Observables for bound orbital motion in axially symmetric space-times

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The periastron shift and the Lense-Thirring effect of bound orbital motion in a general axially symmetric space-time given by Pleban´ski and Demian´ski are analyzed. We also define a measure for the conicity of the orbit and give analytic expressions for the observables in terms of hyperelliptic integrals and Lauricella's F_D function. For an interpretation of these analytical expressions, we perform a post-Schwarzschild and a post-Newton expansion of these quantities. This clearly shows the influence of the different space-time parameters on the considered observables and allows one to characterize Kerr, Taub-NUT, Schwarzschild–de Sitter, or other space-times.

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I. INTRODUCTION

Axially symmetric vacuum solutions of the Einstein field equations are used to describe a wide range of black holes appearing in the Universe. The most prominent of these solutions is the Kerr space-time [[1\]](#page-16-0), which describes an axially symmetric rotating black hole. Generalizations of these are the charged Kerr-Newman black holes, the Kerr–de Sitter space-time which incorporates a nonvanishing cosmological constant, the Kerr-Taub-NUT space-time which includes the NUT charge, or some combination of these. All these space-times can be seen as special cases of the family of electrovac space-times of Petrov type D given by the Plebański-Demiański class of solutions [\[2](#page-16-1)]. They are characterized by seven parameters: the mass, the rotation around the symmetry axis, the electric and magnetic charge, the NUT charge, the cosmological constant, and the acceleration of the gravitating source.

Shortly after the introduction of general relativity, Einstein found that for every revolution of a planetary orbit the point of the shortest distance between particle and central object, the periapsis, is shifted in the direction of rotation of the test particle [[3\]](#page-16-2). Indeed, the explanation of the anomalous shift of Mercury's perihelion, together with the observation of light deflection, constituted the breakthrough of general relativity. Already in 1918, Lense and Thirring found that bound orbital motion around an axially symmetric rotating black hole is perturbed not only compared to the Newtonian case but also to the nonrotating case [\[4\]](#page-16-3). The rotation of the central object causes an additional shift of the periapsis and a precession of the orbital plane. For the more exotic Taub-NUT space-time Misner and Taub [\[5\]](#page-16-4) showed that the geodesic motion takes place

on a cone which, if slit open and flattened, defines the orbital plane as it would be for vanishing NUT charge. These three examples show that the parameters of the space-time affect the observables of bound orbital motion and that in turn these observables may be used to characterize a space-time. In this paper we will investigate these orbital effects starting from the general Pleban´ski-Demiański space-time with vanishing acceleration of the gravitating source.

The only way to get access to these characteristics of black holes is through the orbits of particles and light around these black holes. They are given as solutions of the geodesic equation describing the motion of test particles and light rays in a given space-time. For a thorough investigation of the physical properties of orbits analytical solutions of the geodesic equation are most useful. Hagihara [\[6](#page-16-5)] was the first to find an analytical solution of the geodesic equation in Schwarzschild space-times using elliptic functions. He also classified all possible types of orbits in terms of the energy and the angular momentum of the test particle. The complete set of solutions for charged particle motion in Reissner-Nordström spacetimes has been presented only recently by Grunau and Kagramanova [\[7\]](#page-16-6). Shortly after the discovery of the Kerr solution in 1963 [\[1\]](#page-16-0), a number of authors studied the geodesic motion in this space-time. Their results were reviewed and extended by Chandrasekhar [[8](#page-16-7)]. A treatise of the geodesics in Kerr space-time of the same complete-ness as [[6](#page-16-5)] was given only recently by Slezáková [[9](#page-16-8)] and for the motion of charged particles in Kerr-Newman spacetimes by Xu [[10](#page-16-9)]. Geodesics in even more complicated space-times like the one incorporated in the Plebanski-Demian´ski space-time are beyond the methods introduced by Hagihara. However, in 2008 a method has been found to analytically integrate the geodesic equation in Schwarzschild–de Sitter space-times using the theory of

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hyperelliptic functions [\[11](#page-16-10)[,12\]](#page-16-11) and to classify all types of possible orbits in terms of energy and angular momentum of the particles as well as the cosmological constant. This approach was then used to find solutions of the geodesic equations in general Plebański-Demiański space-times with vanishing acceleration of the gravitating source [\[13\]](#page-16-12) and of some special cases [[14](#page-16-13),[15](#page-16-14)].

Bound stable orbits far away from the black hole are not very different from Newtonian orbits but in the vicinity of the black holes they exhibit some peculiar features. These are caused by mismatches between the periodicities of the radial, polar, and azimuthal motion, which are all equal to 2π in the Keplerian case but may differ greatly near black holes. A systematic approach to handle these different periodicities for Kerr space-time was worked out by Schmidt [\[16\]](#page-16-15) and continued by Drasco and Hughes [\[17\]](#page-16-16) using a technique for decoupling the radial and polar motion suggested by Mino [[18](#page-16-17)]. They showed that arbitrary functions of Kerr black hole orbits can be described in the frequency domain and how these expansions may be computed explicitly. Later, Fujita and Hikida [[19](#page-16-18)] derived analytical expressions for the frequencies of radial, polar, and azimuthal motion in Kerr space-time and also for bound timelike orbits in Kerr space-time.

In this paper, we will find analytic expressions of the fundamental frequencies defined by Schmidt [[16](#page-16-15)], Drasco and Hughes [\[17\]](#page-16-16), as well as Fujita and Hikida [\[19\]](#page-16-18) for the general Plebański-Demiański space-time with vanishing acceleration of the gravitating source. We will then investigate how these frequencies are influenced by the parameters of the black hole. To this end, we expand the analytic expression in terms of the parameters in a Taylor series up to first order. This clearly shows the influence of different parameters on the orbits of the black hole and leads to conclusions about which kinds of orbits may belong to a given black hole.

The paper is organized as follows. In Sec. [II,](#page-1-0) we review the equations of motion in Plebański-Demiański spacetimes decoupled by the method introduced by Mino. These equations are then used in Sec. [III](#page-2-0) to define the fundamental frequencies and observables for Plebański-Demiański space-time following closely the arguments of Fujita and Hikida. All necessary quantities in Schwarzschild space-time as well as their post-Newtonian expansions are computed in Sec. [IV.](#page-3-0) These quantities will serve as a reference for the comparison with more complicated space-times later on. Section [V](#page-5-0) is the most technical one, where the linear correction to the fundamental frequencies are computed. These corrections are given in terms of elementary function or in terms of complete Jacobian elliptic integrals. In Sec. [VI](#page-10-0) we use these results to compute the post-Schwarzschild and post-Newtonian corrections to the periastron shift, the Lense-Thirring effect, and the conicity and compare them to earlier results. A discussion and outlook closes the paper.

II. GEODESIC MOTION IN PLEBAŃSKI-DEMIAŃSKI SPACE-TIME

The axially symmetric Pleban^{ski}-Demian^{ski} spacetimes are characterized by the seven parameters mass, rotation, acceleration, cosmological constant, NUT parameter, electric charge, and magnetic charge. It can be shown that the Hamilton-Jacobi equation for these spacetimes is separable and the geodesic equation integrable, if and only if the acceleration vanishes or null geodesics are considered. In this paper we will consider the case of geodesic motion of massive test particles in a Plebański-Demiański space-time with vanishing acceleration. We will also assume that the test particles are neutral, i.e. without electric or magnetic charge.

The six-parameter Plebański-Demiański space-times considered here are then given by the metric [\[2](#page-16-1)[,20,](#page-16-19)[21\]](#page-16-20) (we use units where $c = 1 = G$)

$$
ds^2/M^2 = \frac{\Delta_r}{p^2}(dt - Ad\varphi)^2 - \frac{p^2}{\Delta_r}dr^2
$$

$$
-\frac{\Delta_\theta}{p^2}\sin^2\theta(adt - Bd\varphi)^2 - \frac{p^2}{\Delta_\theta}d\theta^2, \quad (1)
$$

where $p^2 = r^2 + (n - a\cos\theta)^2$, $A = a\sin^2\theta + 2n\cos\theta$, $B = r^2 + a^2 + n^2$,

$$
\Delta_r = (r^2 + a^2 - n^2)(1 - \Lambda(r^2 + 3n^2))
$$

- 2r + Q_e² + Q_m² - 4\Lambda n²r², (2)

$$
\Delta_{\theta} = 1 + a^2 \Lambda \cos^2 \theta - 4\Lambda a n \cos \theta. \tag{3}
$$

Here a , the angular momentum per mass of the gravitating source, the NUT parameter *n*, the electric charge Q_e , and the magnetic charge Q_m as well as the coordinates r and t are normalized with respect to M , where M is the mass of the gravitating object. The dimensionless parameter Λ denotes the cosmological constant divided by three and normalized by multiplication with M^2 .

The equations of motion for massive test particles in these space-times are given by

$$
\left(\frac{dr}{d\lambda}\right)^2 = P(r)^2 - \Delta_r(r^2 + C) =: R(r),\tag{4}
$$

$$
\left(\frac{d\theta}{d\lambda}\right)^2 = \Delta_\theta (C - (n - a\cos\theta)^2) - \frac{O(\theta)^2}{\sin^2\theta} =: \Theta(\theta), \tag{5}
$$

$$
\frac{d\varphi}{d\lambda} = \frac{a}{\Delta_r} P(r) + \frac{O(\theta)}{\Delta_\theta \sin^2 \theta} =: \Phi(r, \theta), \tag{6}
$$

$$
\frac{dt}{d\lambda} = \frac{r^2 + a^2 + n^2}{\Delta_r} P(r) + \frac{a\sin^2\theta + 2n\cos\theta}{\Delta_\theta \sin^2\theta} O(\theta)
$$

=: $T(r, \theta)$, (7)

where

$$
P(r) = (r^2 + a^2 + n^2)E - aL,
$$
 (8)

$$
O(\theta) = L - (a\sin^2\theta + 2n\cos\theta)E.
$$
 (9)

The constants of motion E , L , and C have the meaning of energy, angular momentum in direction of the symmetry axes, and Carter constant, each per unit mass. The two constants L and C are additionally normalized by division by M and M^2 , respectively, such that they are dimensionless. The affine parameter λ is the Mino time [\[18\]](#page-16-17), also normalized to M, $d\lambda = d\tau/(Mp^2)$ with the eigentime τ . Note that R depends quadratically on *n*, Q_e , and Q_m and that Θ does not depend on Q_e and Q_m .

Equations [\(4\)](#page-1-1)–([7](#page-1-2)) can be solved analytically [\[13](#page-16-12)[,22,](#page-16-21)[23\]](#page-16-22). However, in this paper we are interested in the periods of the motion, which can be used to define observables related to geodesic motion in these space-times.

III. OBSERVABLES FOR BOUND ORBITS

In Newtonian gravity bound geodesic motion is described by a fixed ellipse defining the orbital plane. This is no longer true in the framework of general relativity. In the spherically symmetric Schwarzschild space-time, the orbital plane remains fixed while the ellipse precesses resulting in a periastron shift. In the axially symmetric Kerr space-time additionally the orbital plane itself precesses what is known as the Lense-Thirring effect [[24](#page-16-23)]. In the more exotic Taub-NUT space-time the geodesic motion of test particles does not lie in a plane at all but on a cone; see e.g. [\[25](#page-16-24)].

The precession of the orbital ellipse and the orbital plane is induced by mismatches of the periods of the motion in the r and θ coordinates compared to the average secular increase of the angle φ about the symmetry axes. These effects were discussed in the framework of a Kerr space-time by Schmidt [\[16\]](#page-16-15), Drasco and Hughes [\[17\]](#page-16-16), and Fujita and Hikida [[19\]](#page-16-18). In the following sections, we will use the procedure in [\[19](#page-16-18)] to analyze the first-order corrections to the periastron shift and the Lense-Thirring effect due to the parameters a , n , and Λ . (As Q_e and Q_m appear only quadratically, there are no linear effects due to electric or magnetic charge on neutral test particles.) Below we review their line of argument for the convenience of the reader. In addition, we will characterize the deviation from an orbital plane, the conicity, due to the parameter n .

