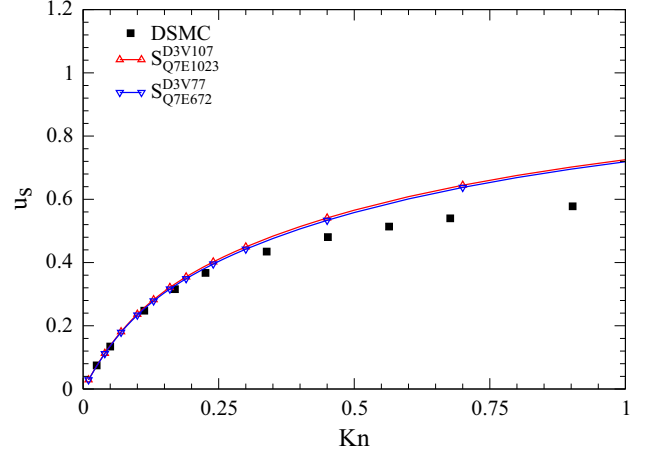


FIG. 6. Slip velocity of selected \mathcal{S}_3 models ($\Lambda_{index} = 16$).

with the DSMC results due to the wall constraint $\sigma_{zxx} = 0$. Additionally, the Knudsen minimum at $\text{Kn} \approx 1$ is very well reproduced. For the slip velocity (Fig. 6) we observe slight differences to the reference data which may be caused by remaining inaccuracies of other wall moments.

C. DVMs of set \mathcal{S}_4 with wall constraints

Another set of wall moment LB models, \mathcal{S}_4 , is characterized by a Gaussian quadrature order $Q = 7$ and scattering stencils which guarantee that all even wall moments up to the quadrature order $Q = 7$ are represented exactly,

$$\sigma = \sigma_{i_1 i_2} = \sigma_{i_1 \dots i_4} = \sigma_{i_1 \dots i_6} = 0. \quad (98)$$

Consequently, the wall accuracy index has the value $\Lambda_{index} = 6245$; see Eq. (81). Similar to the model sets \mathcal{S}_1 and \mathcal{S}_3 discussed previously, we observe for \mathcal{S}_4 models a strong correlation between the wall moment W_{zxx} and the flow results as well. Figure 7 shows the correlation between σ_{zxx} and the mass flow and Fig. 8 shows the correlation between σ_{zxx} and

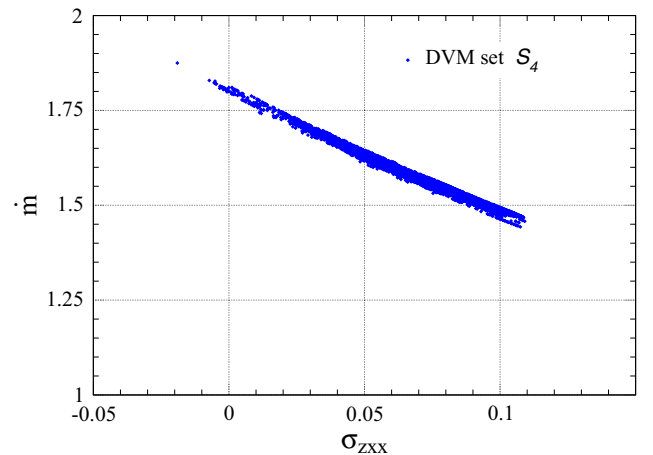


FIG. 7. Correlation between σ_{zxx} and \dot{m} for $\text{Kn} = 0.4514$. Every dot represents a DVM within \mathcal{S}_4 .

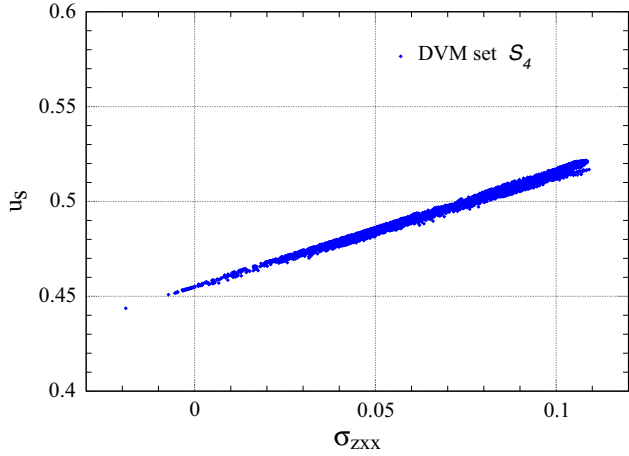


FIG. 8. Correlation between σ_{zxx} and u_s for $Kn = 0.4514$. Every dot represents a DVM within \mathcal{S}_4 .

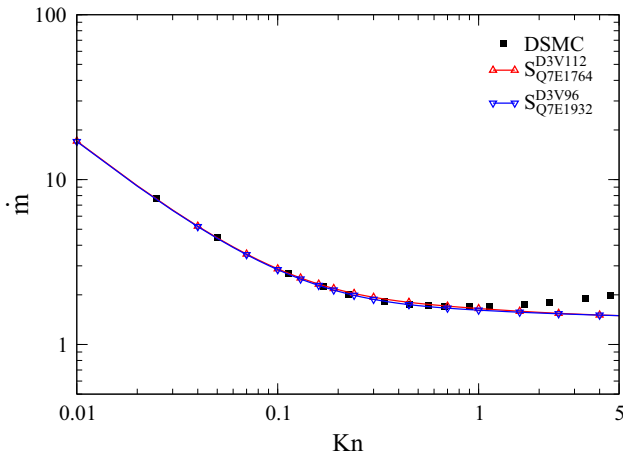


FIG. 9. Mass flow rate of selected \mathcal{S}_4 models ($\Lambda_{index} = 6245$).

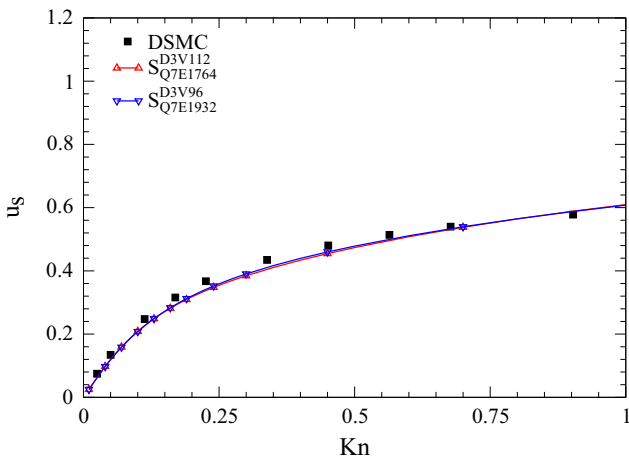


FIG. 10. Slip velocity of selected \mathcal{S}_4 models ($\Lambda_{index} = 6245$).

TABLE IV. Wall moment errors of some exclusive \mathcal{S}_4 models, where Q is the quadrature order.

LB model	Q	σ	σ_z	σ_{xx}	σ_{zxx}	σ_Σ
$S_{Q7E1764}^{D3V112}$	7	0.0	0.0373	0.0	0.000005	0.0090
$S_{Q7E1932}^{D3V96}$	7	0.0	0.0386	0.0	0.0114	0.0101

the slip velocity. Because of the strengths of these correlations we select the models $S_{Q7E1764}^{D3V112}$ and $S_{Q7E1932}^{D3V96}$ with small errors $|\sigma_{zxx}|$ (see Table IV). The Poiseuille flow results for these models are shown in Figs. 9 and 10. Both the mass flow rate and the slip velocity at the wall are in excellent agreement with the DSMC results up to $Kn = 1$. The crucial point for the success of these models is very low errors for all wall moments up to the quadrature order ($Q = 7$). However, these models cannot predict the Knudsen minimum which was reproduced by models of groups \mathcal{S}_1 and \mathcal{S}_3 .

D. Velocity profiles

Figures 11–20 show the streamwise velocities of the considered LB models for several Knudsen numbers. At $Kn = 0.05$ all models agree well with the DSMC results; however, $D3Q19$ and S_{Q7E399}^{D3V38} slightly overpredict the slip velocity at the walls. For higher Kn the models $D3Q19$, $D3Q121$, and S_{Q7E399}^{D3V38} no longer perform well due to errors of the wall moments. On the other hand, the models S_{Q7E447}^{D3V77} , S_{Q7E672}^{D3V96} , and $S_{Q7E1932}^{D3V96}$ show a significantly better prediction of the velocity field because of more accurate wall moments, yielding a more precise realization of the diffuse Maxwell boundary condition.

