Controlling the configuration space topology of mechanical structures

M. Berry[®],¹ David Limberg,² M. E. Lee-Trimble[®],³ Ryan Hayward,⁴ and C. D. Santangelo^{1,*}

¹Department of Physics, Syracuse University, Syracuse, New York 13244, USA

²Department of Polymer Science and Engineering, University of Massachusetts, Amherst, Massachusetts 01003, USA

³Department of Physics, University of Massachusetts, Amherst, Massachusetts 01003, USA

⁴Department of Chemical and Biological Engineering, University of Colorado, Boulder, Colorado 80309, USA

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Linkages are mechanical devices constructed from rigid bars and freely rotating joints studied both for their utility in engineering and as mathematical idealizations in a number of physical systems. Recently, there has been a resurgence of interest in designing linkages in the physics community due to the concurrent developments of mechanical metamaterials, topological mechanics, and the discovery of anomalous rigidity in fiber networks and vertex models. These developments raise a natural question: to what extent can the motion of a linkage or mechanical structure be designed? Here, we describe a method to design the topology of the configurations space of a linkage by first identifying the manifold of critical points, then perturbing around such critical configurations. Unlike other methods, our methods are tractable and provide a simple visual toolkit for mechanism design. We demonstrate our procedure by designing a mechanism to gate the propagation of a soliton in a Kane-Lubensky chain of interconnected rotors.

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I. INTRODUCTION

Linkages have been pervasive in engineering since the beginning of automation. In science, they also serve as prototypical mechanical models for many different physical systems, including animal limbs and joints [1–3], polymer physics [4], protein allostery [5–9], DNA rigidity [10,11] origami [12–14], and jamming [15–17]. At a basic level, a linkage is a graph whose edges have a fixed length but whose vertices are otherwise freely rotating joints. Yet this superficial simplicity belies behavior that can be surprisingly complex and rich. This richness is apparent in the recent rise of mechanical metamaterials [18], materials whose mechanical response arises from their geometry, the discovered topological protection of soft modes in isostatic lattices [19], and the anomalous rigidity of massively undercoordinated spring and fiber networks [20].

One of the first important mathematical results was Kempe's universality theorem, which showed that a linkage can be designed such that a given vertex traces out a portion of any rational algebraic curve [21,22]. The results of following the proof's design procedures can be unwieldy for even simple curves. A planar curve of degree d requires no more than 3d + 2 edges, where d is the degree of the curve [23] (there is a different bound in three dimensions [22]), but these edges must be arranged in very specific geometries and, consequently, the tolerances required in their fabrication are unclear. Moreover, if one wants to change the motion of an existing mechanism, there is no guarantee that the result can be achieved without changing the connectivity of the linkage itself.

This paper arises from the following observation: there are many applications where the precise motion of the vertices of a linkage is less important than the motion's qualitative features. In these cases, not only can we achieve the desired effect, but we can do so with simpler linkages. Moreover, we demonstrate that select small changes in the linkage geometry can drive large, qualitative changes in the behavior of the mechanism, without changing connectivity.

In this paper, we introduce an approach to linkage design that takes advantage of the critical points of a configuration space [24,25]. At a critical point, a mechanism has an anomalously large class of potential linear motions available to it, but higher-order corrections from the mechanical constraints restrict the motion to a subset of these motions [26–29]. Many linkages of this type have branched configuration spaces, meaning that many different qualitative motions are accessible. For example, generic origami and kiragami mechanisms have highly branched configuration spaces, leading to pluripotency [30,31]. Similar branched configuration spaces have been used to design mechanical logic devices [32] and kinematotropic mechanisms that can change how many degrees of freedom they can access [33,34].

Critical points are delicate; even small perturbations of the mechanism geometry will destroy them. However, by controlling those perturbations, we demonstrate that the configuration space topology of a mechanism can be controlled. As a result, we obtain linkages with switchable motion using only a small number of tunable components. We demonstrate this extreme switchability in the celebrated Kane-Lubensky (KL) chain [19], a series of rotors joined by springs, which supports the propagation of a soliton called a "spinner" in which each rotor, in turn, rotates a full 360° [35,36]. We will apply our design methodology to the KL chain. By replacing one of the unit cells with a designed mechanism, we show

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^{*}cdsantan@syr.edu

that the propagation of the spinner soliton can be controllably gated. Because our method relies on topological programming, the mechanisms are robust to flexibility and fabrication imperfections.

In Sec. II, we review relevant parts of rigidity theory and mechanisms. In Sec. III, we describe mathematical tools that provide a geometrical interpretation to critical points. This interpretation will provide the basis of our design methodology, which we will illustrate with an example containing five bars. Finally, in Sec. IV, we will use our formalism to explicitly design a mechanism to gate the KL chain. Importantly, the operation of the resulting gate is robust with respect to small perturbations. Finally, we conclude with a brief discussion highlighting new directions enabled by this work.

II. CRITICAL POINTS IN MECHANISMS

A. Mathematical rigidity

In this section, we review the basic mathematical description of mechanisms. Though we focus on linkages, which are constructed entirely from free-rotating joints and inextensible bars, the formalism can be generalized to mechanisms with other holonomic constraints. We define a linkage as a collection of *V* vertices in *d* dimensions joined by *E* rigid bars. The configuration of a linkage can then always be represented by a point **u** in the space of vertex positions, which we will denote as \mathcal{M} , and has dimension M = Vd. We assume there are *E* bars in the linkage and denote the length of the α th bar, $\ell_{\alpha}(\mathbf{u})$. The configuration space of the linkage can then be represented by the family of equations,

$$\ell_{\alpha}^2(\mathbf{u}) = L_{\alpha}^2,\tag{1}$$

where L_{α} is the target length of the α th bar. Note that Eq. (1) is written using the square of $\ell_{\alpha}(\mathbf{u})$ so that it can be an analytic function everywhere. By replacing $\ell_{\alpha}(\mathbf{u})$ with a more general class of functions in Eq. (1), we can also describe mechanisms with more complex components beyond rigid bars.

Rather than analyzing the configuration space for specific values of L_{α} , we will instead analyze the entire family of configuration spaces that can occur with a fixed network topology by changing the L_{α} . Between Kempe's universality theorem and the potential arbitrariness of $\ell_{\alpha}(\mathbf{u})$, however, it is indeed difficult to say a great deal more about the configuration space with any kind of generality. Therefore, we assume that $\ell^2(\mathbf{u})$ is an analytic function of \mathbf{u} and that $E \leq Vd$. With these assumptions, the Jacobian matrix, whose components are

$$J_{\alpha i}(\mathbf{u}) = \frac{\partial \ell_{\alpha}^{2}(\mathbf{u})}{\partial u_{i}}, \qquad (2)$$

provides critical information about the mechanism. Naively, one would expect the configuration space of the mechanism to be D = M - E (for M > E). Indeed, the inverse function theorem implies that the configuration space is a smooth *D*-dimensional manifold in any open set of \mathcal{M} in which the Jacobian matrix is full rank. At such a configuration **u**, the tangent space coincides with the right null space of $J_{\alpha i}(\mathbf{u})$,

$$\sum_{i} J_{\alpha i}(\mathbf{u}) \delta u_{i} = 0.$$
(3)

The solutions δu_i of Eq. (3) are called zero modes.

Any point \mathbf{u}_C at which the Jacobian fails to be full rank, on the other hand, we call a critical point, and the corresponding edge lengths $\ell_{\alpha}^2(\mathbf{u}_C)$ we call a critical value. Critical points are characterized by self stresses σ_{α} , which are elements of the left null space of $J_{\alpha i}(\mathbf{u}_C)$,

$$\sum_{\alpha} \sigma_{\alpha} J_{\alpha i}(\mathbf{u}_C) = 0. \tag{4}$$

Because of their relation to critical points, we will see that self-stresses play an important role in the topology of the configuration space.