For bound orbits, the radial and polar components r and θ vary between a minimal and maximal value given by the turning points $\frac{dr}{d\lambda} = 0$ and $\frac{d\theta}{d\lambda} = 0$. The periods Λ_r of r and Λ_{θ} of θ with respect to the Mino time λ are then defined by a revolution from maximum to minimum and back to the maximal value. This means that Λ_r and Λ_θ are defined by the smallest nonzero real value with $r(\lambda + \Lambda_r) = r(\lambda)$ and $\theta(\lambda + \Lambda_{\theta}) = \theta(\lambda)$ giving

$$
A_r = \oint_{\alpha_r} \frac{dr}{\sqrt{R(r)}} = 2 \int_{r_p}^{r_a} \frac{dr}{\sqrt{R(r)}},
$$
 (10)

$$
A_{\theta} = \oint_{\alpha_{\theta}} \frac{d\theta}{\sqrt{\Theta(\theta)}} = 2 \int_{\theta_{\min}}^{\theta_{\max}} \frac{d\theta}{\sqrt{\Theta(\theta)}},
$$
(11)

where r_p is the periapsis and r_a the apoapsis. In general these are hyperelliptic integrals, which are a generalization of elliptic integrals. [The closed integration path α_r (α_θ) refers to the integration on the Riemann surface of the algebraic curve defined by $y^2 = R(r)$ ($y^2 = \Theta(\cos\theta)$). It runs around the branch cut connecting r_p and r_a ($\cos\theta_{\text{min}}$) and $cos\theta_{\text{max}}$). On the Riemann surface the two branches of the square root are glued to one analytic function.] From these two periods conjugate fundamental frequencies can be defined by

$$
Y_r := \text{sgn} L \frac{2\pi}{\Lambda_r}, \qquad Y_\theta := \text{sgn} L \frac{2\pi}{\Lambda_\theta}.
$$
 (12)

The sign of L is included in this definition to indicate the direction in which the particle travels around the gravitating object relative to $a > 0$. This means that $L > 0$ corresponds to a prograde and $L < 0$ to a retrograde orbit.

For vanishing n , Eq. [\(5](#page-1-3)) is symmetric with respect to the equatorial plane $\theta = \frac{\pi}{2}$. However, in the more general space-time considered here this is not true, and the deviation from this symmetry can be measured by the difference between $\frac{1}{2}(\theta_{\min} + \theta_{\max})$ and $\frac{\pi}{2}$. In the case of a Taub-NUT space-time, where all parameters except mass and NUT parameter vanish, this implies that the particle moves on an orbital cone rather than an orbital plane [\[5](#page-16-4)]. Later we will see that this phenomenon also appears in the weak field limit. For these reasons, we will refer to this quantity as the conicity, $\Delta_{\text{conicity}} := \pi - (\theta_{\min} + \theta_{\max})$. This means that $\Delta_{\text{conicity}} > 0$ corresponds to a cone opened in the northern direction and $\Delta_{\text{conicity}} < 0$ to a cone opened in the southern direction.

The nature of Eqs. [\(6\)](#page-1-4) and ([7\)](#page-1-2) is somewhat different from ([4\)](#page-1-1) and ([5\)](#page-1-3) as they cannot be solved by periodic functions. They depend on both r and θ but can be separated in an r -dependent and a θ -dependent part,

$$
\Phi(r,\theta) = \Phi_r(r) + \Phi_\theta(\theta),\tag{13}
$$

$$
T(r,\theta) =: T_r(r) + T_\theta(\theta). \tag{14}
$$

The solutions $\varphi(\lambda)$ and $t(\lambda)$ of Eqs. [\(6](#page-1-4)) and [\(7](#page-1-2)) can be written as an averaged part linear in λ plus perturbations in r and θ ,

$$
\varphi(\lambda) = \langle \Phi(r,\theta) \rangle_{\lambda} \lambda + \Phi_{\text{osc}}^r(r) + \Phi_{\text{osc}}^{\theta}(\theta), \qquad (15)
$$

$$
t(\lambda) = \langle T(r,\theta) \rangle_{\lambda} \lambda + T_{\text{osc}}^r(r) + T_{\text{osc}}^{\theta}(\theta), \quad (16)
$$

where

$$
\langle \cdot \rangle_{\lambda} := \lim_{(\lambda_2 - \lambda_1) \to \infty} \frac{1}{2(\lambda_2 - \lambda_1)} \oint_{\lambda_1}^{\lambda_2} \cdot d\lambda \tag{17}
$$

is an infinite time average with respect to λ , and $X_{\csc}^r(r)$ and $X_{\text{osc}}^{\theta}(\theta)$ (with $X = \Phi$ or $X = T$) represent oscillatory deviations from this average. They are defined by

$$
X_{\text{osc}}^r(r) = \int X_r(r) d\lambda - \langle X_r(r) \rangle_{\lambda} \lambda,\tag{18}
$$

$$
X_{\text{osc}}^{\theta}(\theta) = \int X_{\theta}(\theta) d\lambda - \langle X_{\theta}(\theta) \rangle_{\lambda} \lambda \tag{19}
$$

and have periods Λ_r and Λ_θ . Therefore, the average secular increase of φ and t with respect to λ is given by

$$
\Gamma := \langle T(r,\theta) \rangle_{\lambda} = \langle T_r(r) \rangle_{\lambda} + \langle T_\theta(\theta) \rangle_{\lambda}, \tag{20}
$$

$$
\Upsilon_{\varphi} := \langle \Phi(r,\theta) \rangle_{\lambda} = \langle \Phi_r(r) \rangle_{\lambda} + \langle \Phi_{\theta}(\theta) \rangle_{\lambda}.
$$
 (21)

As X_r and X_θ (again with $X = \Phi$ or $X = T$) are periodic functions with respect to λ because r and θ are periodic functions, their integrals,

$$
\oint_{\lambda_1}^{\lambda_2} X_r(r(\lambda)) d\lambda,
$$

and

$$
\oint_{\lambda_1}^{\lambda_2} X_\theta(\theta(\lambda)) d\lambda,
$$

in the definition of the infinite time average can be reduced to an integral over one period Λ_r or Λ_θ , respectively. This yields

$$
\Upsilon_{\varphi} = \frac{2}{\Lambda_r} \int_{r_{\rm p}}^{r_{\rm a}} \frac{\Phi_r(r) dr}{\sqrt{R(r)}} + \frac{2}{\Lambda_\theta} \int_{\theta_{\rm min}}^{\theta_{\rm max}} \frac{\Phi_\theta(\theta) d\theta}{\sqrt{\Theta(\theta)}},\qquad(22)
$$

$$
\Gamma = \frac{2}{\Lambda_r} \int_{r_{\rm p}}^{r_{\rm a}} \frac{T_r(r) dr}{\sqrt{R(r)}} + \frac{2}{\Lambda_\theta} \int_{\theta_{\rm min}}^{\theta_{\rm max}} \frac{T_\theta(\theta) d\theta}{\sqrt{\Theta(\theta)}}.
$$
 (23)

The corresponding frequencies with respect to the coordinate time t are then given by

$$
\Omega_r := \frac{Y_r}{\Gamma}, \qquad \Omega_\theta := \frac{Y_\theta}{\Gamma}, \qquad \Omega_\varphi := \frac{Y_\varphi}{\Gamma}.
$$
 (24)

These are frequencies as seen by an observer at infinity.

In the limit of weak gravitational fields the mismatch of the frequency of the φ - and r-motion, $\Omega_{\rm P} = \Omega_{\varphi} - \Omega_r$ can be interpreted as the precession of the orbital ellipse, and the mismatch of the frequency of the φ - and θ -motion, $\Omega_{LT} = \Omega_{\varphi} - \Omega_{\theta}$ as a precession of the orbital plane. In strong gravitational fields the orbits are in general of such irregularity that orbital planes or ellipses can no longer be identified.

IV. REFERENCE ORBIT

In the next section we will analyze the influence of the parameters a , Λ , and n on the observables of a bound reference orbit, which is neither circular nor polar. (Polar orbits may be considered as a special case but we leave this out here. For a discussion of these orbits in Kerr spacetime, see [\[26\]](#page-16-25).) This will be done by post-Schwarzschild and post-Newton expansions, where we assume the constants of motions to be fixed. But first we will introduce and characterize the reference orbit.

If all space-time parameters except the mass vanish, the Pleban´ski-Demian´ski space-time is identical to the Schwarzschild space-time and the functions $R(r)$ and $\Theta(\theta)$ of Eqs. ([4](#page-1-1)) and [\(5\)](#page-1-3) reduce to

$$
R_0(r) := (E^2 - 1)r^4 + 2r^3 - Cr^2 + 2Cr,
$$
 (25)

$$
\Theta_0(\theta) := C - \frac{L^2}{\sin^2 \theta}.
$$
 (26)

A necessary condition for the existence of a noncircular bound orbit in a Schwarzschild space-time is that $R_0(r)$ has four real zeros, $0 = r_{01} < r_{02} < r_{03} < r_{04} < \infty$, where $R_0(r) > 0$ for $r_{03} < r < r_{04}$; cf. [[23](#page-16-22)]. Therefore, our bound reference orbit has turning points r_{03} and r_{04} . Note that instead of using E^2 and C all formulas can also be expressed in terms of the turning points: A comparison of coefficients in $(E^2 - 1)r^4 + 2r^3 - Cr^2 + 2Cr = (E^2 1)r(r - r_{02})(r - r_{03})(r - r_{04})$ yields

$$
E^2 - 1 = \frac{-2}{r_{02} + r_{03} + r_{04}},\tag{27}
$$

$$
C = \frac{r_{02}r_{03}r_{04}}{r_{02} + r_{03} + r_{04}},\tag{28}
$$

$$
r_{02} = \frac{-2r_{03}r_{04}}{2r_{03} + 2r_{04} - r_{03}r_{04}}.\tag{29}
$$

A nonpolar orbit requires $C \geq L^2$ and lies in an orbital plane with inclination $\arcsin\frac{|L|}{\sqrt{C}}$ (or $\frac{\pi}{2}$ – $\arcsin\frac{|L|}{\sqrt{C}}$ if measured from the equatorial plane). Therefore, the θ -motion of the reference orbit is symmetric with respect to the equatorial plane and confined to $[\theta_{01}, \theta_{02}]$ with $\theta_{01} =$ $\arcsin\frac{|L|}{\sqrt{C}} \in [0, \frac{\pi}{2}]$ and $\theta_{02} = \pi - \arcsin\frac{|L|}{\sqrt{C}} \in [\frac{\pi}{2}, \pi]$. In particular, the orbit lies in the equatorial plane if $C = L^2$. In terms of the inclination and the turning points the constant of motion L is given by

$$
L = \pm \sqrt{C} \sin \theta_{01} = \pm \sqrt{\frac{r_{02} r_{03} r_{04}}{r_{02} + r_{03} + r_{04}}} \sin \theta_{01}.
$$
 (30)

In the following we will calculate the expressions Y_r , Y_θ , Υ_{ϕ} , and Γ for the reference orbit.

A. Frequency of r

Let us first calculate the r period for our reference orbit. For an orbit bound between r_{03} and r_{04} we get

$$
\Lambda_{r,0} := 2 \int_{r_{03}}^{r_{04}} \frac{dr}{\sqrt{R_0(r)}}.
$$
\n(31)

This is a complete elliptic integral of the first kind which can be easily transformed to the Legendre form giving

$$
A_{r,0} = \frac{4K(k)}{\sqrt{(1 - E^2)r_{03}(r_{04} - r_{02})}},
$$
(32)

with $k^2 = r_{02}(r_{04} - r_{03})/(r_{03}(r_{04} - r_{02}))$. For general information on the complete elliptic integral of the first kind,

$$
K(k) = \int_0^1 \frac{dt}{\sqrt{(1 - t^2)(1 - k^2 t^2)}},
$$
(33)

see e.g. [\[27\]](#page-16-26); for fast numerical computation, see e.g. [[28\]](#page-16-27). In computer algebra systems like MATHEMATICA or MAPLE the complete elliptic integrals are usually implemented and, therefore, $\Lambda_{r,0}$ can be computed easily. The conjugate fundamental frequency $Y_{r,0}$ is then given by

$$
Y_{r,0} = \frac{2\pi}{4K(k)} \operatorname{sgn} L \sqrt{(1 - E^2) r_{03} (r_{04} - r_{02})}.
$$
 (34)

B. Frequency of θ

The θ period of the reference orbit is given by

$$
\Lambda_{\theta,0} := 2 \int_{\theta_{01}}^{\theta_{02}} \frac{d\theta}{\sqrt{\Theta_0(\theta)}} = 2 \int_{\xi_{02}}^{\xi_{03}} \frac{d\xi}{\sqrt{C(1-\xi^2) - L^2}},
$$
\n(35)

where we substituted $\xi = \cos\theta$ and $\cos\theta_{01} = \sqrt{1 - \frac{L^2}{C}} =$: ξ_{03} , $\cos\theta_{02} = \xi_{02} = -\xi_{03}$. This can be solved by

$$
\Lambda_{\theta,0} = \frac{2\pi}{\sqrt{C}} = \frac{2\pi\sqrt{r_{02} + r_{03} + r_{04}}}{\sqrt{r_{02}r_{03}r_{04}}},\tag{36}
$$

from which we infer

$$
Y_{\theta,0} = \text{sgn}L \frac{2\pi}{\Lambda_{\theta,0}} = \text{sgn}L\sqrt{C}.
$$
 (37)

Note that for the case $\theta_{01} = \frac{\pi}{2} = \theta_{02}$ it is $\theta(\lambda) \equiv \frac{\pi}{2}$ and, therefore, $\Lambda_{\theta,0}$ is undefined. However, we can treat this as the limiting case $C \rightarrow L^2$, which gives the same results for $A_{\theta,0}$ and $\widetilde{Y}_{\theta,0}$ as above.

