Photon emissions from near-horizon extremal and near-extremal Kerr equatorial emitters

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We consider isotropic and monochromatic photon emissions from equatorial emitters moving along future-directed timelike geodesics in the near-horizon extremal Kerr (NHEK) and near-horizon near-extremal Kerr (near-NHEK) regions, to asymptotic infinity. We obtain numerical results for the photon escaping probability (PEP) and derive analytical expressions for the maximum observable blueshift (MOB) of the escaping photons, both depending on the emission radius and the emitter's proper motion. In particular, we find that for all antiplunging or deflecting emitters that can eventually reach to asymptotic infinity, the PEP is greater than 50% while for all plunging emitters the PEP is less than 55%, and for the bounded emitters in the (near-)NHEK region, the PEP is always less than 59%. In addition, for the emitters on unstable circular orbits in the near-NHEK region, the PEP decreases from 55% to 50% as the orbital radius decreases from the one of the innermost stable circular orbit to the one of the horizon. Furthermore, we show how the orientation of the emitter's motion along the radial or azimuthal direction affects the PEP and the MOB of the emitted photons.

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I. INTRODUCTION

Since the release of the first image of the supermassive black hole M87* by the Event Horizon Telescope Collaboration [1] in 2019, there have been ongoing progresses in the community to explore physical information of the black hole. This M87* image matches very well with the simulation of a rotating Kerr black hole based on Einstein's theory of general relativity. Very recently, an improved image with polarization information was released, which would help us to explore the structure of the magnetic fields near the black hole [2]. Understanding these observations relies not only on the study of photon emissions and propagations in the black hole backgrounds, but also on the rich physics in the surroundings of the black holes, such as the magnetohydrodynamics in plasmas, the distribution of magnetic fields, and even the nature of dark matter halo. Consequently the images of the black holes may allow us to investigate various problems in the strong gravitational field region.

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Besides the silhouette of a black hole shadow, there are other possible observational signatures with interesting features; for example, the image of a hot spot surrounding a black hole [3–6], light rings (LRs) [7–9], and photon rings [10–15]. Theoretically, as the first step, one has to understand clearly the photon emissions around the black hole, especially near the black hole horizon. On the other hand, studying the photon emissions from the near-horizon area is important to exploring various high-energy processes [16–26] happening in the near-horizon region (inside the ergosphere).

Recently, the photon escaping probability (PEP) and maximum observable blueshift (MOB)¹ of photon emissions from near-horizon sources around rotating black holes has been studied by several groups. The PEP of a zero-angular momentum source (ZAMS) near a Kerr-Newmann black hole was first studied in [27], and later the study was generalized to the Kerr-Sen spacetime in [28]. The PEP of a ZAMS approaching the event horizon is

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¹We use the term "MOB" for the reason that the photons with maximum net blueshift can always reach at infinity and thus are observable in principle. But, in practice, these escaping photons may not be observed by a distant observer since the MOB could be negative and the PEP could also be vanishingly small.

zero for a nonextremal Kerr black hole and tends to ~29% for an extremal Kerr black hole. Moreover, the PEP and MOB of photon emissions from equatorial (stable) circular orbiters near a Kerr black hole was studied in [29] and [30,31] with different methods clarifying the photon escaping condition. The PEP for a circular orbiter increases as the orbital radius increases from the innermost stable circular orbit (ISCO), while it decreases as the black hole spin parameter increases at a given orbital radius. Remarkably, the value of PEP for an emitter sitting at the ISCO of an extremal Kerr black hole is still about 55%, and the proper motion of the emitter blueshifts the energy of the escaped photons. Very recently, photon emissions from the plunging orbiters² starting from the ISCO of a black hole with arbitrary spin was studied in [32], and it was found that the PEP is always more than 50% for the emitter at approximately halfway between the ISCO and the event horizon. Moreover, the effects of the proper motion of a plunging emitter on the PEP and the MOB were discussed in [32]. While these studies applied to the emitter of arbitrary radius and the black hole of arbitrary spin, special attentions had been paid to the near-horizon and nearextremal cases in [30] since such cases can be dealt with analytically. In a previous paper [33], we reinvestigated the photon emissions from a ZAMS near a Kerr black hole by adapting the method proposed in [30], and we explored the observability of the very deep region in the near-horizon throat of a high-spin black hole. To that end, we analytically computed the PEP and the MOB of the ZAMSs both in the near-horizon extremal Kerr (NHEK) and the near-horizon near-extremal Kerr (near-NHEK) geometries, and found that the PEP tends to $\sim 13\%$ at the radius of the innermost photon shell.

In this paper, we would like to study the effects of the emitter's proper motion on the PEP and the MOB of the escaping photons, with the ZAMS as a "static" reference [27,33]. We will focus on the equatorial emitters that follow timelike geodesics in the (near-)NHEK³ geometries. We will consider not only the marginally plunging emitters starting from the ISCO [32], but also other possible plunging emitters. Moreover, we will consider the emitters with other geodesic motions as well, including the antiplunging, the deflecting, and the bounded motions. In [32], only the marginally plunging emitters which have the critical angular momentum $l_s = l_{\rm ISCO}$ and critical energy $\omega_s = \omega_{\rm ISCO}$ have been studied, while the black holes were allowed to have arbitrary spin. Here, instead, we will only focus on the high-spin black hole but consider the emitters with more possible values of l_s and ω_s . Note that in the high-spin case the energy ω_s of the emitter is constrained to be near the superradiant bound $\omega_* = l_s/2M$ with the corrections appearing at order $\mathcal{O}(\epsilon^q)$ [see Eqs. (2.10) and (2.11)]. The timelike geodesics in the (near-)NHEK regions were well classified in [34-37] based on the conserved quantities (2.18) in the (near-)NHEK geometries. As we restrict the emitters on the equatorial plane, the Carter constant for these emitters is set to zero, $Q_s = 0$; thus each orbit of an emitter is specifically labeled by its (near-)NHEK energy E_s , angular momentum L_s , and radial orientation s_r . We will consider the behaviors of the PEP and the MOB of the escaping photons emitted from the orbit labeled by (L_s, E_s, s_r) and study how they are affected when each of these parameters varies. Our main results are summarized in Eqs. (4.18)–(4.20) and in Fig. 2. In addition, we provide detailed illustrations and discussions for the results in Figs. 3–6.

The remaining part of this paper is organized as follows. In Sec. II, we briefly introduce the Kerr geometry and the geodesics in it, as well as its near-horizon and (near-) extremal limits. In Sec. III, we define the problem of photon emissions from an arbitrary near-horizon equatorial emitter, and find the equations to read the PEP and the MOB. In Sec. IV, we perform the computations for the (near-)NHEK emitters. In Secs. V and VI, we display our main results with plots and discuss them in detail. In Sec. VII, we summarize and conclude this work.

II. GEOMETRY AND GEODESICS

In this section, we review the timelike and the null geodesics in the Kerr, NHEK, and near-NHEK geometries [34–38]. In particular, we review the classifications of the (near-)NHEK geodesics based on their constants of motions. The timelike geodesics provide the orbits for the photon emitters with various motions. For the null geodesics, we focus on the unstable spherical photon orbits, which correspond to the threshold between the escaping photons and the captured ones.

A. Kerr geometry and geodesics

The Kerr metric in the Boyer-Lindquist coordinates (t, r, θ, ϕ) is given by

$$ds^{2} = -\frac{\Sigma\Delta}{\Xi}dt^{2} + \frac{\Sigma}{\Delta}dr^{2} + \Sigma d\theta^{2} + \frac{\Xi\sin^{2}\theta}{\Sigma} \left(d\phi - \frac{2Mar}{\Xi}dt\right)^{2},$$
(2.1)

where

$$\Delta = r^2 - 2Mr + a^2, \qquad \Sigma = r^2 + a^2 \cos^2 \theta,$$

 $\Xi = (r^2 + a^2)^2 - a^2 \Delta \sin^2 \theta.$ (2.2)

This metric describes a rotating black hole with a mass M and a spin parameter $a \equiv J/M$. The event horizons of the black hole are located at $r_{\pm} = M \pm \sqrt{M^2 - a^2}$.

²The appearance of a bright hot spot falling into a nonspinning black hole and its relation with LRs were studied in Ref. [9]. We use "(near-)NHEK" for both NHEK and near-NHEK.

A free particle possesses four conserved quantities of motion: the mass μ , the energy ω , the axial angular momentum l, and the Carter constant Q. Using the Hamilton-Jacobi method, one can derive the four-momentum p^{μ} of this particle [38]:

$$p^{r} = \frac{1}{\Sigma} \pm_{r} \sqrt{\mathcal{R}(r)}, \qquad (2.3)$$

$$p^{\theta} = \frac{1}{\Sigma} \pm_{\theta} \sqrt{\Theta(\theta)}, \tag{2.4}$$

$$p^{\phi} = \frac{1}{\Sigma} \left[\frac{a}{\Delta} \left[\omega (r^2 + a^2) - al \right] + \frac{l}{\sin^2 \theta} - a\omega \right], \quad (2.5)$$

$$p^{t} = \frac{1}{\Sigma} \left[\frac{r^2 + a^2}{\Delta} \left[\omega (r^2 + a^2) - al \right] + a(l - a\omega \sin^2 \theta) \right], \tag{2.6}$$

where

$$\mathcal{R}(r) = [\omega(r^2 + a^2) - al]^2 - \Delta[Q + (l - a\omega)^2 + \mu^2 r^2],$$
(2.7)

$$\Theta(\theta) = Q + a^2(\omega^2 - \mu^2)\cos^2\theta - l^2\cot^2\theta \qquad (2.8)$$

are the radial and angular potentials, respectively, and \pm_r and \pm_θ denote the radial and polar orientations, respectively.