Sard's theorem ensures that critical values (but not necessarily critical points) are a set of measure zero. In that sense, most choices of edge lengths lead to a configuration space that is a smooth *D*-dimensional manifold. Consequently, any change in the configuration space's topology that occurs as the L_{α} change must happen at a critical point. Thus, these critical points also govern the overall topology of the configuration space of a mechanism.

In the next section, we proceed to analyze the geometry of the configuration space at and near such critical points.

B. Shape of the configuration space at critical points

To understand the shape of the configuration space, we expand $\ell_{\alpha}^{2}(\mathbf{u} + \delta \mathbf{u})$ for small deformations, $\delta \mathbf{u}$ having components δu_{i} , around the critical point, obtaining

$$0 = \sum_{i} J_{\alpha i} \delta u_{i} + \frac{1}{2} \sum_{ij} \frac{\partial^{2} \ell_{\alpha}^{2}(\mathbf{u}_{C})}{\partial u_{i} \partial u_{j}} \delta u_{i} \delta u_{j} + \mathcal{O}(\delta u^{3}).$$
(5)

It is common at this stage to write a formal series expansion, $\delta \mathbf{u} = \delta \mathbf{u}^{(1)} + \delta \mathbf{u}^{(2)} + \cdots$, and substitute it into Eq. (5). One finds $\delta u^{(1)}$ is a zero mode of the Jacobian satisfying [28]

$$\frac{1}{2}\sum_{\alpha}\sum_{ij}\sigma_{\alpha}^{(n)}\frac{\partial^{2}\ell_{\alpha}^{2}(\mathbf{u}_{C})}{\partial u_{i}\partial u_{j}}\delta u_{i}^{(1)}\delta u_{j}^{(1)}=0,$$
(6)

where $\{\sigma_{\alpha}^{(1)}, \sigma_{\alpha}^{(2)}, ...\}$ is a basis for the space of self-stresses at \mathbf{u}_{C} .

To proceed, we make further assumptions. The most important of these is that Eq. (6) completely characterizes the local geometry of the critical point. It is well known that if no solution to Eq. (6) exists, then the linkage is rigid, but the converse does not necessarily hold. There are mechanisms whose rigidity is only visible at higher order, as well as mechanisms that are rigid at order larger than two but, nevertheless, are mobile [27]. Experience suggests that these examples are rarer than the better behaved examples we consider here, but we are unaware of any results quantifying their rarity or even a simple means to determine when Eq. (6) is sufficient to describe the geometry of the critical point accurately. For the scope of this paper, it will prove sufficient to assume we can safely truncate our expansion of $\delta \mathbf{u}$ at second order and check, *post hoc*, that the results produced by our design procedure satisfy our assumptions.

We will make three other assumptions as well:

(i) All critical points \mathbf{u}_C lying on a configuration space of constant L_{α} are isolated. There are linkages for which this fails and for which the entire configuration space lies along a sequence of critical points (see, for example, [37]). Note,

however, that there are also mechanisms with large D which do satisfy this assumption [31,38,39]. In this paper, we will ultimately focus on example mechanisms with only a single degree of freedom, so this will not prove a particularly strong assumption, but in this section we allow D to be general and only specialize to D = 1 subsequently.

(ii) All critical points have exactly one self-stress. This assumption is certainly not always true. It fails, for example, in flat origami mechanisms [31]. Generally, however, we will see that, qualitatively, critical points with several self-stresses appear to require more fine-tuning. This assumption implies that there will be D + 1 zero modes at each critical point by the rank-nullity theorem applied to the Jacobian matrix at \mathbf{u}_C .

(iii) The matrix

$$\sum_{\alpha} \sigma_{\alpha} \frac{\partial^2 \ell_{\alpha}^2(\mathbf{u}_C)}{\partial u_i \partial u_j}$$

has nonzero eigenvalues when restricted to the zero modes at \mathbf{u}_C . This assumption allows us to simplify the characterization of the critical points. Notice that without assumption 2, this characterization would be more difficult because Eq. (6) would yield a system of quadratic equations rather than a single equation.

While all of these assumptions will play a role in our analysis, one could relax some of them at the expense of complicating the design procedure. Additionally, they may not be mutually exclusive from our assumption zero that Eq. (6) completely characterizes the local geometry of the critical point. Our examples will satisfy them, however, and we leave it for future work to understand which are truly required and which are conveniences.

Suppose we choose a basis for the zero modes at \mathbf{u}_C , $\{\boldsymbol{\zeta}_1, \ldots, \boldsymbol{\zeta}_{D+1}\}$, writing $\delta u_i^{(1)} = \sum_n c_n \zeta_{n,i}$. Then, Eq. (6) becomes

$$\sum_{nm} \mathcal{Q}_{nm} c_n c_m = 0, \tag{7}$$

where Q_{nm} is a symmetric matrix given by

$$Q_{nm} = \sum_{ij} \sum_{\alpha} \zeta_{n,i} \zeta_{m,j} \sigma_{\alpha}^{(1)} \partial^2 \ell_{\alpha}^2(\mathbf{u}_C) / \partial u_i \partial u_j.$$
(8)

Under our assumptions, there are just two possibilities. If Q_{nm} is either positive- or negative-definite, the linkage is rigid: there is no solution to Eq. (7) other than $c_n = 0$. If Q_{nm} has a combination of positive and negative eigenvalues, however, the geometry of the configuration space at \mathbf{u}_C is that of a cone. This is precisely what happens in single-vertex flat origami [31,40] (Fig. 1). We call such a point \mathbf{u}_C a branch point, though this space of possible zero modes is sometimes called a kinematic tangent cone [41].

C. Shape of the configuration space near critical points

We next ask what happens to the configuration space of a mechanism when the lengths are deformed from their critical values, $L_{\alpha} = L_{\alpha}^{(c)} + \delta L_{\alpha}$. A lengthy calculation shows (see Appendix A)

$$\sum_{nm} \mathcal{Q}_{nm}(c_n - \delta c_n)(c_m - \delta c_m) = \Delta, \qquad (9)$$

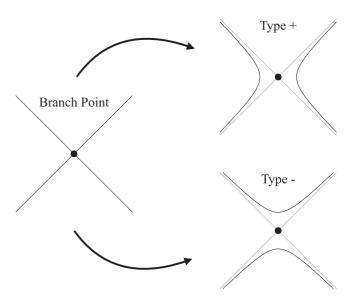


FIG. 1. Schematic of how a configuration space with a branch point split into one of two types of smooth, disconnected configuration spaces. The choice of sign is arbitrary.

where the deformation is along the zero modes at \mathbf{u}_C , $\sum_n c_n \zeta_{n,i}$ as before, and δc_n and Δ are quantities whose value depends linearly on the length changes, δL_{α} , to lowest order.

We first consider what happens when $\Delta = 0$. In that case, when $\delta c_n = 0$, Eq. (9) recovers the results from the previous section: there is either a rigid point or a branch point at $c_n = 0$ corresponding to the critical point \mathbf{u}_C . When $\delta c_n \neq 0$, however, the critical point itself moves by $\approx \sum_n \delta c_n \zeta_{n,i}$.

When $\Delta \neq 0$ and Q has only positive eigenvalues (the critical point is second-order rigid), we have two possibilities: (1) $\Delta > 0$ implies the solution to Eq. (9) is an ellipsoid in D + 1 dimensions (it is almost rigid [42]), and (2) $\Delta < 0$ implies there is no solution to Eq. (9). The opposite occurs if Q_{nm} has only negative eigenvalues.