C. Frequency of φ

For the reference orbit Eq. [\(6\)](#page-1-4) simplifies to

$$
\frac{d\varphi}{d\lambda} = \frac{L}{\sin^2 \theta} \tag{38}
$$

and Y_{φ} can be calculated to

$$
\begin{split} \Upsilon_{\varphi,0} &= \frac{2}{\Lambda_{\theta,0}} \int_{\theta_{01}}^{\theta_{02}} \frac{L}{\sin^2 \theta \sqrt{\Theta_0(\theta)}} = \frac{2\pi \operatorname{sgn} L}{\Lambda_{\theta,0}} \\ &= \operatorname{sgn} L \sqrt{C}, \end{split} \tag{39}
$$

where we used the substitution $\xi = \cos\theta$ as in Sec. [IV B](#page-4-0). Although $\Lambda_{\theta,0}$ is not defined for $\theta(\lambda) \equiv \frac{\pi}{2}$ we get in this case $\varphi(\lambda) = L\lambda$ and, therefore, the same result $Y_{\varphi,0} =$ $L = \text{sgn}L\sqrt{C}.$

D. Frequency of t

The expression for Γ_0 is the most complicated in this section as it involves an elliptic integral of the third kind. For the reference orbit Eq. [\(7](#page-1-2)) simplifies to

$$
\frac{dt}{d\lambda} = \frac{r^3 E}{r - 2},\tag{40}
$$

which leads to

$$
\Gamma_0 = \frac{2}{\Lambda_{r,0}} \int_{r_{02}}^{r_{03}} \frac{r^3 E dr}{(r-2)\sqrt{R_0(r)}}
$$

=
$$
\frac{E}{2K(k)} \left(-r_{03}r_{04}K(k) + r_{03}(r_{04} - r_{02})E(k) + \frac{2r_{04}(3-2E^2)}{1-E^2} \Pi(n_1, k) + \frac{8r_{04}}{r_{04}-2} \Pi(n_2, k) \right), (41)
$$

where $E(k)$ and $\Pi(n, k)$ are the complete elliptic integrals of second and third kind,

$$
E(k) = \int_0^1 \frac{(1 - k^2 t^2)dt}{\sqrt{(1 - t^2)(1 - k^2 t^2)}},
$$
(42)

$$
\Pi(n,k) = \int_0^1 \frac{dt}{(1 - nt^2)\sqrt{(1 - t^2)(1 - k^2 t^2)}},\qquad(43)
$$

and the parameters n_1 , n_2 are given by

$$
n_1 = \frac{r_{03} - r_{04}}{r_{03}} < 0, \qquad n_2 = \frac{2(r_{04} - r_{03})}{r_{03}(r_{04} - 2)} < 0. \tag{44}
$$

E. Observables

Let us now collect the results so far in this section. As expected for $a = 0$, the frequencies $Y_{\theta,0}$ and $Y_{\varphi,0}$ coincide and the Lense-Thirring effect vanishes,

$$
\Omega_{LT,0} = \frac{Y_{\varphi,0} - Y_{\theta,0}}{\Gamma_0} = 0.
$$
\n(45)

Likewise, for $n = 0$ the particle moves on an orbital plane,

$$
\Delta_{\text{conicity},0} = \pi - (\theta_{01} + \theta_{02}) = 0. \tag{46}
$$

The periastron shift $\Omega_{P,0} = (\Upsilon_{\varphi,0} - \Upsilon_{r,0})\Gamma_0^{-1}$ is given in terms of elliptic integrals

$$
\Omega_{P,0} = \frac{\text{sgn}L}{E} \frac{2\sqrt{C}K(k) - \pi\sqrt{r_{03}(r_{04} - r_{02})(1 - E^2)}}{(-r_{03}r_{04}K(k) + r_{03}(r_{04} - r_{02})E(k) + \frac{2r_{04}(3 - 2E^2)}{1 - E^2}\Pi(n_1, k) + \frac{8r_{04}}{r_{04} - 2}\Pi(n_2, k))}.
$$
\n(47)

The usual result for the periastron shift in terms of radians or degrees can be found in the following way: Using the averaged $\lambda(\varphi) = \varphi \Upsilon_{\varphi}^{-1}$ we get the period of the *r*-motion in terms of φ , $r_{\varphi} := r \circ \lambda$: $\varphi \mapsto r(\varphi Y_{\varphi}^{-1})$, by observing that $r_{\varphi}(\varphi + Y_{\varphi} \tilde{\Lambda}_r) = r(\varphi Y_{\varphi}^{-1} + \Lambda_r) = r_{\varphi}(\varphi)$. From this period we get for the difference between the angle of the periapsis and 2π after one revolution,

$$
\Delta_{P,0} = Y_{\varphi,0} \Lambda_{r,0} - 2\pi \operatorname{sgn} L
$$

= $\operatorname{sgn} L \Biggl(\frac{4\sqrt{C}K(k)}{\sqrt{(1 - E^2)r_{03}(r_{04} - r_{02})}} - 2\pi \Biggr).$ (48)

If this shift should be expressed in terms of radians per time, it can be referred to the time needed for a revolution from some fixed φ_0 to φ_0 again, i.e. the sidereal period, or for a revolution from some fixed r_0 to r_0 again, i.e. the anomalistic period. The first choice corresponds to the usual notion of the periastron shift in arc seconds per century whereas the second corresponds to the definition of $\Omega_{\rm P}$: The time elapsed for a revolution from, say, periapsis to periapsis is given as the period of $r_t := r \circ$ $\lambda: t \mapsto r(t\Gamma^{-1}),$ which is ΓA_r . If $\Delta_p = \Upsilon_p A_r - 2\pi \text{sgn}L$ is divided by this period we obtain $\Omega_{\rm P} = \Delta_{\rm P}/(\Gamma \Lambda_r)$.

Note that ([47](#page-5-1)) and ([48](#page-5-2)) are exact and, therefore, more complicated than the post-Newtonian formula given e.g. in [\[29\]](#page-16-28). If we consider the weak field approximation by assuming that the periapsis r_{03} and the apoapsis r_{04} become large, we recover Eq. (51) of [\[29\]](#page-16-28) with vanishing solar quadrupole momentum and parameters corresponding to general relativity,

$$
\Delta_{P,0} \approx \frac{6\pi \operatorname{sgn}L}{d(1-\epsilon^2)}M,\tag{49}
$$

where $d = \frac{M}{2}(r_a + r_p)$ is the semimajor axis and $\epsilon = \frac{M}{2d} \times$ $(r_a - r_p)$ the eccentricity with the apoapsis $r_a = r_{04}$ and the periapsis $r_p = r_{03}$. Here we used ([27](#page-3-1))–[\(29\)](#page-3-2) to perform the series expansion. Note that the angular momentum of the test particle is usually chosen to be positive and, therefore, its sign does not appear in Eq. (51) of [\[29\]](#page-16-28). But we include it here as we choose the sign of L relative to the angular momentum of the gravitating source, which will be nonzero later on. The post-Newtonian expression for $\Omega_{\rm P}$ is given by

$$
\Omega_{P,0} \approx \frac{3 \text{ sgn} L}{d^{(5/2)}(1 - \epsilon^2)} M^{(5/2)},\tag{50}
$$

which includes the perturbation of the time needed for one revolution from periapsis to periapsis.

Let us test our results for the orbital motion of Mercury. From [\[30\]](#page-17-0) we take for peri- and aphelion the values

$$
r_{\min} = (307\,500.7 \pm 3.0) \times 10^{-6} \text{ AU},\tag{51}
$$

$$
r_{\text{max}} = (466\,696.6 \pm 2.6) \times 10^{-6} \text{ AU},\tag{52}
$$

and use $M = 1476.62528$ m for the mass of the Sun to determine $r_{03} = \frac{r_{\text{min}}}{M}$ and $r_{04} = \frac{r_{\text{max}}}{M}$. With Eqs. [\(27\)](#page-3-1)–[\(29\)](#page-3-2) and the results of this section we obtain

$$
\Omega_{\text{P},0} = 3.252308 \times 10^{-19} \pm 5.5 \times 10^{-24}. \quad (53)
$$

In order to express this dimensionless quantity in terms of arc seconds per century we have to multiply it by $\frac{c}{M}$ where $c = 299792458 \text{ m s}^{-1}$ is the speed of light. Then we get

$$
\Omega_{\text{P},0} \frac{c}{M} = 42.98048 \pm (0.73 \times 10^{-3}) \, (\text{arc sec}) / \text{cy}, \quad (54)
$$

in consistency with observations; cf. [\[31\]](#page-17-1). However, it is not the usual result in the sense that it describes the perihelion shift per revolution from periapsis to periapsis rather than from 0 to 2π .

The usual result for the perihelion shift in terms of arc seconds per century can by found using $\Delta_{\rm P}$,

$$
\Delta_{P,0} = 5.018\,648\,5 \times 10^{-7} \pm 3.7 \times 10^{-12}.\tag{55}
$$

In [\[30\]](#page-17-0) a revolution of Mercury is given as 87.969 257 days. We get with $t_{\varphi} := t \circ \lambda$: $\varphi \mapsto \Gamma_0 Y_{\varphi,0}^{-1} \varphi$ for a revolution of 2π

$$
y_{\rm M} := \frac{2\pi\Gamma_0}{\Upsilon_{\varphi,0}} \frac{M}{c} = 87.96925 \pm (8.5 \times 10^{-4}) \text{ days}, \quad (56)
$$

which agrees to the given accuracy with observation. This yields for the perihelion shift

$$
\frac{\Delta_{P,0}}{y_M} = 42.98048 \pm (0.73 \times 10^{-3}) \, (\text{arc sec}) / \text{cy}, \quad (57)
$$

in accord with observations. We see here that this value coincides with ([54](#page-5-3)) within the given accuracy and, therefore, the two different definitions cannot be distinguished.

V. FIRST-ORDER CORRECTIONS

In the following we will calculate the linear post-Schwarzschild corrections for all quantities used to define the observables $\Omega_{\rm P}$ and $\Omega_{\rm LT}$ as well as the conicity $\Delta_{\rm conicity}$ due to the parameters a, n, and Λ . As the parameters Q_{e} and Q_m appear only quadratically in Eqs. ([4](#page-1-1))–[\(7\)](#page-1-2), we will not study them here. However, this would be a totally analogous procedure. We assume the constants of motion to be fixed but let the zeros r_i of R and θ_i of Θ vary. By this procedure we will reduce the hyperelliptic integrals appearing in the definitions of Λ_r , Λ_θ , Λ_φ , and $\overline{\Gamma}$ for the

general Plebański-Demiański space-time to elliptic integrals and elementary expressions.

The observables $\Omega_{\rm P}$ and $\Omega_{\rm LT}$ are defined through the frequencies Y_r , Y_θ , Y_φ , and Γ . Whereas Y_r and Y_θ depend only on the variable indicated in the index, Y_{φ} and Γ can be separated in an r- and a θ -dependent part, $Y_{\varphi} = Y_{\varphi r} + Y_{\varphi \theta}$ and $\Gamma = \Gamma_r + \Gamma_{\theta}$, where

$$
\begin{aligned}\n\Upsilon_{\varphi r} &:= \frac{2}{\Lambda_r} \int_{r_\rho}^{r_a} \frac{\Phi_r(r) dr}{\sqrt{R(r)}} =: \frac{I_{\varphi r}}{\Lambda_r}, \\
\Upsilon_{\varphi \theta} &:= \frac{2}{\Lambda_\theta} \int_{\theta_{\text{min}}}^{\theta_{\text{max}}} \frac{\Phi_\theta(\theta) d\theta}{\sqrt{\Theta(\theta)}} =: \frac{I_{\varphi \theta}}{\Lambda_\theta},\n\end{aligned} \tag{58}
$$

$$
\Gamma_r := \frac{2}{\Lambda_r} \int_{r_\rho}^{r_a} \frac{T_r(r)dr}{\sqrt{R(r)}} =: \frac{I_{tr}}{\Lambda_r},
$$
\n
$$
\Gamma_\theta := \frac{2}{\Lambda_\theta} \int_{\theta_{\text{min}}}^{\theta_{\text{max}}} \frac{T_\theta(\theta) d\theta}{\sqrt{\Theta(\theta)}} =: \frac{I_{t\theta}}{\Lambda_\theta}.
$$
\n(59)

In terms of these expressions the periastron shift and the Lense-Thirring effect can be restated as

$$
\Omega_{\rm P} = \frac{Y_{\varphi} - Y_r}{\Gamma} = \frac{I_{\varphi r} A_{\theta} + I_{\varphi \theta} A_r - \text{sgn} L \, 2\pi A_{\theta}}{I_{tr} A_{\theta} + I_{t\theta} A_r},\tag{60}
$$

$$
\Omega_{LT} = \frac{Y_{\varphi} - Y_{\theta}}{\Gamma} = \frac{I_{\varphi r} A_{\theta} + I_{\varphi \theta} A_{r} - \text{sgn} L 2\pi A_{r}}{I_{tr} A_{\theta} + I_{t\theta} A_{r}}.
$$
 (61)

A. Standard form of hyperelliptic integrals

All integrals appearing here have the form

$$
\int_{x_1}^{x_2} \frac{f(x)dx}{\sqrt{P_6(x)}},\tag{62}
$$

where P_6 is a polynomial of degree 6 in $x = r$ or $x = \xi =$ $\cos\theta$, $f(x)$ is a rational function, and x_1 , x_2 are zeros of P_6 . This type of integral is called a hyperelliptic integral. The functions P_6 and f as well as the zeros x_1, x_2 depend on the parameters a , n , and Λ . For a Taylor expansion of such an integral it is of advantage to first reduce it to a standard form similar to the elliptic integrals which appeared in the calculations for the reference orbit. However, to our knowledge such a standard form does not exist in the literature. A straightforward generalization of the Legendre standard form of elliptic integrals of the first kind can be obtained by an additional term $(1 - k_2^2 t^2)$ under the square root but this yields again an elliptic integral as a substitution $s = t^2$ shows. A better choice is to generalize the Riemann form of elliptic integrals,

$$
\int_0^1 \frac{dt}{\sqrt{t(1-t)(1-k^2t)}},\tag{63}
$$

to the hyperelliptic form

$$
\int_0^1 \frac{(At+B)dt}{\sqrt{t(1-t)(1-k_1^2t)(1-k_2^2t)(1-k_3^2t)}},\qquad(64)
$$

which was also used in [[26](#page-16-25)] for the calculation of the periastron shift of equatorial orbits in Kerr–de Sitter space-times. As pointed out in [\[26\]](#page-16-25), the form ([64](#page-6-0)) can also be expressed in terms of Lauricella's hypergeometric F_D function; see also [A](#page-13-0)ppendix A.

