Prandtl and Rayleigh number dependence of the Reynolds number in turbulent thermal convection

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The Prandtl and Rayleigh number dependences of the Reynolds number in turbulent thermal convection following from the unifying theory by Grossmann and Lohse [J. Fluid Mech. **407**, 27 (2000); Phys. Rev. Lett. **86**, 3316 (2001)] are presented and compared with various recent experimental findings. This dependence Re(Ra,Pr) is more complicated than a simple global power law. For Pr=5.5 and $10^8 < Ra < 10^{10}$ the effective or local power law exponent of Re as a function of Ra is definitely less than 0.50, namely, $Re \sim Ra^{0.45}$, in agreement with Qiu and Tong's experimental findings [Phys. Rev. E **64**, 036304 (2001)]. We also calculated the kinetic boundary layer width. Both in magnitude and in Ra scaling it is consistent with the data.

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I. INTRODUCTION

Recently, Qiu and Tong [1] measured the Reynolds number Re= UL/ν of the large scale velocity ("wind velocity") U in thermal convection as a function of the Rayleigh number Ra= $\alpha_p g L^3 \Delta/\nu\kappa$ for a fixed Prandtl number Pr= ν/κ = 5.5 (water). Here α_p is the isobaric thermal expansion coefficient, g the gravitational acceleration, Δ the temperature difference between the warmer bottom plate and the colder top plate, L the height of the aspect ratio Γ =1 cell, ν the kinematic viscosity, and κ the thermal diffusivity.

Depending on the precise definition of the wind velocity U one has to consider slightly different Reynolds numbers. Qiu and Tong [1] take a Reynolds number Re_{γ} , which is based on a direct measurement of the shear γ throughout the cell. Other Reynolds numbers have been based on the frequency peak in the temperature power spectrum [2,3] which has been associated with the large scale convection roll. In the Ra range 10^8-10^{10} Qiu and Tong [1,4] find the approximate power law $\text{Re}_{\gamma}=0.085\text{Ra}^{0.455}$ [5]. This exponent is surprising since most hitherto measured data for this Prandtl number or for Pr=0.7 (helium gas) give a power law exponent closer to 0.50 [2,3,6].

Qiu and Tong [1] also measured the oscillation frequency f_0^{θ} in the time dependence of the temperature crosscorrelation function between the temperatures measured at two points on opposite locations in the Rayleigh-Bénard (RB) cell. These oscillations are considered as a signature of plume transport with the circulating wind, i.e., as a measure of the recurrence rate U/2L and thus of the wind velocity U[1]. With this frequency they define a Reynolds number $\operatorname{Re}_{f^{\theta}} \equiv 2f_0^{\theta}L^2/\nu$. With $f_0^{\theta} \approx U/2L$ it can be connected with the standard definition of the Reynolds number $\operatorname{Re}_{f^{\theta}} \approx UL/\nu$ $= \operatorname{Re}$. This Reynolds number $\operatorname{Re}_{f^{\theta}}$ turns out to agree with the above Reynolds number based on the shear rate $\operatorname{Re}_{\gamma}$ $\equiv \gamma L^2/(2\nu)$ (see Fig. 2 of Ref. [1]). Indeed, with $\gamma \approx 2U/L$ one gets $\operatorname{Re}_{\gamma} \approx \operatorname{Re}$. In a very recent report Lam *et al.* [7] also present experimental results for the Rayleigh number dependence of the Reynolds number, but in addition also for its Prandtl number dependence. For $10^8 < \text{Ra} < 3 \times 10^{10}$ they find the effective scaling law Re~Ra^{0.43}Pr^{-0.76} for a Reynolds number based on the frequency f_0^u extracted from the autocorrelation function of the velocity fluctuations, $4f_0^u L^2/\nu$. Note that this expression differs by a factor of 2 from that of Ref. [1]. To allow for easier comparison, we drop this factor 2 and define $\text{Re}_{f^u} \equiv 2f_0^u L^2/\nu$ which, using the data from [7], is $\text{Re}_{f^u} = 0.545 \text{Ra}^{0.43} \text{Pr}^{-0.76}$. The *Ra* dependence is consistent with Qiu and Tong's findings $\text{Re}_{\gamma} = \text{Re}_{f^{\theta}} = 0.085 \text{Ra}^{0.455}$ [1,4]. Also the absolute values agree rather well: For water (Pr=5.5) at Ra=10⁹ one has $\text{Re}_{f^u} = 1106$ according to [7] and $\text{Re}_{f^{\theta}} = 1058$ according to [1].