Finally, we consider the case of a branch point, for which Q_{nm} has eigenvalues of opposite sign. To develop intuition, it is useful to consider the special case of a branch point when D = 1. Then, Q_{nm} is a 2×2 matrix with two eigenvalues of opposite sign. The solutions to Eq. (9) take the form of two hyperbolas in the plane spanned by the zero modes at \mathbf{u}_C whose precise configuration depends on the sign of Δ (see Fig. 1 for characteristic examples for both signs of Δ). For D > 1, branch points also break up into smooth surfaces but do so, presumably, in a more complex way that depends on the signature of Q_{nm} (see Ref. [40] for an example in origami).

As an illustrative example, we turn to the well-studied four-bar linkage shown in Fig. 2(a). The four-bar linkage is constructed from two rotors of length L_1 and L_3 pinned at one end and joined at the other by a bar of length L_2 . The system configuration can be parameterized as a point in four dimensions with coordinates (x_1, y_1, x_2, y_2) , and the configuration space is one dimensional. When $L_1 = L_2 = L_3 = a$, there are three branch points each having a single self-stress and two zero modes. The configuration space is shown in Fig. 2(b) in terms of the two rotor angles θ_1 and θ_2 . By slightly increasing the length of $L_2 > a$, the branch points all split into a pair

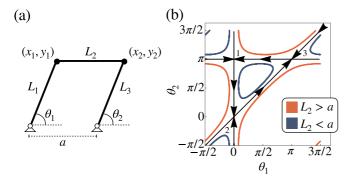


FIG. 2. (a) Schematic of the planar, four-bar linkage with variables defined. (b) Projection of the configuration space of the two-rotor mechanism with $L_1 = L_2 = L_3 = a$ projected into the (θ_1, θ_2) plane (black). This choice of lengths has three branchlike critical points. Deforming the length of L_2 results in a smooth configuration space with either one (red) or two (blue) components. The arrows indicate the direction of the tangent form $t_i(\mathbf{u})$ from Eq. (10).

of hyperbolas oriented opposite each other in the quadrants spanned by the configuration space when $L_2 = a$. On the other hand, $L_2 < a$ results in the branch point splitting into a pair of hyperbolas in the other pair of quadrants. As a result of switching the orientation of the hyperbolas, the configuration space goes from having a single component for $L_2 > a$ to two disconnected components when $L_2 < a$.

III. CONTROLLING CONFIGURATION SPACE TOPOLOGY

We noted earlier that the topology of the configuration space cannot change without passing through an intermediate critical point. If it could, this would contradict the notion that the configuration space is smooth when the Jacobian $J_{\alpha i}$ is full rank. This fact and the analysis of Sec. II suggests a method for controlling the topology of the configuration space: (1) find a set of lengths L_{α} for which the configuration space has many branch points, and (2) perturb the lengths, $L_{\alpha} \rightarrow L_{\alpha} + \delta L_{\alpha}$, such that the branch points split into smooth hyperbolas in the desired configuration. For the four-bar linkage in Fig. 2(b), for example, if we could control how each of the three branch points splits independently, we would have complete control over how the configuration space winds around the torus defined by the angles (θ_1, θ_2) as well as the number of components in the configuration space.

A. The geometry of the critical configuration set

Since we are interested in understanding how to choose edge lengths L_{α} to control the topology of the configuration space of a linkage, we will consider all possible mechanisms that have the same connectivity but arbitrary values of L_{α} . To do so, we define an antisymmetric tensor,

$$t_{i_1\cdots i_D}(\mathbf{u}) = \sum_{j_1\cdots j_E} \epsilon_{i_1\cdots i_D j_1\cdots j_E} \frac{\partial \ell_1^2(\mathbf{u})}{\partial u_{j_1}} \cdots \frac{\partial \ell_E^2(\mathbf{u})}{\partial u_{j_E}}, \qquad (10)$$

where $\epsilon_{i_1 \cdots i_D j_1 \cdots j_E}$ is the antisymmetric Levi-Civita tensor. Importantly, $t_{i_1 \cdots i_D}(\mathbf{u}) = 0$ if and only if \mathbf{u} is a critical point. This is because the components of $t^{i_1 \cdots i_D}$ are the $E \times E$ minors

of the Jacobian matrix. When these all vanish, the Jacobian matrix has lower rank (see Appendix B for a more detailed discussion). Thus, Eq. (10) identifies all possible critical points in mechanisms sharing the same connectivity. Versions of Eq. (10) have been studied to identify singularities in robot manipulators [43–47].

The tangent form allows us to define the *critical configuration set* as the locus of points for which

$$t_{i_1\cdots i_D}(\mathbf{u}) = 0. \tag{11}$$

In many practical cases, and in all of the cases we consider in this paper, it is possible to solve Eq. (11) analytically. Note, however, that the solutions to Eq. (11) only provide the configurations where the Jacobian of the mechanism is not full rank. Therefore, some of the solutions may not satisfy all of our assumptions from Sec. II B. We conjecture that our assumptions are valid on all but a set of measure zero of the critical configuration set, but are not aware of or able to produce a proof of this.

To help understand the geometry of the critical configuration set, we return to our previous example, the planar, four-bar linkage from Fig. 2(a). In this example, D = 1 but M = 4 since the mechanism configurations are specified by points (x_1, y_1, x_2, y_2) . If the two pinned vertices are located at (0,0) and (a, 0) and we restrict $L_{\alpha} > 0$ (so no bars have zero length), this critical set is described by the two-dimensional manifold of configurations in which all vertices are colinear, $y_1 = y_2 = 0$.

For one degree of freedom mechanisms (D = 1), Eq. (10) provides another way to understand how the configuration space topology changes with changing lengths near a critical point \mathbf{u}_C . In that case, $t_i(\mathbf{u})$ is a vector field everywhere tangent to the zero modes of the mechanism, which follows from the simple fact that it is always orthogonal to the constraints (Appendix B). Thus, $t_i(\mathbf{u})$ can be thought of as a local vector field whose integral curves trace out curves of constant L_{α} . That is, when $t_i(\mathbf{u}) \neq 0$, curves of constant L_{α} can be parameterized by the solutions

$$\frac{du_i(s)}{ds} = t_i[\mathbf{u}(s)]. \tag{12}$$

We show this in Fig. 2(b) using arrows pointing along t_i projected onto the rotor angles. Because $t_i(\mathbf{u})$ is divergence free (Appendix B), each branch point has two arrows pointing in and two arrows pointing out. Note that $t_i(\mathbf{u})$ provides a way to think about the mechanism configuration space as a dynamical system. This dynamical system should not be confused with the motions of the physical mechanism, however, which can move either parallel or antiparallel to $t_i(\mathbf{u})$ equally well. This is also distinct from the dynamical system approach obtained for a periodic (or nearly periodic) mechanism as an iterated map [48,49].

Equation (12) also provides an intuitive way to understand the hyperbolas formed by the configuration space near branch points that arise from Eq. (9). We project $t_i(\mathbf{u})$ near \mathbf{u}_C onto the plane spanned by the two zero modes, ζ_1 and ζ_2 . Since $t_i(\mathbf{u})$ is tangent to the configuration spaces, we expect that the trajectories approach this plane as they approach \mathbf{u}_C . After projection, we obtain a two-dimensional (2D) vector field whose components are

$$T_n(c_1, c_2) = \sum_i \zeta_{n,i} t_i (\mathbf{u}_C + c_1 \zeta_1 + c_2 \zeta_2).$$
(13)

The integral curves of T_n then trace the projection of the configuration space onto the plane spanned by the zero modes near the branch point.