The transformation of the form [\(62\)](#page-6-1) to the form [\(64\)](#page-6-0) depends on the range of integration $[x_1, x_2]$. In the case of $P_6(x) = R(r)$, we have four real zeros $r_1 < r_2 < r_3 < r_4$ and another two zeros r_0 , r_5 which may be complex and which tend to infinity for vanishing Λ . Here $x_1 = r_3$ and $x_2 = r_4$ result in a transformation $r = (At + B)^{-1} + r_1$, which yields

$$
\int_{r_3}^{r_4} \frac{dr}{\sqrt{R(r)}} = \frac{1}{\sqrt{D}} \int_0^1 \frac{(At + B)dt}{\sqrt{t(1 - t)(1 - k_1^2 t)(1 - k_2^2 t)(1 - k_3^2 t)}}
$$

$$
= \frac{A}{\sqrt{D}} \frac{\pi}{2} F_D \left(\frac{3}{2}, \vec{\beta}_1, 2, \vec{m}\right)
$$

$$
+ \frac{B}{\sqrt{D}} \pi F_D \left(\frac{1}{2}, \vec{\beta}_1, 1, \vec{m}\right), \tag{65}
$$

where $B = 1/(r_4 - r_1) =: B_r$,

$$
A = \frac{r_4 - r_3}{(r_4 - r_1)(r_3 - r_1)} =: A_r,
$$
 (66)

$$
D = \Lambda \frac{(r_3 - r_1)(r_4 - r_2)}{(r_4 - r_1)^2} (r_4 - r_0)(r_5 - r_4) =: D_r, \quad (67)
$$

$$
k_1^2 = \frac{(r_4 - r_3)(r_2 - r_1)}{(r_3 - r_1)(r_4 - r_2)} = k_{1r}^2,
$$
 (68)

$$
k_2^2 = -\frac{(r_4 - r_3)(r_5 - r_1)}{(r_3 - r_1)(r_5 - r_4)} =: k_{2r}^2,
$$
 (69)

$$
k_3^2 = -\frac{(r_4 - r_3)(r_1 - r_0)}{(r_3 - r_1)(r_4 - r_0)} =: k_{3r}^2,
$$
 (70)

and $\vec{\beta}_1 = (\frac{1}{2}, \frac{1}{2}, \frac{1}{2}), \vec{m} = (k_{1r}^2, k_{2r}^2, k_{3r}^2) =: \vec{m}_{0r}$. For the case of $P_6(x) = \Theta_{\xi}(\xi)$ [see Eq. ([90](#page-8-0))], there are two real zeros $\xi_2 < \xi_3$ and four other maybe complex zeros ξ_0 , ξ_1 , ξ_4 , ξ_5 , where ξ_1 , ξ_4 tend to infinity for vanishing a and ξ_0 , ξ_5 for vanishing a or Λ . With $x_1 = \xi_2$ and $x_2 = \xi_3$ we get the same form [\(65\)](#page-6-2) but with $B = 1/(\xi_3 - \xi_1) =: B_{\xi}$,

$$
A = \frac{(\xi_3 - \xi_2)}{(\xi_2 - \xi_1)(\xi_3 - \xi_1)} =: A_{\xi},
$$
\n(71)

$$
D = a^4 \Lambda \frac{(\xi_2 - \xi_1)(\xi_4 - \xi_3)}{(\xi_3 - \xi_1)^2} (\xi_3 - \xi_0)(\xi_3 - \xi_5) =: D_{\xi}, \tag{72}
$$

$$
k_1^2 = -\frac{(\xi_3 - \xi_2)(\xi_4 - \xi_1)}{(\xi_2 - \xi_1)(\xi_4 - \xi_3)} =: k_{1\xi}^2, \tag{73}
$$

$$
k_2^2 = -\frac{(\xi_3 - \xi_2)(\xi_5 - \xi_1)}{(\xi_2 - \xi_1)(\xi_5 - \xi_3)} =: k_{2\xi}^2, \tag{74}
$$

$$
k_3^2 = -\frac{(\xi_3 - \xi_2)(\xi_1 - \xi_0)}{(\xi_2 - \xi_1)(\xi_3 - \xi_0)} =: k_{3\xi}^2, \tag{75}
$$

and $\vec{m} = (k_{1\xi}^2, k_{2\xi}^2, k_{3\xi}^2) =: \vec{m}_{0\xi}$.

B. Linear correction to period of r

In this section we will calculate the linear corrections to the reference orbit due to the parameters a , n , and Λ . The exact formula for the r period is given by

$$
A_r = 2 \int_{r_3}^{r_4} \frac{dr}{\sqrt{R(r)}}
$$

= $\frac{2}{\sqrt{D_r}} \int_0^1 \frac{(A_r t + B_r) dt}{\sqrt{t(1 - t)(1 - k_{1r}^2 t)(1 - k_{2r}^2 t)(1 - k_{3r}^2 t)}}$
= $\frac{2A_r}{\sqrt{D_r}} \frac{\pi}{2} F_D \left(\frac{3}{2}, \vec{\beta}_1, 2, \vec{m}_{0r}\right) + \frac{2B_r}{\sqrt{D_r}} \pi F_D \left(\frac{1}{2}, \vec{\beta}_1, 1, \vec{m}_{0r}\right).$ (76)

All quantities indexed with an r are given as some combination of the zeros of $R(r)$ and, therefore, depend on the parameters a , n , and Λ . This means that for a Taylor expansion of Λ_r , we need to explicitly know how the zeros r_i of $R(r)$ depend on the parameters. To see this, we compare the coefficients of the equation

$$
R(r) = \Lambda \left(r - \frac{\tilde{r}_0}{\sqrt{\Lambda}} \right) (r - r_1)(r - r_2)(r - r_3)
$$

$$
\times (r - r_4) \left(r - \frac{\tilde{r}_5}{\sqrt{\Lambda}} \right), \tag{77}
$$

where $\tilde{r}_0 = r_0 \sqrt{\Lambda}$, $\tilde{r}_5 = r_5 \sqrt{\Lambda}$ do not have a singularity at $\Lambda = 0$. As the coefficient of r^5 vanishes in $R(r)$ we immediately see that

$$
-(r_2 + r_3 + r_4 + r_1)\Lambda - (\tilde{r}_0 + \tilde{r}_5)\sqrt{\Lambda} = 0 \qquad (78)
$$

and, thus, \tilde{r}_0 and \tilde{r}_5 have to be expanded in terms of $\sqrt{\Lambda}$. Therefore, we introduce $l = \sqrt{\Lambda}$ as an expansion parameter instead of Λ . Expanding the right-hand side of ([77](#page-7-0)) in terms of a , n , and l gives a system of equations which can be solved for the derivatives of all r_i with respect to the parameters.

The Taylor expansion of Λ_r in $\vec{p} = (a, n, l)$ near $\vec{p} = \vec{0}$ then reads

$$
A_r \approx A_r(\vec{p} = \vec{0}) + \sum_{i=1}^{3} A_{r,i}(\vec{p} = \vec{0})p_i
$$
\n
$$
= A_{r,0} + 2\sum_{i=1}^{3} \left\{ \frac{-D_{r,i}}{2D_r^{(3/2)}} \left[A_r \frac{\pi}{2} F_D \left(\frac{3}{2}, \vec{\beta}_1, 2, \vec{m}_{0r} \right) + B_r \pi F_D \left(\frac{1}{2}, \vec{\beta}_1, 1, \vec{m}_{0r} \right) \right]
$$
\n
$$
+ \frac{1}{\sqrt{D_r}} \left[A_{r,i} \frac{\pi}{2} F_D \left(\frac{3}{2}, \vec{\beta}_1, 2, \vec{m}_{0r} \right) + B_{r,i} \pi F_D \left(\frac{1}{2}, \vec{\beta}_1, 1, \vec{m}_{0r} \right) \right] + \frac{1}{\sqrt{D_r}} \sum_{j=1}^{3} \frac{(k_{jr}^2)_{,i}}{2k_{jr}^2} \left[A_r \frac{\pi}{2} \left(F_D \left(\frac{3}{2}, \vec{\beta}_1, 2, \vec{m}_{0r} \right) - F_D \left(\frac{3}{2}, \vec{\beta}_1, 2, \vec{m}_{0r} \right) \right) \right]
$$
\n
$$
- F_D \left(\frac{3}{2}, \vec{\beta}_1, 2, \vec{m}_{0r} \right) + B_r \pi \left(F_D \left(\frac{1}{2}, \vec{\beta}_1, 1, \vec{m}_{0r} \right) - F_D \left(\frac{1}{2}, \vec{\beta}_1, 1, \vec{m}_{0r} \right) \right) \right] \bigg\} \bigg|_{\vec{p} = \vec{0}} p_i,
$$
\n(80)

where we used $\frac{\partial}{\partial x_i}F_D(\alpha, \vec{\beta}, \gamma, \vec{x}) = \frac{\beta_i}{x_i}(F_D(\alpha, \vec{\beta}^i, \gamma, \vec{x}) - F_D(\alpha, \vec{\beta}, \gamma, \vec{x}))$ which can be found in [\[32\]](#page-17-2). The symbol $\vec{\beta}^j$ means that the jth component of $\vec{\beta}$ is increased by one [e.g. $\vec{\beta}_1^2 = (\frac{1}{2}, \frac{3}{2}, \frac{1}{2})$], and by X_i we denote the derivative of X with respect to p_i . Note that

$$
k_{2r}^2 = \frac{r_{03} - r_{04}}{r_{03}} = k_{3r}^2 \quad \text{for } \vec{p} = \vec{0},\tag{81}
$$

and, therefore, the Lauricella function F_D reduces to an elliptic integral in this case. Here r_{0i} are again the zeros of R_0 , i.e. the turning points of the reference orbit.