In addition, Lam *et al.* [7] measure a Reynolds number based on the velocity maximum in the cell, $\text{Re}_{U,max} \equiv U_{max}L/\nu$. The *Ra* exponent of that Reynolds number appears to vary as a function of the Prandtl number from 0.50 to 0.68 as Pr increases from 3 to 1200. The power law exponent is definitely steeper than that for the velocity autocorrelation based Reynolds number Re_{f^u} .

II. THE REYNOLDS NUMBER

These experimental findings have triggered us to finally present the Ra and Pr number dependence of Re, as it follows from our unifying theory of thermal convection [8,9].

The main idea of that theory is to decompose the energy dissipation rate ϵ_u and the thermal dissipation rate ϵ_{θ} into their boundary layer (BL) and bulk contributions,

$$\boldsymbol{\epsilon}_{u} = \boldsymbol{\epsilon}_{u,BL} + \boldsymbol{\epsilon}_{u,bulk}, \qquad (1)$$

$$\boldsymbol{\epsilon}_{\theta} = \boldsymbol{\epsilon}_{\theta, BL} + \boldsymbol{\epsilon}_{\theta, bulk} \,. \tag{2}$$

For the left-hand sides the exact relations $\epsilon_u = (\nu^3/L^4)$ (Nu -1)RaPr⁻² and $\epsilon_{\theta} = \kappa (\Delta^2/L^2)$, Nu are employed, where κ is the thermal diffusivity and Δ the temperature difference between the bottom and top plates. The individual contributions on the right-hand sides of Eqs. (1) and (2) are modeled

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in terms of the large scale velocity U, the temperature difference Δ , the height L, and the widths λ_u and λ_θ of the kinetic and thermal boundary layers (see Refs. [8,9]). Note that $\epsilon_{u,BL}$ contains contributions from both the top and bottom plates as well as from the sidewalls. From visualizations of the thermal boundary layers we conclude that $\epsilon_{\theta,BL}$ mainly contains top and bottom plate contributions as the sidewalls are thermally isolated. The thermal dissipation in the sidewall boundary layers scales like the bulk dissipation, which is proportional to RePr, and therefore can be included into $\epsilon_{\theta,bulk}$ [10]. For the thicknesses of the thermal and kinetic boundary layers we assume

 $\lambda_{\theta} = L/(2Nu)$

and

(3)

$$\lambda_u = aL/\sqrt{\text{Re}},\tag{4}$$

respectively. The latter expression holds in laminar boundary layers of Prandtl-Blasius type [see [11] or [12], Sec. 39, Eq. (39.12)]. In Refs. [8,9] we adopted a = 1/4 for the dimensionless prefactor that determines the absolute width of the kinetic boundary layer. The final result is [see Eqs. (13) and (14) of Ref. [9]]:

Nu Ra Pr⁻² =
$$c_1 \frac{\text{Re}^2}{g(\sqrt{\text{Re}_c/\text{Re}})} + c_2 \text{Re}^3$$
, (5)

$$\operatorname{Nu} = c_{3} \operatorname{Re}^{1/2} \operatorname{Pr}^{1/2} \left[f \left(\frac{2a \operatorname{Nu}}{\sqrt{Re_{c}}} g \left(\sqrt{\frac{\operatorname{Re}_{c}}{\operatorname{Re}}} \right) \right) \right]^{1/2} + c_{4} \operatorname{Pr} \operatorname{Re} f \left(\frac{2a \operatorname{Nu}}{\sqrt{\operatorname{Re}_{c}}} g \left(\sqrt{\frac{\operatorname{Re}_{c}}{\operatorname{Re}}} \right) \right).$$
(6)

The crossover functions *f* and *g* model the crossover from the thermal boundary layer nested in the kinetic one towards the inverse situation and from Eq. (4) toward $\lambda_u \sim L$, respectively; for details, see Ref. [9]. The four prefactors c_i , i = 1,2,3,4, and the transition Reynolds number Re_c to the large Pr regime (cf. [9]) were fitted to 151 experimental data points Nu(Ra,Pr), measured by Ahlers and Xu [13] for an aspect ratio $\Gamma = 1$ cell, still assuming a = 0.25.