For a massive particle, we have $\mu > 0$ and $p^{\mu} = \mu \frac{\partial x^{\mu}}{\partial \tau}$ with τ being the proper time. For a massless particle (photon), we have $\mu = 0$ and $p^{\mu} = \frac{\partial x^{\mu}}{\partial \tau}$ with τ being an affine parameter. For null geodesics, the energy ω may be scaled out from the expressions of p^{μ} under a reparametrization and it is convenient to introduce a pair of impact parameters for the null geodesics:

$$\lambda = \frac{l}{\omega}, \qquad \eta = \frac{Q}{\omega^2}.$$
 (2.9)

B. (Near-)NHEK geometry and geodesics

In order to study the near-horizon geometry of a (near-) extremal Kerr black hole, it is convenient to consider the near-extremal limit [34,39,40],

$$a = M\sqrt{1 - (\epsilon \kappa)^2}, \qquad 0 < \epsilon \ll 1, \qquad \kappa \text{ finite}, \quad (2.10)$$

and work with the Bardeen-Horowitz coordinates [34,39–41],

$$T = \epsilon^q \frac{t}{2M}, \quad R = \frac{r - M}{\epsilon^q M}, \quad \Phi = \phi - \frac{t}{2M}, \quad 0 < q \le 1.$$
 (2.11)

Under the transformations (2.11), and taking the limit $\epsilon \to 0$ for 0 < q < 1, the Kerr metric (2.1) yields the NHEK metric [41]

$$ds^{2} = 2M^{2}\Gamma \left[-R^{2}dT^{2} + \frac{dR^{2}}{R^{2}} + d\theta^{2} + \Lambda^{2}(d\Phi + RdT)^{2} \right],$$
(2.12)

$$\Gamma(\theta) = \frac{1 + \cos^2 \theta}{2}, \qquad \Lambda(\theta) = \frac{2 \sin \theta}{1 + \cos^2 \theta}.$$
 (2.13)

In the NHEK geometry, the event horizon is mapped to R=0 and the ISCO is mapped to $R=2^{1/3}$. However, since the NHEK metric is invariant under a dilation $(R,T) \rightarrow (\varepsilon R, T/\varepsilon)$ for an arbitrary constant ε , the ISCO can actually stay anywhere in the intrinsic NHEK geometry and $R=2^{1/3}$ gets meaningful only after the NHEK region is glued onto the asymptotic far region [35,40].

Under the transformations (2.11) and taking the limit $\epsilon \to 0$ for q=1, we find that the Kerr metric (2.1) yields the near-NHEK metric⁴ [40,42]

$$ds^{2} = 2M^{2}\Gamma \left[-(R^{2} + \kappa^{2})dT^{2} + \frac{dR^{2}}{(R^{2} + \kappa^{2})} + d\theta^{2} + \Lambda^{2}(d\Phi + RdT)^{2} \right].$$
(2.14)

In the near-NHEK geometry, the event horizon is mapped to $R = \kappa$ and the ISCO is mapped to $R = \infty$ [35,40].

Note that the NHEK metric (2.12) is equivalent to the near-NHEK metric (2.14) with zero near-horizon temperature, $\kappa=0$ [34,42]. Therefore, in the following we will work with Eq. (2.14) for both the NHEK case (0 < q < 1 and $\kappa=0$) and near-NHEK case (q=1 and $\kappa\neq0$), for simplicity.

The four-momentum p^{μ} of a free particle in the (near-) NHEK limit becomes [34,36]

$$p^{R} = \pm_{r} \frac{\sqrt{\mathcal{R}_{n}(R)}}{2M^{2}\Gamma},$$
(2.15a)

$$p^{\theta} = \pm_{\theta} \frac{\sqrt{\Theta_n(\theta)}}{2M^2 \Gamma},\tag{2.15b}$$

$$p^{\Phi} = \frac{1}{2M^2\Gamma} \left(-\frac{R(E+LR)}{R^2 - \kappa^2} + \frac{L}{\Lambda^2} \right),$$
 (2.15c)

⁴We will use the same coordinates (T, R, θ, Φ) [Eq. (2.11)] to describe the near-NHEK geometry which has already been used for the NHEK metric (2.12). In the following, we will also use the same notations for the conserved quantities (E, L, C, λ, η) both in the NHEK and near-NHEK geometries. However, these should be distinguishable from the context.

$$p^{T} = \frac{1}{2M^{2}\Gamma} \left(\frac{E + LR}{R^{2} - \kappa^{2}} \right),$$
 (2.15d)

where

$$\mathcal{R}_n(R) = -CR^2 + 2ELR + E^2 + (C + L^2)\kappa^2, \quad (2.16)$$

$$\Theta_n(\theta) = C + \left(1 - \frac{1}{\Lambda^2}\right)L^2 - 2M^2\Gamma\mu^2$$
 (2.17)

are, respectively, the radial and angular potentials in the NHEK region, with C, E, and L being the (near-)NHEK conserved quantities: the Casimir, the energy, and the angular momentum, respectively. The (near-)NHEK conserved quantities are related to the Kerr conserved quantities by [34,36]

$$l = L, \quad \omega = \frac{L}{2M} \left(1 + \epsilon^q \frac{E}{L} \right), \quad Q = C + \frac{3}{4} L^2 - \mu^2 M^2.$$
 (2.18)

For the null geodesics we have $\mu = 0$, and then we may define a pair of shifted impact parameters,

$$\lambda_0 = \frac{E}{L}, \qquad \eta_0 = \frac{C}{L^2},$$
 (2.19)

which are related to the Kerr impact parameters (2.9) by⁵

$$\lambda = 2M(1 - \epsilon^q \lambda_0)$$
 $\eta = M^2(3 + 4\eta_0).$ (2.20)

It is convenient to introduce a critical angular momentum [35,36]

$$L^* = l^* = \frac{2}{\sqrt{3}} \sqrt{M^2 \mu^2 + Q},$$
 (2.21)

which is for the geodesic at the innermost stable spherical orbit (ISSO). In terms of this critical angular momentum, the Casimir *C* can be rewritten as

$$C = -\frac{3}{4}[L^2 - (L^*)^2]. \tag{2.22}$$

We will restrict to the future-directed geodesics, which require $p^T > 0$ [Eq. (2.15d)], i.e.,

$$E + LR > 0. (2.23)$$

Next, by analyzing the root structures of the radial potential (2.16), the equatorial orbits are classified in the phase space (L, E) regarding their radial motions [34,36].

TABLE I. Future-directed geodesics in NHEK [34,36].

Classification	L	Е	\pm_r	Range
ISSO Marginal Plunging Antiplunging Deflecting	$L = L^*$ $L > L^*$ $L \ge L^*$ $L \ge L^*$ $L > L^*$	E = 0 E = 0 E > 0 E > 0 E < 0	±1 ±1 -1 +1 ±1	$\begin{array}{c} 0 \leq R \leq \infty \\ R_{+} \leq R \leq \infty \end{array}$
Bounded	$-L^* < L < L^*$ $L = -L^*$ $L < -L^*$	E > 0 $E > 0$ $E > 0$	±1 ±1 ±1	$\begin{array}{l} 0 \leq R \leq R_+ \\ 0 \leq R \leq R_0 \\ 0 \leq R \leq R \end{array}$

TABLE II. Future-directed geodesics in near-NHEK [34,36].

Classification	L	E	\pm_r	Range
Spherical	$L > L^*$	$E = -\kappa\sqrt{-C}$	±1	$R = \frac{\kappa L}{\sqrt{-C}}$
Plunging	$L = L^*$ $L > L^*$	$E \ge 0$ $E > -\kappa \sqrt{-C}$		$\kappa \le R \le \infty$ $\kappa \le R \le \infty$
Antiplunging	$L = L^*$ $L > L^*$	$E \ge 0$ $E > -\kappa \sqrt{-C}$		$\kappa \le R \le \infty$ $\kappa \le R \le \infty$
Deflecting	$L > L^*$	$E<-\kappa\sqrt{-C}$	±1	$R_+ \le R \le \infty$
Bounded	$-L^* < L < L^*$ $L = L^*$ $L = -L^*$ $L < -L^*$	$-\kappa L < E < 0$	$\pm 1 \\ \pm 1$	$\kappa \le R \le R_0$

The classifications⁶ for the NHEK orbits and for the near-NHEK orbits are collected in Tables I and II, respectively, where R_{\pm} and R_0 are the root(s) for the corresponding cases:

$$R_{\pm} = \frac{E}{C}L \pm \frac{1}{|C|}\sqrt{(L^2 + C)(E^2 + \kappa^2 C)},$$

$$R_0 = -\frac{E^2 + \kappa^2 L^2}{2EL}.$$
(2.24)

The meaning of these distinct orbits are as follows [35,36]. Each ISSO orbits at a constant radius. A plunging orbit enters the NHEK geometry from the (near-)NHEK boundary and eventually crosses the horizon in finite affine time. An antiplunging orbit emerges out of the horizon and eventually leaves the (near-)NHEK boundary in finite affine time. A deflecting orbit falls in from the (near-)NHEK boundary, bounces off at the radial turning point, and returns to the boundary in finite affine time. Marginal (plunging/antiplunging/deflecting) orbits exist only in the NHEK geometry and leave the NHEK geometry in infinite

⁵This means that the photon emissions are near the superradiant bound. Similar parameters have also been introduced in [3].

⁶Note that there was a forgotten region for the near-NHEK deflecting orbits in [36], which was corrected in [37]. Note also that we use "antiplunging" for the "outward" orbits in [36].

affine time. A bounded orbit stays in the (near-)NHEK region between the horizon and the radial turning point. The antiplunging and deflecting orbits can escape from the (near-)NHEK to infinity while the plunging and the bounded ones can not. Note that all near-NHEK spherical orbits are unstable while all NHEK spherical orbits are ISSO [36].

III. GENERAL SETUP

In this section, we consider the photon emissions from an equatorial source near a Kerr black hole and discuss the near-horizon and near-extremal limits of this problem. In the following, we use k^{μ} and p^{μ} for the four-momentums of an emitter and a photon, respectively, and we use the notation that the quantities with a subscript "s" are for an emitter (source) while the quantities without subscript are for the photons. In addition, we let $s_{r/\theta} = \pm_{r/\theta} = \pm 1$ denote the orientations of the emitters and let $\sigma_{r/\theta} = \pm_{r/\theta} = \pm 1$ denote the orientations of the photons.

We consider photon emissions from the equatorial emitters that move along timelike geodesics. For such emitters we have $\theta_s = \pi/2$, thus $Q_s = 0$ and

$$L_s^* = l_s^* = \frac{2}{\sqrt{3}} M\mu. \tag{3.1}$$

The photon emissions are described by null geodesics, which have

$$L^* = l^* = \frac{2}{\sqrt{3}}\sqrt{Q}. (3.2)$$

We will first introduce a pair of local emission angles of the photons in the emitter's local rest frame (LRF), which are expressed in terms of the impact parameters of the photons. Next, we consider the unstable spherical photon orbits, which make up the photon shell and correspond to critical emission angles on the emitter's sky. The critical emission angles will line up into a closed critical curve on the emitter's sky, which allows us to distinguish the photons captured by the central black hole from those escaping to asymptotic infinity [30]. We then define the PEP, whose computation is based on integrating over the interior region of the critical curve, and we find the expression for the MOB.

A. Local emission angles on the emitter's sky

We use x^{μ} with $\mu = 0, 1, 2, 3$ to represent both the Boyer-Lindquist coordinates (t, r, θ, ϕ) for the Kerr metric (2.1) and the Bardeen-Horowitz coordinates (T, R, θ, Φ) for the NHEK (2.12) and near-NHEK (2.14) metrics.