In particular, the $S_{Q7E1932}^{D3V96}$ model guarantees a high accuracy for all wall moments up to the Gaussian quadrature order $Q = 7$ and therefore remains quantitatively accurate at least up to $Kn \approx 1$. Only for Knudsen numbers higher than $Kn = 2$ do we observe a slight overestimation of the slip velocity.

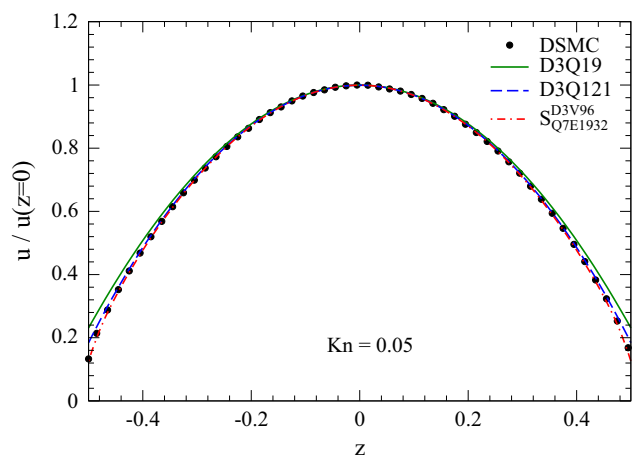


FIG. 11. Streamwise velocity profile for $Kn = 0.05$.

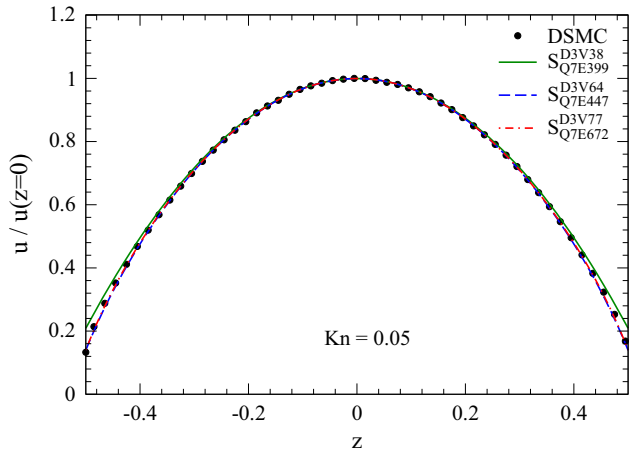


FIG. 12. Streamwise velocity profile for $Kn = 0.05$.

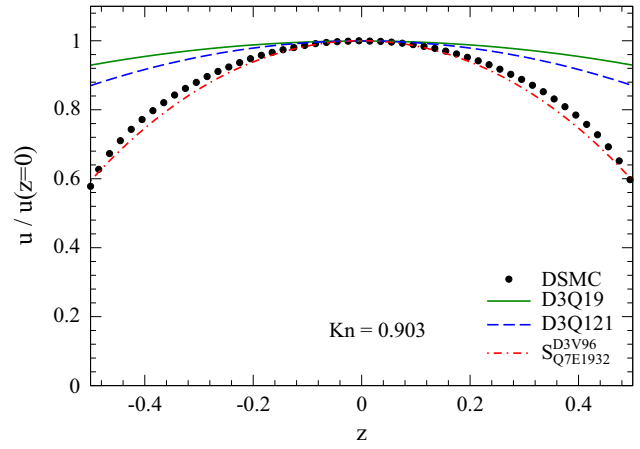


FIG. 15. Streamwise velocity profile for $Kn = 0.903$.

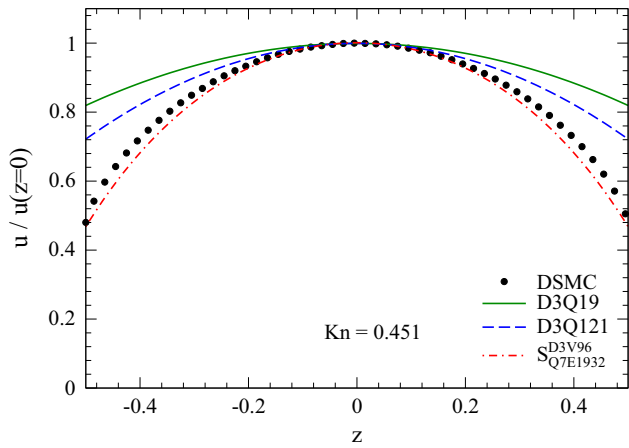


FIG. 13. Streamwise velocity profile for $Kn = 0.451$.

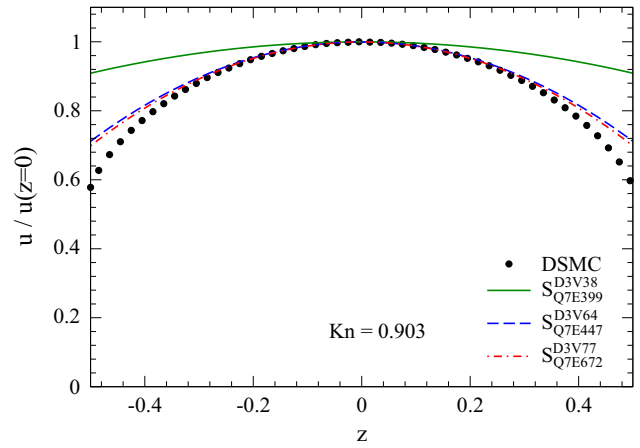


FIG. 16. Streamwise velocity profile for $Kn = 0.903$.

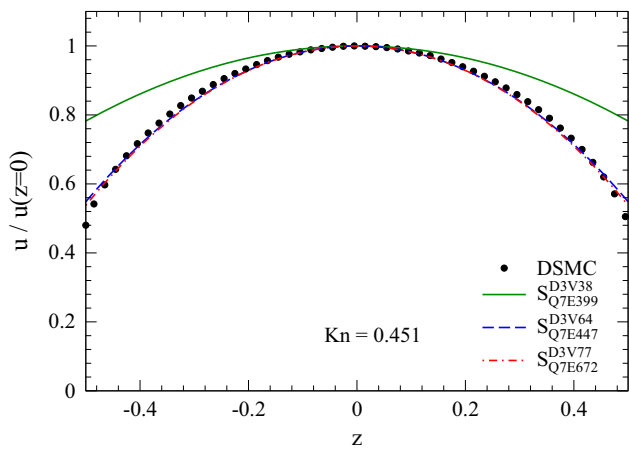


FIG. 14. Streamwise velocity profile for $Kn = 0.451$.

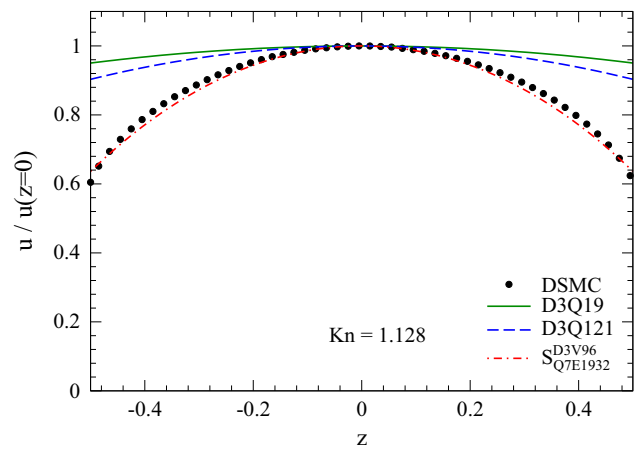


FIG. 17. Streamwise velocity profile for $Kn = 1.128$.