Note that once *a* is chosen and the c_i and Re_c are fixed through the fit, not only is Nu(Ra,Pr) determined for all Ra and Pr, but also Re(Ra,Pr). However, while Nu(Ra,Pr) does *not* depend on the choice of a = 1/4, the wind amplitude Re(Ra, Pr) *does* depend on the value of *a*. The reason is the invariance of Eqs. (5) and (6) under the following rescaling transformation: Re $\rightarrow \alpha$ Re, Re_c $\rightarrow \alpha$ Re_c, $a \rightarrow \alpha^{1/2}a$, $c_1 \rightarrow c_1/\alpha^2$, $c_2 \rightarrow c_2/\alpha^3$, $c_3 \rightarrow c_3/\alpha^{1/2}$, and $c_4 \rightarrow c_4/\alpha$.

As meanwhile reliable results for Re(Ra,Pr) are known, we use this amplitude rescaling freedom in Eqs. (5) and (6) to fix the prefactor *a* and thus the width of the kinetic boundary layer. This procedure does not affect the functional dependence of Re on Ra and Pr. It only fixes an overall factor for Re(Ra,Pr). The Nusselt number, in contrast, is not affected at all.



FIG. 1. Re/Ra^{1/2} vs Ra for Pr=5.5 from the experiments of Ref. [1] (thick dots and dashed line as best fit Re=0.085Ra^{0.455}) and from the theory of Ref. [9] with a=0.482 (solid line). A local power law fit to the theoretical result in the Ra regime shown gives Re=0.102 Ra^{0.447}.

To determine the prefactor *a*, we take Qiu and Tong's experimental finding $\text{Re}=Re_{\gamma}=0.085 \text{ Ra}^{0.455}$ [1]; see Fig. 1. The result is a=0.482 rather than a=0.25 as was suggested in [12] and also assumed in our earlier publications [8,9]. With this choice of *a*, the coefficients in Eqs. (5) and (6) are $c_1=8.7$, $c_2=1.45$, $c_3=0.46$, $c_4=0.013$, and $\text{Re}_c=1.0.^1$

In Fig. 1 we show Re/Ra^{1/2} vs Ra as resulting from the theory together with the measured data of Refs. [1,4]. Indeed, the theoretical data for Re(Ra) also display a local power law exponent which is significantly smaller than 1/2. A power law fit in the Ra regime 10^8-10^{10} gives Re = 0.102 Ra^{0.447}, with a power law exponent very close to what has been observed by Qiu and Tong [1].

For completeness we also show again the phase diagram in the Ra-Pr space (see Fig. 2). This is identical to Fig. 1 of Ref. [9], with two exceptions. (i) Some additional experimental data points have been added. (ii) As we now took a = 0.482 rather than a = 1/4 in relation (4) in order to adapt the overall factor in Re to the measured values of Fig. 1, the shear Reynolds number of the kinetic boundary layer

$$\operatorname{Re}_{s} = \lambda_{u} U / \nu = a \sqrt{\operatorname{Re}}$$
⁽⁷⁾

also changes as compared to [9]. For a given wind Reynolds number Re the shear Reynolds number Re_s increases with *a*. Now, since according to [12], Sec. 41, the kinetic BL suffers a transition from laminar to turbulent for $\text{Re}_s \approx 420$, the larger choice of *a* implies that the onset of shear triggered turbulence in the BL happens already for smaller Re. Therefore the dotted line in the phase diagram, which signals this onset of turbulence in the kinetic BL, shifts upward.² The

¹Applying the above rescaling of Re, Re_c, and the c_i , these numbers exactly correspond to the numbers given in [9]. The rescaling factor to be taken is $\alpha = (a/0.25)^2 = 3.72$.