In terms of the components of the spacetime metric $g_{\mu\nu}$, the locally nonrotating frame (LNRF) is given by [43]

$$e_{(0)} = \zeta(\partial_0 + \varsigma \partial_3), \qquad e_{(1)} = \xi_1 \partial_1,$$

 $e_{(2)} = \xi_2 \partial_2, \qquad e_{(3)} = \xi_3 \partial_3,$ (3.3)

where

$$\zeta = \sqrt{\frac{g_{33}}{g_{03}^2 - g_{00}g_{33}}}, \qquad \zeta = -\frac{g_{03}}{g_{33}}, \qquad \xi_1 = \frac{1}{\sqrt{g_{11}}},$$

$$\xi_2 = \frac{1}{\sqrt{g_{22}}}, \qquad \xi_3 = \frac{1}{\sqrt{g_{33}}}.$$
(3.4)

The LRF of an equatorial emitter can be obtained by performing a Lorentz transformation on the LNRF, which is given by

$$\sigma_{[0]} = \gamma [e_{(0)} + v^{(1)}e_{(1)} + v^{(3)}e_{(3)}]|_{x^i = x_s^i}, \tag{3.5a}$$

$$\sigma_{[1]} = \frac{1}{v} [v^{(3)} e_{(1)} - v^{(1)} e_{(3)}]|_{x^i = x^i_s}$$
(3.5b)

$$\sigma_{[2]} = e_{(2)}|_{x^i = x^i_s},\tag{3.5c}$$

$$\sigma_{[3]} = \gamma \left[v e_{(0)} + \frac{1}{v} (v^{(1)} e_{(1)} + v^{(3)} e_{(3)}) \right] |_{x^i = x_s^i}, \quad (3.5d)$$

where v and $v^{(i)}$ are the 3-velocity and its components of the emitter relative to the LNRF, and γ is the boost factor. For an emitter at x_s^{μ} with momentum k^{μ} , these are given by

$$v^{(i)} = \frac{k^{\mu} e_{\mu}^{(i)}}{k^{\mu} e_{\mu}^{(0)}} \Big|_{x^{i} = x_{s}^{i}}, \qquad (i = 1, 2, 3),$$

$$v = \sqrt{(v^{(1)})^{2} + (v^{(3)})^{2}}, \qquad \gamma = \frac{1}{\sqrt{1 - v^{2}}}. \tag{3.6}$$

In order to study photon emissions from an equatorial emitter, we can define a pair of local emission angles (α, β) on the emitter's sky (see Fig. 1) [27,30],

$$\alpha = \arccos\left[\frac{p_s^{[3]}}{p_s^{[0]}}\right] \in [0, \pi],$$

$$\beta = \arcsin\left[\frac{p_s^{[1]}}{\sqrt{(p_s^{[1]})^2 + (p_s^{[2]})^2}}\right] \in \left[-\frac{\pi}{2}, \frac{\pi}{2}\right], \quad (3.7)$$

where $p_s^{[a]} = p^\mu \sigma_\mu^{[a]}|_{x^i=x^i_s}$ are the components of the photon's f in its LRF. Note that the setup is symmetric about the equatorial plane and only σ_θ^2 appears in the local angles, therefore we may only focus on a half-sphere of the emitter's sky by choosing either $\sigma_\theta = 1$ or $\sigma_\theta = -1$ in $p_s^{[2]}$. Then the half-sphere is uniquely parametrized by (α, β) .

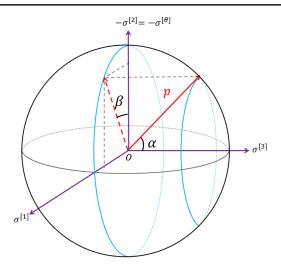


FIG. 1. Emitter's sky parametrized by local emission angles α and β .

B. Photon shell and critical emission angles

The photon shell consists of unstable spherical photon orbits in the Kerr spacetime, which satisfy

$$\mathcal{R}(\tilde{r}) = \mathcal{R}'(\tilde{r}) = 0. \tag{3.8}$$

By solving these equations, we can obtain the critical photon impact parameters [38],

$$\tilde{\lambda}(\tilde{r}) = a + \frac{\tilde{r}}{a} \left[\tilde{r} - \frac{2(\tilde{r}^2 - 2M\tilde{r} + a^2)(\tilde{r})}{\tilde{r} - M} \right], \quad (3.9a)$$

$$\tilde{\eta}(\tilde{r}) = \frac{\tilde{r}^3}{a^2} \left[\frac{4M(\tilde{r}^2 - 2M\tilde{r} + a^2)(\tilde{r})}{(\tilde{r} - M)^2} - \tilde{r} \right]. \tag{3.9b}$$

We have $\tilde{\eta}(\tilde{r}) \ge 0$ for the photons crossing the equatorial plane, then these orbits lie in the range $\tilde{r}_1 \le \tilde{r} \le \tilde{r}_2$, where

$$\tilde{r}_1 = 2M \left[1 + \cos\left(\frac{2}{3}\arccos\left[-\frac{a}{M}\right]\right) \right], \quad (3.10)$$

$$\tilde{r}_2 = 2M \left[1 + \cos\left(\frac{2}{3}\arccos\left[\frac{a}{M}\right]\right) \right].$$
 (3.11)

Hereafter, the quantities adorned with a tilde are evaluated for the critical photon emissions.

Plugging (3.9) into (3.7) gives the critical emission angles, which line up into a closed critical curve on an emitter's sky [30],

$$C = \{ (\tilde{\alpha}(\tilde{r}), \tilde{\beta}(\tilde{r})) | \tilde{r}_1 < \tilde{r} < \tilde{r}_2 \}. \tag{3.12}$$

This critical curve separates the emitter's sky into two distinct regions. To distinguish these regions, it is useful to introduce the "direction to the black hole center" on the emitter sky [Eq. (3.7)], which corresponds to [30]

$$\lambda = \eta = 0, \qquad \sigma_r = -1. \tag{3.13}$$

Then we call the region containing the "direction to the black hole center" the captured region, and call the other complementary one the escaping region.

In the extremal limits (2.10), the intrinsic (near-)NHEK calculations for unstable spherical photon orbits, obtained by solving $\mathcal{R}_n(\tilde{R}) = \mathcal{R}'_n(\tilde{R}) = 0$, are not sufficient to resolve the whole photon shell described by (3.9). The reason is that the whole photon shell stretches over various near-horizon limits for a near-extremal black hole [30]. In the extremal limit $\epsilon \to 0$, the radii of the innermost photon shell (3.10) and the outermost photon shell (3.11) become

$$\tilde{r}_1 = M[1 + \epsilon \tilde{R}_1 + \mathcal{O}(\epsilon^2)], \tag{3.14}$$

$$\tilde{r}_2 = 4M[1 + \mathcal{O}(\epsilon^2)], \tag{3.15}$$

where $\tilde{R}_1 = 2\kappa/\sqrt{3}$. We can see that the radius \tilde{r}_1 is in the near-NHEK p=1 region while the radius \tilde{r}_2 is in the Extreme Kerr "p=0" region [see Eq. (2.11)] [30,34]. Therefore, one must rely on the calculations in the Kerr geometry to obtain the whole photon shell. By expanding the critical impact parameters of the Kerr photon shell [Eqs. (3.9)] in ϵ under the transformations [30]

$$a = M\sqrt{1 - (\epsilon \kappa)^2}, \quad r_s = M(1 + \epsilon^q R_s), \quad \tilde{r} = M(1 + \epsilon^p \tilde{R}),$$

$$(3.16)$$

we can obtain the shifted critical impact parameters $(\tilde{\lambda}_0, \tilde{\eta}_0)$. For the NHEK case we choose q = 2/3 and for the near-NHEK case we have q = 1. We will discuss the expansions for the (near-)NHEK cases in Sec. IV.

C. Photon escaping probabilities and net blueshift

We assume that the emitter emits monochromatic photons isotropically in its LRF (3.5). The redshift factor g and net blueshift z of a photon that reaches to asymptotic infinity are defined by

$$g \equiv \frac{\omega}{p_s^{[0]}}, \qquad z \equiv \frac{g-1}{g}. \tag{3.17}$$

Let A_e/A_c be the area of the photon escaping/captured region in the emitter's sky of unit radius, respectively. The photon escaping probability is defined by [27,30]

$$\mathcal{P}_e \equiv \frac{\mathcal{A}_e}{4\pi} = \frac{1 - \mathcal{A}_c}{4\pi}.\tag{3.18}$$

In order to compute the PEP, we introduce a pair of planar polar coordinates [31],

$$\rho = \sqrt{2(-\cos\alpha + 1)}, \qquad \varphi = \frac{\pi}{2} + \beta, \quad (3.19)$$

such that the area element dA on the (ρ, φ) -plane is equal to the area element on the sphere of the emitter's sky $d\Omega$,

$$d\mathcal{A} = \rho d\rho \wedge d\varphi = \sin \alpha d\alpha \wedge d\beta = d\Omega. \tag{3.20}$$

Regarding the position of the direction to the black hole center on the emitter's sky, the area of the interior region of the closed critical curve (3.12) may correspond to either A_e or A_c , which can be computed by [30]

$$\mathcal{A}_{\rm in} = \int_{\rm in} \tilde{\rho} d\tilde{\rho} d\tilde{\varphi} = \int_{\rm in} \frac{1}{2} \tilde{\rho}^2 d\tilde{\varphi}. \tag{3.21}$$

IV. PHOTON EMISSIONS FROM (NEAR-)NHEK EMITTERS

In Sec. II B, we have reviewed the (near-)NHEK geometries and the geodesics for the massive particles and for the photons. In this section, we compute the PEP and the MOB for the photon emissions from the emitters at (near-)NHEK radius R_s , whose motions are characterized by (L_s, E_s, s_r) . The relevant formulae are just defined in Sec. III. We have q = 2/3 in the Bardeen-Horowitz

coordinates [Eq. (2.11)] for the NHEK geometry, and we have q=1 in the Bardeen-Horowitz coordinates for the near-NHEK geometry.