For the linear correction with respect to a we obtain

$$
\Lambda_{r,a}(\vec{0}) = -4EL \frac{\zeta_1 (1 - E^2)^2 (r_{03} - r_{02}) K(k) + 2r_{03} (4 - 3C + 3CE^2) E(k)}{(1 - E^2)^3 \sqrt{r_{03} (r_{04} - r_{02}) (1 - E^2)} (r_{04} - r_{02}) (r_{03} - r_{02})^2 (r_{04} - r_{03})^2},
$$
\n(82)

where $k^2 = \frac{r_{02}(r_{04}-r_{03})}{r_{03}(r_{04}-r_{02})}$ as for the reference orbit and ζ_1 is a nonsymmetric function of r_{02} , r_{03} , and r_{04} ,

$$
\zeta_1 = r_{02}r_{03} + r_{02}r_{04} - r_{03}^2 - r_{04}^2. \tag{83}
$$

The linear corrections with respect to n vanishes,

$$
A_{r,n}(\vec{0}) = 0 \tag{84}
$$

and also the linear correction due to $l = \sqrt{\Lambda}$, as expected. To determine the linear correction due to Λ we use

$$
\frac{df}{d\Lambda}(a, n, \Lambda)|_{\Lambda=0} = \left[\frac{df}{dl}(a, n, l)\frac{1}{2l}\right]_{l=0}.
$$
\n(85)

In this way the linear correction due to Λ can be calculated to

$$
A_{r,\Lambda}(\vec{0}) = \frac{1}{\sqrt{r_{03}(r_{04} - r_{02})(1 - E^2)} \left[\frac{r_{03}r_{04}(\gamma_1 - 2C\gamma_2)}{(r_{02} - r_{04})(r_{03} - r_{04})^2 (r_{02} - r_{03})} K(k) - \frac{4r_{03}C(-16C + 76CE^2 - 57CE^4 - C^2 + C^2E^4 - 48)}{(r_{02} - r_{04})(r_{02} - r_{03})^2 (r_{03} - r_{04})^2 (1 - E^2)^5} E(k) + \frac{6r_{04}}{(1 - E^2)^2} \Pi(n_1, k) \right]},
$$
(86)

where γ_1 and γ_2 are nonsymmetric functions of r_{02} , r_{03} , and r_{04} ,

$$
\gamma_1 = 2r_{03}^2r_{04}^2 - 3r_{03}^3r_{04} - 3r_{04}^3r_{03} + 3r_{03}^3r_{02} + 3r_{04}^3r_{02} - r_{03}^2r_{02}^2 + 2r_{03}r_{04}r_{02}^2 - r_{04}^2r_{03}r_{02} - r_{04}^2r_{02}^2 - r_{03}^2r_{04}r_{02},
$$

$$
\gamma_2 = r_{04}r_{02} + r_{03}r_{02} - 2r_{04}r_{03}.
$$
 (87)

C. Linear correction to period of θ

The linear correction to the θ period can be determined analogously to the foregoing subsection. The exact formula for the period of the θ -motion is

$$
A_{\theta} = \frac{2A_{\xi}}{\sqrt{D_{\xi}}} \frac{\pi}{2} F_D \left(\frac{3}{2}, \vec{\beta}_1, 2, \vec{m}_{0\xi} \right) + \frac{2B_{\xi}}{\sqrt{D_{\xi}}} \pi F_D \left(\frac{1}{2}, \vec{\beta}_1, 1, \vec{m}_{0\xi} \right).
$$
\n(88)

Again, we have to determine the dependence of the zeros ξ_i , $i = 0, ..., 5$ on the parameters a, n, and Λ . This time we use the ansatz

$$
\Theta_{\xi}(\xi) = a^4 \Lambda (\xi - \xi_2)(\xi - \xi_3) \left(\xi - \frac{\tilde{\xi}_1}{a}\right) \left(\xi - \frac{\tilde{\xi}_4}{a}\right)
$$

$$
\times \left(\xi - \frac{\tilde{\xi}_0}{a\sqrt{\Lambda}}\right) \left(\xi - \frac{\tilde{\xi}_5}{a\sqrt{\Lambda}}\right),\tag{89}
$$

where $\Theta_{\xi}(\xi)$ is the right-hand side of [\(5](#page-1-3)) with the substitution $\dot{\xi} = \cos \theta$,

$$
\left(\frac{d\xi}{d\lambda}\right)^2 = \Theta_{\xi}(\xi)
$$

= $(1 + a^2\Lambda\xi^2 - 4\Lambda a n\xi)(C - (n - a\xi)^2)(1 - \xi^2)$
 $-(L - (a(1 - \xi^2) + 2n\xi)E)^2,$ (90)

and $\tilde{\xi}_1 = a\xi_1$, $\tilde{\xi}_4 = a\xi_4$, $\tilde{\xi}_0 = a\sqrt{\Lambda}\xi_0$, $\tilde{\xi}_5 = a\sqrt{\Lambda}\xi_5$ behave regularly in the limit $a = 0$, $\Lambda = 0$. By comparing the coefficient of ξ^5 it can be seen that ξ_0 and ξ_5 expand in terms of $l = \sqrt{\Lambda}$ like r_0 and r_5 in the foregoing section. Solving the system of equations given by [\(89\)](#page-8-1), we obtain $\tilde{\xi}_0 \approx -i\sqrt{1 - E^2}$, $\tilde{\xi}_5 \approx i\sqrt{1 - E^2}$ and, therefore, $\tilde{\xi}_0$ and $\tilde{\xi}_5$ are complex conjugate.

For the Taylor expansion of Λ_{θ} we obtain the same formula [\(80\)](#page-7-1) with r replaced by ξ . However, in this case we get

$$
k_{1\xi}^2 = k_{2\xi}^2 = k_{3\xi}^2 = 0 \quad \text{for } \vec{p} = 0,
$$
 (91)

which means that the Lauricella function F_D reduces to an elementary function in the limit $\vec{p} = 0$. For the linear correction with respect to a we get

$$
\Lambda_{\theta,a}(\vec{0}) = -\frac{2\pi LE}{C^{(3/2)}}.
$$
\n(92)

The corrections with respect to *n* and *l* vanish, $\Lambda_{\theta,n}(\vec{0}) =$ $0 = \Lambda_{\theta,l}(\vec{0})$, as well as the correction for Λ due to $\Lambda_{\theta,l}(\vec{0}) = \mathcal{O}(l^2).$

D. Linear correction to frequency of φ -motion

In this section we will calculate the Taylor expansions of the integrals $I_{\varphi r}$ and $I_{\varphi \theta}$ defined in [\(58\)](#page-6-3),

$$
I_{\varphi r}(\vec{p}) = 2 \int_{r_p}^{r_a} \frac{\Phi_r(r) dr}{\sqrt{R(r)}}, \qquad I_{\varphi\theta}(\vec{p}) = 2 \int_{\theta_{\min}}^{\theta_{\max}} \frac{\Phi_\theta(\theta) d\theta}{\sqrt{\Theta(\theta)}}.
$$

The conversion of these two integrals to the standard form is more involved than in the case of Λ_r and Λ_θ because an additional function Φ_r or Φ_θ appears in the integrand here. Let us first consider $\Phi_r(r) = a \frac{P(r)}{\Delta_r}$. The poles of Φ_r are given by the horizons $\Delta_r = 0$, which we denote by h_i , $1 \le$ $i \leq 4$. Let $h_2 \approx 0$ and $h_3 \approx 2$ correspond to the Schwarzschild case, whereas $h_1 \approx -l^{-1} - 1$ and $h_4 \approx$ $l^{-1} - 1$ tend to infinity for vanishing Λ . Then Φ_r can be rewritten as

$$
\Phi_r(r) = \frac{aP(r)}{l^2(r - \frac{\tilde{h}_1}{l})(r - h_2)(r - h_3)(r - \frac{\tilde{h}_4}{l})},\tag{93}
$$

where $\tilde{h}_1 = lh_1$, $\tilde{h}_4 = lh_4$. Now Φ_r can be decomposed in partial fractions and the resulting integrals transformed to the standard form by the substitution outlined in Sec. [VA](#page-6-4). This yields an integral of the form

$$
I_{\varphi r}(\vec{p}) = 2 \sum_{i=0}^{4} \frac{c_{i\varphi r}}{\sqrt{D_r}} \int_0^1 \frac{(A_r t + B_r) dt}{(1 - N_{i\varphi r} t) \sqrt{t(1 - t)(1 - k_{1r}^2 t)(1 - k_{2r}^2 t)(1 - k_{3r}^2 t)}}
$$
(94)

$$
=2\sum_{i=0}^{4}\frac{c_{i\varphi r}}{\sqrt{D_r}}\bigg[A_r\frac{\pi}{2}F_D\bigg(\frac{3}{2},\vec{\beta}_2,2,\vec{m}_{i\varphi r}\bigg)+B_r\pi F_D\bigg(\frac{1}{2},\vec{\beta}_2,1,\vec{m}_{i\varphi r}\bigg)\bigg],\tag{95}
$$

where $c_{i\varphi r}$ and $N_{i\varphi r}$ are some constants with $N_{0\varphi r} = 0$, $\vec{\beta}_2 = (\frac{1}{2}, \frac{1}{2}, \frac{1}{2}, 1)$, and $\vec{m}_{i\varphi r} = (k_{1r}^2, k_{2r}^2, k_{3r}^2, N_{i\varphi r})$. The Taylor expansion of this can be done analogously to Sec. [V B](#page-7-2) and gives

$$
I_{\varphi r}(\vec{p}) \approx \frac{4Er_{04}\Pi(n_2, k)}{(r_{04} - 2)\sqrt{r_{03}(r_{04} - r_{02})(1 - E^2)}}a
$$
\n(96)

as the other corrections with respect to n, l, and Λ vanish. Here r_{0i} , k, and n_2 correspond to the reference orbit; see Sec. [IV](#page-3-0) for a definition.

Now let us consider $I_{\varphi\theta}$. With the substitution $\xi = \cos\theta$, the integral can be transformed to

$$
I_{\varphi\theta}(\vec{p}) = 2 \int_{\xi_2}^{\xi_3} \frac{(L - (a(1 - \xi^2) + 2n\xi)E)d\xi}{(\xi - 1)(\xi + 1)(\xi - \frac{1}{l}(2nl + f(n, l)))(\xi - \frac{1}{l}(2nl - f(n, l)))\sqrt{\Theta_{\xi}(\xi)}},
$$
(97)

where $f(n, l) = \sqrt{4l^2n^2 - 1}$. Then the standard form can be obtained with the substitution described in Sec. [VA](#page-6-4) giving the integral [\(95\)](#page-9-0) with r replaced by ξ . Here again $c_{i\varphi\xi}$ and $N_{i\varphi\xi}$ are constants with $N_{0\varphi\xi} = 0$. The Taylor expansion is given by

with vanishing corrections due to n , l , and Λ .

$$
I_{\varphi\theta}(\vec{p}) \approx 2\pi \,\text{sgn}L - \frac{2\pi E}{\sqrt{C}}a,\tag{98}
$$

E. Linear correction to frequency of t-motion

The two integrals to be considered for determining the correction to the t-motion are

$$
I_{tr}(\vec{p}) = 2 \int_{r_p}^{r_a} \frac{T_r(r) dr}{\sqrt{R(r)}}, \qquad I_{t\theta}(\vec{p}) = 2 \int_{\theta_{\min}}^{\theta_{\max}} \frac{T_{\theta}(\theta) d\theta}{\sqrt{\Theta(\theta)}}.
$$

The procedure will be analogous to the foregoing subsection. The poles of T_r and T_θ are the same as of Φ_r and Φ_θ respectively and, therefore, the integrals can be expressed similar to ([95](#page-9-0)). We get

$$
I_{tr}(\vec{p}) = 2 \sum_{i=0}^{4} c_{itr} \bigg[A_r \frac{\pi}{2} F_D \bigg(\frac{3}{2}, \vec{\beta}_2, 2, \vec{m}_{itr} \bigg) + B_r \pi F_D \bigg(\frac{1}{2}, \vec{\beta}_2, 1, \vec{m}_{itr} \bigg) \bigg] \approx I_{tr}(\vec{0}) + I_{tr,a}(\vec{0})a + I_{tr,n}(\vec{0})n + I_{tr,\Lambda}(\vec{0})\Lambda,
$$
\n(99)

where c_{itr} and N_{itr} are constants with $N_{0tr} = 0$, $\vec{\beta}_2 = (\frac{1}{2}, \frac{1}{2}, \frac{1}{2}, 1)$, $\vec{m}_{itr} = (k_{1r}^2, k_{2r}^2, k_{3r}^2, N_{itr})$, and $I_{tr}(\vec{0}) = I_{tr,0}$ as for the reference orbit. The linear correction due to a is given by

$$
I_{tr,a}(\vec{0}) = \frac{4Lr_{03}^2}{(r_{03}(r_{04} - r_{02})(1 - E^2))^{(3/2)}(r_{04} - r_{03})^2(r_{03} - r_{02})} \left[\frac{r_{04}\eta_1 E^2 K(k)}{(r_{04} - 2)(r_{03} - 2)} + \frac{2C(8 + 2C - 3CE^2)}{(r_{03} - r_{02})(1 - E^2)^2}E(k)\right],
$$
(100)

where η_1 is a nonsymmetric function of r_{02} , r_{03} , and r_{04} ,

$$
\eta_1 = 2r_{04}^2 r_{02} - r_{03}r_{04}^2 r_{02} - 2r_{04}^2 r_{03} + 2r_{03}^2 r_{04}^2 - 2r_{03}^2 r_{04} - r_{03}^2 r_{04} r_{02} + 2r_{03}^2 r_{02}.
$$
\n(101)