²Note again that all heat current plots Nu(Ra,Pr) of Ref. [9] remain completely unchanged through the modification of the value a in Eq. (4).



FIG. 2. Phase diagram in the Ra-Pr plane according to the theory of Ref. [9] The upper solid line means $\text{Re}=\text{Re}_c$, the lower nearly parallel solid line is $\epsilon_{u,BL} = \epsilon_{u,bulk}$, the curved solid line is $\epsilon_{\theta,BL} = \epsilon_{\theta,bulk}$, and the long-dashed line is the line $\lambda_u = \lambda_{\theta}$. The dotted line indicates where the laminar kinetic BL becomes turbulent, based on a critical shear Reynolds number $\text{Re}_s^* \approx 420$ of the kinetic BL. Beyond that line, the definition (4) of λ_u does not make sense anymore, and therefore no line $\lambda_u = \lambda_{\theta}$ is drawn there. Data points where Nu has been measured or numerically calculated have been included (for several aspect ratios): squares Chavanne *et al.* [14]; diamonds; Cioni *et al.* [15]; circles Niemela *et al.* [16]; stars; Ahlers and Xu [13]; down triangles, Xia *et al.* [17]; up triangles, Verzicco and Camussi (numerical simulations) [18].

location of this (dotted) turbulence transition line is discussed at the end of this paper in more detail.

The phase diagram Fig. 2 offers an understanding of why the scaling exponent of Re with Ra is smaller than 0.50 for Pr = 5.5. The exponent 0.447 originates in the combined contributions from the regimes IV_{μ} (with the pure power law exponent 4/9), IV_1 (with the pure power law exponent 1/2), and II_{μ} and II_{I} (with the pure power law exponents 2/5). Incidentally, the (unweighted) mean of the scaling exponents 4/9, 1/2, 2/5, and 2/5 in the Ra range $10^8 - 10^{10}$ gives 0.44, very close to the numerical finding in Fig. 1. For larger Ra, beyond 10^{10} , the influence of the regimes $II_{l,u}$ will decrease and that of $IV'_{u,l}$ will become stronger. Thus we expect a larger local exponent. If IV_l (with 1/2), IV_u (with 4/9), and IV'_{μ} (with 1/2) dominate, the unweighted mean is 0.48. We emphasize that generally there is no pure power law behavior but, instead, a more general dependence of Re on Ra, cf. Fig. 3.

For other Pr our theory predicts a *local* exponent 1/2. In Fig. 3(a) we have plotted Re/Ra^{1/2} vs Ra for various Pr. The local exponent 1/2 can be noticed, e.g., for small Pr $\approx 10^{-3} - 10^{-4}$ and large Ra $\gtrsim 10^{13}$ (regime IV_u), or for large Pr $\approx 10^2$ and small Ra $\approx 10^6$ (regime I_u).

It is evident from Fig. 3(a) that there is no pure power law throughout the whole Ra regime. As stressed in Refs. [8,9], power laws only apply *locally*. The same holds for the dependence of Re on Pr. In Fig. 3(b) we show this dependence



FIG. 3. (a) Re/Ra^{1/2} vs Ra for various Pr and (b) Re vs Pr for various Ra, according to our unifying theory. The dashed parts of the lines indicate the region beyond the expected transition of the kinetic BL into the turbulent state. There the II'_l, IV'_l, and IV'_u exponents will influence and change the curves: Nu \propto Ra^{1/2}Pr^{1/2} and Re \propto Ra^{1/2}Pr^{-1/2}, as in IV_l.

as it results from our theory. The Reynolds number strongly decreases with increasing Pr. This reflects the increasing effects of the viscosity on the wind as Pr grows.