A. Critical emission angles and critical curve

In the last section, we obtained the LRF [Eq. (3.5)] and the local emission angles [Eq. (3.7)] of an equatorial emitter at arbitrary radius around a black hole of arbitrary spin. Now we compute the LRF and the local emission angles for a (near-)NHEK emitter. At the emitter's position, the LNRF (3.3) in the (near-)NHEK geometry is given by

$$\begin{split} e_{(T)} &= \frac{\partial_T - R_s \partial_{\Phi}}{M \sqrt{R_s^2 - \kappa^2}}, \qquad e_{(R)} &= \frac{\sqrt{R_s^2 - \kappa^2}}{M} \partial_R, \\ e_{(\theta)} &= \frac{1}{M} \partial_{\theta}, \qquad e_{(\Phi)} &= \frac{1}{2} \partial_{\Phi}. \end{split} \tag{4.1}$$

The components of 3-velocity [Eq. (3.6)] of an emitter relative to the LNRF are

$$v^{(R)} = \frac{s_r \sqrt{\mathcal{R}_n(R_s)}}{E_s + L_s R_s}, \qquad v^{(\Phi)} = \frac{L_s \sqrt{R_s^2 - \kappa^2}}{2(E_s + L_s R_s)}, \quad (4.2)$$

where \mathcal{R}_n is defined in (2.16). Then the local emission angles [Eq. (3.7)] of an emitter are given by

$$\alpha(\lambda_0, \eta_0) = \arccos\left[\frac{-2v^2(\lambda_0 + R_s) + 2\sigma_r\sqrt{\mathcal{R}_{n0}(R_s)}v^{(R)} + v^{(\Phi)}\sqrt{R_s^2 - \kappa^2}}{v[2(\lambda_0 + R_s) - 2\sigma_rv^{(R)}\sqrt{\mathcal{R}_{n0}(R_s)} - v^{(\Phi)}\sqrt{R_s^2 - \kappa^2}]}\right],\tag{4.3a}$$

$$\beta(\lambda_0, \eta_0) = \arcsin \left[\frac{2\sigma_r v^{(\Phi)} \sqrt{\mathcal{R}_{n0}(R_s)} - v^{(R)} \sqrt{R_s^2 - \kappa^2}}{\sqrt{4v^2 (R_s^2 - \kappa^2)\Theta_{n0}(\frac{\pi}{2}) + [v^{(R)} \sqrt{R_s^2 - \kappa^2} - 2\sigma_r v^{(\Phi)} \sqrt{\mathcal{R}_{n0}(R_s)}]^2}} \right], \tag{4.3b}$$

where

$$\mathcal{R}_{n0}(R) = \frac{\mathcal{R}_n(R)}{L^2} = -\eta_0(R^2 - \kappa^2) + 2\lambda_0 R + \lambda_0^2 + \kappa^2,$$
(4.4)

$$\Theta_{n0}\left(\frac{\pi}{2}\right) = \frac{\Theta_n(\theta)}{L^2}\bigg|_{\theta = \frac{\pi}{2}} = \eta_0 + \frac{3}{4}.\tag{4.5}$$

As discussed in Sec. III B, the (near-)NHEK critical parameters $(\tilde{\lambda}_0, \tilde{\eta}_0)$ [Eq. (2.20)] can be obtained by expanding the Kerr critical impact parameters $(\tilde{\lambda}, \tilde{\eta})$ [Eq. (3.9)]. To be specific, let us consider the expansions (3.16) for the emitter's radius r_s with q = 2/3 or 1, and for the photon shell radii \tilde{r} with $0 . Then, to the leading order as <math>\epsilon \to 0$, we obtain

$$\tilde{\lambda}_0^+ = \frac{1}{2}\tilde{R}^2, \qquad \tilde{\eta}_0^+ = 0, \quad \text{for } p = p^+ = \frac{q}{2},$$
 (4.6a)

$$\tilde{\lambda}_{0}^{-} = -\frac{\kappa^{2}}{\tilde{R}}, \qquad \tilde{\eta}_{0}^{-} = -\frac{\kappa^{2}}{\tilde{R}^{2}}, \quad \text{for } p = p^{-} = 1,$$
 (4.6b)

$$\tilde{\lambda}_0^{m+} = \infty, \quad \tilde{\lambda}_0^{m+} = 0, \quad \text{for } p = p^{m+} \in \left(0, \frac{q}{2}\right),$$
 (4.6c)

$$\tilde{\lambda}_0^{m-} = 0, \quad \tilde{\lambda}_0^{m-} = 0, \quad \text{for } p = p^{m-} \in \left(\frac{q}{2}, 1\right),$$
 (4.6d)

where $\tilde{R}(+) \in (0, \infty)$ and $\tilde{R}(-) \in [\tilde{R}_1, \infty)$. Here, the superscripts " \pm " and "m" represent the outer/inner and middle limits of the photon shell, respectively. The critical emission angles and the critical curve (3.12) can be

obtained by plugging the critical impact parameters (4.6) into the local emissions angles (4.3) with $\sigma_r = \text{sign}(\tilde{r} - r_s)$. Note that the limits of $p = p^{\pm}$ could complete the critical curve, while those of $p = p^{m\pm}$ only contribute the points that connect the p^{\pm} curves [30,33].

Note also that, if we take the transformation $\tilde{R}^2 \to R_s \tilde{R}^2$ for $(\tilde{\alpha}^+, \tilde{\beta}^+)$, then a NHEK critical curve is independent of R_s once the 3-velocity components (4.2) are fixed, and so does the PEP. This is due to the dilational symmetry of the NHEK geometry. In contrast, a nonzero near-horizon temperature κ in the near-NHEK geometry breaks the dilational symmetry. Therefore, in contrast with the NHEK case, a near-NHEK critical curve does depend on R_s even when the 3-velocity components (4.2) are fixed, and so does the PEP.

B. Photon escaping probability

Let us now compute the PEP (3.18) for the photon emissions from a free (near-)NHEK emitter labeled by (L_s, E_s, s_r) [Eq. (2.18)] at radius R_s . To do so, we need to compute the area inside the critical curve (3.12) in the (ρ, φ) -plane (3.19). From (2.20) and (3.13), the "direction to the black hole center" is expressed in terms of the shifted critical impact parameters as

$$\lambda_0 = \infty, \qquad \eta_0 = -\frac{3}{4}. \tag{4.7}$$

These parameters correspond to a point in the plane which is contained in the photon captured region. Then the interior area \mathcal{A}_{in} [Eq. (3.21)] can be computed numerically as

$$\mathcal{A}_{\text{in}} = \int_{\text{in}} \frac{1}{2} \tilde{\rho}^2 d\tilde{\varphi}
= \left| \int_{\tilde{R}_1}^{\infty} \tilde{\rho}^2(\tilde{\alpha}^-) \frac{d\tilde{\varphi}(\tilde{\beta}^-)}{d\tilde{R}} d\tilde{R} + \int_0^{\infty} \tilde{\rho}^2(\tilde{\alpha}^+) \frac{d\tilde{\varphi}(\tilde{\beta}^+)}{d\tilde{R}} d\tilde{R} \right|.$$
(4.8)

Note that we need to integrate over both of the half-spheres with $\sigma_{\theta} = \pm 1$ on the emitter's sky, thus the factor 1/2 in front of $\tilde{\rho}^2$ has been canceled out. We will present the results in Secs. V and VI, and discuss them in detail there.

C. Maximum observable blueshift

As $\epsilon \to 0$, the redshift factor (3.17) for the photon emissions from a (near-)NHEK emitter becomes

$$\begin{split} g(\lambda_{0},\eta_{0}) &= \frac{\sqrt{R_{s}^{2} - \kappa^{2}}}{\gamma[2(R_{s} + \lambda_{0}) - v^{(\Phi)}\sqrt{R_{s}^{2} - \kappa^{2}} - 2\sigma_{r}v^{(R)}\sqrt{\mathcal{R}_{\kappa 0}(R_{s})}]} \\ & (4.9) \end{split}$$

The MOB $z_{\rm mob}$ is obtained when the redshift factor reaching its maximum value $g_{\rm max}$. In order to find the maximum value for the redshift, we analyze the ranges of the impact parameters (λ_0,η_0) for the photons that can reach to infinity.⁷ We have reviewed the null geodesics in the (near-)NHEK region in Sec. II B, and summarized the classifications in Tables I and II for the NHEK and near-NHEK cases, respectively. For the future-directed photon emissions we have $L^* = \frac{2}{\sqrt{3}}\sqrt{Q} \ge 0$ and $\lambda_0 > -R$, and for the photons arrived at infinity we have $\sigma_r = 1$.

In the NHEK region we always have $r_s > \tilde{r}_1$. In the near-NHEK region we may either have $r_s > \tilde{r}_1$ or have $r_s \leq \tilde{r}_1$. For $r_s > \tilde{r}_1$, both the reflecting and the antiplunging photon trajectories can reach to infinity, while for $r_s \leq \tilde{r}_1$, only the antiplunging photon trajectories can reach to infinity. Then from Tables I and II we have $-1 \leq \eta_0 \leq 0$ and $\lambda_0 > \lambda_{0-}(\eta_0)$ for the antiplunging orbits, and $\lambda_0 < \lambda_{0-}(\eta_0)$ for the deflecting orbits, where

$$\lambda_{0-}(\eta_0) = -\kappa \sqrt{-\eta_0}. (4.10)$$

Moreover, the positivity of the potentials (4.4) and (4.5) requires

$$\lambda_0 \ge \lambda_{0t}(\eta_0) = -R_s + \sqrt{(1+\eta_0)(R_s^2 - \kappa^2)}, \qquad \eta_0 \ge -\frac{3}{4}.$$
(4.11)

For the photon emissions from the infalling emitters with $s_r = -1$ in Eq. (4.2), we have $\partial_{\eta_0} g > 0$ and $\partial_{\lambda_0} g < 0$. Note also that we have $\lambda_0 > \lambda_{0t}$ for $r_s > \tilde{r}_1$ and $\lambda_0 > \lambda_{0-}$ for $r_s \le \tilde{r}_1$. Then we obtain

$$g_{\text{max},i} = \begin{cases} g_i[\lambda_{0t}(-3/4), -3/4], & \text{if } r_s > \tilde{r}_1, \\ g_i[\lambda_{0-}(-3/4), -3/4], & \text{if } r_s \le \tilde{r}_1, \end{cases}$$
(4.12)

where

$$g_{i}[\lambda_{0t}(-3/4), -3/4] = \frac{1}{\gamma[1 - v^{(\Phi)}]} = \frac{\sqrt{3}L_{s}^{*}\sqrt{R_{s}^{2} - \kappa^{2}}}{(2R_{s} - \sqrt{R_{s}^{2} - \kappa^{2}})L_{s} + 2E_{s}}, \quad (4.13)$$

Note that σ_r will flip from -1 to +1 at the turning point.

⁷The MOB of the marginally plunging emitters from the ISCO was discussed in [32] by analyzing the corresponding parameters of the photon emissions.

$$g_{i}[\lambda_{0-}(-3/4), -3/4] = \frac{\sqrt{R_{s}^{2} - \kappa^{2}}}{\gamma \left[2R_{s} - \sqrt{3}\kappa - v^{(\Phi)}\sqrt{R_{s}^{2} - \kappa^{2}} - v^{(R)}\sqrt{3R_{s}^{2} - 4\sqrt{3}R_{s}\kappa + 4\kappa^{2}} \right]}.$$
 (4.14)

Hereafter, we use subscript "i/o" to represent $s_r = \mp 1$ in Eq. (4.2), respectively.