The correction due to *n* vanishes, $I_{tr,n}(\vec{0}) = 0$, and the correction due to Λ is given by

$$
I_{tr,\Lambda}(\vec{0}) = \frac{E}{\sqrt{r_{03}(r_{04} - r_{02})(1 - E^2)}} \Big\{ -r_{03}r_{04}K(k) \Big[\Big(\frac{(\alpha_2 + 12C\alpha_3)}{24(1 - E^2)(r_{04} - r_{03})^2(r_{03} - r_{02})(r_{04} - r_{02})} + \frac{\alpha_1}{12} \Big) (r_{04} - 2)^{-1} (r_{03} - 2)^{-1} \Big] \Big\} + \frac{2}{(1 - E^2)^3} \Big] + \frac{r_{03}(r_{04} - r_{02})}{(1 - E^2)} E(k) \Big[\frac{(\alpha_4 + 12C\alpha_5)}{24S} - \frac{8 - 75E^2 + 149E^4 - 116E^6 + 32E^8}{E^2(1 - E^2)^2} - \frac{4C}{3} \Big] \Big\} + \frac{r_{04}}{1 - E^2} \Pi(n_1, k) \Big[\frac{238 - 754E^2 + 952E^4 - 560E^6 + 2^7E^8}{(1 - E^2)^3} + \frac{2C(2E^2 - 5)}{1 - E^2} + \frac{\alpha_6 + 4C\alpha_7}{8(r_{04} - r_{03})^2(r_{03} - r_{02})(r_{04} - r_{02})} \Big] \Big] + \frac{r_{02} + r_{03} - 3r_{04}}{(1 - E^2)^2} \Big[r_{03}r_{04}K(k) - r_{03}(r_{04} - r_{02})E(k) - \frac{2r_{04}(3 - 2E^2)}{1 - E^2} \Pi(n_1, k) \Big] + \frac{256r_{04}}{r_{04} - 2} \Pi(n_2, k) \Big\}, \tag{102}
$$

where $S = (r_{04} - r_{03})^2 (r_{03} - r_{02})^2 (r_{04} - r_{02})^2 (r_{04} - 2)$ × $(r_{03} - 2)(r_{02} - 2)$ and α_i , $i = 1, ..., 7$, are nonsymmetric functions of r_{02} , r_{03} , and r_{04} given in Appendix [B.](#page-13-1)

The Taylor expansion for the θ -dependent part is much easier. The zero-order term and all linear corrections except $I_{t\theta,a}$ vanish, $I_{t\theta}(\vec{0}) = I_{t\theta,n}(\vec{0}) = I_{t\theta,\Lambda}(\vec{0}) = 0$. The remaining term is given by

$$
I_{t\theta,a}(\vec{0}) = \frac{2\pi L}{\sqrt{C}}.\tag{103}
$$

VI. OBSERVABLES

In Sec. [III](#page-2-0) we defined the general expressions for the fundamental frequencies Ω_r , Ω_θ , and Ω_φ in Plebanski-Demiański space-time. The analytic expressions of these quantities are given in terms of complete hyperelliptic integrals on a Riemann surface of genus 2 [see Eqs. [\(76\)](#page-7-3), [\(88\)](#page-8-2), ([95](#page-9-0)), and ([99\)](#page-9-1)], whose numerical evaluation is quite cumbersome. By a post-Schwarzschild expansion of the fundamental frequencies we will reduce the hyperelliptic integrals to elementary expressions and elliptic integrals, which can be handled easily with computer algebra systems like MATHEMATICA or MAPLE. In addition, we will perform the post-Newtonian expression for comparison with other results and use the parameter values of Mercury to get an idea of the order of magnitude of the corrections.

A. Periastron shift

In Sec. [IV](#page-3-0) we calculated the expression for $\Omega_{\rm P}$ for the Schwarzschild case [see Eq. [\(47\)](#page-5-1)] and noted that as long as a particle is considered to move in an orbital plane usually the difference angle $\Delta_{\rm P}$ [see ([48](#page-5-2))] is used instead of $\Omega_{\rm P}$. Let us now consider the linear corrections of $\Omega_{\rm P}$ and $\Delta_{\rm P}$ due to a, n, and Λ . As all linear corrections due to n so far vanished, the linear correction due to this parameter also vanishes for $\Omega_{\rm P}$ and $\Delta_{\rm P}$. The linear correction due to a for the frequency $\Omega_{\rm P}$ is given by

$$
\Omega_{P,a}(\vec{0}) = \frac{I_{\varphi r,a}(\vec{0}) + \text{sgn}L\sqrt{C}\Lambda_{r,a}(\vec{0}) - E\Lambda_{r,0}(1 - \frac{|L|}{\sqrt{C}})}{I_{rr,0}} + \text{sgn}L\frac{(L\Lambda_{r,0} + I_{rr,a}(\vec{0}))(2\pi - \sqrt{C}\Lambda_{r,0})}{I_{rr,0}^2}
$$
\n
$$
= \frac{2}{Z} \bigg[\frac{r_{04}}{r_{04} - 2} \Pi(n_2, k) + \sqrt{C}|L|\frac{\zeta_1(r_{02} - r_{03})(E^2 - 1)^2 K(k) - 2r_{03}(4 - 3C(E^2 - 1))E(k)}{(r_{02} - r_{03})^2(r_{02} - r_{04})(r_{03} - r_{04})^2(E^2 - 1)^3} - K(k)\bigg(1 - \frac{|L|}{\sqrt{C}}\bigg) \bigg]
$$
\n
$$
+ \frac{|L|}{E^2 Z^2} \bigg[K(k) + \bigg(\frac{r_{03}r_{04}\eta_1 E^2 K(k)}{(r_{04} - 2)(r_{03} - 2)} + r_{03} \frac{2C(8 + 2C - 3CE^2)E(k)}{(r_{03} - r_{02})(1 - E^2)^2} \bigg) ((1 - E^2)(r_{04} - r_{02})(r_{04} - r_{03})^2(r_{03} - r_{02}))^{-1} \bigg]
$$
\n
$$
\times \bigg[2\pi \sqrt{r_{03}(r_{02} - r_{04})(E^2 - 1)} - 4\sqrt{C}K(k) \bigg], \tag{104}
$$

where $Z = -r_{03}r_{04}K(k) + r_{03}(r_{04} - r_{02})E(k) + 2r_{04}\frac{3-2E^2}{1-E^2}\Pi(n_1, k) + \frac{8r_{04}}{r_{04}-2}\Pi(n_2, k)$ and ζ_1 , η_1 are nonsymmetric functions of r_{02} , r_{03} , and r_{04} defined in ([83](#page-7-4)) and ([101\)](#page-9-2). Here r_{0i} , k, and n_i correspond to the reference orbit and are defined in Sec. [IV.](#page-3-0) Note that $\Omega_{P,q}$ does not depend on the sign of L, i.e. on whether the particle travels on a prograde or retrograde orbit.

The correction of
$$
\Delta_p
$$
 due to the angular momentum of the gravitating source reads
\n
$$
\Delta_{P,a}(\vec{0}) = \frac{4E\sqrt{C}}{\sqrt{(1 - E^2)r_{03}(r_{04} - r_{02})} \left\{ \left[\frac{|L|\zeta_1}{(r_{04} - r_{03})^2(r_{03} - r_{02})(r_{04} - r_{02})(E^2 - 1)} - \frac{1}{\sqrt{C}} + \frac{|L|}{C} \right] K(k) - \frac{2|L|r_{03}(4 - 3C - 3CE^2)}{(1 - E^2)^3(r_{04} - r_{03})^2(r_{03} - r_{02})^2(r_{04} - r_{02})} E(k) + \frac{r_{04}}{r_{04} - 2} \Pi(n_2, k),
$$
\n(105)

which also does not depend on the sign of L . These formulas are exact in M and, therefore, quite complicated. If we consider the post-Newtonian approximation of these terms by using [\(27\)](#page-3-1)–([30](#page-3-3)) and assuming that r_{03} and r_{04} , the turning points of the reference orbit in Schwarzschild space-time, become large we obtain

$$
\Omega_{P,a}(\vec{0})a = 2\frac{1 - 6\sin\theta_{01}}{d^3(1 - \epsilon^2)^{(3/2)}}JM^2 + \mathcal{O}(JM^3), \quad (106)
$$

$$
\Delta_{P,a}(\vec{0})a = 4\pi \frac{1 - 6\sin\theta_{01}}{d^{(3/2)}(1 - \epsilon^2)^{(3/2)}} J M^{(1/2)} + \mathcal{O}(J M^{(3/2)}),
$$
\n(107)

where d is again the semimajor axis, ϵ the eccentricity, and $J = aM$ is the non-normalized specific angular momentum of the gravitating source. These terms are similar to the expression derived by Lense and Thirring [[4\]](#page-16-3) for the precession of the longitude of pericenter, to which it differs only in the second term containing the inclination θ_{01} . However, the fundamental frequency $\Omega_{\rm P}$ is not identical to the precession of longitude of the pericenter: As Ω_r does not contribute to the first order correction of $\Omega_{\rm P}$ the value of $\Omega_{P,a}$ is calculated by the correction due to $\Omega_{\varphi} = \frac{Y_{\varphi}}{\Gamma}$ only, which is the averaged $\frac{d\varphi}{dt}$. The longitude of pericenter on the other hand contains the argument of the pericenter which is defined not on the reference plane but the orbital plane.

Note that the expressions for $\Omega_{\rm P}$ and $\Delta_{\rm P}$ depend on the inclination of the orbital plane. For an inclination of θ_{01} < $\arcsin\frac{1}{6} \approx 0.053\pi$ the periastron shift is perturbed in the direction of the rotation of the gravitating source whereas for $\theta_{01} > \arcsin\frac{1}{6}$ the perturbation is in the opposite direction. (Here an inclination of $\theta_{01} = \frac{\pi}{2}$ corresponds to the equatorial plane and $\theta_{01} \leq \frac{\pi}{2}$ by definition; see Sec. [IV.](#page-3-0)) In particular, if the unperturbed test particle moves on the equatorial plane, the periastron shift is perturbed against the direction of rotation. This seems to be counterintuitive as a particle radially approaching the gravitational source is dragged along the direction of the rotation. However, this does not mean that the shape of the orbit is affected in the same way: The rotation of the gravitating source acts as an repulsive force which also causes the peri- and apoapsis to increase; see [\(4](#page-1-1)).

Let us now consider the linear correction of $\Omega_{\rm P}$ and $\Delta_{\rm P}$ due to Λ . For $\Omega_{\rm P}$ we get a very complicated expression given in Appendix [B](#page-13-1). In terms of the expression derived in the previous section we get

$$
\Omega_{\text{P},\Lambda} = \text{sgn} L \sqrt{C} \bigg[\frac{\Lambda_{r,\Lambda}}{I_{tr,0}} + \bigg(\frac{2\pi}{\sqrt{C}} - \Lambda_{r,0} \bigg) \frac{I_{tr,\Lambda}}{I_{tr,0}^2} \bigg]. \tag{108}
$$

The correction of Δ_P due to Λ is much simpler,

$$
\Delta_{P,\Lambda} = \frac{\text{sgn}L\sqrt{C}}{\sqrt{(1 - E^2)r_{03}(r_{04} - r_{02})(1 - E^2)}}\times \left\{ \frac{2C(r_{02}r_{03} + r_{02}r_{04} - 2r_{03}r_{04}) - \gamma_1}{(r_{04} - r_{03})^2(r_{04} - r_{02})(r_{03} - r_{02})}r_{03}r_{04}K(k) + \frac{48 + 16C - 76CE^2 + 57CE^4 + C^2 - C^2E^4}{16 + 8C - 36CE^2 + 27CE^4 + C^2 - C^2E^2} \times r_{03}(r_{04} - r_{02})E(k) + \frac{6r_{04}}{1 - E^2}\Pi(n_1, k)\right\}.
$$
 (109)

These complicated expressions can be simplified by considering the post-Newtonian approximation, which reads

$$
\Omega_{P,\Lambda} \Lambda = \frac{1}{2} \operatorname{sgn} L \, d^{(3/2)} \sqrt{1 - \epsilon^2} \hat{\Lambda} M^{(1/2)} + \mathcal{O}(\hat{\Lambda} M^{(3/2)}),
$$
\n(110)

$$
\Delta_{P,\Lambda}\Lambda = \pi \operatorname{sgn} L \, d^3 \sqrt{1 - \epsilon^2} \frac{\hat{\Lambda}}{M} + 2\pi \operatorname{sgn} L \, d^2 \frac{2 - \epsilon^2}{\sqrt{1 - \epsilon^2}} \hat{\Lambda}
$$

+ $\mathcal{O}(\hat{\Lambda}M),$ (111)

where $\hat{\Lambda} = \frac{3\Lambda}{M^2}$ is the usual non-normalized cosmological constant and d , ϵ are the semimajor axis and eccentricity. The first term of Eq. ([111](#page-11-0)) coincides with the result in [[33\]](#page-17-3). Interestingly, although we assumed both $\hat{\Lambda}$ and M to be small we assumed nothing about their ratio. This raises the question whether the first term of (111) is indeed the largest of the expansion. But if we divide the second term by the first, this is proportional to $\frac{M}{d}$, which is assumed to be small. This proves that the terms of the above expansions in fact decrease in magnitude for increasing powers of M.