The absence of global scaling exponents is highlighted in Fig. 4 in which we present the local slopes $d \log_{10}(\text{Re})/d \log_{10}(\text{Ra})$ and $d \log_{10}(\text{Nu})/d \log_{10}(\text{Ra})$ resulting from our theory: Indeed, neither one of these local scaling exponents is constant over a larger range for hitherto experimentally achievable Rayleigh and Prandtl numbers.

We now compare our results for Re(Ra,Pr) with the recent experimental findings by Lam *et al.* [7] in the regime 10^8 <Ra $<3\times10^{10}$ and 3<Pr<1200. For this Prandtl number regime we find [cf. Fig. 3(b)] effective (local) power laws Re \sim Pr $^{-0.70}$ for Ra= 10^{10} and Re \sim Pr $^{-0.73}$ for Ra= 10^{8} .³ The corresponding experimental local power law exponent of the Reynolds number Re_{fu} based on the oscillation frequency f_0^u

³These exponents can be understood as originating from the physics in regime IV_u (with the Prandtl dependence exponent -2/3) together with some influence of the regimes I_u and II_u (with Pr exponents -5/6 and -2/3).



FIG. 4. Local slopes $d \log_{10}(\text{Re})/d \log_{10}(\text{Ra})$ (a) and $d \log_{10}(\text{Nu})/d \log_{10}(\text{Ra})$ (b) for five Prandtl numbers $\text{Pr}=10^{-4}$, $\text{Pr}=10^{-2}$, $\text{Pr}=10^0$ (thick line), $\text{Pr}=10^2$, and $\text{Pr}=10^4$ as they result from our theory. As in Fig. 3 the dashed parts of the lines indicate the region beyond the expected transition of the kinetic BL into the turbulent state.

is -0.76 ± 0.01 . For the maximum velocity based Reynolds number Re_{*U,max*} the corresponding exponent varies between -0.88 and -0.94. Vice versa, for the Rayleigh number regime $10^8 < \text{Ra} < 3 \times 10^{10}$ we find [see Fig. 3(a)] effective (local) power laws Re~Ra^{0.44} for Pr=3 and Re~Ra^{0.48} for Pr = 1200. Lam *et al.*'s [7] experimental local exponents are 0.43 for an oscillation based Reynolds number Re_{*f*^u} (consistent with the findings from Qiu and Tong for Re_{*f*^u}) and between 0.50 and 0.68 for the maximum velocity based Reynolds number Re_{*U,max*}.

To summarize, our theory is in good agreement with (i) the Reynolds number Re_{f^u} based on the circulation frequency f_0^u as measured in [7], (ii) the Reynolds number $\operatorname{Re}_{f^\theta}$ based on the circulation frequency f_0^θ as measured in [1,4], and (iii) the Reynolds number $\operatorname{Re}_{\gamma}$ based on the shear rate γ_h as measured in [1] [which agrees with the method (ii) according to [1]]. In contrast, the Reynolds number $\operatorname{Re}_{U,max}$ based on the maximum velocity as measured in [7] shows a slightly stronger dependence on both Ra and Pr than is suggested by our theory.

Our theory (as yet) does not distinguish between these various Reynolds numbers. For $Ra \rightarrow \infty$ one would expect that they all show the same scaling behavior, but at finite and



FIG. 5. $\lambda_u(\text{Ra,Pr})/L$, according to our unifying theory, in the same regime $10^8 < \text{Ra} < 3 \times 10^{10}$ (a) and 3 < Pr < 1200 (b) as experimentally examined by Lam *et al.* [7]. The numbers on the curves are the local power law exponents of the solid curves. The dashed line is the experimental finding $\lambda_u/L = 3.6\text{Ra}^{-0.26}$ [19] for the sidewall thickness of the kinetic boundar layer for $\text{Pr} \approx 7$. The data points measured in that publication [19] are given as filled circles.

relatively small $Ra \approx 10^8 - 10^{10}$ deviations might occur, which apparently is the case here.