For the outgoing emitter with $s_r = -1$ in Eq. (4.2), we also have $\partial_{\eta_0} g < 0$, and we have $\partial_{\lambda_0} g \gtrless 0$ for $\lambda_0 \lessgtr \lambda_{0c}$, where

$$\lambda_{0c}(\eta_0) = -R_s + \gamma_R \sqrt{1 + \eta_0} \sqrt{R_s^2 - \kappa^2}$$
 (4.15)

with $\gamma_R = 1/\sqrt{1 - (v^{(R)})^2}$. Then we obtain

$$g_{\text{max},o} = \begin{cases} g_o[\lambda_{0c}(-3/4), -3/4], & \text{if } \lambda_{0-} < \lambda_{0c}, \\ g_o[\lambda_{0-}(-3/4), -3/4], & \text{if } \lambda_{0-} > \lambda_{0c}, (r_s \le \tilde{r_1} \text{only}), \end{cases}$$

$$(4.16)$$

where $g_o[\lambda_{0-}(-3/4), -3/4]$ has the same form as $g_i[\lambda_{0-}(-3/4), -3/4]$ [Eq. (4.14)] but with $s_r = 1$ in $v^{(R)}$, and

$$g_o[\lambda_{0c}(-3/4), -3/4] = \frac{1}{\gamma[1/\gamma_R - v^{(\Phi)}]}$$

$$= \frac{\sqrt{3}L_s^*}{-L_s + \sqrt{3}(L_s^*)^2 + L_s^2}. \quad (4.17)$$

From (4.13) we can see that, for an infalling emitter, an inward radial motion and a retrograde angular motion will redshift the energy of an escaping photon while a prograde angular motion will blueshift the energy of an escaping photon. From (4.17) we can see that, for an outgoing emitter, the energy of an escaping photon is affected by its velocity in a complex way but is determined only by its angular momentum with a simple relation. We also find that $g_{\text{max},o}$ is always greater than $g_{\text{max},i}$ with the same parameters (L_s, E_s) , which means that an outward radial motion will enhance the energy of escaping photons.

In terms of z_{mob} , the MOB for a (near-)NHEK emitter with $r_s > \tilde{r}_1$ can be summarized as

$$z_{\text{mob},o} = 1 - \frac{1}{\sqrt{3}L_s^*} \left(\sqrt{L_s^2 + 3(L_s^*)^2} - L_s \right),$$
 (4.18)

$$z_{\text{mob},i} = 1 - \frac{1}{\sqrt{3}L_s^*} \left(\frac{2(L_s R_s + E_s)}{\sqrt{R_s^2 - \kappa^2}} - L_s \right), \quad (4.19)$$

while the MOB for a (near-)NHEK emitter with $r_s \le \tilde{r}_1$ can be summarized as

$$z_{\text{mob}} = \begin{cases} 1 - \frac{1}{g[\lambda_{0-}(-3/4), -3/4]}, & \text{if } s_r = -1, \text{ or } s_r = 1 \& \lambda_{0-} > \lambda_{0c}, \\ 1 - \frac{1}{g[\lambda_{0c}(-3/4), -3/4]}, & \text{if } s_r = 1 \& \lambda_{0-} < \lambda_{0c}. \end{cases}$$

$$(4.20)$$

This is one of our main results and we will analyze the expressions of the MOB for specific emitters in Sec. V and plot them in Figs. 2 and 4–6.

V. MOB AND PEP FOR DIFFERENT EMITTERS

In Fig. 2, we display the results of $z_{\text{mob}}(R_s)$ and $\mathcal{P}_e(R_s)$ of the escaping photons emitted from different orbits labeled by (L_s, E_s, s_r) . We can see that the behaviors of the PEP are closely related to that of the MOB. In addition, we also present more examples of the PEPs in Fig. 3 for the emitters along the plunging, the antiplunging, the bounded, and the deflecting orbits, respectively. We take a ZAMS as the reference emitter which has $v^{(R)} = 0$ and $v^{(\Phi)} = 0$. Note that the ZAMS does not follow a geodesic and that we will use the term "orbits" only for the geodesics. We take the emitters moving on circular orbits as the references, as well. We use red and blue dotted curves to show the results for ZAMSs and circular orbits, respectively. The dotted curves with other colors are used as additional reference emitters whose motions are described in the plots. The dashed and solid curves are, respectively, for the infalling $(s_r = -1)$ and the outgoing $(s_r = 1)$ emitters along several orbits labeled by (L_s, E_s) . We show these curves with the rescaled radial coordinates

$$s_{\text{NK}} = \log(R+1), \quad s_{\text{nNK}} = \log\left[R + \sqrt{R^2 - \kappa^2}\right], \quad (5.1)$$

in terms of which, the proper radial distances in the NHEK and near-NHEK regions are [34]

$$ds_{NK}(\bar{R}_1, \bar{R}_2) = s_{NK2} - s_{NK1},$$

 $ds_{nNK}(R_1, R_2) = s_{nNK2} - s_{nNK1}, \qquad \bar{R} = R + 1,$ (5.2)

while the proper radial distance between the NHEK and near-NHEK regions scales as $\log |\epsilon|$ [34,43]. Note that from the perspective of near-NHEK geometry, the $R_{\rm nNK} \to \infty$ limit glues onto the NHEK region. Here the subscripts

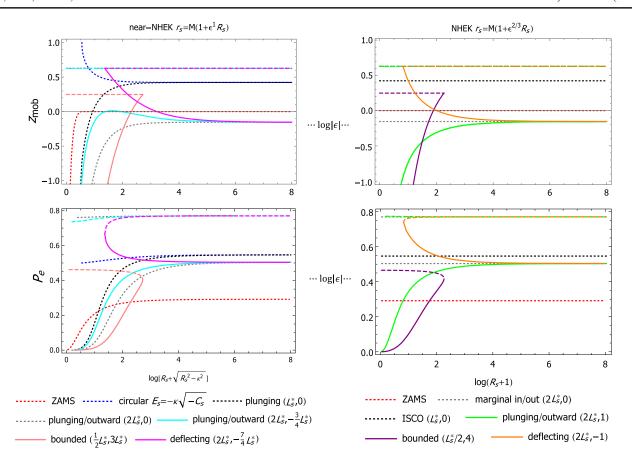


FIG. 2. The MOB $z_{\text{mob}}(R_s)$ and the PEP $\mathcal{P}_e(R_s)$ for photon emissions from several representative emitters moving along (near-) NHEK equatorial geodesics. The conserved quantities (L_s, E_s) label the emitters. We show these with the rescaled radial coordinates, representing the proper radial distance [34]. The dotted curves are used as the reference emitters: the red and blue dotted curves are for ZAMS (which does not follow a geodesic) and the emitters at circular geodesics, respectively, the black and gray dotted curves are for the emitters on the marginal (plunging/antiplunging/deflecting) orbits with $s_r = \pm 1$. The dashed and solid curves are for the emitters on the outgoing $(s_r = +1)$ and ingoing $(s_r = -1)$ orbits, respectively. We have set M = 1, $\mu = 1$, and $\kappa = 1$ (for the near-NHEK case).

"NK" and "nNK" represent the quantities in the NHEK and near-NHEK geometries, respectively. In practice, we choose $M=1, \mu=1$ for the emitter's mass, and $\kappa=1$ for the near-NHEK case.

Before describing the main results of this paper, we recall the relevant results for photon emissions from a ZAMS and from the emitters on circular geodesics. For photon emissions from a ZAMS, we have

$$z_{\text{mob}}(R_s)|_{\text{ZAMS}} = \begin{cases} 0, & \text{if } r_s > \tilde{r}_1, \\ 1 - \frac{2R_s - \sqrt{3}\kappa}{\sqrt{R_s^2 - \kappa^2}}, & \text{if } r_s \le \tilde{r}_1. \end{cases}$$
 (5.3)

It was found in [27,33] that the PEP for a NHEK ZAMS is about 29% and the PEP for a near-NHEK ZAMS decreases as the source radius decreases from near-NHEK infinity towards horizon radius and reaches approximately 13% at \tilde{r}_1 . For photon emissions from the emitters on circular orbits in the range $r_s > \tilde{r}_1$ (\tilde{r}_1 is also the last allowed radius for a circular orbit [43], at which $L_s \to \infty$), we have

$$z_{\text{mob}}(R_s)|_{\text{cir}} = \begin{cases} 1 - \frac{1}{\sqrt{3}}, & \text{if } r_s = M(1 + \epsilon^{\frac{2}{3}}R_s), & (\text{ISCO}), \\ 1 - \frac{1}{\sqrt{3}L_s^*} \left(\frac{2(L_sR_s - \kappa\sqrt{-C_s})}{\sqrt{R_s^2 - \kappa^2}} - L_s\right), & \text{if } r_s = M(1 + \epsilon R_s), & R_s > \tilde{R}_1. \end{cases}$$
(5.4)

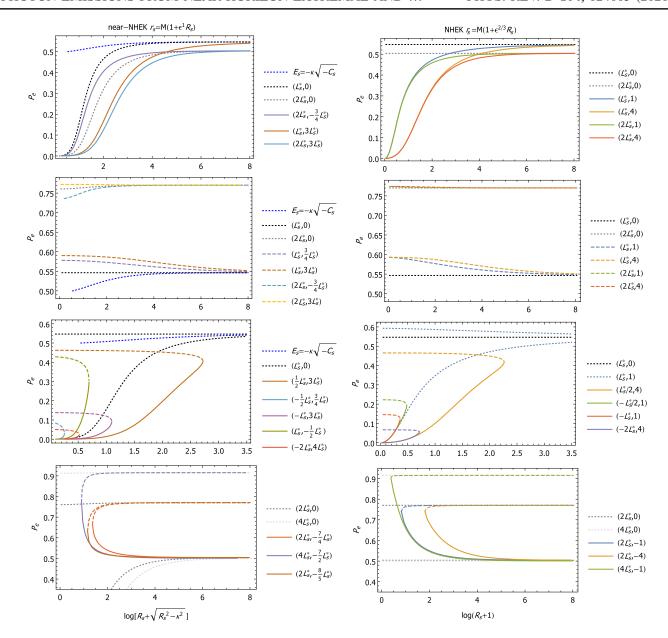


FIG. 3. Top to bottom: examples for the PEPs for the emitters along the plunging, the antiplunging, the bounded, and the deflecting orbits, respectively. The conserved quantities (L_s, E_s) [Eq. (2.18)] label the emitters. The dashed and solid curves are for the sources on the outgoing $(s_r = +1)$ and infalling $(s_r = -1)$ orbits, respectively. The dotted curves are displayed as the references.