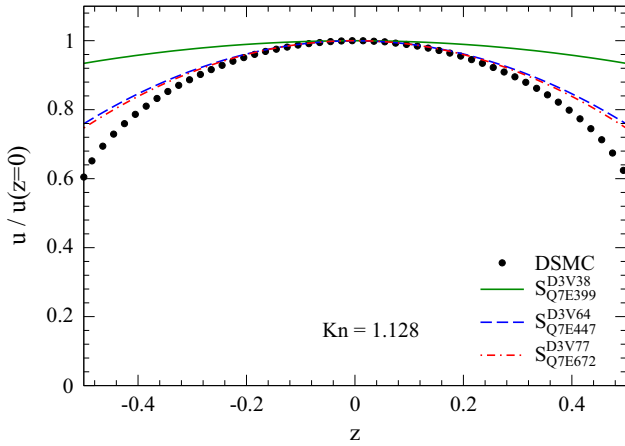


FIG. 18. Streamwise velocity profile for $Kn = 1.128$.

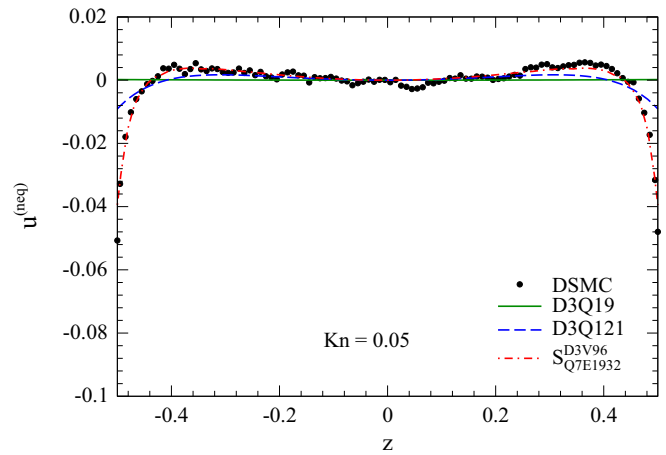


FIG. 21. Nonequilibrium velocity profile for $Kn = 0.05$.

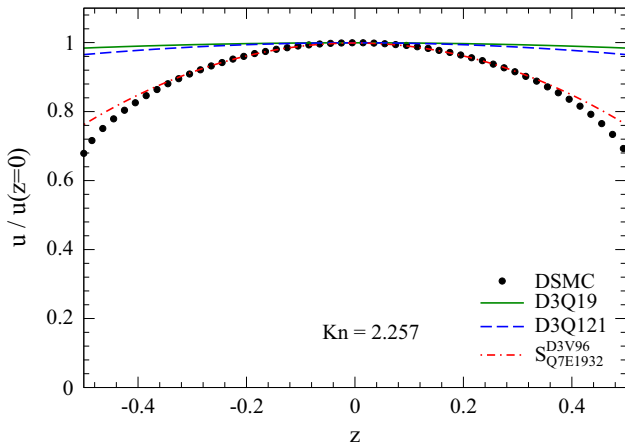


FIG. 19. Streamwise velocity profile for $Kn = 2.257$.

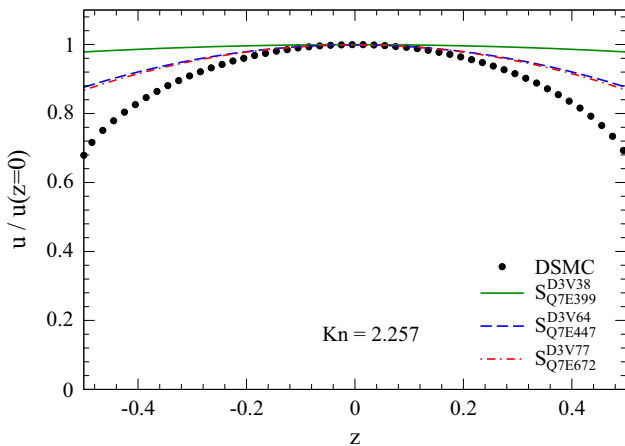


FIG. 20. Streamwise velocity profile for $Kn = 2.257$.

E. Knudsen layer

An even stronger requirement for the DVMs than correct evaluation of mass flow and slip velocity is to yield the correct velocity profile beyond the Navier-Stokes flow regime. For this purpose, we define for each solution $u(z)$ a quadratic velocity profile by

$$u^{(NS)}(z) = u(0) \left\{ 1 - 3 \left[1 - \frac{1}{l_0} \int_{-l_0/2}^{l_0/2} dz' \frac{u(z')}{u(0)} \right] \left(\frac{z}{l_0/2} \right)^2 \right\}, \tag{99}$$

which gives the same mass flow as $u(z)$. The nonequilibrium content of $u(z)$ is measured by its deviation from the quadratic profile (99),

$$u^{(neq)}(z) = u(z) - u^{(NS)}(z). \tag{100}$$

Using the DSMC data for $u(z)$, we find a reference function $u^{(neq)}(z)$ for each Kn number the DVMs compete with. As shown for $Kn = 0.05$ in Fig. 21, the DSMC data show a small velocity defect at the vicinity of the wall—the Knudsen layer—such that $u^{(neq)} < 0$.

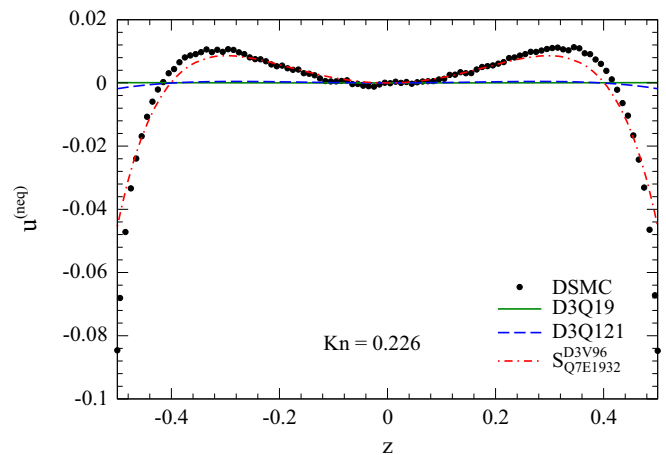


FIG. 22. Nonequilibrium velocity profile for $Kn = 0.226$.

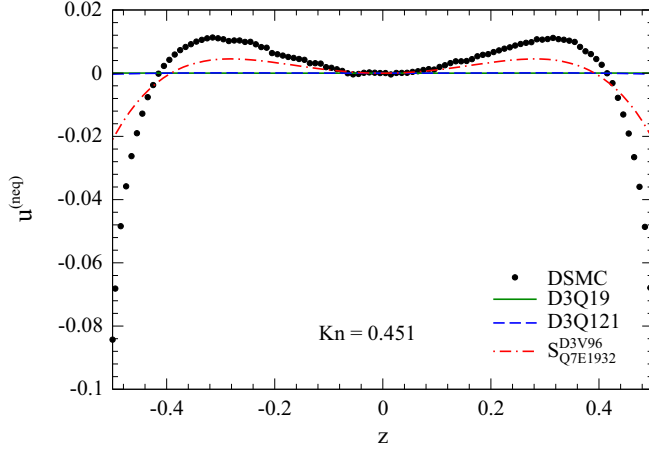


FIG. 23. Nonequilibrium velocity profile for $\text{Kn} = 0.451$.

In turn, there is an excess of nonequilibrium velocity in the bulk, $u^{(neq)} > 0$, by definition of Eq. (99). The standard model $D3Q19$ is not able to show a Knudsen layer, due to its low quadrature order, $Q = 5$. It yields a strictly quadratic profile with $u^{(neq)}(z) = 0$. The model $D3Q121$ with higher quadrature order, $Q = 9$, shows some velocity defect; however, it does not quite match the DSMC result for its wall moments are inaccurate. On the other hand, the DVM $S_{Q7E1932}^{D3V96}$, which has both a high quadrature order and (almost) correct wall moments, shows excellent agreement with DSMC data in the Knudsen layer for $\text{Kn} = 0.05$.

This hierarchy of DVMs remains for higher Kn numbers. For $\text{Kn} = 0.226$, the Knudsen layer is more pronounced; see Fig. 22. While the models $D3Q19$ and $D3Q121$ are unsatisfactory, the model $S_{Q7E1932}^{D3V96}$ is able to describe the Knudsen layer. Increasing the Kn number further, high-order flow regimes gain more importance, causing the DVMs presented here to cease to be valid for evaluating the nonequilibrium velocity profile. This is exemplified for $\text{Kn} = 0.451$ in Fig. 23.

VII. CONCLUSION

In this paper we presented high-order LB models for solving the Boltzmann-BGK equation for finite Kn number flows.