In this projection, the constant L_{α} trajectories are quite limited in how they can appear. We know that $T_n(0, 0) =$ 0, but because we assume branch points are isolated, the projected tangent vector $T_n(c_1, c_2) \neq 0$ elsewhere. Now suppose that the critical point is a branch point. The projection of the configuration space on the plane of zero modes will have the form of a hyperbolic fixed point, with a stable and unstable manifold associated with the configuration space branches that solve Eq. (7) (Fig. 1). Thus, we would generically expect the trajectories near the branch point to appear hyperbolic when projected onto the plane of zero modes. Though we do not work with second-order rigid points here, these considerations also limit what the trajectories do near such rigid points [42].

B. The geometry of the critical value set

For any point \mathbf{u}_C in the critical configuration set, $\ell_{\alpha}^2(\mathbf{u}_C)$ gives its corresponding critical value: the set of squared bar lengths that would be required for the system to be in configuration \mathbf{u}_C . We will call the image of the critical configuration set in the space of squared lengths the *critical value set*. We again illustrate with the four-bar linkage: the set of critical values is a self-intersecting surface $(L_1^2, L_2^2, L_3^2) = [x_1^2, (x_2 - x_1)^2, (a - x_2)^2]$. In Fig. 3, we show the critical value set in terms of (L_1, L_2, L_3) rather than the squared lengths to make the surface slightly more compact and easier to understand visually. Note that we have included additional leaves on either the $L_1 = 0$, $L_2 = 0$, or $L_3 = 0$ plane which happen to contain only rigid critical points; this is a natural consequence of the fact that any mechanism with two pinned vertices and one edge having zero length must always be rigid.

Were we to choose the L_{α} to lie anywhere along the portion of the critical value set in Fig. 3, the resulting linkage would have one or more critical points. It is also apparent that Fig. 3 self-intersects. At such a self-intersection, there will be multiple critical configurations \mathbf{u}_{C} , corresponding to the same choices of edge lengths L_{α} . Thus, if we choose the L_{α} along a line of self-intersection, there are two branch points [Fig. 3(d)]. If we choose L_{α} along a smooth portion of the critical value set, there is only one critical point [Fig. 3(e)]. Interestingly, Fig. 3 shows that at $(L_1, L_2, L_3) = (a, a, a)$, three individual sheets self-intersect. Therefore, we expect that that choice of L_{α} is the unique place where three branch points coincide [Fig. 3(c)]. An animation demonstrating the branch splitting for the four-bar linkage is included in the Supplemental Material [50].

The critical value set contains more information than just the location of critical values. If the critical value set is locally a smooth manifold, the self-stresses at such a critical point are always normal to the critical values (see Appendix B). Though the converse is not generally true—normals need not also be self-stresses—if the critical value set is a manifold of



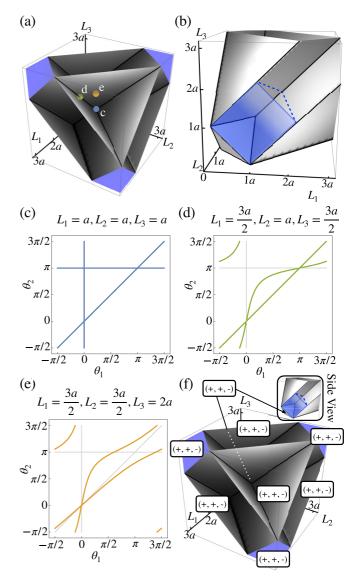


FIG. 3. (a) The critical value set for the four-bar linkage, plotted in terms of (L_1, L_2, L_3) in units of *a*. The points (c)–(e) show the locations of the configuration spaces in (c)–(e). (b) A different view of the surface with a cutout on the $L_1 = 0$ plane showing the shape of one of the enclosed volumes. There is one critical point in the configuration space along any smooth portion of the set (e). Selfintersecting lines indicate choices with two critical points (d) and the triply self-intersection point at $(L_1, L_2, L_3) = (a, a, a)$ (c) is the unique choice with three critical points. (c)–(e) The configuration spaces corresponding to the points in (a), showing how to split different branch points. (f) The volumes labeled with their branch point types, corresponding to the standard engineering classifications.

codimension one, it must necessarily coincide with the single self-stress at that point and there can be no other self-stresses. We also see that splitting a branch point amounts to choosing δL_{α} transverse to the critical value set. On one side of the surface in Fig. 3, a branch point splits into one pair of smooth branches; on the other side, it splits into the opposite pair. This endows the calculation of how branch points split under small perturbations of the lengths with a concise geometrical meaning.

With this understanding of the critical value set, we can classify the distinct configuration spaces of the four-bar linkage in terms of the $2^3 = 8$ individual volumes enclosed by the surface in Fig. 3. For completeness, we note that these volumes correspond to standard results for the four-bar linkage found in the engineering literature, which can be classified by the sign of three functions [51],

$$\tau_1 = a - L_1 + L_2 - L_3, \quad \tau_2 = a - L_1 - L_2 + L_3,$$

$$\tau_3 = L_2 + L_3 - a - L_1, \quad (14)$$

derived from limits on the angles θ_1 and θ_2 . When one of the τ_i is equal to zero, the configuration space contains the corresponding critical point from Fig. 2(b). Thus, the critical value set in Fig. 3 agrees with the surfaces computed in Refs. [52,53]. The regions are labeled in Fig. 3(f).

C. Three-rotor system

Finally, in this section, we will use these considerations to describe a design procedure for configuration space topology. To be concrete, it is helpful to consider a specific example, the three-rotor linkage in Fig. 4(a). The three-rotor linkage has three pinned joints attached to three bars of length r_1 , r_2 , and r_3 (the rotors) and whose opposite ends are joined by bars of length L_1 and L_2 . Therefore, **u** is a six-component vector and the five bars provide constraints, $\ell_{\alpha}^2(\mathbf{u}) = L_{\alpha}^2$, that limit the configuration space to a single degree of freedom generically.

Since the mechanism has five bars, it is difficult to visualize the critical set and critical value set. Nevertheless, we can still gain insight by restricting ourselves to the cross section of \mathcal{M} for which $r_1 = r_2 = r_3 = a$. We plot the cross section of the critical value set with the (L_1, L_2) plane in Fig. 4(b). While this is a cross section, the open regions in Fig. 4(b) still correspond to structures with different configuration space topologies, with the transitions from one distinct region to another through the critical value set occurring through a branch point. However, it is still a cross section of a higherdimensional space and care must be taken when interpreting the intersections of the critical value set. Choosing $L_1 = L_2 =$ a leads to a configuration space with 12 interconnected branch points, though it appears that only two lines meet at $L_1 =$ $L_2 = a$ in Fig. 4(b). The proliferation of branch points in this example can be understood from the fact that this linkage contains two pairs of four-bar linkages. Choosing all bars to have length a, therefore, maximizes the branch points of each individual submechanism.