All circular orbits in the NHEK region are the ISCO while the ones in the near-NHEK region are unstable. For a given unstable circular orbit in the near-NHEK region, the orbital radius is uniquely determined by $R_c = \kappa L_s / \sqrt{-C_s}$, and then we obtain $z_{\text{cir},u} = z_{\text{mob}}(R_c)|_{\text{cir}} = z_{\text{mob},o}$. Here, the subscript "u" represents "unstable." For photon emissions from the ISCO, the PEP is about 55% [29,30]. We find that, for photon emission from unstable circular orbits,

the PEP decreases from $\sim 55\%$ to $\sim 50\%$ as the orbital radius decreases from the one of the ISCO towards \tilde{r}_1 .

Let us first look at the MOB (see Fig. 2) for a (near-) NHEK emitter labeled by (L_s, E_s, s_r) . For photon emissions from an outgoing emitter $(L_s, E_s, +1)$, the MOB is a constant along a given orbit, i.e., the MOB does not depend on R_s . For photon emissions from a marginal infalling emitter which exists only in the NHEK region, the MOB does not depend on R_s as well, and we have

$$z_{\text{mar},i} = z_{\text{mob},i}|_{\text{marginal}} = 1 - \frac{L_s}{\sqrt{3}L_s^*}.$$
 (5.5)

⁹For photon emissions from stable circular orbits outside the ISCO of an extremal Kerr black hole, the PEP decreases monotonously as the orbital radius decreases and reaches approximately 55% at the ISCO [29,30].

For photon emissions from a bound/deflecting emitter along the infalling part of its orbit, $z_{\text{mob},i}(R_s)$ decreases/increases monotonously as the emitter moves from (near-) NHEK infinity towards horizon. For photon emissions from the plunging emitters, the behavior of $z_{\text{mob},i}(R_s)$ between the NHEK and near-NHEK cases are different: for photon emissions from a NHEK plunging emitter or from a near-NHEK plunging emitter with $E_s \geq 0$, $z_{\text{mob},i}(R_s)$ decreases monotonously as the emitter moves from (near-)NHEK infinity towards horizon, while for photon emissions from a near-NHEK plunging emitter with $-\kappa\sqrt{-C_s} < E_s < 0$, $z_{\text{mob},i}(R_s)$ increases at the beginning as the emitter leaves from the near-NHEK infinity until reaching an extreme value at

$$R_e = -\frac{\kappa^2 L_s}{E_s}. ag{5.6}$$

Then $z_{\text{mob},i}(R_s)$ begins to decrease as the emitter falls towards the horizon, and at R_e we find $z_{\text{mob},i} = z_{\text{mob},o} = z_{\text{cir},u}$ as $E_s \to -\kappa \sqrt{-C_s}$. Moreover, at the (near-)NHEK infinity, we have $z_{\text{mob},i}(R_{\infty}) = z_{\text{mar},i}$ for a plunging or a deflecting emitter, and at the turning point of a bounded or a deflecting orbit, we have $z_{\text{mob},i} = z_{\text{mob},o}$. These are summarized in Tables IV and VI.

Next, we turn to the PEP (see Figs. 2 and 3) for a (near-) NHEK emitter labeled by (L_s, E_s, s_r) . For photon emissions from a plunging emitter, $\mathcal{P}_{e}(R_s)$ decreases monotonously along its orbit as the emitter moves from (near-) NHEK infinity towards the horizon. For photon emissions from a bounded emitter, $\mathcal{P}_e(R_s)$ decreases monotonously along its orbit as the emitter moves outward from the horizon towards the turning point and then bounces off back to the horizon. For photon emissions from a deflecting emitter, $\mathcal{P}_e(R_s)$ increases monotonously along its orbit as the emitter moves inward from (near-)NHEK infinity towards the turning point and then bounces off back towards (near-)NHEK infinity. For photon emissions from the antiplunging emitters, there are differences between the NHEK and near-NHEK cases: for a NHEK antiplunging emitter and for a near-NHEK antiplunging emitter with $E_s > \kappa \sqrt{-C_s}$, $\mathcal{P}_e(R_s)$ decreases monotonously along its orbit from the horizon towards infinity; while for a near-NHEK antiplunging emitter with $E_s < |\kappa \sqrt{-C_s}|$, $\mathcal{P}_e(R_s)$ increases monotonously along its orbit from the horizon towards infinity. Furthermore, we find that for all (antiplunging and deflecting) emitters that can eventually leave from the NHEK infinity we always have $P_e > 50\%$, while for all plunging emitters that fall into the black hole we always have $\mathcal{P}_{e} < 55\%$. In addition, for the bounded emitters in the (near-)NHEK region, we always have $\mathcal{P}_e < 59\%$. These are also summarized in Tables IV and VI.

VI. INFLUENCE OF THE EMITTER'S PROPER MOTION

From Fig. 2, we can also see that the PEP and MOB are clearly affected by the proper motion of an emitter. For an emitter with proper motion labeled by (L_s, E_s, s_r) , the behaviors of the PEP and MOB along its orbit are clarified in the previous section. Next, we consider the behaviors of the PEP and MOB as the orbital constants (L_s, E_s, s_r) vary. To do so, we study the functions $z_{\text{mob}}(L_s, E_s, s_r)$ and $\mathcal{P}_e(L_s, E_s, s_r)$ at several representative points along each emitter's orbit.

In Sec. V, we have found that $z_{\text{mob},o}$ [Eq. (4.18)] depends only on L_s and tends to 1 (infinite blueshift) as $L_s \to \infty$ while tends to $-\infty$ (infinite redshift) as $L_s \to -\infty$. Moreover, $z_{\text{mob},i}$ [Eq. (4.19)] of an infalling emitter is always less than that of an outgoing one with the same parameters (L_s, E_s) , and $z_{\text{mob},i}$ depends on both L_s and E_s . We now consider only photon emissions with $z_{\text{mob}} \ge 0$ and make a cutoff along a given orbit (if relevant) at R_z where $z_{\text{mob}}(R_z) = 0$. In other words, we focus on the prograde orbits $(L_s > 0)$ for the emitters, since no photon emissions from the retrograde emitters $(L_s < 0)$ are blueshifted.

A. NHEK emitters

Table III shows the components of the 3-velocities of the NHEK emitters relative to the ZAMS at the endpoints of these emitters' orbits or at the cutoff point

$$R_z = \frac{2E_s}{\sqrt{3}L_s^* - L_s}. (6.1)$$

We find that the velocities are independent of E_s at these points, and a change of E_s will only extend or shrink the range with $z_{\text{mob}} \geq 0$ by shifting the points R_z or R_+ . As $|E_s|$ increases, R_+ and R_z increase as well. For a bounded orbit, the radial range $R \in (0, R_+)$ is extended while the entire region $R \in (R_z, R_+)$ appears farther away from the horizon. For a deflecting orbit, the radial range $R \in (R_+, \infty)$ is shrunk while the entire region $R \in (R_+, R_z)$ appears closer to the horizon. For a plunging emitter, the radial range $R \in (R_z, \infty)$ is shrunk. We find that the PEP and MOB are also independent of E_s at the endpoints or at the cutoff point.

In Table IV, we display distinct behaviors of $z_{\text{mob}}(L_s, E_s, s_r)$ and $\mathcal{P}_e(L_s, E_s, s_r)$ between the endpoints of the part of an emitter's orbit with $z_{\text{mob}} \geq 0$. Meanwhile, we display in Fig. 4 the corresponding values of z_{mob} and \mathcal{P}_e . From Fig. 4, we can see that the maximum PEP is increased monotonously as L_s is increased from 0. For $L_s = 0$, the maximum PEP is about 33% and the PEP at R_z is about 29%. For $L_s = L_s^*$, the maximum PEP for the antiplunging emitters is about 59%, the minimum/maximum PEP for the antiplunging/plunging emitters is about 55%, and the PEP for the plunging emitters at R_z is about

TABLE III. The 3-velocity components at the endpoints $(0, R_+, \text{ or } R_\infty)$ or the cutoff point (R_z) of a NHEK equatorial orbit.

R_s	0	R_z	R_{+}	R_{∞}
$v^{(R)}$	±1	$\pm\sqrt{2\sqrt{3}L_s^*L_s}/(\sqrt{3}L_s^*+L_s)$	0	$\pm\sqrt{-C_s}/L_s$
$v^{(\Phi)}$	0	$L_s/(\sqrt{3}L_s^* + L_s)$	$L_s/\sqrt{3(L_s^*)^2+L_s^2}$	1/2

TABLE IV. Behaviors of $z_{\text{mob}}(L_s, E_s, s_r)$ and $\mathcal{P}_e(L_s, E_s, s_r)$ between the endpoints of the part of each NHEK equatorial orbit with $z_{\text{mob}} \geq 0$, where $z_{\text{mar},i}$ and $z_{\text{mob},o}$ are given in Eqs. (5.5) and (4.18), respectively.

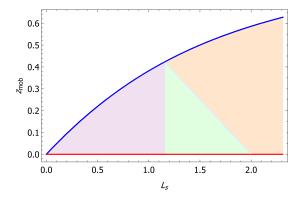
L_s	E_s	s_r	$z_{ m mob}$	${\cal P}_e$	Motion
$0 < L_s < L_s^*$	$E_s > 0$	+	$z_{\text{mob},o}$	$\mathcal{P}_e(R_+) < \mathcal{P}_e < \mathcal{P}_e(0)$	Bounded
$0 < L_s < L_s^*$	$E_s > 0$	_	$(0, z_{\text{mob},o})$	$\mathcal{P}_e(R_z) < \mathcal{P}_e < \mathcal{P}_e(R_+)$	Bounded
$L_s = L_s^*$	$E_s = 0$		$1 - 1/\sqrt{3}$	~55%	ISCO
$L_s \ge L_s^*$	$E_s > 0$	+	$z_{{ m mob},o}$	$\mathcal{P}_e(R_\infty) < \mathcal{P}_e < \mathcal{P}_e(0)$	Antiplunging
$L_s \ge L_s^*$	$E_s > 0$	_	$(0, z_{\text{mar},i})$	$\mathcal{P}_e(R_z) < \mathcal{P}_e < \mathcal{P}_e(R_\infty)$	Plunging
$L_s > L_s^*$	$E_s = 0$	\pm	$z_{\text{mob},o}/z_{\text{mar},i}$	$\mathcal{P}_{ ext{mar},o}/\mathcal{P}_{ ext{mar},i}$	Marginal
$L_s > L_s^*$	$E_s < 0$	+	$z_{{ m mob},o}$	$\mathcal{P}_e(R_+) < \mathcal{P}_e < \mathcal{P}_e(R_\infty)$	Deflecting
$L_s^* < L_s < 2$	$E_s < 0$	_	$(0, z_{\text{mob},o})$	$\mathcal{P}_e(R_z) < \mathcal{P}_e < \mathcal{P}_e(R_+)$	Deflecting
$L_s > 2$	$E_s < 0$	_	$(z_{\text{mar},i}, z_{\text{mob},o})$	$\mathcal{P}_e(R_\infty) < \mathcal{P}_e < \mathcal{P}_e(R_+)$	Deflecting

39%. For $L_s = \sqrt{3}L_s^*$, the PEP for the antiplunging emitters is in a narrow range around 73% and the PEP for the plunging emitters in (R_z, R_∞) is about 50%. However, as $L_s \to \sqrt{3}L_s^*$, the cutoff point R_z tends to infinity and thus the radial region with $z_{\text{mob}} \ge 0$ is actually shrunk to zero. For $L_s > \sqrt{3}L_s^*$, photon emissions from the antiplunging and deflecting orbits always have a PEP greater than 50% while photon emissions from the plunging orbits never have a PEP greater than 50%.