It was shown how to derive new discrete velocity models (DVMs) for any quadrature order using an efficient algorithm. The energy of a stencil was bounded from above in order to be able to define complete sets of minimal DVMs. These sets comprise more than 50 000 models with seventh quadrature order in $D = 3$ spatial dimensions. In future investigations, the collection of minimal discrete velocity models for high-order LB simulations can be systematically extended.

We analytically derived a theorem via the Chapman-Enskog expansion which enables us to identify the recovered flow regimes of any LB model, i.e., Navier-Stokes, Burnett, super-Burnett, and so on. For isothermal flows, this theorem rigorously relates the velocity space discretization error of a velocity moment for low Ma values with the quadrature order Q of a LB model. Thus, one can tell from the quadrature order Q which flow regime is exactly recovered by the LB model for $\text{Ma} \rightarrow 0$. In particular, it was shown for isothermal flows that even the standard models with $Q = 5$ recover the Burnett

momentum dynamics for $\text{Ma} \rightarrow 0$. The seventh-order LB models we focused on here recover the momentum flux tensor of nonequilibrium gas flows up to the fourth flow regime for $\text{Ma} \rightarrow 0$.

Aside from nonequilibrium effects in the bulk, wall-bounded high- Kn -number flows require a LB model to correctly describe the gas-surface interaction. We observed that several high-order LB models show significant deviations from reference results because of their poor ability to recover the diffuse Maxwell boundary condition accurately. In order to characterize this capability we defined an analytical constraint for a DVM for the exact implementation of the diffuse Maxwell boundary condition. It was shown how to generate in a systematic way high-order DVMs with the inherent ability to fulfill the diffuse Maxwell boundary condition. Alternatively, it is possible to arrive at discrete velocity models which obey the constraint for even velocity moments at the wall by simply using scattering stencils. The wall accuracy index Λ_{index} was put forward to label the velocity moments at the wall evaluated correctly by a discrete velocity model. The seventh-order model $S_{Q7E1932}^{D3V96}$, e.g., was found with a wall accuracy index $\Lambda_{index} = 6245$ and an almost exact representation of the wall moments up to the seventh order. The circumstance that models with even numbers of discrete velocities perform better than for odd numbers can be due to the low net weight of erroneous wall-parallel (and zero) velocities in half-space quadratures. For an exact integration of particle velocities emitted by the wall, we derived a sufficient condition on the wall accuracy order Λ .

At finite Kn , we compared our Poiseuille flow LB results with those of DSMC. We found that the agreement is strongly correlated with the wall constraint on the one hand and requires the use of high-order DVMs ($Q \geq 7$) on the other. Therefore, the correct evaluation of high- Kn -number Poiseuille flow requires both a sufficiently high quadrature order and an exact representation of the diffuse Maxwell boundary condition. Consequently, the model $S_{Q7E1932}^{D3V96}$ yields an excellent agreement with DSMC for both the mass flow and the slip velocity, using Kn numbers up to $\text{Kn} \simeq 1$. In the velocity profiles one can clearly recognize that the discrete velocity model $S_{Q7E1932}^{D3V96}$ recovers the Knudsen layer in the vicinity of the wall for $\text{Kn} \lesssim 0.3$, while standard discrete velocity models utterly fail. The increased computational effort compared to the commonly used standard model $D3Q19$ is about a factor of 5–6. Nevertheless, this is still much faster than any DSMC simulation, especially for low-Mach-number flows, where DSMC methods require a large number of samples to reduce statistical errors. Thus, we recommend using $S_{Q7E1932}^{D3V96}$ for finite Kn number flows.

For $\text{Kn} \gtrsim 0.3$, the results for the nonequilibrium velocity profile departs from the DSMC results, because the flow regimes are only recovered up to the fourth Chapman-Enskog level and there are remaining errors in high-order wall moments. However, we expect that for LB models with a sufficiently high quadrature order and wall accuracy order, a more precise representation of the Knudsen layer phenomenon can be obtained. This will be part of our future investigations. Moreover, it remains to be seen whether the Knudsen layer turns out to be a nonperturbative phenomenon in the sense of the Chapman-Enskog expansion.

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APPENDIX A: THREE-DIMENSIONAL DVMS

Here we present DVMS generated by the algorithm shown in Sec. II and used in the introductory example and the numerical test case in Sec. VI (see Tables V–XV). The stencil's symbol is explained in Eq. (24), c is the lattice speed, and g counts the stencil groups S_g generated by the symmetries of the lattice from a single velocity $\xi^{(g)}$, yielding V_g velocities. Each velocity in S_g is weighted by \bar{w}_g [see Eq. (18)] when using the quadrature prescription (10).

APPENDIX B: TWO-DIMENSIONAL DVMS

In this appendix, we present minimal DVMS for $D = 2$ (see Tables XVI–XVIII). As shown in Table I above, there are three sets of DVMS concerned with $D = 2$. The y component is absent. For all but the wall accuracy index Λ_{index} , the definitions are the same as for $D = 3$. Since there are no y components of wall moments, we have to set $m_y = 0$ in Eq. (63). The wall accuracy index is then defined as before [see Eq. (73)], but there are only $L = 14$ components in the vector,

$$\mathbf{T}^\alpha = (1, \xi_{\alpha z}, \xi_{\alpha x}^2, \xi_{\alpha z}^3, \xi_{\alpha z} \xi_{\alpha x}^2, \xi_{\alpha x}^4, \xi_{\alpha z}^5, \xi_{\alpha z}^3 \xi_{\alpha x}^2, \xi_{\alpha z} \xi_{\alpha x}^4, \xi_{\alpha x}^6, \xi_{\alpha z}^7, \xi_{\alpha z}^5 \xi_{\alpha x}^2, \xi_{\alpha z}^3 \xi_{\alpha x}^4, \xi_{\alpha z} \xi_{\alpha x}^6), \quad (\text{B1})$$

whereas b_k are the components of the vector

$$\mathbf{b} = (W, W_z, W_{xx}, W_{zz}, W_{zx}, W_{xxx}, W_{zzzz}, W_{zzzx}, W_{zxxx}, W_{xxxx}, W_{zzzzzz}, W_{zzzzzx}, W_{zzzxxx}, W_{zxxxxx}); \quad (\text{B2})$$

cf. Eqs. (70) and (68).

The set S_5 comprises all 1188 minimal DVMS with quadrature order $Q = 7$ in the energy sphere given by $E \leq 250$. These are discrete velocity models with the wall accuracy index being $\Lambda_{index} = 0$. The DVM with lowest velocity count $V_{min} = 16$ is S_{Q7E58}^{D2V16} and it is shown in Table XVI. Increasing the quadrature order to $Q = 9$, we find 592 DVMS in the set S_6 , where $E_{max} = 300$. The DVM with the lowest velocity count in S_6 is S_{Q7E944}^{D2V32} (see Table XVII), and it yields $V = 33$. Finally, we identified 21 952 DVMS with quadrature order $Q = 7$ which have scattering stencils, i.e., their wall accuracy index is $\Lambda_{index} = 549$. The stencil energy is bounded for these models by $E_{max} = 1000$ to define the set S_7 . The DVM with

TABLE V. DVM used for the introductory example.

S_{Q5E24}^{D3V15}			
$c = 1.2247448713915890$			
g	$\xi^{(g)}/c$	V_g	\bar{w}_g
1	(0,0,0)	1	$3.888888888888889 \times 10^{-1}$
2	(2,0,0)	6	$2.777777777777778 \times 10^{-2}$
3	(1,1,-1)	8	$5.555555555555556 \times 10^{-2}$

TABLE VI. DVM with minimal velocity count within the set S_1 .

S_{Q7E399}^{D3V38}			
$c = 0.7500000000000000$			
g	$\xi^{(g)}/c$	V_g	\bar{w}_g
1	(1,0,0)	6	$7.2239858906525573 \times 10^{-2}$
2	(4,0,0)	6	$6.3648834019204390 \times 10^{-3}$
3	(2,2,0)	12	$4.3895747599451303 \times 10^{-2}$
4	(6,0,0)	6	$4.1805473904239336 \times 10^{-5}$
5	(4,4,4)	8	$1.7146776406035665 \times 10^{-4}$

TABLE VII. DVM with minimal velocity count within the set S_1 .