To identify these branch points, we solve $t_i(\mathbf{u}) = 0$ subject to the length constraints $\ell_{\alpha}^2(\mathbf{u}) = L_{\alpha}^2$ using *Mathematica* (Wolfram). At each critical point, we then solve Eq. (7) to obtain the tangents to the configuration space. The trajectories in Fig. 4(c) are obtained by first stepping along one of the obtained tangent vectors, then stepping along the configuration space in the direction indicated by $t_i(\mathbf{u})$ with a step size proportional to its magnitude. The step size is adjusted to maintain the edge lengths to less than one-percent strain. Finally, the integration for each segment is terminated when the magnitude of $t_i(\mathbf{u})$ falls below a critical threshold, indicating that the integration has reached a point close to the next critical point. Once terminated, we minimize $\sum_i t_i^2$ with respect to the configuration to verify that the integration

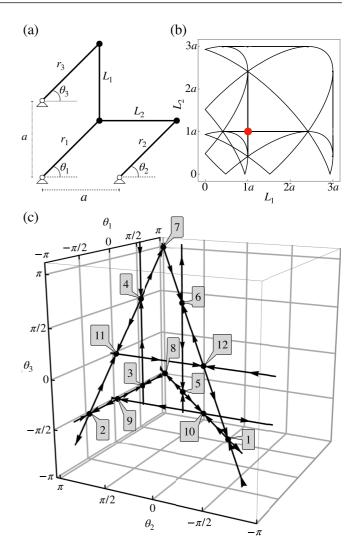


FIG. 4. (a) Schematic of the planar, three-rotor linkage with variables defined. (b) A cross section of the critical value set for $r_1 = r_2 = r_3 = a$. (c) The 3D configuration space of the three-rotor linkage with $r_1 = r_2 = r_3 = L_{13} = L_{23} = a$, corresponding to the red (gray) point in (b), contains 12 individual critical points. Arrows indicate the orientation of each configuration space segment.

has found the next branch point. The directions of integration inherited from $t_i(\mathbf{u})$ are indicated by the arrows in Fig. 4(c).

Note, however, that the branch points shown in Fig. 4(c) are not all independent. Projecting the configuration space onto the θ_1 - θ_2 plane must give the configuration space of the equivalent four-bar linkage found by ignoring the third rotor. In contrast, removing the first rotor is equivalent to the projection onto the θ_3 - θ_2 plane. Consequently, any branch points that overlap in one of these two projections must, after a deformation, still be identical in projection and such overlapping branches appear or disappear together. From Fig. 4, this implies that branch points are paired {(1, 2), (3, 5), (4, 6), (7, 8), (9, 11), (10, 12)}.

We finally consider how to "program" the configuration space by adjusting the lengths of r_1 , L_1 , and L_2 away from their critical values. For each critical point, we plot the domain



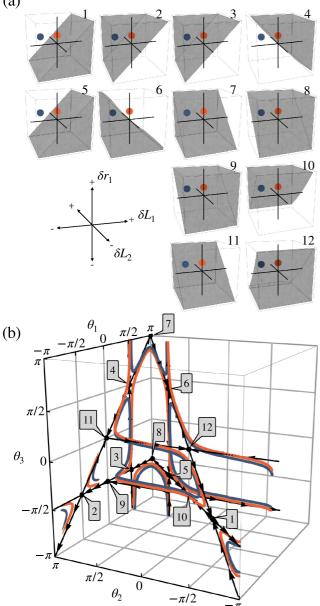


FIG. 5. A schematic of programming the three-rotor system. (a) A map showing how changing the length of r_1 , L_1 , and L_2 leads to different ways to split the branch points from Fig. 4. The shaded and unshaded regions correspond to the two ways for the branch points to split. The red (light gray) dot, corresponding to a change in length $r_1 = 1.05a$, leads to the red (light gray) curve in (b). In order to change the topology of the configuration space by changing how branch points 4 and 6 split, an additional change to $L_1 = 0.9a$ can be effected, shown by the blue (dark gray) dot. The new configuration space is shown in (b) as a blue (dark gray) curve.

over which $\Delta > 0$ in Fig. 5 as a table for each branch point. We next choose lengths according to the red (light gray) dot in Fig. 5(a), which increases r_1 at constants L_1 and L_2 . The resulting configuration space is shown as the red (light gray) curve in Fig. 5(b). Note that the red (light gray) point was chosen so that the configuration space is smooth but passes near the branch points. If the red (light gray) curve has the

topology we want already, we can stop now. If we instead wanted to switch the sign of the branch point pair (4,6) to obtain a particular configuration space topology, from Fig. 5(a)we see that the three lengths (r_1, L_1, L_2) distinguish this pair of branch points from the rest. Inspection of Fig. 5(a) suggests that an additional change in L_1 would switch the way only those two branch points split. The result of this perturbation is the blue (dark gray) curve in Fig. 5(b). Note that Fig. 5(b) shows that since each branch point has one self-stress, the hyperbolas approach the plane spanned by the two zero modes as expected.

If we limit ourselves to perturbing only the bar lengths (r_1, L_1, L_2) , Fig. 5(a) shows even more redundancy in how the branch points split than expected from our previous analysis that branch points split in pairs. That is, just three control lengths are not sufficient to obtain full control over the way the configuration space splits at the branch points. While it would be difficult to plot Fig. 5 using all five bar lengths, there seems to be no mathematical obstacle to generalizing the analysis to distinguish all six pairs of branch points independently.

IV. THE GATED KANE-LUBENSKY CHAIN

We finally apply our design methodology to design a mechanism that gates the propagation of a soliton in the Kane-Lubensky (KL) chain [19]. The KL chain is a topologically polarized lattice of rotors that has a zero mode on either the left edge or the right edge, depending on the choice of bar lengths. It was later discovered that the KL chain actually supported two distinct families of propagating solitons, the "flipper" and the "spinner" [35], that allowed a continuous pathway between the left and right edge modes. The spinner soliton, however, is topologically protected by the shape of the configuration space [35,36]. In this section, we modify a single unit cell of a spinner-supporting KL chain with rotor length r = 3a/2 and $\ell = 3a/2$ by adding an additional two bars and one pinned vertex [Fig. 6(a)].

In the spinner phase of the KL chain, a full cycle consists of the soliton traveling back and forth across the chain once, and the KL chain returning to its initial configuration. After one full cycle, each rotor in the KL chain has rotated by 2π , with each rotor rotating by π each time the soliton passes. Here, we will show that these additional components can act as a gate by opening a gap in the full 2π rotation of the KL chain rotors, thereby obstructing the passage of the soliton.

In addition to the length of the two additional bars, L_1 and L_2 , we also allow the location of the pinned vertex to be set an arbitrary distance D from the KL chain. In order to allow for different positions of the third pinned vertex, we augment **u** to include the y coordinate of the third vertex, but also augment the constraint functions to pin that vertex's y position. Thus, we use a constraint map,

$$f_{\alpha}(\mathbf{u}) = \begin{pmatrix} \ell_1^2(\mathbf{u}) \\ \vdots \\ \ell_E^2(\mathbf{u}) \\ D^2(\mathbf{u}) \end{pmatrix}, \qquad (15)$$

where $D(\mathbf{u})$ is the function that determines the distance between pinned vertex 3 and 2. Thus, the generalized

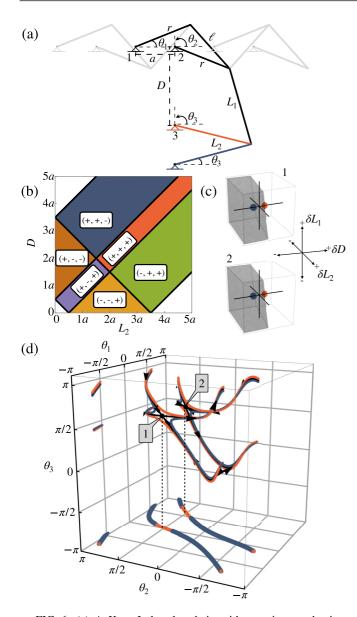


FIG. 6. (a) A Kane-Lubensky chain with a gating mechanism attached. The gating state is controlled by the length *D*; the red (light gray) bar and the blue (dark gray) bar correspond to an ungated and gated state, respectively. (b) A cross section of the critical set with $r = \ell = 3a/2$ and $L_1 = 2a$. There is a critical point at $L_2 = 2a$ and D = 5a/2. (c) At the critical point, changing the position of the third rotor or the lengths of two beams can switch the type of branch split (gray and transparent regions). The red (light gray) point corresponds to branch splitting such that the mechanism is ungated, while the blue (dark gray) point demonstrates that changing *D* can switch to a gated configuration. (d) The configuration space at and near the critical point as a function of the three rotor angles, and the projection of that configuration, and thus the chain is gated.

constraint $f_{\alpha}(\mathbf{u})$ is a smooth function whose solution allows us to pin vertex 3 by setting the length D in Fig. 6(a) to an arbitrary value.