As expected, an antiplunging or deflecting emitter can be observed (i.e., with considerable PEP and by carrying enough energy at infinity) by a distant observer, since the emitter itself can also reach to an asymptotic far region. However, for a plunging emitter that eventually drops into the central black hole, it is interesting to ask to what depth in the near-horizon throat can the emitter still be observed. We find that, among all plunging emitters, the one with critical parameters $(L_s^*, 0)$ has a maximum PEP and the range of (R_z, R_∞) extends to the entire NHEK region.

B. Near-NHEK emitters

Unlike those for the NHEK emitters, the 3-velocity components relative to the ZAMS for an near-NHEK plunging emitter with $-\kappa\sqrt{-C_s} < E_s < 0$ are no longer monotonous along its orbits, and neither does the MOB of such an emitter. Instead, there is an extreme velocity along



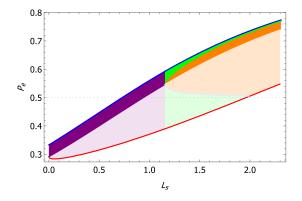


FIG. 4. Values of $z_{\text{mob}}(L_s)$ and $\mathcal{P}_e(L_s)$ between the endpoints of the part of each NHEK equatorial orbit with $z_{\text{mob}} \geq 0$. The radial ranges are illustrated explicitly in Table IV. The blue solid line on the left is for $z_{\text{mob},o}$ and the red solid line has $z_{\text{mob}} = 0$. The light-purple, light-green, and light-orange ranges are for photon emissions from the infalling $(s_r = -1)$ bounded, plunging, and deflecting emitters, respectively, while the purple, green, and orange ranges are for photon emissions from the outgoing $(s_r = +1)$ bounded, antiplunging, and deflecting emitters, respectively.

TABLE V. The 3-velocity components at the endpoints $(\kappa, R_+, \text{ or } R_\infty)$, the cutoff point (R_z) , or the critical point (R_e) of a near-NHEK equatorial orbit.

R_s	κ	R_z	R_e	R_{+}	R_{∞}
$v^{(R)}$	±1	$\pm\sqrt{2\sqrt{3}L_s^*L_s}/(\sqrt{3}L_s^*+L_s)$	$v^{(R)}(R_e)$	0	$\pm\sqrt{-C_s}/L_s$
$v^{(\Phi)}$	0	$L_s/(\sqrt{3}L_s^*+L_s)$	$v^{(\Phi)}(R_e)$	$L_s/\sqrt{3(L_s^*)^2+L_s^2}$	1/2

each of these orbits at the critical point R_e [Eq. (5.6)]. Table V shows the components of an emitter's 3-velocity at the endpoints and the critical point of the emitter's orbit, or the cutoff point

$$R_z = \frac{2E_s L_s \pm (\sqrt{3}L_s^* + L_s)\sqrt{E_s^2 + \kappa^2(C_s + \frac{\sqrt{3}}{2}L_s^*L_s)}}{2C_s + \sqrt{3}L_s^*L_s},$$
(6.2)

where the plus/minus sign is taken as $L_s \leq \sqrt{3}L_s^*$, respectively. In the table, the 3-velocity components at the critical points are

$$v^{(R)}(R_e) = -\sqrt{(E_s^2 + \kappa^2 C_s)/(E_s^2 - \kappa^2 L_s^2)},$$

$$v^{(\Phi)}(R_e) = \kappa L_s / \sqrt{4(E_s^2 - \kappa^2 L_s^2)}.$$
(6.3)

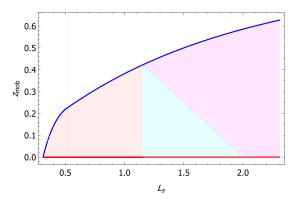
Note that, as $E_s \to -\kappa \sqrt{-C_s}$, the 3-velocity components obtained at R_e have the same forms as those obtained at the turning point R_+ of a bounded or deflecting orbit. We find that the velocities do not depend on E_s at the endpoints $(\kappa, R_+, \text{ or } R_\infty)$ and at the cutoff point R_z , but do depend on E_s at the critical point R_e . As in the NHEK cases, a change of E_s will extend or shrink the range with $z_{\text{mob}} \geq 0$ in a similar way. However, for the near-NHEK case, the PEP and MOB depend on E_s at the endpoints, the cutoff point, and the critical point. And in the limit $|E_s| \to \infty$, the PEP

and MOB tend to the values of the corresponding NHEK cases.

Table VI displays distinct behaviors of z_{mob} and P_e between the endpoints of the part of an emitter's orbit with $z_{\text{mob}} \ge 0$. $z_{\text{mob},o}$ is given in Eq. (4.18) if $r_s \ge \tilde{r}_1$, or in Eq. (4.20) if $r_s < \tilde{r}_1$. $z_{\text{mar},i}$ is given in Eq. (5.5) and $z_{\text{mob},i}(R_e) = 1 + \frac{L_s}{\sqrt{3}L_s^*} - \frac{2\sqrt{\kappa^2 L_s^2 - E_s^2}}{\sqrt{3}\kappa L_s^*}$. The value of L_0 is obtained for $z_{\text{mob},o} = 0$. Meanwhile, we display in Fig. 5 the corresponding values of z_{mob} and \mathcal{P}_e , in which E_s is chosen to be 0 for the bounded, the plunging, and the antiplunging emitters, and E_s is chosen to be $-\kappa L_s$ for the deflecting emitters. In order to see the effects of E_s , we display $z_{\text{mob}}(R_s)$ and $\mathcal{P}_e(R_s)$ for the emitters with $L_s = \sqrt{3}L_s^*$ and $\frac{4}{5}L_s^*$, and each with several values of E_s in Fig. 6. The effects of E_s can also be seen from Fig. 3. Figure 5 shows that, if E_s is fixed, $z_{\text{mob}}(L_s)$ and $\mathcal{P}_e(L_s)$ for the near-NHEK emitters have similar behaviors as those for the NHEK emitters. However, the values of PEP and MOB do depend on E_s and tend to the NHEK values as $|E_s| \to \infty$. For $L_s \to L_0 = \frac{L_s^*}{\sqrt{s}}$ and $E_s = 0$, the turning point R_+ approaches \tilde{r}_1 and the maximum PEP is about 31%. For $L_s \ge L_s^*$ and $E_s \to E_c = -\kappa \sqrt{-C_s}$, the PEP for photon emissions from unstable circular orbits at $R_c = \kappa L_s / \sqrt{-C_s}$ can be formally written as $\mathcal{P}_{\text{cir},u} = \mathcal{P}_e(L_s)$. We find that $\mathcal{P}_{\text{cir.}u}$ is the same as the PEP of the marginal infalling emitters in the NHEK region, $\mathcal{P}_{\text{mar},i}$, that is $\mathcal{P}_{\text{cir},u} = \mathcal{P}_{\text{mar},i}$ (see the dotted blue curve on the right in Fig. 5 and the light-blue curve above the light-green region on the right in

TABLE VI. Behaviors of $z_{\text{mob}}(L_s, E_s, s_r)$ and $\mathcal{P}_e(L_s, E_s, s_r)$ between the endpoints of the part of each near-NHEK equatorial orbit with $z_{\text{mob}} \geq 0$.

L_s	E_s	s_r	$z_{ m mob}$	\mathcal{P}_{e}	Motion
$\overline{L_0 < L_s < L_s^*}$	$E_s > -\kappa L_s$	+	$z_{\text{mob},o}$	$\mathcal{P}_e(R_+) < \mathcal{P}_e < \mathcal{P}_e(\kappa)$	Bounded
$L_0 < L_s < L_s^*$	$E_s > -\kappa L_s$	_	$(0, z_{\text{mob},o})$	$\mathcal{P}_e(R_z) < \mathcal{P}_e < \mathcal{P}_e(R_+)$	Bounded
$L_s = L_s^*$	$E_s \geq 0$	+	$z_{{ m mob},o}$	$\mathcal{P}_e(R_\infty) < \mathcal{P}_e < \mathcal{P}_e(\kappa)$	Antiplunging
$L_s > L_s^*$	$E_s > \kappa \sqrt{-C_s}$				
$L_s > L_s^*$	$E_s < \kappa \sqrt{-C_s} $	+	$z_{\text{mob},o}$	$\mathcal{P}_e(\kappa) < \mathcal{P}_e < \mathcal{P}_e(R_\infty)$	Antiplunging
$L_s > L_s^*$	$-\kappa\sqrt{-C_s} < E_s < 0$	_	$(0, z_{\text{mob},i}(R_e))$	$\mathcal{P}_e(R_z) < \mathcal{P}_e < \mathcal{P}_e(R_\infty)$	Plunging
$L_s \ge L_s^*$	$E_s \geq 0$	_	$(0, z_{\text{mar},i})$	$\mathcal{P}_e(R_z) < \mathcal{P}_e < \mathcal{P}_e(R_\infty)$	Plunging
$L_s > L_s^*$	$E_s = -\kappa \sqrt{-C_s}$		$z_{\text{mob},o}$	${\cal P}_{{ m cir},u}$	Circular
$L_s > L_s^*$	$E_s < -\kappa\sqrt{-C_s}$	+	$z_{\text{mob},o}$	$\mathcal{P}_e(R_+) < \mathcal{P}_e < \mathcal{P}_e(R_\infty)$	Deflecting
$L_s^* < L_s < 2$	$E_s < -\kappa\sqrt{-C_s}$	_	$(0, z_{\text{mob},o})$	$\mathcal{P}_e(R_z) < \mathcal{P}_e < \mathcal{P}_e(R_+)$	Deflecting
$L_s > 2$	$E_s < -\kappa\sqrt{-C_s}$	_	$(z_{\text{mar},i}, z_{\text{mob},o})$	$\mathcal{P}_e(R_\infty) < \mathcal{P}_e < \mathcal{P}_e(R_+)$	Deflecting