S_{Q7E219}^{D3V38}			
$c = 0.86602540378443865$			
g	$\xi^{(g)}/c$	V_g	\bar{w}_g
1	(1,0,0)	6	$6.7724867724867725 \times 10^{-2}$
2	(2,0,0)	6	$5.555555555555556 \times 10^{-2}$
3	(2,2,2)	8	$4.6296296296296296 \times 10^{-3}$
4	(2,2,0)	12	$1.8518518518518519 \times 10^{-2}$
5	(6,0,0)	6	$1.7636684303350970 \times 10^{-4}$

TABLE VIII. Selection of DVMS within the set S_1 . See Sec. VI for a discussion.

S_{Q7E408}^{D3V59}			
$c = 0.74685634388439233$			
g	$\xi^{(g)}/c$	V_g	\bar{w}_g
1	(0,0,0)	1	$2.0080700829205231 \times 10^{-2}$
2	(1,1,1)	8	$8.9344539381631413 \times 10^{-2}$
3	(4,0,0)	6	$1.8287317703258027 \times 10^{-3}$
4	(3,3,3)	8	$5.2273139486813077 \times 10^{-4}$
5	(3,3,0)	12	$2.3272910797607214 \times 10^{-3}$
6	(3,1,1)	24	$9.2533853908214560 \times 10^{-3}$

TABLE IX. Selection of DVMS within the set S_1 . See Sec. VI for a discussion.

S_{Q7E447}^{D3V64}			
$c = 0.69965342816864754$			
g	$\xi^{(g)}/c$	V_g	\bar{w}_g
1	(2,0,0)	6	$5.9646397884737016 \times 10^{-3}$
2	(1,1,1)	8	$8.0827437008387392 \times 10^{-2}$
3	(5,0,0)	6	$1.1345266793939999 \times 10^{-3}$
4	(3,3,3)	8	$9.5680047874015889 \times 10^{-4}$
5	(3,3,0)	12	$3.9787631334632013 \times 10^{-3}$
6	(3,1,1)	24	$1.0641080987258957 \times 10^{-2}$

TABLE X. DVM with minimal velocity count within the set \mathcal{S}_2 . For a description of the symbols, see Eq. (24).

S_{Q9E471}^{D3V79}		$c = 1.0000000000000000$	
g	$\xi^{(g)}/c$	V_g	\bar{w}_g
1	(0,0,0)	1	$1.0570987654320988 \times 10^{-1}$
2	(1,0,0)	6	$3.8095238095238095 \times 10^{-2}$
3	(2,0,0)	6	$2.8645833333333333 \times 10^{-2}$
4	(1,1,1)	8	$4.9479166666666667 \times 10^{-2}$
5	(2,2,2)	8	$5.2083333333333333 \times 10^{-4}$
6	(2,2,0)	12	$5.2083333333333333 \times 10^{-3}$
7	(6,0,0)	6	$2.7557319223985891 \times 10^{-6}$
8	(3,3,3)	8	$9.6450617283950617 \times 10^{-6}$
9	(3,1,1)	24	$1.3020833333333333 \times 10^{-3}$

TABLE XI. Common DVM ($D3Q121$ [24]) within the set \mathcal{S}_2 . For a description of the symbols, see Eq. (24).

S_{Q9E594}^{D3V121}		$c = 1.1969797703930744$	
g	$\xi^{(g)}/c$	V_g	\bar{w}_g
1	(0,0,0)	1	$3.0591622029486006 \times 10^{-2}$
2	(0,0,-1)	6	$9.8515951037263392 \times 10^{-2}$
3	(1,1,-1)	8	$2.7525005325638124 \times 10^{-2}$
4	(0,0,-3)	6	$3.2474752708807381 \times 10^{-4}$
5	(2,2,-2)	8	$1.8102175157637424 \times 10^{-4}$
6	(2,0,-2)	12	$4.2818359368108407 \times 10^{-4}$
7	(1,0,-2)	24	$6.1110233668334243 \times 10^{-3}$
8	(3,3,-3)	8	$6.9287508963860285 \times 10^{-7}$
9	(1,1,-3)	24	$1.0683400245939109 \times 10^{-4}$
10	(2,0,-3)	24	$1.4318624115480294 \times 10^{-5}$

TABLE XII. Selection of DVMs within the set \mathcal{S}_3 . See Sec. IV C for a discussion.

S_{Q7E672}^{D3V77}		$c = 0.62590566441325041$	
g	$\xi^{(g)}/c$	V_g	\bar{w}_g
1	(2,0,0)	6	$6.5178619315224175 \times 10^{-3}$
2	(1,1,1)	8	$5.8638132347907334 \times 10^{-2}$
3	(6,0,0)	6	$1.0066312290789269 \times 10^{-3}$
4	(3,3,3)	8	$2.1962967289288607 \times 10^{-3}$
5	(3,3,0)	12	$4.8042518606023833 \times 10^{-3}$
6	(3,1,1)	24	$1.5528601406828448 \times 10^{-2}$
7	(0,0,0)	1	$2.9560614762917757 \times 10^{-2}$
8	(4,4,0)	12	$6.8996146397277359 \times 10^{-4}$

TABLE XIII. Selection of DVMs within the set \mathcal{S}_3 . See Sec. IV C for a discussion.

$S_{Q7E1023}^{D3V107}$		$c = 0.61887631323925978$	
g	$\xi^{(g)}/c$	V_g	\bar{w}_g
1	(1,1,1)	8	$5.4242093013777495 \times 10^{-2}$
2	(2,1,0)	24	$2.4637212114133877 \times 10^{-3}$
3	(5,0,0)	6	$1.5469959015615954 \times 10^{-3}$
4	(3,3,3)	8	$1.9425678925800591 \times 10^{-3}$
5	(3,3,0)	12	$7.2337497640759286 \times 10^{-3}$
6	(3,1,1)	24	$1.4384326790070621 \times 10^{-2}$
7	(0,0,0)	1	$4.4998866720663948 \times 10^{-2}$
8	(6,2,2)	24	$2.1182172560744541 \times 10^{-4}$

TABLE XIV. Selection of DVMs within the set \mathcal{S}_4 . See Sec. IV C for a discussion.

$S_{Q7E1932}^{D3V96}$		$c = 0.37787639086813054$	
g	$\xi^{(g)}/c$	V_g	\bar{w}_g
1	(1,1,1)	8	$1.2655649299880090 \times 10^{-3}$
2	(3,3,3)	8	$2.0050978770655310 \times 10^{-2}$
3	(3,1,1)	24	$2.7543347614356814 \times 10^{-2}$
4	(4,4,4)	8	$4.9712543563172566 \times 10^{-3}$
5	(7,1,1)	24	$3.6439016726158895 \times 10^{-3}$
6	(6,6,1)	24	$1.7168180273737716 \times 10^{-3}$

TABLE XV. Selection of DVMs within the set \mathcal{S}_4 . See Sec. IV C for a discussion.

$S_{Q7E1764}^{D3V112}$		$c = 0.40531852273291520$	
g	$\xi^{(g)}/c$	V_g	\bar{w}_g
1	(1,1,1)	8	$3.3503407500643648 \times 10^{-3}$
2	(3,1,1)	24	$2.8894128958152456 \times 10^{-2}$
3	(4,4,4)	8	$4.5930345162087793 \times 10^{-3}$
4	(3,2,2)	24	$4.4163148398082762 \times 10^{-3}$
5	(7,1,1)	24	$2.3237070220062610 \times 10^{-3}$
6	(5,5,1)	24	$3.3847240912752922 \times 10^{-3}$

TABLE XVI. DVM with the lowest velocity count $V_{min} = 16$ in the set \mathcal{S}_5 , where $D = 2$, $Q = 7$, and $E \leq 250$ (see Table I).

S_{Q7E58}^{D2V16}		$c = 0.86602540378443865$	
g	$\xi^{(g)}/c$	V_g	\bar{w}_g
1	(1,0)	4	$1.5802469135802469 \times 10^{-1}$
2	(2,0)	4	$6.1728395061728395 \times 10^{-2}$
3	(2,2)	4	$2.7777777777777778 \times 10^{-2}$
4	(4,0)	4	$2.4691358024691358 \times 10^{-3}$

TABLE XVII. DVM with the lowest velocity count $V_{min} = 33$ in the set \mathcal{S}_6 , where $D = 2$, $Q = 9$, and $E \leq 300$ (see Table I).