The exact formula for the perihelion shift assuming a nonvanishing cosmological constant was derived in [\[12](#page-16-11)[,34\]](#page-17-4). We will use the formalism of $[12]$ $[12]$ $[12]$ [see Eq. ([70](#page-6-5))] to compute the perihelion shift of Mercury in Schwarzschild–de Sitter space-times and compare it with our approximate formula. For this, we use $M =$ 1476:62528 m and, by using ([27](#page-3-1))–[\(30\)](#page-3-3), we determine averaged values for the energy and angular momentum from the apo- and periapsis data given by [[30](#page-17-0)] and written down in Sec. [IV E.](#page-4-1) With the exact formulas given in [\[12\]](#page-16-11) the difference between the perihelion shift for $\Lambda = 0$ and with a cosmological constant of $\hat{\Lambda} = 3\Lambda/M^2 =$ 3×10^{-52} m⁻² is given by

$$
\Delta_{P, (\Lambda = 10^{-52})} / y_M - \Delta_{P, (\Lambda = 0)} / y_M
$$

= 1.038 833 075 425 928 × 10⁻¹⁴ $\frac{\text{arc sec}}{\text{cy}}$, (112)

where y_M is the Mercury year as calculated in [\(56](#page-5-4)). If we insert the same values in our formula [\(106\)](#page-11-1) we get

$$
\Delta_{P}/y_{M} - \Delta_{P,0}/y_{M}
$$

= 1.038832108177831 × 10⁻¹⁴ $\frac{\text{arc sec}}{\text{cy}}$ + $\mathcal{O}(\Lambda^{2})$, (113)

which agrees very well with [\(112\)](#page-11-2).

B. Lense-Thirring effect

The Lense-Thirring effect, which can be identified with a precession of the orbital plane in the weak field limit, is up to first order not influenced by any other parameter than the rotation a. This is because all linear corrections to Y_{θ}

and Y_{φ} due to *n* and Λ vanish. The linear correction due to a is given by

$$
\Omega_{LT,a} = \frac{1}{Z} \left[\frac{2r_{04}}{(r_{04} - 2)} \Pi(n_2, k) - 2K(k) \right],\tag{114}
$$

where $Z = -r_{03}r_{04}K(k) + r_{03}(r_{04} - r_{02})E(k) + 2r_{04}\frac{3-2E^2}{1-E^2} \times$ $\Pi(n_1, k) + \frac{8r_{04}}{r_{04}-2} \Pi(n_2, k)$ and r_{0i} , k, and n_i correspond to the reference orbit; see Sec. [IV.](#page-3-0)

Let us consider now the weak field limit by assuming that r_{03} and r_{04} become large. Using ([27\)](#page-3-1)–[\(30\)](#page-3-3), we get

$$
\Omega_{LT,a}a = \frac{2}{d^3(1 - \epsilon^2)^{(3/2)}} J M^2 + \mathcal{O}(J M^3),\tag{115}
$$

where $J = aM$ is the non-normalized angular momentum of the gravitating source per unit mass, d is the semimajor axis, and ϵ the eccentricity. This formula is identical to the precession rate of the longitude of the ascending node as given by Lense and Thirring; cf. [[4](#page-16-3)].

C. Conicity

The effect of n can be observed by a deviation from the symmetry of the geodesic motion with respect to equatorial plane. In Sec. [III](#page-2-0) we defined $\Delta_{\text{conicity}} = \pi - (\theta_{\text{min}} + \theta_{\text{max}})$ as a measure for this deviation, which is given by

$$
\Delta_{\text{conicity}} \approx \text{sgn}L \frac{4E}{\sqrt{C}} n \tag{116}
$$

up to first order. Linear corrections due to a and Λ vanish. This deviation from the symmetry to the equatorial plane also implies that for $n \neq 0$ a test particle cannot move on an orbital plane and, in particular, not in the equatorial plane (if $E \neq 0$). Instead it moves on a cone with opening angle $\pi - \frac{4E}{\sqrt{C}}n + \mathcal{O}(n^2)$. As θ_{01} and θ_{02} are perturbed by the same value $\alpha = -\text{sgn}L\frac{2E}{\sqrt{C}}n$ (up to first order) the symmetry axis of this cone coincides with one of the two normals of the orbital plane in the Schwarzschild case, i.e. $\theta_n = \theta_{02} - \frac{\pi}{2} \in (0, \frac{\pi}{2})^2$ if $\alpha < 0$ or $\theta_n = \theta_{01} + \frac{\pi}{2} \in (\frac{\pi}{2}, \pi)$ if $\alpha > 0$.

In the post-Newtonian limit, the conicity becomes

$$
\Delta_{\text{conicity}} \approx \frac{4 \text{ sgn} L}{\sqrt{d(1 - \epsilon^2)}} \frac{\hat{n}}{M^{(1/2)}} - \frac{8 \text{ sgn} L}{d^{(3/2)}(1 - \epsilon^2)^{(3/2)}} \hat{n} M^{(1/2)},\tag{117}
$$

where $\hat{n} = nM$ is the non-normalized NUT parameter. Here d and ϵ are the semimajor axis and the eccentricity, as before. Note that as we assumed $n = \frac{\hat{n}}{M}$ to be small in the first place, the first term in the above equation will indeed tend to zero for $M \rightarrow 0$ (but is still the largest of the expansion).

The conicity can be used to determine an upper bound for the value of n . As the error tolerance of the inclination of an orbiting object places a bound for Δ_{conicity} , this can be used in one of the above equations to get an estimate of n . From [\[30\]](#page-17-0) we obtain for the difference of maximal and minimal inclination of Mercury over a time span of 10 yr $|\Delta_{\text{conicity}}| \leq 4.2001$ arc sec or $|\Delta_{\text{conicity}}| \leq 2.04 \times 10^{-5}$ rad. Inserted in [\(116](#page-12-0)) this yields $|n| \le 0.032$.

VII. CONCLUSIONS AND OUTLOOK

In this paper we derived analytic expressions for the periastron shift and Lense-Thirring effect assuming neutral test particles in the general axially symmetric Plebanski-Demiański electrovac space-time with vanishing acceleration of the gravitating source. We determined the direct dependence of these observables on the parameters of the space-time by expanding them in a Taylor series up to first order. In addition, we defined the conicity of an orbit and analyzed its dependence on the space-time parameters by also expanding it in a Taylor series up to first order. Within the linear approximation we used the osculating orbital elements of Mercury to derive an upper bound for the dimensionless NUT parameter n.

From the six parameters of a Pleban^{ski}-Demian^{ski} space-time with vanishing acceleration, only the mass M, the rotation a , the NUT charge n , and the cosmological constant Λ have any linear effects on neutral test particles. For Schwarzschild space-time $(0 = a = n = \Lambda = Q_e$ Q_m), the only effect is the periastron shift, whereas the Lense-Thirring effect vanishes and the mean value of the polar coordinate coincides with the equator. In the following we summarize the effects due to the other parameters compared to Schwarzschild space-time:

- (i) Taub-NUT space-time $(0 = a = \Lambda = Q_e = Q_m)$: The mean value of the polar motion deviates from the equator and the motion takes place on a cone rather than a plane.
- (ii) Kerr space-time $(0 = n = \Lambda = Q_e = Q_m)$: The periastron shift is changed and the Lense-Thirring effect is nonzero. The latter can be interpreted as a precession of the orbital plane in the weak field approximation. Both are independent from the direction of rotation of the particle.
- (iii) Kerr-Taub-NUT space-time ($0 = \Lambda = Q_e = Q_m$): In addition to Kerr, the mean value of the polar motion deviates from the equator. The combination with a nonvanishing Lense-Thirring effect causes a precession of the orbital cone.
- (iv) Kerr-Taub-NUT–de Sitter space-time ($0 = Q_e$ = Q_m): In addition to Kerr-Taub-NUT, the periastron shift is changed by a nonvanishing Λ . However, due to the smallness of Λ the effect is tiny.

The effects of nonvanishing space-time parameters may also be analyzed compared to any other space-time covered by the general Pleban^{ski}-Demian^{ski} metric using the same methods as presented in this paper. In particular the Kerr

space-time may be used as reference. In this case $Y_{\theta,0}$ and $Y_{\varphi,0}$ are also given in terms of elliptic integrals instead of elementary expressions.

The mathematical framework presented in this paper is based on the theory of hyperelliptic and elliptic functions. We defined a standard form for hyperelliptic integrals based on the Riemann form of elliptic integrals and expressed it in terms of the hypergeometric F_D function introduced by Lauricella (see also [\[26\]](#page-16-25)). The methods presented in this paper may also be used to analyze the linear (and higher order) effects of the six parameters of Plebański-Demiański space-time with vanishing acceleration on charged particles with a complete analogous procedure. Another interesting question is whether additional effects arise for a higher order approximation, for example, the quadratic effect of n on r -dependent expressions or the coupling of a to the other parameters. Similarly to the analysis presented here, observables for unbound orbits like light deflection or the deflection of massive particles may be considered. This will be postponed to a later publication.

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APPENDIX A: LAURICELLA'S F_D FUNCTION

The four functions F_A , F_B , F_C , and F_D of Lauricella are hypergeometric functions of multiple variables generalizing the hypergeometric functions of Gauss and Appell. They were introduced in 1893 [[35](#page-17-5)] and given as a hypergeometric series,

$$
F_D(\alpha, \vec{\beta}, \gamma, \vec{x}) = \sum_{\vec{\iota}=0}^{\infty} \frac{(\alpha)_{|\vec{\iota}|} (\vec{\beta})_{\vec{\iota}}}{(\gamma)_{|\vec{\iota}|} \vec{\iota}!} \vec{x}^{\vec{\iota}}, \tag{A1}
$$

where $\vec{\iota}$ is a multi-index, $|x_{\iota_n}| \leq 1$ for all *n*, and $(\cdot)_{|\vec{\iota}|}$ the Pochhammer symbol. Here $|\vec{\iota}| = \sum_{n} \iota_n$, $\vec{\iota} = \prod_{n} \iota_n!$, and $(\vec{\beta})_{\vec{\iota}} = \prod_n (\beta_n)_{\iota_n}$. The function F_D can be extended to other values of \vec{x} by analytic continuation.

In this paper the F_D function is used because it can be represented in an integral form,

$$
F_D(\alpha, \vec{\beta}, \gamma, \vec{x}) = \frac{\Gamma(\gamma)}{\Gamma(\alpha)\Gamma(\gamma - \alpha)} \int_0^1 t^{\alpha - 1} (1 - t)^{\gamma - \alpha - 1} \times \prod_n (1 - x_n t)^{-\beta_n} dt,
$$
\n(A2)

for $Re(\gamma) > Re(\alpha) > 0$, which is exactly the form of all hyperelliptic integrals appearing in this paper.

APPENDIX B: DETAILS OF $\Omega_{\rm{RA}}$

The calculation of the linear correction of the periastron shift due to the cosmological constant yields a very cumbersome expression given by

$$
\Omega_{P,\Lambda} = \frac{2EsgnL\sqrt{C}}{(E^2-1)Z} \left[\frac{\gamma_1 - 2C(r_{02}r_{03} + r_{02}r_{04} - 2r_{03}r_{04})}{(r_{02} - r_{04})(r_{02} - r_{03})(r_{03} - r_{04})^2} r_{03}r_{04}K(k) \right. \\
\left. + \frac{-16C + 76CE^2 - 57CE^4 - C^2 + C^2E^4 - 48}{16 + 8C - 36CE^2 + 27CE^4 + C^2 - C^2E^2} r_{03}(r_{04} - r_{02})E(k) + \frac{6r_{04}}{E^2 - 1} \Pi(n_1, k) \right] \\
- \frac{sgnL}{2EZ^2} [2\sqrt{C}K(k) - \pi\sqrt{r_{03}(r_{02} - r_{04})(E^2 - 1)}] \left[\frac{r_{02} + r_{03} - 3r_{04}}{(E^2 - 1)^2} \left[r_{03}r_{04}K(k) - r_{03}(r_{04} - r_{02})E(k) \right] \right. \\
\left. - 2r_{04} \frac{3 - 2E^2}{1 - E^2} \Pi(n_1, k) \right] 3 + \left[\frac{2}{(E^2 - 1)^3} - \frac{\alpha_1}{12(r_{04} - 2)(r_{03} - 2)} \right. \\
\left. + \frac{\alpha_2 + 12C\alpha_3}{24(r_{04} - 2)(r_{03} - 2)(r_{02} - r_{04})(r_{02} - r_{03})(r_{03} - r_{04})^2(E^2 - 1)} \right] r_{03}r_{04}K(k) \\
+ \left[\frac{4C}{3(E^2 - 1)^2} - \frac{\alpha_4 + 12C\alpha_5}{24(r_{04} - 2)(r_{03} - 2)(r_{02} - 2)(r_{02} - r_{04})^2(r_{02} - r_{03})^2(r_{03} - r_{04})^2} {E^2(E^2 - 1)^4} + \frac{-75E^2 + 8 + 149E^4 - 116E^6 + 32E^8}{E^2(E^2 - 1)^4} \right] r_{03}(r_{04} - r_{02
$$

where $Z = -r_{03}r_{04}K(k) + r_{03}(r_{04} - r_{02})E(k) + 2r_{04}\frac{3-2E^2}{1-E^2}\Pi(n_1, k) + \frac{8r_{04}}{r_{04}-2}\Pi(n_2, k)$ and $\gamma_1, \alpha_1, ..., \alpha_7$ are nonsymmetric expressions of r_{02} , r_{03} , and r_{04} . These constants read