III. KINETIC BOUNDARY LAYER THICKNESS

Next, we compare with the measured thickness of the *top* and bottom kinetic boundary layers in an aspect ratio $\Gamma = 1$ cell [7]. For the regime $10^8 < \text{Ra} < 3 \times 10^{10}$ and 3 < Pr< 1200, the experimental results can be parametrized in the local power law $\lambda_u/L = 0.65 \text{Ra}^{-0.16 \pm 0.02} \text{ Pr}^{0.24 \pm 0.01}$. Our theory gives steeper dependences on both Ra and Pr (see Fig. 5). The correspinding theoretical exponents are around -0.23 and 0.35. For the Ra dependence the discrepancy between the experimental Ra scaling exponent -0.16 and the theoretical Ra scaling exponent -0.23 (for fixed $\text{Pr}\approx 6$) was already identified in [8], when compared to older experimental results [20].

However, note that the kinetic boundary layer thickness measured at the *sidewalls* does agree with the prediction from our theory. For that thickness Qiu and Xia [19] found $\lambda_u/L = 3.6 \text{Ra}^{-0.26 \pm 0.03}$ for Pr=7, which we also drew in Fig. 5. The agreement with the theoretical prediction is very

good. Note that the theoretical overall prefactor was adapted to a different quantity (namely, Re; see Fig. 1) in a different experiment (cf. [1]) and therefore the quality of the agreement not only of the slope but also of the absolute size must be emphasized.

As the theoretical Ra scaling exponent for the BL thickness is closer to the experimental result for the sidewall BL rather than to that of the top or bottom BL, it seems that the main contribution to the kinetic energy dissipation in the boundary layers comes from the sidewalls and not from the top or bottom plates. Given that for an aspect ratio $\Gamma = 1$ (cylindrical) cell (i.e., the diameter equals its height L) the sidewalls contribute twice the top plus bottom area, this may not be too surprising. If we weigh the experimental scaling exponents of side and top/bottom walls according to the relative areas, we obtain $(-0.16-2 \times 0.26)/3 = -0.23$ (for Pr =7), very close to our theoretical value. Since our theory does not distinguish between the top/bottom plates and the sidewalls in Eqs. (1) and (4), there is no freedom for different exponents in the theory and an averaged value results. A refinement of the theory, including a possible physical explanation of the different Ra scaling of the sidewall and the bottom/top kinetic boundary layer widths (as an effect of the geometrical shape of the RB cell) and a first understanding of the aspect ratio dependence (as coming from the observation that the boundary layer probes the streamwise extension of the surface rather than the perpendicular length scale of the cell; cf. [11]) will be published in a forthcoming paper [10].

IV. BREAKDOWN OF THE LAMINAR KINETIC BOUNDARY LAYER

One expects that at large enough Ra the laminar kinetic BLs will become turbulent due to the shear triggered transition. Since the sidewall and top/bottom BL thicknesses seem to scale differently, this may in fact happen at different Ra. Note also that the different Ra scaling of the sidewall and the top/bottom BL thicknesses implies an aspect ratio dependence of the kinetic laminar BL breakdown.

Our theory does not distinguish between sidewalls and top/bottom plates, and the onset of the shear driven turbulence, indicated by the dotted line in the phase diagram Fig. 2, is based on only one λ_u [Eq. (4)]. As onset Reynolds number we took a critical shear Reynolds number of Re^{*}_s = 420, which is the value advocated in Ref. [12] for a semiinfinite plane boundary layer. For Pr=1 this means the onset of shear turbulence at Ra= 3×10^{14} . For the finite and closed RB geometry other values for Re^{*}_s might be more realistic. If we take the critical shear Reynolds number Re^{*}_s only half as large (i.e., Re^{*}_s=210), the onset of shear driven turbulence in the boundary layers takes place already at Ra= 10^{13} (again for Pr=1). Or, for fixed Ra, reducing Re^{*}_s by a factor of 1/2 leads to an upward shift of the dotted line in the Pr direction by a factor of about 10.