S_{Q9E132}^{D2V33}		$c = 1.1587791906520175$	
g	$\xi^{(g)}/c$	V_g	\bar{w}_g
1	(0,0)	1	$1.6198651186147246 \times 10^{-1}$
2	(1,0)	4	$1.4320396528198750 \times 10^{-1}$
3	(1,1)	4	$3.3883996404301766 \times 10^{-2}$
4	(2,0)	4	$5.5611157082744134 \times 10^{-3}$
5	(2,2)	4	$8.4479885070276616 \times 10^{-5}$
6	(3,0)	4	$1.1325437650467775 \times 10^{-3}$
7	(2,1)	8	$1.2816907733721003 \times 10^{-2}$
8	(4,4)	4	$3.4555225091487045 \times 10^{-6}$

the lowest velocity count gives $V_{min} = 20$. Optimizing for both σ_{zx} and σ_Σ yields the DVM S_{Q7E944}^{D2V32} shown in Table XVIII. The latter is expected to give back excellent results for planar Poiseuille flow.

APPENDIX C: ERROR ESTIMATE OF EQUILIBRIUM MOMENTS

For a detailed analysis of the quadrature error of an equilibrium moment $\Delta M_{i_1 \dots i_k}^{(n)} = M_{i_1 \dots i_k}^{(n)} - M_{i_1 \dots i_k}^{(n)DVM}$ we distinguish four different cases.

Case 1: $k \leq N$ and $k + N \leq Q$

The Hermite order N is high enough to capture all contributions of the Hermite polynomials $\mathcal{H}_{i_1 \dots i_k}^{(k)}, \mathcal{H}_{i_1 \dots i_{k-2}}^{(k-2)}, \dots$ and the quadrature order guarantees an exact evaluation of all terms in Eq. (50). Thus, the moment $M_{i_1 \dots i_k}^{(0)}$ is exactly recovered and

$$\Delta M_{i_1 \dots i_k}^{(0)} = 0. \quad (C1)$$

Case 2: $k \leq N$ and $k + N > Q$

Although the Hermite order is high enough, the quadrature order $Q = n_0 + k$, with $n_0 < N$, is not high enough to evaluate $M_{i_1 \dots i_k}^{(0)}$ exactly. For low Ma values, the leading term of the quadrature error in Eq. (50) is the smallest value of n exceeding n_0 , due to the relation (48). This term is the one with $n =$

TABLE XVIII. DVM with optimal values for the wall errors in the set \mathcal{S}_7 , where $D = 2$, $Q = 7$, $E \leq 1000$, and $\sigma_{i_1 \dots i_{2n}} = 0$. (see Table I).

S_{Q7E944}^{D2V32}		$c = 0.34040702226615838$	
g	$\xi^{(g)}/c$	V_g	\bar{w}_g
1	(1,1)	4	$8.4201053650845727 \times 10^{-2}$
2	(2,2)	4	$5.0708714918479963 \times 10^{-2}$
3	(5,1)	8	$4.4541157350549509 \times 10^{-2}$
4	(6,4)	8	$1.2646044225450934 \times 10^{-2}$
5	(12,3)	8	$3.5791413933671213 \times 10^{-4}$

$n_0 + 1$ in Eq. (50) and we can write

$$\begin{aligned} \Delta M_{i_1 \dots i_k}^{(0)} &= \frac{1}{(Q - k + 1)!} a_{i_1 \dots i_{Q-k+1}}^{(0)} \sum_{\alpha} w_{\alpha} \mathcal{H}_{i_1 \dots i_{Q-k+1}}^{(Q-k+1)}(\xi_{\alpha}) \\ &\quad \times \mathcal{H}_{i_1 \dots i_k}^{(k)}(\xi_{\alpha}) + \text{subleading terms} \\ &= O(\text{Ma}^{Q-k+1}), \end{aligned} \quad (C2)$$

where the subleading terms contain errors with a higher Ma power compared to the first term. Obviously, the quadrature error $\Delta M_{i_1 \dots i_k}^{(0)}$ is finite for low Ma values; see Eq. (50). Consequently, we obtain $\Delta M_{i_1 \dots i_k}^{(0)} = O(\text{Ma}^0)$ for $Q - k + 1 < 0$.

Case 3: $k > N$ and $k + N \leq Q$

The Hermite order N is not high enough to capture the moment $M_{i_1 \dots i_k}^{(0)}$ completely, whereas the quadrature order guarantees the exact evaluation of all terms in Eq. (50). Therefore, errors are produced by the absence of terms in the equilibrium function which are beyond the Hermite order N . The leading term in the quadrature error is thus $\sim a_{i_1 \dots i_{N+1}}^{(0)}$ and with Eq. (48) we get

$$\Delta M_{i_1 \dots i_k}^{(0)} = O(\text{Ma}^{N+1}). \quad (C3)$$

Case 4: $k > N$ and $k + N > Q$

The Hermite order N is not high enough to recover $M_{i_1 \dots i_k}^{(0)}$ and the quadrature order Q does not guarantee an exact evaluation. The relevant quadrature error in the sense of the Ma expansion of the equilibrium function $f^{(0)}$ is determined analogously to case 2 by

$$\Delta M_{i_1 \dots i_k}^{(0)} = O(\text{Ma}^{Q-k+1}) \quad (C4)$$

for $Q - k + 1 \geq 0$ and $\Delta M_{i_1 \dots i_k}^{(0)} = O(\text{Ma}^0)$ otherwise.

These cases can be summarized into a more compact relation:

$$\Delta M_{i_1 \dots i_k}^{(0)} = \begin{cases} 0 & \text{for } k + N \leq Q \text{ and } k \leq N, \\ O(\text{Ma}^{N+1}) & \text{for } k + N \leq Q \text{ and } k > N, \\ O(\text{Ma}^{Q-k+1}) & \text{for } k - 1 \leq Q < k + N, \\ O(\text{Ma}^0) & \text{for } Q < k - 1. \end{cases} \quad (C5)$$

APPENDIX D: PROOF OF THE QUADRATURE ERROR THEOREM

In order to prove the theorem in Sec. III C, we assume that the moment $M_{i_1 \dots i_k}^{(n)}(\partial^{(0)} \rho, \partial_{j_1}^{(1)} \rho, \partial_{j_1 j_2}^{(2)} \rho, \dots, \partial^{(0)} u_i, \partial_{j_1}^{(1)} u_i, \partial_{j_1 j_2}^{(2)} u_i, \dots)$ is an algebraic expression of ρ, u_i , and their spatial derivatives, where $\partial^{(0)} = 1$ and $\partial_{j_1 \dots j_q}^{(q)} = \partial_{j_1} \dots \partial_{j_q}$. By using the relations (32) for the multiple-scale time derivatives, the equation for $M_{i_1 \dots i_k}^{(n)}$

can be written as

$$M_{i_1 \dots i_k}^{(n)} = -\tau \left[\partial_t^{(0)} M_{i_1 \dots i_k}^{(n-1)} - \frac{1}{\rho} \sum_{m=1}^{n-1} \sum_q \frac{\partial M_{i_1 \dots i_k}^{(n-m-1)}}{\partial (\partial_{j_1 \dots j_q}^{(q)} u_s)} \partial_{j_1 \dots j_q}^{(q)} \right. \\ \left. \times (\partial_r \Pi_{rs}^{(m)}) + \partial_j M_{i_1 \dots i_k}^{(n-1)} \right]. \quad (\text{D1})$$

The sum on $q \geq 0$ in Eq. (D1) accounts for all derivatives of ρ and u_i the moment $M_{i_1 \dots i_k}^{(n)}$ depends on.