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$$
\gamma_{1} = -2r_{03}^{2}r_{04}^{2} + 3r_{03}^{3}r_{04} + 3r_{04}^{3}r_{03} - 3r_{03}^{3}r_{02} - 3r_{04}^{3}r_{02} + r_{03}^{2}r_{02} - 2r_{03}r_{04}r_{02}^{2} + r_{04}^{2}r_{03}r_{02} + r_{04}^{2}r_{02}^{2} + r_{03}^{2}r_{04}r_{02}
$$
 (B2)
\n
$$
\alpha_{1} = 1536 - 288r_{04} - 288r_{03} + 96r_{02} + 20r_{02}^{2} - 8r_{04}^{2}r_{02} + 4r_{03}^{2}r_{04}r_{02} + 4r_{04}^{2}r_{03}r_{02} + 5r_{03}r_{04}r_{02}^{2}
$$
\n
$$
+ 8r_{02}r_{03}r_{04} + 14r_{03}^{2}r_{04} + 15r_{04}^{3}r_{03} - 10r_{03}r_{02}^{2} - 10r_{04}r_{02}^{2} - 30r_{03}^{3} - 8r_{03}^{2}r_{02} + 14r_{04}^{2}r_{03}
$$
\n
$$
+ 14r_{03}^{2}r_{04} + 56r_{03}r_{04} - 32r_{02}r_{04} - 32r_{02}r_{03} - 84r_{03}^{2} + 15r_{03}^{3}r_{04} - 30r_{04}^{3} - 84r_{04}^{2},
$$
 (B3)
\n
$$
\alpha_{2} = 864r_{03}r_{04}^{2}r_{02}^{2} - 4r_{03}^{4}r_{04}r_{02}^{3} - 336r_{03}r_{04}^{3}r_{02} + 384r_{03}^{2}r_{04}r_{02} - 20r_{03}^{2}r_{04}r_{02}^{3} + 384r_{04}^{2}r_{03}r_{02} + 36r_{03}^{3}
$$

$$
\alpha_3 = 32r_{04}^2r_{02} + 2r_{03}r_{04}^2r_{02}^2 + 14r_{03}r_{04}^3r_{02} - 32r_{03}^2r_{04}r_{02} - 16r_{03}r_{04}r_{02}^2 + 4r_{03}^2r_{02}r_{04}^3 + 3r_{03}r_{02}^2r_{04}^3 - 4r_{03}^3r_{04}^2r_{02} \n+ 14r_{03}^3r_{04}r_{02} - 11r_{03}r_{02}r_{04}^4 - 3r_{03}^4r_{04}r_{02} + 2r_{03}^2r_{02}^2r_{04} + 3r_{03}^3r_{04}r_{02}^2 + 3r_{03}^4r_{04}^2 + 24r_{03}^2r_{04}^2 + 11r_{03}^2r_{04}^4 - 22r_{04}^4r_{03} \n- 18r_{03}^2r_{04}^3 - 44r_{04}^3r_{02} - 12r_{03}^3r_{02} + 6r_{03}^4r_{02} - 2r_{03}^3r_{04}^2 + 44r_{04}^3r_{03} + 32r_{03}^2r_{02} - 32r_{04}^2r_{03} - 32r_{03}^2r_{04} - 4r_{02}^2r_{03}^2r_{04}^2 \n- 6r_{03}^4r_{04} + 12r_{04}^2r_{02}^2 + 12r_{03}^2r_{02}^2 + 12r_{03}^3r_{04} + 22r_{04}^4r_{02} - 2r_{03}^3r_{04}^3 - 6r_{04}^3r_{02}^2 - 6r_{02}^2r_{03}^3, \tag{B5}
$$

$$
\alpha_4 = 204r_{03}^7r_{04}^2 - 128r_{03}^5r_{02}^4 + 348r_{02}^7r_{04}^2 + 80r_{03}^6r_{02}^4 + 32r_{02}^5r_{04}^3 + 560r_{03}^4r_{04}r_{03}^3 - 192r_{02}^5r_{04}r_{03} - 1536r_{03}^2r_{04}r_{02}^3 \n- 1344r_{03}^3r_{04}r_{02}^3 + 1344r_{02}^3r_{04}r_{03}^2 + 2400r_{03}^4r_{04}r_{02}^2 + 864r_{02}^4r_{04}r_{03}^2 - 1440r_{02}^4r_{04}^2r_{04}r_{03} + 1728r_{03}r_{02}^3r_{04}^3r_{04}^3 \n- 4800r_{03}^2r_{02}^2r_{04}^3 - 1536r_{03}^2r_{04}r_{03} - 1536r_{03}^3r_{02}r_{04}^2 - 1536r_{03}^3r_{04}^2r_{04}^2 - 1536r_{03}^3r_{04}^2r_{04} + 864r_{04}^4r_{02}r_{03}^2 - 3264r_{03}r_{04}^5r_{04}^2r_{02} \n+ 3168r_{03}r_{04}^4r_{02}^2 - 1728r_{03}^5r_{04}r_{02} - 2208r_{03}^4r_{03}r_{04}^2r_{04} - 1536r_{03}^3r_{04}^2r_{02} + 1728r_{03}^3r_{04}^3r_{02} + 1344r_{03}^3r_{04}^2r_{02}^2 \n- 1088r_{03}^5r_{02}r_{04} + 1792r_{03}^5r_{04}^2r_{02} + 1536r_{03}r_{02}r_{04}^4 + 1536r_{
$$

$$
-384r_{02}^{2}r_{03}^{2}r_{04}^{4} + 828r_{04}^{2}r_{02}^{2} - 164r_{03}^{6}r_{02}r_{04}^{2} + 64r_{02}^{5}r_{04}^{2}r_{03} - 8r_{03}^{6}r_{04}^{3}r_{02}^{2} - 114r_{03}^{6}r_{04}r_{02}^{3} + 176r_{03}^{4}r_{02}^{4} + 192r_{03}^{5}r_{02}^{2}r_{04}^{2}
$$

\n
$$
-18r_{03}^{6}r_{04}^{3}r_{02} + 20r_{03}^{6}r_{02}^{2}r_{04}^{2} - 148r_{02}^{6}r_{04}^{2}r_{03}^{2} - 544r_{03}^{4}r_{02}^{3}r_{04}^{3}r_{02}^{3} - 354r_{03}^{3}r_{04}^{6}r_{02} - 1096r_{02}^{2}r_{03}^{3}r_{04}^{4} - 784r_{02}^{5}r_{04}^{3}r_{03}^{3}
$$

\n
$$
-260r_{03}^{2}r_{04}^{6}r_{02} - 174r_{02}^{7}r_{04}^{2}r_{03}^{2} - 1896r_{02}^{2}r_{04}^{6} - 262r_{03}^{4}r_{03}^{4} + 74r_{03}^{4}r_{04}^{2}r_{02}^{2} + 102r_{03}^{7}r_{04}r_{03}^{2} + 87r_{02}^{7}r_{04}^{3}r_{03}^{3}
$$

\n
$$
+51r_{03}^{7}r_{04}^{3}r_{02} - 102r_{03}^{7}r_{04}^{2}r_{02}^{2} + 174r_{02}^{7}r_{04}^{2}r_{03} + 194r_{03}^{4}r_{02}^{2}r_{04}^{2} + 216r_{03}^{4}r_{03}^{2} + 112r_{02}^{6
$$

$$
\alpha_{5} = -18r_{02}^{5}r_{04}^{3} + 128r_{03}r_{04}^{2}r_{02}^{2} - 16r_{03}^{4}r_{02}^{3} - 128r_{03}r_{04}^{3}r_{02} - 48r_{02}^{5}r_{04}r_{03} + 24r_{03}^{2}r_{04}r_{02}^{3} + 24r_{03}^{3}r_{04}r_{02}^{3} + 32r_{02}^{3}r_{04}^{2}r_{03}^{2} + 32r_{03}^{3}r_{04}^{2}r_{03}^{2} + 24r_{03}^{4}r_{04}r_{02}^{2} - 40r_{02}^{4}r_{04}r_{03}^{2} - 8r_{02}^{4}r_{04}^{2}r_{03} + 56r_{03}r_{02}^{3}r_{04}^{3} + 16r_{03}^{2}r_{02}^{2}r_{04}^{3} + 120r_{03}^{2}r_{02}r_{04}^{3} - 40r_{02}^{3}r_{04}^{2}r_{03} - 168r_{03}r_{04}^{2}r_{02} - 16r_{03}^{3}r_{04}^{2}r_{02} - 128r_{03}^{3}r_{04}r_{02} + 24r_{03}r_{04}r_{02} - 24r_{03}^{5}r_{04}r_{02} - 24r_{03}^{4}r_{02}r_{04}^{2} + 120r_{03}^{3}r_{04}^{2}r_{02} + 120r_{03}^{3}r_{04}^{2}r_{03} - 32r_{03}^{3}r_{04}^{3}r_{02} - 16r_{03}^{3}r_{04}^{2}r_{02} - 128r_{03}^{3}r_{04}r_{02} + 6r_{03}^{5}r_{02}^{2}r_{04} + 6r_{03}^{5}r_{04}^{2}r_{02} + 176r_{03}r_{02}r_{04}^{4} + 32r_{02}^{4}r_{04}^{2}r_{03} +
$$

$$
\alpha_{6} = 32r_{03}r_{04}^{3}r_{02} - 58r_{02}^{5}r_{04}r_{03} + 128r_{03}^{2}r_{04}r_{02} - 28r_{03}^{2}r_{04}r_{02}^{3} + 128r_{04}^{2}r_{03}r_{02} + 5r_{03}^{3}r_{04}r_{02}^{3} + 26r_{02}^{3}r_{04}r_{03}^{2}
$$
\n
$$
- 35r_{03}^{4}r_{04}r_{02}^{2} - 18r_{02}^{4}r_{04}r_{03}^{2} - 256r_{03}r_{04}r_{02}^{2} + 18r_{02}^{4}r_{04}^{2}r_{04}^{2}r_{03} - 67r_{03}r_{02}^{3}r_{04}^{2} - 64r_{02}^{3}r_{04}r_{03} + r_{03}^{2}r_{04}^{2}r_{04}^{3}
$$
\n
$$
+ 128r_{03}^{2}r_{02}r_{04}^{3} + 68r_{02}^{3}r_{04}r_{03} - 260r_{03}r_{02}^{2}r_{04}^{3} + 5r_{04}^{4}r_{02}r_{03}^{2} + 49r_{03}r_{04}^{5}r_{02} + 5r_{03}r_{04}^{4}r_{02} + 9r_{03}^{5}r_{04}r_{02}
$$
\n
$$
- 39r_{03}^{4}r_{02}r_{04} - 160r_{03}^{3}r_{04}r_{02} + 30r_{03}^{3}r_{04}r_{02} + 5r_{03}^{3}r_{04}r_{02} + 32r_{03}^{3}r_{04}r_{02} + 140r_{03}r_{02}r_{04}^{4} - 88r_{02}^{4}r_{04}r_{03}
$$
\n
$$
- 52r_{03}^{4}r_{04}r_{02} - 4r_{03}^{3}r_{04}r_{02}^{2} + 29r_{03}^{2}r_{
$$

$$
\alpha_7 = -12r_{02}^3r_{04}r_{03} + 9r_{03}^2r_{02}^3 + 9r_{02}^3r_{04}^2 - 16r_{03}r_{04}r_{02}^2 - r_{03}r_{04}^2r_{02}^2 - 9r_{03}^2r_{02}^2r_{04} + 16r_{03}^2r_{02}^2 + 2r_{02}^2r_{03}^3 + 16r_{04}^2r_{02}^2 - 6r_{04}^3r_{02}^2 + 6r_{03}r_{04}^3r_{02} - 32r_{04}^3r_{02} - 3r_{03}^4r_{02} + 8r_{04}^2r_{03}r_{02} + 2r_{03}^2r_{04}^2r_{02} - 24r_{03}^2r_{04}r_{02} - 11r_{04}^4r_{02} - 2r_{03}^3r_{04}r_{02} + 3r_{03}^4r_{04} + 5r_{03}^2r_{04}^3 - 3r_{03}^3r_{04}^2 + 11r_{04}^4r_{03} + 32r_{04}^3r_{03}.
$$
 (B9)

These complicated expressions are exact and may be simplified by an additional approximation. For example, for large peri- and apoapsis only the higher orders of r_{03} and r_{04} may be taken into account. In particular, in the post-Newtonian case these long expressions are reduced to ([110](#page-11-3)).

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