Using this generalized formulation, we can compute a cross section of the critical value set with $r = \ell = 3a/2$ and

 $L_1 = 2a$ [Fig. 6(b)]. Figure 6(b) shows that there are six distinct regions separated by critical points. The labels on each region correspond to the sign of τ_1 , τ_2 , and τ_3 from Eq. (14) with respect to the four-bar linkage between vertices two and three. For concreteness, we choose $L_2 = 2a$ and D = 5a/2, on the boundary between the blue (+,+,-) and red (+,+,+)regions, as the initial lengths for our gate, resulting in a configuration space with two critical points [Fig. 6(d)]. When D < 5a/2, the system is in the red (light gray) regime, and when D > 5a/2, it is in the blue (dark gray) regime. This choice determines whether or not the KL chain rotors wind around fully. Note that the projection of the configuration space in the θ_1 - θ_2 plane [Fig. 5(d)] never changes shape, but that the change in how the branch points split into hyperbolas determines whether or not the full range of angles is accessible to the system. To verify that the red (light gray) and blue (dark gray) regimes correspond to ungated and gated behavior of the KL chain device, we use the *mechanisms* package [54] in Mathematica (Wolfram) to calculate the infinitesimal motions of the linkage and animate those motions. As shown in Fig. 7, changing D controls whether or not the soliton can complete a full cycle along the KL chain.

To test our design, we constructed the gated KL chain in Fig. 6(a) numerically and constructed a single unit cell and gate from LEGO pieces (The Lego Group). The design of the LEGO gate was chosen to be compatible with the LEGO realization of a KL chain shown in [35]. When testing different examples, we pushed on the various bars and rotors in the device to move it through all possible configurations. We tracked how rotors 1 and 2 moved to determine if the gate was preventing a soliton from propagating. Figure 7 shows a comparison between the simulated and LEGO chains with both D larger and smaller than 5a/2. Movies of both chains in the gated and ungated states are provided in the Supplemental Material [50]. In the case of an ungated chain, the soliton propagates from one end of the chain to the other (and back); for a gated chain, the soliton propagates up to the location of the gate but is reflected.

Interestingly, the size of the gap in the projection of the blue (dark gray), gated configuration space in Fig. 6(d) is important for determining how the soliton is reflected from the gate. For very small gaps, which occurs when D is close to its critical value, the soliton can, temporarily, pass the gate but is, ultimately, prevented from completing an entire cycle. For larger gaps, when D is farther from its critical value, the soliton appears to reflect from the gate. From Fig. 6(c), the same effect can be achieved by changing the size of L_2 instead of D, since the plane dividing the gray and transparent regions, which corresponds to the two ways to split the branch, is slightly angled in that direction. Movies of both simulated and LEGO chains that switch between the gated and ungated states by changing L_2 are also provided in the Supplemental Material [50].

Our analysis shows that the presence of a gap in the (θ_1, θ_2) plane blocks soliton propagation. In the example of Fig. 6(a), changing the length *D* moves the device from the gated [blue, (+,+,-)] region to the ungated [red, (+,+,+)] region of Fig. 6(b). However, this is not the only pair of regions that produces a functioning gate. Indeed, the regions indicated in Fig. 6(b) as (+, +, +) (red) and (-, -, +) (yellow) are

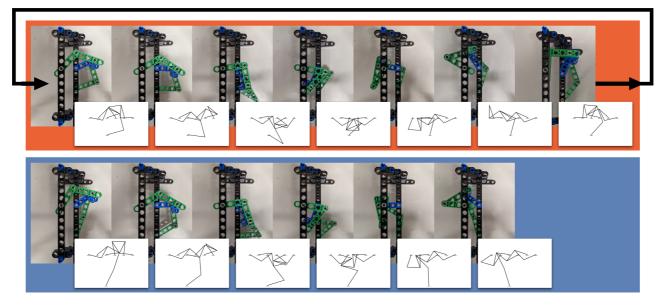


FIG. 7. Top row (red): Ungated device made from LEGOs with the corresponding simulation. This device can continue rotating and return back to its initial position, as indicated by the arrow. Bottom row (blue): Gated device made from LEGOs with the corresponding simulation. This device gets stuck in the configuration shown in the last frame and is forced to reverse direction in order to continue moving.

ungated with respect to propagation of the soliton, whereas the remaining regions are gated. Numerical experiments further show that if we had chosen L_1 to change length as well, we would have found even more regions of both gated and ungated behavior as we extended Fig. 6(b). It becomes clear that there is a great deal of flexibility when choosing D, L_1 , and L_2 to produce the desired dynamics of the final KL chain and gate system.

V. CONCLUSIONS

In this paper, we have described a procedure to design the topology of the configuration space of mechanical linkage. The idea rests on the ability to identify critical points and, especially, branch points—singular configurations of a linkage in which several pathways meet. By analyzing the shape of the configuration space near these branch points, we are able to design perturbations to the lengths and positions of a fixed set of vertices that change the shape of the topology of the configuration space in well-defined ways. As a demonstration, we used our techniques to design a gate for the propagation of the spinner soliton in a Kane-Lubensky chain. While we applied our approach to linkages with fixed edge length, there is no reason they would not also apply more generally to other systems with holonomic constraints.

Because the design procedure works by controlling configuration space topology, the resulting mechanisms should be quite robust to fabrication errors and the tolerance of the joints, so long as one chooses lengths L_{α} sufficiently far from the critical value set.

It would be interesting to extend this work in a few further directions. First, when bars are no longer rigid but elastic, there arises the possibility of a snap through transition between the different hyperbolas on either side of a branch point. Indeed, tuning various branches close to or farther from a branch point could be used to tune the ease of initiating a snap through transition. This could potentially lead to mechanical structures and mechanical metamaterials whose mechanical response can be reprogrammed *in situ*.

A second interesting extension would be to consider mechanisms built from responsive materials that are sensitive to external stimuli. In that case, the dynamic increase or decrease in the lengths of the bars could be used to drive the pathway of a mechanism in an environmentally dependent manner. This could also be affected if the positions of certain pinned vertices could be made to depend on the external environment or the state of a second input mechanism. This would enable the realization of simple mechanical logic that is robust to some damage because it relies only on the topology of a configuration space [55,56].

Finally, we note that our design principle exploits the fact that the configuration space topology can only change at critical points—configurations where the Jacobian of the constraints fails to be full rank. Our approach is somewhat reminiscent of Morse theory, in which the extrema of a scalar function can be related to the topology of the space on which that function is defined [57]. Morse theory has been used to study the configuration spaces of spherical (and other) linkages [38,39], but we leave it to future work to make this connection more precise.