In order to discuss the relevant truncation error for a CE moment $M_{i_1 \dots i_k}^{(n)}$, we show in the following that the multiple-scale time derivative $\partial_t^{(0)}$ in Eq. (D1) does not alter the Ma power of the truncation error of an inaccurate CE moment. For this purpose, we consider a CE moment $M_{i_1 \dots i_k}^{(n)}$ which is recovered by a Gauss-Hermite quadrature up to an error

$$\Delta M_{i_1 \dots i_k}^{(n)} = O(\text{Ma}^m), \quad (\text{D2})$$

with Ma power m . The time derivative $\partial_t^{(0)}$ of $M_{i_1 \dots i_k}^{(n)}$ is given by

$$\partial_t^{(0)} M_{i_1 \dots i_k}^{(n)} = \partial_t^{(0)} (M_{i_1 \dots i_k}^{(n)DVM} + \Delta M_{i_1 \dots i_k}^{(n)}) \\ = \partial_t^{(0)} M_{i_1 \dots i_k}^{(n)DVM} - \sum_q \frac{\partial \Delta M_{i_1 \dots i_k}^{(n)}}{\partial (\partial_{j_1 \dots j_q}^{(q)} \rho)} \partial_{j_1 \dots j_q}^{(q)} (\partial_j \rho u_j) \\ - \sum_q \frac{\partial \Delta M_{i_1 \dots i_k}^{(n)}}{\partial (\partial_{j_1 \dots j_q}^{(q)} u_j)} \partial_{j_1 \dots j_q}^{(q)} \left(u_s \partial_s u_j + \frac{1}{\rho} \partial_j \rho \right), \quad (\text{D3})$$

which implies

$$\Delta \partial_t^{(0)} M_{i_1 \dots i_k}^{(n)} = - \sum_q \frac{\partial \Delta M_{i_1 \dots i_k}^{(n)}}{\partial (\partial_{j_1 \dots j_q}^{(q)} \rho)} \partial_{j_1 \dots j_q}^{(q)} \partial_j (\rho u_j) \\ - \sum_q \frac{\partial \Delta M_{i_1 \dots i_k}^{(n)}}{\partial (\partial_{j_1 \dots j_q}^{(q)} u_j)} \partial_{j_1 \dots j_q}^{(q)} \left(\underbrace{u_s \partial_s u_j}_{O(\text{Ma}^2)} + \frac{1}{\rho} \underbrace{\partial_j \rho}_{O(\text{Ma})} \right). \quad (\text{D4})$$

For finite Kn flows density gradients $\partial_j \rho$ can be of order $O(\text{Ma})$, even in the low-Ma flow regime. Consequently, we find that the time derivative $\partial_t^{(0)}$ does not lower the order in Ma number of the error term (D2)

$$\Delta \partial_t^{(0)} M_{i_1 \dots i_k}^{(n)} = \partial_t^{(0)} \Delta M_{i_1 \dots i_k}^{(n)} = O(\text{Ma}^m). \quad (\text{D5})$$

The error of the time derivative of a moment equals the time derivative of the moment error.

Using the recurrence relation (D1) for a CE moment $M_{i_1 \dots i_k}^{(n)}$, the error estimate (52) for equilibrium moments and relation (D5) for the multiple-scale derivative $\partial_t^{(0)}$, we prove the theorem in Sec. III C by induction. Based on Eq. (D1), the truncation error of $M_{i_1 \dots i_k}^{(1)}$ is determined by

$$\Delta M_{i_1 \dots i_k}^{(1)} = -\tau [\Delta \partial_t^{(0)} M_{i_1 \dots i_k}^{(0)} + \partial_{j_1} \Delta M_{i_1 \dots i_k j_1}^{(0)}]. \quad (\text{D6})$$

Because of Eq. (52) the relevant error with respect to the Ma power is produced by the highest equilibrium moment, which

is contained in the second term on the right-hand side. The first term does not contribute to the dominant truncation error, because it consists of a lower equilibrium moment and the multiple-scale derivative $\partial_t^{(0)}$ does not change the Ma power of this error term. Consequently, we find

$$\Delta M_{i_1 \dots i_k}^{(1)} = -\tau \partial_{j_1} \Delta M_{i_1 \dots i_k j_1}^{(0)} + \text{subleading terms} \quad (\text{D7})$$

and with Eq. (52) the estimate

$$\Delta M_{i_1 \dots i_k}^{(1)} = \begin{cases} 0 & \text{for } 2(k+1) + 1 \leq Q, \\ O(\text{Ma}^{Q-k}) & \text{for } k \leq Q < 2(k+1) + 1, \\ O(\text{Ma}^0) & \text{for } Q < k. \end{cases} \quad (\text{D8})$$

For the next CE moment we obtain from Eq. (D1)

$$M_{i_1 \dots i_k}^{(2)} = -\tau \left[\partial_t^{(0)} M_{i_1 \dots i_k}^{(1)} - \frac{1}{\rho} \sum_q \frac{\partial M_{i_1 \dots i_k}^{(0)}}{\partial (\partial_{j_1 \dots j_q}^{(q)} u_s)} \partial_{j_1 \dots j_q}^{(q)} (\partial_r \Pi_{rs}^{(1)}) \right. \\ \left. + \partial_j M_{i_1 \dots i_k j}^{(1)} \right]. \quad (\text{D9})$$

The error of the first term on the right-hand side is given by Eq. (D8) where the time derivative $\partial_t^{(0)}$ does not affect the Ma power,

$$\Delta \partial_t^{(0)} M_{i_1 \dots i_k}^{(1)} = \partial_t^{(0)} \Delta M_{i_1 \dots i_k}^{(1)} \\ = \begin{cases} 0 & \text{for } 2(k+1) + 1 \leq Q, \\ O(\text{Ma}^{Q-k}) & \text{for } k \leq Q < 2(k+1) + 1, \\ O(\text{Ma}^0) & \text{for } Q < k. \end{cases} \quad (\text{D10})$$

With respect to Eq. (52) we get for the error of the next term

$$\Delta \frac{\partial M_{i_1 \dots i_k}^{(0)}}{\partial (\partial_{j_1 \dots j_q}^{(q)} u_s)} = \begin{cases} 0 & \text{for } 2k + 1 \leq Q, \\ O(\text{Ma}^{Q-k}) & \text{for } k \leq Q < 2k + 1, \\ O(\text{Ma}^0) & \text{for } Q < k, \end{cases} \quad (\text{D11})$$

where we have to take into account that the Ma power of the error term is decreased by one by the derivative $\partial/\partial(\partial_{j_1 \dots j_q}^{(q)} u_s)$. Based on Eq. (D8), we obtain

$$\Delta \Pi_{rs}^{(1)} = \begin{cases} 0 & \text{for } 7 \leq Q, \\ O(\text{Ma}^{Q-2}) & \text{for } 2 \leq Q < 7, \\ O(\text{Ma}^0) & \text{for } Q < 2, \end{cases} \quad (\text{D12})$$

and for the last term in Eq. (D9),

$$\Delta M_{i_1 \dots i_k j}^{(1)} = \begin{cases} 0 & \text{for } 2(k+2) + 1 \leq Q, \\ O(\text{Ma}^{Q-k-1}) & \text{for } k+1 \leq Q < 2(k+2) + 1, \\ O(\text{Ma}^0) & \text{for } Q < k+1. \end{cases} \quad (\text{D13})$$

Due to $k \geq 2$, the error estimate of all terms in Eq. (D9) implies that the relevant quadrature error of $M_{i_1 \dots i_k}^{(2)}$ is contained in the last term of Eq. (D9) and thus we get

$$\Delta M_{i_1 \dots i_k}^{(2)} = -\tau \partial_{j_2} \Delta M_{i_1 \dots i_k j_2}^{(1)} + \text{subleading terms} \\ = (-\tau)^2 \partial_{j_1} \partial_{j_2} \Delta M_{i_1 \dots i_k j_1 j_2}^{(0)} + \text{subleading terms}. \quad (\text{D14})$$

In the last step of the proof we have to show the relation (53) for an integer n assuming that the relation is

valid for $n - 1, n - 2, \dots, 1$. We analyze the error of all terms on the right-hand side of Eq. (D1),

$$\Delta M_{i_1 \dots i_k}^{(n)} = -\tau \left[\Delta \partial_t^{(0)} M_{i_1 \dots i_k}^{(n-1)} - \Delta \left\{ \frac{1}{\rho} \sum_{m=1}^{n-1} \sum_q \frac{\partial M_{i_1 \dots i_k}^{(n-m-1)}}{\partial (\partial_{j_1 \dots j_q}^{(q)} u_s)} \partial_{j_1 \dots j_q}^{(q)} (\partial_r \Pi_{rs}^{(m)}) \right\} + \partial_j \Delta M_{i_1 \dots i_k}^{(n-1)} \right]. \quad (\text{D15})$$