The breakdown of the top and bottom laminar kinetic BLS is expected to lead to an enhanced heat transfer. In the of Chavanne *et al.* experiments [14,21] such an enhanced heat transfer is indeed observed around $Ra=10^{11}$. In con-



FIG. 6. The kinetic energy ϵ_u (solid), the kinetic energy in the bulk $\epsilon_{u,bulk}$ (dashed), and the kinetic energy in all kinetic boundary layers $\epsilon_{u,BL}$ (dotted) as a function of Ra for Pr=1. As in the text, all three functions are made dimensionless with *L* and ν , e.g., $\epsilon_u/(\nu^3 L^{-4}) = Pr^{-2}Ra(Nu-1)$. The inset shows the ratio $\epsilon_{u,BL}/\epsilon_u$ as a function of Ra for Pr=1, 10, and 100 (bottom to top).

trast, the experiments of Niemala *et al.* [16] do not display such an enhancement. Our theory cannot resolve this discrepancy between the experiments. On the contrary, the phase diagram Fig. 2 suggests that it is the data of [16] in which a breakdown of the laminar kinetic BL should be visible (if at all), as they reach larger Ra. More work will be necessary to resolve this paradox.

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APPENDIX: KINETIC DISSIPATION IN THE BULK AND IN THE BL

In order to give an idea of the *absolute* kinetic energy dissipation rates per mass and their distribution between bulk and boundary layers, we calculated [from Eq. (5)] Fig. 6, showing ϵ_u , $\epsilon_{u,bulk}$, and $\epsilon_{u,BL}$ as functions of Ra (here for Pr=1). With typical values such as $\nu = 10^{-6} \text{ m}^2 \text{s}^{-1}$ and L = 0.5 m one gets, for Ra=10¹⁰, the small dissipation rate $\epsilon_u = 1.6 \times 10^{-5} \text{ m}^2 \text{s}^{-3}$. For Ra=10¹⁸ one a obtains in contrast the large value $\epsilon_u = 8.0 \times 10^5 \text{ m}^2 \text{s}^{-3}$. This cannot, of course, be realized in a laboratory. Very large Ra in laboratory experiments are reached today by pressure induced increase of the density, which in turn reduces the kinetic and thermal diffusivities ν and κ . This leads to reasonably moderate dissipation rates. In the experiment of [16], for instance, for Ra=1.06×10¹⁰ it is $\nu = 5.04 \times 10^{-6} \text{ m}^2 \text{s}^{-1}$, Pr = 0.689, and thus $\epsilon_u = 7 \times 10^{-3} \text{ m}^2 \text{s}^{-3}$, while for Ra=1.26

×10¹⁶ it is $\nu = 3.245 \times 10^{-8} \text{ m}^2 \text{s}^{-1}$, Pr=7.36, and $\epsilon_u = 1.5 \times 10^{-3} \text{ m}^2 \text{s}^{-3}$, still rather moderate. For these relatively small Pr, the boundary layers' contributions to the total kinetic dissipation rates are rather small, as one could anticipate from Fig. 2.

Interestingly enough, if the Ra increase is obtained by

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enlargening the density and so reducing ν and κ , the dissipation rate becomes *smaller* with increasing Ra. If all other parameters defining Ra, such as α_p, g, Δ , and *L* are kept fixed, one evaluates $\epsilon_u \propto [(\alpha_p g \Delta)^3 L]^{1/2} Pr^{-1/2} Ra^{\beta - 1/2}$. Here β denotes the Nu versus Ra scaling exponent. Using $\beta \approx 1/3$ we find $\epsilon_u \propto [(\alpha_p g \Delta)^3 L]^{1/2} Pr^{-1/2} Ra^{-1/6}$.

et al. (Springer, Berlin, 1961), Vol. 2, p. 575. Since we consider aspect ratio $\Gamma = 1$, we can use the cell height *L* instead of the cell width ℓ . In 1908, in his thesis, Paul Richard Heinrich Blasius extended the laminar boundary layer theory to semiinfinite surfaces, $\ell \rightarrow \infty$; see H. Blasius, Z. Math. Phys. 56, 1 (1908). Then $\lambda_u = 1.72\sqrt{\nu x/U}$; the prefactor was also calculated. If the streamwise distance *x* from the edge of the surface is $\approx \ell$, one recovers Prandtl's formula.

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