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APPENDIX A: PERTURBING QUADRATIC CRITICAL POINTS

In this Appendix, we will show that Eq. (9) does, indeed, describe the configuration space near a critical point when the lengths of a linkage are perturbed from their critical values. We assume we have a mechanism with *E* edges and *V* vertices in *d* dimensions with dV > E. We further suppose that the configuration of the mechanism is at a critical point \mathbf{u}_C , with corresponding critical values $(L_{\alpha}^{(c)})^2$. Let $\mathbf{u} = \mathbf{u}_C + \delta \mathbf{u}$ and, correspondingly, $L_{\alpha} = L_{\alpha}^{(c)} + \delta L_{\alpha}$, and expand the squared lengths to quadratic order, using Eq. (1),

$$2L_{\alpha}^{(c)}\delta L_{\alpha} + \delta L_{\alpha}^{2} = \sum_{i} \frac{\partial \ell_{\alpha}^{2}(\mathbf{u}_{C})}{\partial u_{i}} \delta u_{i} + \sum_{ij} \frac{1}{2} \frac{\partial^{2} \ell_{\alpha}^{2}(\mathbf{u}_{C})}{\partial u_{i} \partial u_{j}} \delta u_{i} \delta u_{j}.$$
(A1)

Finally, as in the main text, we assume that Eq. (A1) completely characterizes the critical point, and that there is one self-stress at \mathbf{u}_C , with components σ_{α} , and two zero modes, with components $\zeta_{1,i}$ and $\zeta_{2,i}$.

It will prove convenient to express Eq. (A1) using an orthonormal basis in the space of square lengths, $\{\sigma_{\alpha}, e_{\alpha}^{(1)}, \ldots, e_{\alpha}^{(E-1)}\}$. We similarly write δu_i in an orthonormal basis $\{\zeta_{1,i}, \zeta_{2,i}, \eta_{1,i}, \ldots, \eta_{E-1,i}\}$,

$$\delta u_i = c_1 \zeta_{1,i} + c_2 \zeta_{2,i} + \sum_{I=1}^{E-1} a_I \eta_{I,i}.$$
 (A2)

We first contract Eq. (A1) with σ_{α} , and we obtain an equation that can be expressed as

$$(\mathbf{c}^T \quad \mathbf{a}^T) \begin{pmatrix} \mathcal{Q} & \mathcal{B} \\ \mathcal{B}^T & \mathcal{M} \end{pmatrix} \begin{pmatrix} \mathbf{c} \\ \mathbf{a} \end{pmatrix} = \tilde{\Delta},$$
 (A3)

where the components of the matrices are given by

$$\tilde{\Delta} = \sum_{\alpha} \sigma_{\alpha} \left(2L_{\alpha}^{(c)} \delta L_{\alpha} + \delta L_{\alpha}^2 \right), \tag{A4}$$

$$Q_{nm} = \frac{1}{2} \sum_{\alpha i i} \zeta_{n,i} \zeta_{m,j} \sigma_{\alpha} \frac{\partial^2 \ell_{\alpha}^2}{\partial u_i \partial u_j}, \qquad (A5)$$

$$\mathcal{M}_{nm} = \frac{1}{2} \sum_{\alpha i j} \eta_{n,i} \eta_{m,j} \sigma_{\alpha} \frac{\partial^2 \ell_{\alpha}^2}{\partial u_i \partial u_j}, \tag{A6}$$

and

$$\mathcal{B}_{nm} = \frac{1}{2} \sum_{\alpha i j} \zeta_{n,i} \eta_{m,j} \sigma_{\alpha} \frac{\partial^2 \ell_{\alpha}^2}{\partial u_i \partial u_j}.$$
 (A7)

We also assume that a_I are the components of the vector **a** and that c_1 and c_2 are the components of a two-dimensional vector **c**. Finally, we complete the square in Eq. (A3) to obtain

$$(\mathbf{c} + \mathcal{Q}^{-1}\mathcal{B}\mathbf{a})^T \mathcal{Q}(\mathbf{c} + \mathcal{Q}^{-1}\mathcal{B}\mathbf{a}) = \tilde{\Delta} - \mathbf{a}^T \mathcal{B}^T \mathcal{Q}^{-1}\mathcal{B}\mathbf{a}.$$
 (A8)

Note that Q^{-1} exists because all of the eigenvalues of Q are nonzero by assumption.

Already, Eq. (A3) is in the form of a conic section whose form depends on the eigenvalues of Q. What remains is to show that **a** depends only on the length changes (and not **c**) to lowest order and, ultimately, to find an expression to determine it. To do this, we project Eq. (A1) onto the remaining basis vectors $e_{\alpha}^{(n)}$ in the space of square lengths. We obtain

$$\sum_{m} \sum_{i} \sum_{\alpha} e_{\alpha}^{(n)} \frac{\partial \ell_{\alpha}^{2}(\mathbf{u}_{C})}{\partial u_{i}} \eta_{m,i} a_{m} + \frac{1}{2} \sum_{ij\alpha} e_{\alpha}^{(n)} \frac{\partial^{2} \ell_{\alpha}^{2}(\mathbf{u}_{C})}{\partial u_{i} \partial u_{j}} \delta u_{i} \delta u_{j}$$
$$= \sum_{\alpha} e_{\alpha}^{(n)} (2L_{\alpha}^{(c)} \delta L_{\alpha} + \delta L_{\alpha}^{2}).$$
(A9)

There are E - 1 equations in Eq. (A9) and dV - E + 1 zero modes at the critical point, the space spanned by δu_i^{\perp} is dV - (dV - E + 1) = E - 1 dimensional. The matrix appearing in Eq. (A9) is, consequently, square. Since we have already removed zero modes and self-stresses, it is also invertible. We define a new matrix **M** such that its inverse \mathbf{M}^{-1} is given by the components

$$\mathbf{M}_{nm}^{-1} = \sum_{i} \sum_{\alpha} e_{\alpha}^{(n)} \frac{\partial \ell_{\alpha}^{2}(\mathbf{u}_{C})}{\partial u_{i}} \eta_{m,i}.$$
 (A10)

This then allows us to solve Eq. (A9) in powers of both δL_{α} and **c**. To first order in both, we obtain

$$a_n \approx \sum_m M_{nm} \sum_{\alpha} 2e_{\alpha}^{(m)} L_{\alpha}^{(c)} \delta L_{\alpha} + \mathcal{O}(c\delta L, c^2, \delta L^2).$$
(A11)

We can now put together the results by defining

$$\delta \mathbf{c} = -\mathcal{Q}^{-1} \mathcal{B} \mathbf{a} \tag{A12}$$

and

.

$$\Delta = \tilde{\Delta} - \delta \mathbf{c} \mathcal{Q} \delta \mathbf{c} \tag{A13}$$

to obtain

$$(\delta \mathbf{c} - \delta \mathbf{c})^T \mathcal{Q}(\delta \mathbf{c} - \delta \mathbf{c}) = \Delta, \qquad (A14)$$

where Δ and δc depend linearly on the changes in lengths to lowest order. Therefore, small perturbations of the length are seen to produce trajectories that lie on a 2D conic section with a perturbed center.

While this is a rather intricate derivation, we could have obtained the correct answer up to the order of $\delta \mathbf{u} \sim \delta L^{1/2}$ more simply by assuming $\mathcal{O}(\mathbf{a}) \sim \mathcal{O}(\mathbf{c})$. We have found the full form of Eq. (A14) to be more useful in perturbing larger linkages, however, as it better captures the case that changes in the bar lengths perturb, but do not completely eliminate, critical points in the configuration space of a linkage.