Using Eq. (53) for $n - 1$, we find

$$\begin{aligned} \Delta \partial_t^{(0)} M_{i_1 \dots i_k}^{(n-1)} &= \partial_t^{(0)} \Delta M_{i_1 \dots i_k}^{(n-1)} = (-\tau)^{n-1} \partial_t^{(0)} \partial_{j_1} \dots \partial_{j_{n-1}} \Delta M_{i_1 \dots i_k j_1 \dots j_{n-1}}^{(0)} + \text{subleading terms} \\ &= \begin{cases} 0 & \text{for } 2(k+n-1) + 1 \leq Q, \\ O(\text{Ma}^{Q-k-n+2}) & \text{for } k+n-2 \leq Q < 2(k+n-1) + 1, \\ O(\text{Ma}^0) & \text{for } Q < k+n-2, \end{cases} \end{aligned} \quad (\text{D16})$$

where $\partial_t^{(0)}$ does not affect the Ma power of the error term. Furthermore, we can estimate the error of $\partial M_{i_1 \dots i_k}^{(n-m-1)} / \partial (\partial_{j_1 \dots j_q}^{(q)} u_s)$ for $m = 1, \dots, n - 1$ by using Eq. (53) for $n - m - 1$ in combination with Eq. (52),

$$\Delta \frac{\partial M_{i_1 \dots i_k}^{(n-m-1)}}{\partial (\partial_{j_1 \dots j_q}^{(q)} u_s)} = \begin{cases} 0 & \text{for } 2(k+n-m-1) + 1 \leq Q, \\ O(\text{Ma}^{Q-k-n+m+1}) & \text{for } k+n-m-1 \leq Q < 2(k+n-m-1) + 1, \\ O(\text{Ma}^0) & \text{for } Q < k+n-m-1, \end{cases} \quad (\text{D17})$$

where the derivatives $\partial / \partial (\partial_{j_1 \dots j_q}^{(q)} u_s)$ reduce the Ma power of the error term by one. The error of the quantity $\Pi_{rs}^{(m)}$ for $m = 1, \dots, n - 1$ can be determined by applying the theorem (53) for m ,

$$\begin{aligned} \Delta \Pi_{rs}^{(m)} &= (-\tau)^{m+2} \partial_{j_1} \dots \partial_{j_m} \Delta M_{rs j_1 \dots j_m}^{(0)} + \text{subleading terms} \\ &= \begin{cases} 0 & \text{for } 2(m+2) + 1 \leq Q, \\ O(\text{Ma}^{Q-m-1}) & \text{for } m+1 \leq Q < 2(m+2) + 1, \\ O(\text{Ma}^0) & \text{for } Q < m+1, \end{cases} \end{aligned} \quad (\text{D18})$$

where $m = 1, \dots, n - 1$. The error of the last term of Eq. (D1) is given by

$$\begin{aligned} \Delta M_{i_1 \dots i_k j}^{(n-1)} &= (-\tau)^{n-1} \partial_{j_1} \dots \partial_{j_{n-1}} \Delta M_{i_1 \dots i_k j j_1 \dots j_{n-1}}^{(0)} + \text{subleading terms} \\ &= \begin{cases} 0 & \text{for } 2(k+n) + 1 \leq Q, \\ O(\text{Ma}^{Q-k-n+1}) & \text{for } k+n-1 \leq Q < 2(k+n) + 1, \\ O(\text{Ma}^0) & \text{for } Q < k+n-1, \end{cases} \end{aligned} \quad (\text{D19})$$

where we have used Eq. (52) and applied Eq. (53) for $n - 1$. Because of $k \geq 2$ and $m = 1, \dots, n - 1$ the estimate of all errors occurring in Eq. (D15) implies that the last term on the right-hand side of Eq. (D15) contains the relevant error with respect to the Ma power. This yields

$$\begin{aligned} \Delta M_{i_1 \dots i_k}^{(n)} &= -\tau \partial_{j_n} \Delta M_{i_1 \dots i_k j_n}^{(n-1)} + \text{subleading terms} \\ &= (-\tau)^n \partial_{j_1} \dots \partial_{j_n} \Delta M_{i_1 \dots i_k j_1 \dots j_n}^{(0)} + \text{subleading terms} \end{aligned} \quad (\text{D20})$$

and therefore the theorem is proved for any CE level n .

APPENDIX E: MAXWELL BOUNDARY FACTOR Ψ

We now turn to the evaluation of the factor Ψ [see Eq. (59)] of the diffuse Maxwell boundary condition. Assuming that the CE expansion [cf. Eq. (28)] is applicable, we write

$$\Psi = \sum_{n=0} \Psi^{(n)}, \quad (\text{E1})$$

with

$$\Psi^{(n)} = \frac{\sum_{\mathbf{n} \cdot \xi_\alpha < 0} f_\alpha^{(n)} |\mathbf{n} \cdot \xi_\alpha|}{\sum_{\mathbf{n} \cdot \xi_\beta > 0} f_\beta^{(0)} (\rho_w, 0) |\mathbf{n} \cdot \xi_\beta|}, \quad (\text{E2})$$

where we have set the expansion parameter $\epsilon = 1$. The denominator in Eq. (E2),

$$Z = \sum_{\mathbf{n} \cdot \xi_\beta > 0} f_\beta^{(0)} (\rho_w, 0) |\mathbf{n} \cdot \xi_\beta| = \rho_w n_i W_i^{DVM}, \quad (\text{E3})$$

is captured exactly by the DVM if the wall accuracy order yields $\Lambda \geq 1$. The numerator in Eq. (E2) can be analyzed by using Eq. (31),

$$\Psi^{(0)} = -\frac{n_i}{Z} Y_i^{(0)}, \quad (\text{E4a})$$

$$\Psi^{(1)} = -\frac{n_i}{Z} (-\tau) [\partial_t^{(0)} Y_i^{(0)} + \partial_j Y_{ij}^{(0)}], \quad (\text{E4b})$$

$$\Psi^{(2)} = -\frac{n_i}{Z} [(-\tau)^2 (\partial_t^{(0)} \partial_t^{(0)} Y_i^{(0)} + 2\partial_t^{(0)} \partial_j Y_{ij}^{(0)} + \partial_j \partial_k Y_{ijk}^{(0)}) + (-\tau) \partial_t^{(1)} Y_i^{(0)}], \quad (\text{E4c})$$

$$\dots$$

$$\Psi^{(n)} = \Psi^{(n)}(Y_{i_1}^{(0)} \dots Y_{i_{n+1}}^{(0)}), \quad (\text{E4d})$$

where we have introduced the moments

$$Y_{i_1 \dots i_k}^{(0)} = \sum_{\mathbf{n} \cdot \xi_\alpha < 0} f_\alpha^{(0)}(\rho_w, \mathbf{u}) \xi_{\alpha i_1} \dots \xi_{\alpha i_k}. \quad (\text{E5})$$

The equilibrium (16) expanded up to the Hermite order N can be expressed by

$$f_\alpha^{(0)}(\rho_w, \mathbf{u}) = w_\alpha \sum_{n=0}^N \frac{1}{n!} a_{i_1 \dots i_n}^{(0)} \mathcal{H}_{i_1 \dots i_n}^{(n)}(\xi_\alpha)$$

$$= w_\alpha \sum_{n=0}^N b_{i_1 \dots i_n} \xi_{\alpha i_1} \dots \xi_{\alpha i_n}, \quad (\text{E6})$$

with some coefficients $b_{i_1 \dots i_n}$. Thus, the moments $Y_{i_1 \dots i_k}^{(0)}$ are determined by the equilibrium wall moments,

$$Y_{i_1 \dots i_k}^{(0)} = \sum_{n=0}^N (-1)^{n+k} b_{j_1 \dots j_n} W_{j_1 \dots j_n i_1 \dots i_k}^{DVM}. \quad (\text{E7})$$

In general, the n th CE contribution of Ψ is a function of the moments $Y_{i_1 \dots i_k}^{(0)}$ with $k \leq n+1$ [see Eq. (E4)], which can be directly expressed by the equilibrium wall moments $W_{i_1 \dots i_m}^{DVM}$ with $m \leq N+n+1$. As a consequence, a sufficiently high accuracy of the equilibrium wall moments, characterized by the wall accuracy order Λ , guarantees an exact evaluation of the Maxwell boundary factor Ψ up to the n th CE level with

$$n = \Lambda - N - 1. \quad (\text{E8})$$

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