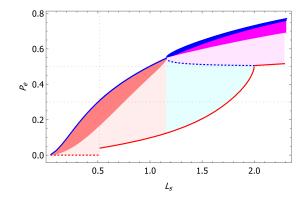
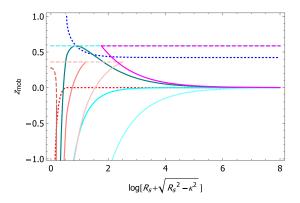


FIG. 5. Values of $z_{\text{mob}}(L_s)$ and $\mathcal{P}_e(L_s)$ between the endpoints of the part of each near-NHEK equatorial orbit with $z_{\text{mob}} \geq 0$. The ranges are illustrated explicitly in Table VI. For the emitters with $L_0 < L_s < \frac{L_s^*}{\sqrt{5}}$ and $E_s = 0$, the MOB z_{mob} is obtained for photon emissions from $\tilde{r} < \tilde{r}_1$; otherwise, it is given in Eqs. (4.18) and (4.19). We have chosen $E_s = 0$ for the bounded, the plunging and the antiplunging emitters, and have chosen $E_s = -\kappa L_s$ for the deflecting emitters. The solid blue line in the left is for $z_{\text{mob},o}$ and the solid red line has $z_{\text{mob}} = 0$, while the dotted red line in the right has $z_{\text{mob}} \to -\infty$. The light-pink, light-blue, and light-magenta ranges are for photon emissions from the infalling ($s_r = -1$) bounded, plunging, and deflecting emitters, respectively, while the pink and magenta ranges are for photon emissions from the outgoing ($s_r = +1$) bounded and deflecting emitters, respectively, and the blue range is the overlap range for photon emissions from the antiplunging and outgoing deflecting emitters.



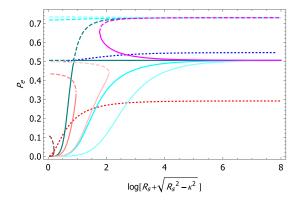


FIG. 6. $z_{\text{mob}}(R_s)$ and $\mathcal{P}_e(R_s)$ for photon emissions from several near-NHEK emitters. The red dotted curve is for the ZMAS, and the blue dotted curve is for the emitters on circular geodesic orbits. The dashed curves are for $s_r=+1$ and the solid curves are for $s_r=-1$. The dark-cyan, cyan, and light-cyan have $L_s=\sqrt{3}L_s^*$ and $E_s=-\kappa\sqrt{-C_s}$, 0, and $2\sqrt{3}L_s^*$, respectively. The magenta curve has $L_s=\sqrt{3}L_s^*$ and $E_s=-\kappa L_s$. The dark-pink, pink, and light-pink curves have $L_s=\frac{4}{5}L_s^*$ and $E_s=-\frac{(8-\sqrt{3})}{10}L_s^*$, 0, and $\frac{\sqrt{3}}{2}L_s^*$, respectively.

Fig. 4). As E_s is perturbed from E_c for $L_s \in [L_s^*, \sqrt{3}L_s^*)$, an unstable circular orbit would become a deflecting/antiplunging/plunging orbit and $\mathcal{P}_e(L_s)$ would start to depart from $\mathcal{P}_{\text{cir},u}$. Among all the plunging emitters with a given $L_s \in [L_s^*, \sqrt{3}L_s^*)$, the PEP takes a maximum value for $E_s \to E_c^+$, which corresponds to the marginal plunging emitter starting from R_c . For the marginal plunging emitter with $L_s = L_s^*$, the orbital radius R_c tends to the radius of the ISCO, and we have $E_c = 0$ and

$$\mathcal{P}_e(R_{\rm ISCO})|_{(L_s^*,0,-1)} \approx 55\%, \qquad \mathcal{P}_e(R_z)|_{(L_s^*,0,-1)} \approx 12\%,$$
(6.4)

where

$$R_z = \sqrt{1 + \frac{2}{\sqrt{3}}}\kappa. \tag{6.5}$$

Here we only give a few selected illustrations for several representative features for photon emissions from the nearhorizon emitters, and a complete picture can be built up from the figures in this paper.

VII. SUMMARY AND CONCLUSION

In this paper, we studied isotropic and monochromatic photon emissions from equatorial emitters moving on timelike geodesics in the (near-)NHEK regions [see Eqs. (2.12) and (2.14)]. The orbits for these geodesics were classified in Tables I and II, whose orbital radii were

written as $r_s = M(1 + \epsilon^q R_s)$. Along each emitter's orbit, we had the conserved quantities for the emitter: the angular momentum l_s and the energy ω_s . For a (near-)NHEK emitter, we introduced $l_s = L_s$ and $\omega_s = \frac{l_s}{2M} + \frac{E_s}{2M} \epsilon^q$. We had $q = \frac{2}{3}$ for the NHEK case in the ISCO scale and q = 1for the near-NHEK case. In Sec. IV we computed the PEP and MOB for photon emissions from these emitters. In general, the results depend on the emitter radius R_s and the constants of the emitter's motion: L_s , E_s , and s_r , with s_r being the radial orientation. Then, in Sec. V we analyzed the behaviors of the PEP and MOB along each given orbit to see their dependence on R_s , and we summarized the results in Tables IV and VI. After that, we analyzed the effects of an emitter's proper motion on the MOB and PEP in Sec. VI. The main analytical result for MOB was summarized in Eqs. (4.18)-(4.20). Moreover, the numerical results for the MOB and PEP were displayed in Figs. 2–6, from which we found that the MOB and PEP had similar behaviors as the parameters of the emitters vary.

In Fig. 2, we showed that the PEP and MOB for the emitters belonging to distinct classes of motions were different. In addition, in Fig. 3 we displayed more examples on the PEP for the emitters belonging to each class of motion. Furthermore, from Figs. 4–6 we could see that the overall observability of an emitter, which said the magnitudes of the PEP and MOB, were determined by the angular momentum l_s of the emitter. This is reasonable since the energy ω_s of a (near-)NHEK emitter is constrained to be near the superradiant bound $\omega_* = \frac{l_s}{2M}$. And, as expected, the photons from the outgoing emitters are brighter than those from the infalling ones that have the same parameters (l_s, ω_s) . As the angular momentum l_s of an emitter increases, the MOB and the maximum of the PEP for the emitters with all allowed motions increase accordingly. On the other hand, for the emitters with the same angular momentum l_s , the corrections of their energies to the critical energy, ¹⁰ $\Delta\omega = \omega_s - \omega_* = \frac{E_s}{2M}\epsilon^q$, distinguish their motions and affect the MOB and PEP for these emitters as well. Simply speaking, while l_s plays a dominant role on the overall feature of MOB and PEP, $\Delta \omega$ starts to play a role after l_s being fixed.

We picked out several characteristic values of l_s to illustrate our results. If $l_s=0$, then $z_{{\rm mob},o}=0$ [see Eqs. (4.18) and (4.20)] which means that no photon emissions from retrograde emitters ($l_s<0$) were blueshifted. We focused on the case $l_s>0$, which has blueshifted photon emissions. Around $l_s=l_s^*$ [see Eq. (3.1)], the emitters start to appear with distinct motions and

different observational features. For $0 < l_s < l_s^*$, the emitters are bounded and require $\Delta \omega < -\epsilon^q \kappa \omega_*$, and the PEPs for them are always less than 59%. In this case, the maximum value is reached as $l_s \rightarrow l_s^*$ and the emitter is bounced outward at the inner boundary of its orbit. For $l_s = l_s^*$ and $\Delta \omega = 0$, the emitters are on the ISCO in the NHEK region and have the marginal (anti)plunging motion in the near-NHEK region, while for $l_s = l_s^*$ and $\Delta\omega \ge 0$, the emitters have the (anti)plunging/ deflecting motions, respectively. And, for the emitter at the ISCO, the PEP is about 55% and the MOB is $1 - 1/\sqrt{3} \approx 0.42$. Similarly, for $l_s > l_s^*$, the emitters with $\Delta \omega = -\epsilon \kappa \omega_* \sqrt{-C_s/l_s^2}$ are on unstable circular orbits while the emitters with $\Delta \omega \ge -\epsilon^q \kappa \omega_* \sqrt{-C_s/l_s^2}$ take the (anti)plunging/deflecting motions, respectively. The PEP for the emitter on unstable circular orbits, $\mathcal{P}_{cir,u}$, decreases from 55% to 50% as the orbital radius decreases from the ISCO radius towards the horizon radius. The PEP for the deflecting and antiplunging emitters are always greater than 50% and the MOB for the outgoing parts of the orbits are always greater than 0.42, which implies that the emitters with $l_s > l_s^*$ and with outward radial motions could be well observed by distant observers. On the other hand, the PEPs for the plunging emitters decrease as the emitters move from the NHEK boundary towards the horizon and are always less than 55% (the value for the emitters at the ISCO). We found that $\mathcal{P}_{\text{mar},i} = \mathcal{P}_{\text{cir},u}$ with $\mathcal{P}_{\text{mar},i}$ being the PEP for the plunging emitters at the NHEK boundary. Moreover, as a plunging emitter moves from NHEK infinity towards the horizon, the inward radial velocity for the emitter with $\Delta \omega > 0$ increases monotonously, while that for the emitter with $\Delta \omega < 0$ decreases at the beginning and then increases to the end. The maximum value of MOB for a plunging emitter with $\Delta \omega > 0$ is obtained at the NHEK infinity, and this maximum value $(z_{\text{mar},i})$ decreases as l_s is increased and $z_{\text{mar},i} = 0$ for $l_s = \sqrt{3}l_s^*$. In addition, as l_s increases for $\Delta\omega \to -\epsilon\kappa\omega_*\sqrt{-C_s/l_s^2}$, the maximum value of MOB for a plunging emitter increases, and this maximum value $(z_{cir,u})$ is obtained at the corresponding orbital radius of an unstable circular orbit, and actually $z_{\text{cir},u} = z_{\text{mob},o}$. Furthermore, we found that the PEP and MOB decreased faster for the plunging emitters with larger $\Delta \omega$, which means that for the plunging emitters, a smaller $\Delta \omega$ would extend the observable range towards the deeper region in the near-horizon throat of a high-spin black hole.

ACKNOWLEDGMENTS

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¹⁰Note that, since in the NHEK region we have $\Delta \omega$ at $\mathcal{O}(\epsilon^{2/3})$ while in the near-NHEK region we have $\Delta \omega$ at $\mathcal{O}(\epsilon)$, the near-NHEK emitters correspond to the NHEK emitters with $\Delta \omega = 0$.

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