APPENDIX B: PROPERTIES OF THE TANGENT FORM

The tangent form is defined as

$$t^{i_1\cdots i_D}(\mathbf{u}) = \sum_{j_1\cdots j_N} \epsilon^{i_1\cdots i_D j_1\cdots j_N} \frac{\partial f_1(\mathbf{u})}{\partial u_{j_1}}\cdots \frac{\partial f_N(\mathbf{u})}{\partial u_{j_N}}, \qquad (B1)$$

where $\epsilon^{i_1 \cdots i_D j_1 \cdots j_N}$ is the antisymmetric Levi-Civita tensor. Next we compute some simple properties of the tangent form. *The tangent form is divergence free.* This can be seen from the following calculation:

$$\frac{\partial t^{i_1\cdots i_D}(\mathbf{u})}{\partial u_{i_1}} = \sum_{j_1\cdots j_N} \epsilon^{i_1\cdots i_D j_1\cdots j_N} \frac{\partial^2 f_1(\mathbf{u})}{\partial u_{i_1} \partial u_{j_1}} \cdots \frac{\partial f_N(\mathbf{u})}{\partial u_{j_N}}$$
$$\times \cdots + \sum_{j_1\cdots j_N} \epsilon^{i_1\cdots i_D j_1\cdots j_N} \frac{\partial f_1(\mathbf{u})}{\partial u_{j_1}} \cdots \frac{\partial}{\partial u_{i_1}} \frac{\partial f_N(\mathbf{u})}{\partial u_{j_N}}$$
$$= 0, \tag{B2}$$

where each term is zero due to the antisymmetry of the Levi-Civita tensor and the symmetry of partial derivatives.

For one degree of freedom mechanisms, the tangent form is a vector tangent to the configuration space away from critical points. First, we note that

$$\sum_{i_1} \frac{\partial f_{\alpha}}{\partial u_{i_1}} t^{i_1 \cdots i_D} [\mathbf{u}(s)] = 0,$$
(B3)

which implies that $\frac{\partial f_a}{\partial u_i} t^i[\mathbf{u}(s)] = 0$. Now suppose that $\mathbf{u}(s)$ traces the configuration space in a region where $t^{i_1 \cdots i_D}[\mathbf{u}(s)]$ is nonzero. Then,

$$\sum_{i} \frac{\partial f_{\alpha}[\mathbf{u}(s)]}{\partial u_{i}} \frac{\partial u_{i}(s)}{\partial s} = 0.$$
(B4)

Hence the configuration space is perpendicular to all of the $\partial f_{\alpha}(\mathbf{u})/\partial u_i$, but $t_i(\mathbf{u})$ is also perpendicular to all of them. Hence, they must be parallel. The more general case for mechanisms with more than one degree of freedom is more subtle but can also be computed.

The tangent form is zero at **u** if and only if **u** is a critical point. Ultimately, this is a consequence of the fact that the components of $t^{i_1 \cdots i_D}(\mathbf{u})$ are the $E \times E$ minors of the Jacobian of $\ell^2(\mathbf{u})$. Nevertheless, we demonstrate it here for completeness. There are *E* functions,

$$\left\{\frac{\partial f_1(\mathbf{u})}{\partial u_i}, \cdots, \frac{\partial f_E(\mathbf{u})}{\partial u_i}\right\}.$$
 (B5)

Since the zero modes are defined by the nonzero solutions, δu_i of

$$\sum_{i} \frac{\partial f_{\alpha}(\mathbf{u})}{\partial u_{i}} \delta u_{i} = 0$$
 (B6)

the zero modes are in the orthogonal complement of the span of the vectors, $\partial f_{\alpha}(\mathbf{u})/\partial u_i$. At a critical point, there must be additional zero modes and so the $\partial f_{\alpha}(\mathbf{u})/\partial u_i$ span a lowerdimensional space and can no longer be linearly independent. Without loss of generality, we can take it to be $\alpha = 1$ so

$$\frac{\partial f_1(\mathbf{u})}{\partial u_i} = \sum_{\beta > 1} c_\alpha \frac{\partial f_\beta(\mathbf{u})}{\partial u_i}.$$
 (B7)

Substituting this into the definition of $t_{i_1\cdots i_D}(\mathbf{u})$ and using Eq. (B3), we immediately obtain $t_{i_1\cdots i_D}(\mathbf{u}) = 0$.

Similarly, if $t_{i_1\cdots i_D}(\mathbf{u}) = 0$, then the $\partial f_{\alpha}(\mathbf{u})/\partial u_i$ cannot all be linearly independent. One way to do see this is to choose *D* vectors \mathbf{v}_n orthogonal to the $\partial f_{\alpha}(\mathbf{u})/\partial u_i$ for all α , as well as to each other. Then,

$$v_{1,i_1} \cdots v_{D,i_D} t_{i_1 \cdots i_D}(\mathbf{u}) = \det \begin{pmatrix} \mathbf{v}_1^T \\ \vdots \\ \mathbf{v}_D^T \\ \nabla f_1(\mathbf{u})^T \\ \vdots \\ \nabla f_E(\mathbf{u})^T \end{pmatrix} = 0, \quad (B8)$$

where ∇ is the gradient in **u** and superscript *T* denotes the transpose. Since the **v**_n are orthogonal to the other vectors, one of the $\nabla f_{\alpha}(\mathbf{u})$ must be linearly dependent on the rest of them. We immediately obtain that there is at least one additional linear independent zero mode.

Self-stresses are orthogonal to the critical value set. The critical set is defined as the set of points \mathbf{u}_C such that $t_{i_1\cdots i_D}(\mathbf{u}_C) = 0$. The critical value set is the image of the critical set under the map $f_{\alpha}(\mathbf{u}_C)$. Suppose that $\mathbf{u}_C(s)$ is a one-parameter path of points in a smooth portion of the critical set. Then consider its image $F_{\alpha}(s)$,

$$f_{\alpha}(\mathbf{u}_{C}(s)) = F_{\alpha}(s). \tag{B9}$$

If the derivative $\partial F_{\alpha}(s)/\partial s$ is nonzero, then it is tangent to the critical value set. Therefore,

$$\frac{\partial F_{\alpha}(s)}{\partial s} = \sum_{i} \frac{\partial f_{\alpha}(\mathbf{u}_{C}(s))}{\partial u_{i}} \frac{\partial u_{C,i}(s)}{\partial s}.$$
 (B10)

If σ_{α} is a self-stress, then $\sum_{\alpha} \sigma_{\alpha} \partial f_{\alpha} / \partial u_I = 0$. Therefore, we obtain

$$\sum_{\alpha} \sigma_{\alpha} \frac{\partial F_{\alpha}(s)}{\partial s} = 0.$$
 (B11)

Since this is true for any path in the critical value set, it follows that all self-stresses are orthogonal to the critical value set.

Though the converse of this is not true—some vectors normal to the critical value set may not be self-stresses—if the critical value set has codimension one, then there can be only one self-stress and the normal vector of the critical value set necessarily corresponds to that self-stress.

Orientation. The tangent form $t_{i_1\cdots i_D}(\mathbf{u})$ carries additional useful geometrical information about the mechanism at regular (noncritical) configurations. When D = 0, $t(\mathbf{u})$ is a scalar whose sign was used to compute a topological index in periodic mechanisms [36]. Beyond this, it endows the configuration space with a natural orientation in any dimension. At a regular point on the configuration space of a mechanism, \mathbf{x} , $t_{i_1\cdots i_D}(\mathbf{x})dx^{i_1} \wedge \cdots \wedge dx^{i_D}$ is a differential form which provides a local orientation: for any basis of tangent vectors $\{\zeta_{1,j}, \cdots, \zeta_{D,j}\}$,

$$\operatorname{sgn} \sum_{i_1 \cdots i_D} \zeta_{1,i_1} \cdots \zeta_{D,i_D} t_{i_1 \cdots i_D}(\mathbf{u}) = \pm 1.$$
 (B12)

However, note that this local orientation is only defined up to an overall sign since we can always take one of the constraint functions to have the opposite sign.

Though we do not make a great deal of use of it in this paper, it is worth noting that if one is able to find two regions in

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